Technical Memorandum 33-736 Volume III

Mission Design Data for Venus, Mars, and Jupiter Through 1990

Andrey B. Sergeyevsky

(NASA-CF-143507) MISSION DESIGN DATA FOR N75-32142
VENUS, MAPS, AND JUFITEP THROUGH 1990,
VOLUME 3 (Jet Propulsion Lab.) 330 p

CSCL 22A Unclas
G3/12 35317

JET PROPULSION LABORATORY

CALIFORNIA INSTITUTE OF TECHNOLOGY

PASADENA, CALIFORNIA

September 1, 1975

REPRODUCED BY
NATIONAL TECHNICAL
INFORMATION SERVICE
U.S. DEPARTMENT OF COMMERCE
U.S. DEPARTMENT OF COMMERCE
U.S. DEPARTMENT OF COMMERCE



I		

Technical Memorandum 33-736 Volume III

Mission Design Data for Venus, Mars, and Jupiter Through 1990

Andrey B. Sergeyevsky

JET PROPULSION LABORATORY

CALIFORNIA INSTITUTE OF TECHNOLOGY

PASADENA, CALIFORNIA

September 1, 1975

Prepared Under Contract No. NAS 7-100 National Aeronautics and Space Administration

PREFACE

This document is divided into three volumes. Volume I comprises the mission design data for Venus, Volume II the data for Mars, and Volume III the data for Jupiter.

ı		

TABLE OF CONTENTS

A.	INTRODUCTION
В.	DESCRIPTION OF TRAJECTORY DATA
C.	DESCRIPTION OF PLANETARY POSITIONAL DATA
REF	PERENCES
	LIST OF FIGURES
1.	Definition of B-Plane
2.	Definition of Cone and Clock Angle
ETS, Jupit	ours of C ₃ and Flight Times, VHP, DLA, ZAL, INC, ZAP, LVI, ZAE, ETE, THA, SG1, SG2, and SG3 for Earth to ter Missions Launched in 1977, 1978, 1979, 1980-81, 1982, 1984, 1985, 1986, 1987, 1988, 1989, 1990 4-1 through 4-182
Junit	ter Positional Data for the Years 1975 to 1995 4-183 through 4-330

ı		

MISSION DESIGN DATA FOR VENUS, MARS AND JUPITER THROUGH 1990

Andrey B. Sergeyevsky

A. INTRODUCTION

This document presents mission design data for direct transfer trajectories from Earth to three planets — Venus, Mars and Jupiter, extending previously published information (see Refs. 1, 2, 3, 4 and 5) through the 1990 departure opportunity.

The primary purpose of this effort is to provide the mission analyst with graphical information, sufficient for preliminary mission design and evaluation. The data follows closely the format of Reference 4 and reflects methods of Reference 2. A specially modified version of the Space Research Conic Program (SPARC) (see Ref. 6) was used to generate the trajectory information presented. The data were automatically contour-plotted on the SC4020 plotter using the General Plot Program (GPP) (see Ref. 7), then hand-retouched and labeled. A special program (VIEWPE) was constructed to provide planetary positional data in graphical form, plotted on the SC4020, and presented in original format.

The data are arranged in three sections by arrival planet, in natural sequence. Each section consists of two parts—the trajectory characteristics for all available opportunities to the particular planet, in chronological order, followed by that planet's positional data for every calendar year, from 1975 to 1995.

The persevering and encouraging insistence of management, especially that of Mr. Willard E. Bollman to carry this effort through to completion, as well as the graphic and editorial support of Mr. Richard W. Rackus are gratefully acknowledged.

B. DESCRIPTION OF TRAJECTORY CHARACTERISTICS DATA

1. General

The data represent trajectory performance information plotted in the departure date/arrival date space, thus defining all possible transfer trajectories between the two bodies, within the time-span considered. Fourteen individual parameters are contour-plotted on the departure energy (C₃) background contour chart, for each opportunity. The following opportunities are presented:

To Venus: 1975, 1976/7, 1978, 1980, 1981, 1983,

1984/5, 1986, 1988, 1989/90.

To Mars: 1979, 1981/2, 1983/4, 1985/6, 1988,

1990.

To Jupiter: 1977, 1978, 1979, 1980/81, 1981/82,

1983, 1984, 1985, 1986, 1987, 1988,

1989, 1990.

2. Definition of Terms

The following parameters are displayed on the contour plots:

C₃ Earth departure energy (km^2/sec^2) ; same as the square of departure hyperbolic excess velocity $V_{\infty}^2 = C_3 = V_1^2 - 2GM/R_1$, where

V_I = conic injection velocity (km/sec)

GM = gravitational constant times mass of the attracting body, from Reference 8:

> $GM_{VENUS} = 0.32486010E6.$ (km³/sec²)

 $GM_{EARTH} = 0.39860115E6$

 $GM_{MARS} = 0.42828444E5$

 $GM_{JUPITER} = 0.12670772E9$

R _l =	=	
		sum of surface radius R _S PLANET
		h _I , where (see Ref. 8):

$$R_{S_{VENUS}} = 6052 \text{ (km)}$$
 $R_{S_{EARTH}} = 6378.16$
 $R_{S_{MARS}} = 3393.4$

$$R_{S_{JUPITER}} = 71372$$

- TF Time of flight (Days)
- CD Earth to planet communication distance at arrival (km)
- VHP Arrival hyperbolic excess velocity

$$V_{\infty} = \sqrt{V^2 - \frac{2GM}{R}}$$
, (km/sec),

where

V = Heliocentric conic arrival velocity at heliocentric radius R (km).

Arrival Planet Orbit insertion velocity increment ΔV , at periapse, may be computed from V_{∞} :

$$\Delta V = \sqrt{V_{\infty}^2 + \frac{GM}{R_p}} - \sqrt{\frac{2GM R_A}{R_p(R_A + R_p)}}$$

where R_p and R_A are planetocentric periapse and apoapse radii (km), respectively. Similarly, if specific capture orbit period P(sec) and periapse radius R_p are desired:

$$\Delta V = \sqrt{V^2 + \frac{2GM}{R_p}} - \sqrt{\frac{2GM}{R_p} - \sqrt[3]{\left(\frac{2GM\pi}{P}\right)^2}}$$

- B-PLANE A plane normal to the incoming V_{∞} vector and passing through the center of planet.
- T-AXIS Axis in B-plane, parallel to ecliptic (Earth mean orbital) plane (see Figure 1).

- DLA Geocentric declination (vs. mean Earth equator of launch date) of the departure V_{∞} vector. May impose launch constraints. (deg)
- ZAL Angle between departure V_{∞} vector and Sun-Earth vector. Equivalent to Earth-probe-Sun angle, several days out. (deg)
- INC Heliocentric inclination of transfer trajectory with mean ecliptic (Earth orbital) plane of launch date. (deg)
- ZAP Angle between arrival V_{∞} vector and the arrival planet-to-Sun vector. Equivalent to planet-probe-Sun angle at far encounter; for subsolar impact would be equal to $180.^{\circ}$ (deg).
- ETS Angle in arrival B-plane, measured from T-axis, clockwise, to projection of Sunto-planet vector. Equivalent to solar occultation region center-line. (deg)
- LVI Planetocentric latitude of vertical impact vs arrival planet equator. Note that Venusian north is below ecliptic, while Mars' and Jupiter's is above. Equivalent to declination of the incoming asymptote (i.e., the negative of incoming V_{∞} vector) in planetary equator system.
- ZAE Angle between arrival V_{∞} vector and the planet-to-Earth vector. Equivalent to planet-probe-Earth angle at far encounter. (Deg.)
- ETE Angle in arrival B-plane, measured from T-axis, clockwise, to projection of Earth-to-planet vector. Equivalent to Earth occultation region centerline. (deg)
- THA Angle in arrival B-plane, from T-axis, clockwise, to major axis of error dispersion ellipse (0 180 deg).
- SG1 Semi-major axis magnitude of B-plane dispersion ellipse, resulting from a spherically distributed V_{co} velocity vector error of 0.1 m/sec on departure asymptote (km).
- SG2 Semi-minor axis of above dispersion eclipse (km).

SG3 Arrival time dispersion, normal to B-plane, for above error model (sec).

YR/M/D Year, Month, Date.

Y-axis label

C. DESCRIPTION OF PLANETARY POSITIONAL DATA

1. General

The data represent planetary geometry-related information plotted versus calendar arrival date at the target planet. Each set of seven plots represents the annual time history of 19 parameters, and may be used for flyby and orbiter missions.

2. Description of Curve Labels

P	Target	planet,	equivalent	to	probe	ap-
	proach	ing or in	orbit abou	it ta	rget pla	net.

- E Earth
- S Sun
- CA Cone Angle, i.e., Sun-probe-object (Earth or Canopus, etc.) angle. (See Figure 2.)
- KA Clock Angle, i.e., angle between projections of the Probe-Canopus and probeobject vectors into the plane normal to the sun-line (for which CA = 90°). (See Figure 2.)
- RISEXX Rise time (GMT) of planet through 6° horizon mask at DSN Station No. XX. (e.g., XX = 14 = GOLDSTONE, 43 = CANBERRA, 63 = MADRID.)
- SETXX Set time (GMT) of planet through 6° horizon mask at DSN Station No. XX.

3. Description of Plots

Plot

a)	DECLIN	Geocentric Earth equatorial declination of planet (P), plan-		
		etocentric planetary equatorial declination of Earth (E)		
		and Sun (S). Note that Venus-		
		ian north is below ecliptic.		

- b) EC.LON Heliocentric ecliptic longitude of planet.
- c) CA,KA Cone (ECA) and Clock (EKA) angle of Earth and cone angle of Canopus (CCA) as seen from a Sun-Canopus oriented spacecraft near target planet, P (see Figure 2).
- d) DISTANCE Sun-Planet distance (SP) and Earth-Planet communication distance (EP) in mill. km.
- e) SUN-EARTH- Sun-Earth-Planet angle (SEP), PLANET indicating times of superior (SEP \simeq 0) and inferior (SEP \simeq 180°) conjunction; SEP > 5° is a communications constraint.
- f) STATION Rise and Set times (GMT) of planet at 3 DSN Stations on Earth, 6° mask.

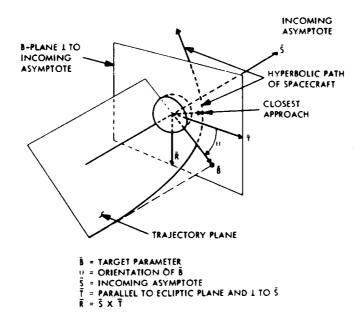


Figure 1. Definition of B-Plane

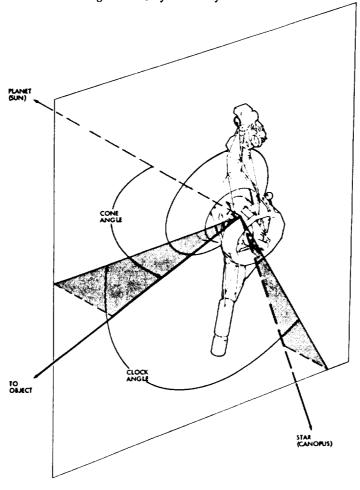


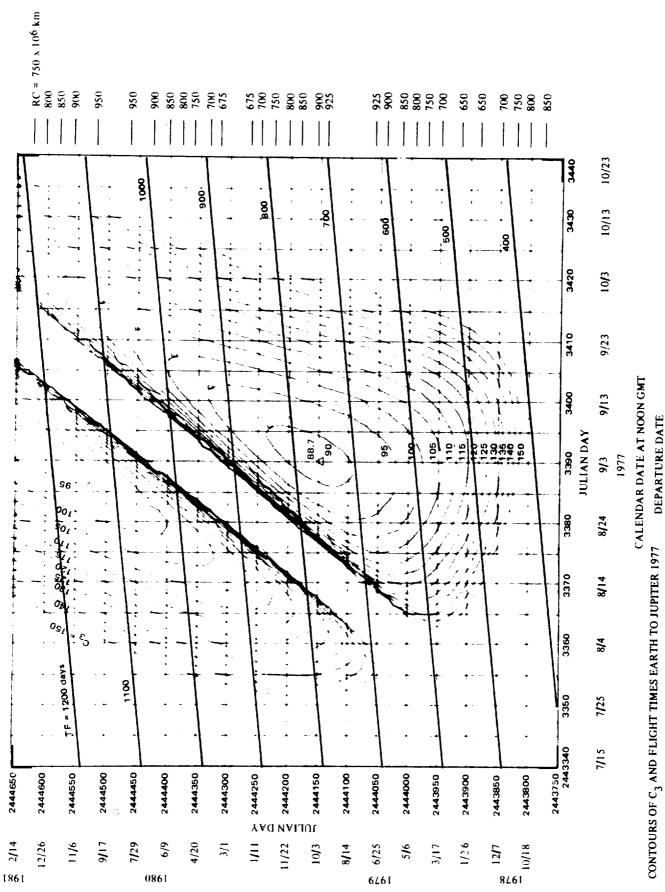
Figure 2. Definition of Cone and Clock Angle

REFERENCES

- Clarke, V. C., Jr., Bollman, W. E., Roth, R. T., Scholey, W. J., "Design Parameters for Ballistic Interplanetary Trajectories Part I. One-way Transfers to Mars and Venus," JPL TR 32-77, January 1963.
- 2. Clarke, V. C., Jr., Bollman, W. E., Feritis, P. H., Roth, R. Y., "Design Parameters for Ballistic Interplanetary Trajectories Part II. One-way Transfers to Mercury and Jupiter," JPL TR 32-77, January 1966.
- 3. Richards, R. J., Roth, R. Y., "Earth-Mars Trajectories," JPL TM 33-100, June 1965.
- Kohlhase, C. E., Bollman, W. E., "Trajectory Selection Considerations for Voyager Missions to Mars During the 1971-1977 Time Period," JPL TM 33-210, September 1965.
 - *JPL Internal Document

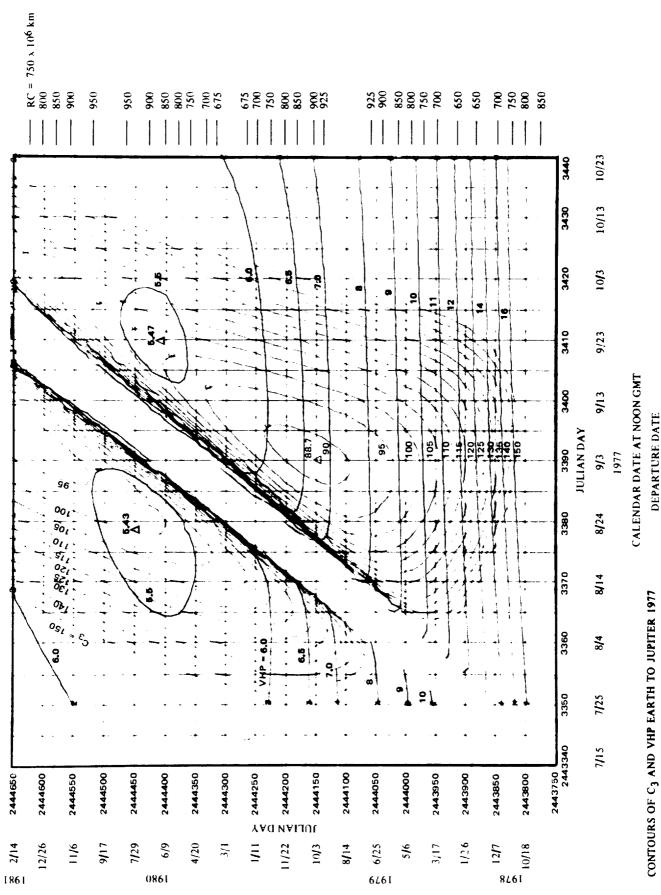
- 5. Wallace, R. A., "Trajectory Considerations for a Mission to Jupiter in 1972," JPL TM 33-375, March 1968.
- 6. Roth, R., Zorian, M. D., "Space Research Conic Program, Phase III," JPL 900-130, Rev. A, May 1969.*
- 7. "General Plot Program," JPL 900-341, Anon., May 1970.*
- 8. Melbourne, W. G., Mulholland, T. D., Sjogren, W. L., Sturms, F. M., Jr., "Constants and Related Information for Astrodynamic Calculations, 1968," JPL TR 32-1306, July 1968.

·			
	ı		

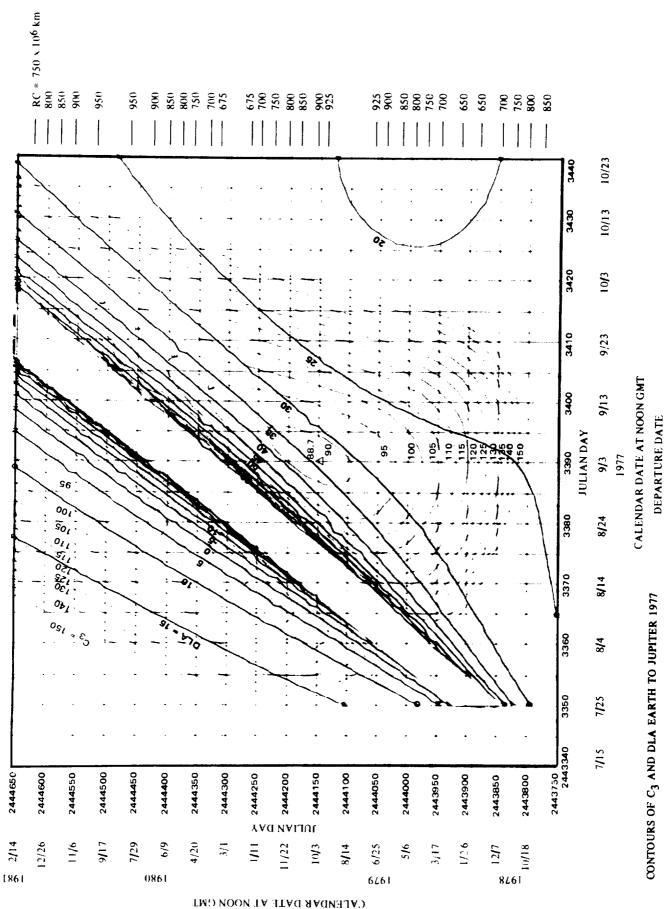


ARRIVAL DATE
CALENDAR DATE AT NOON GMT

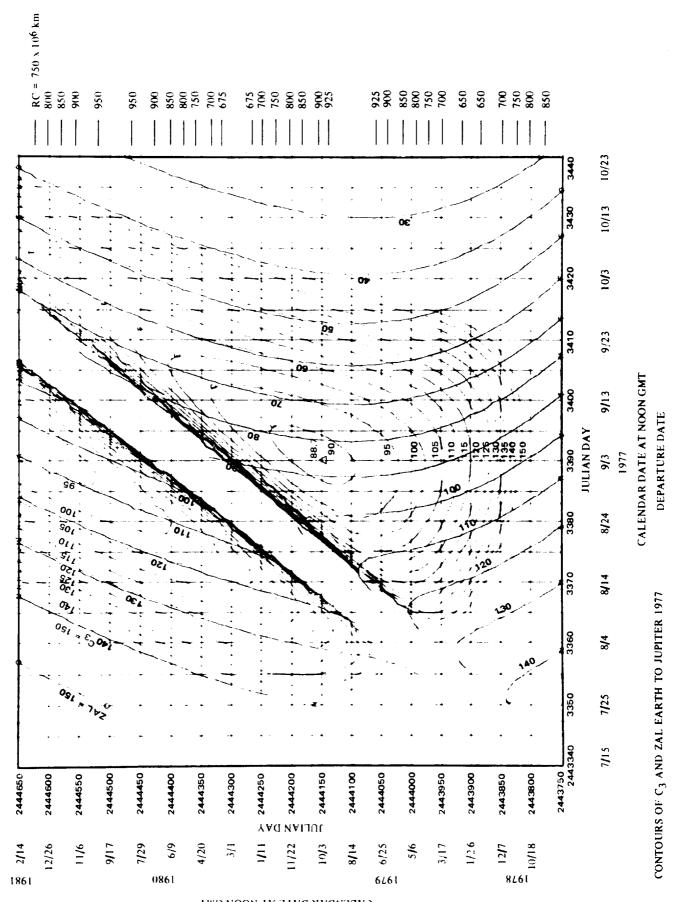
•



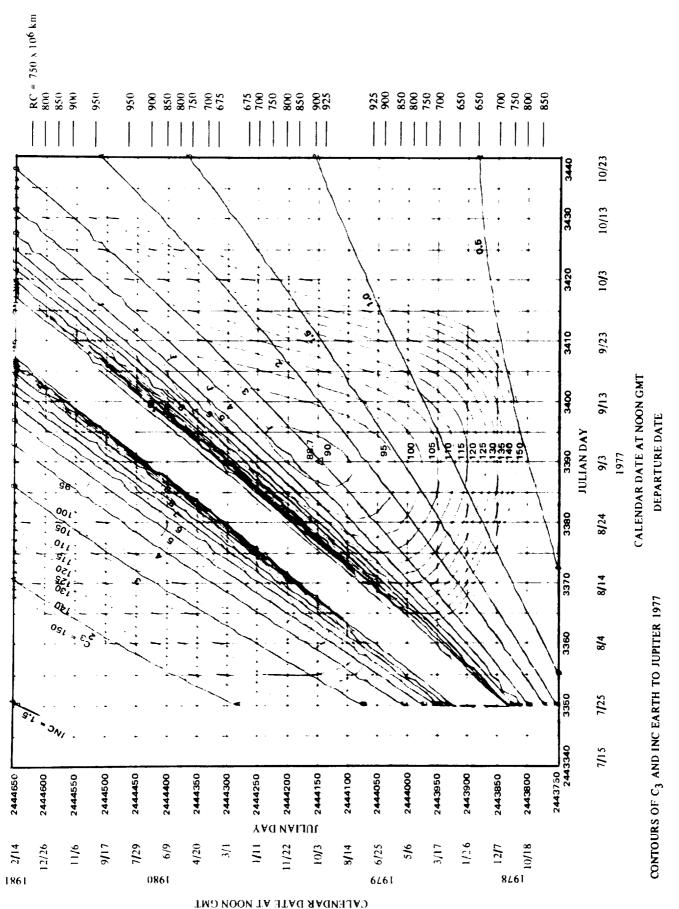
ARRIVAL DATE CALENDAR DATE AT NOON GMT



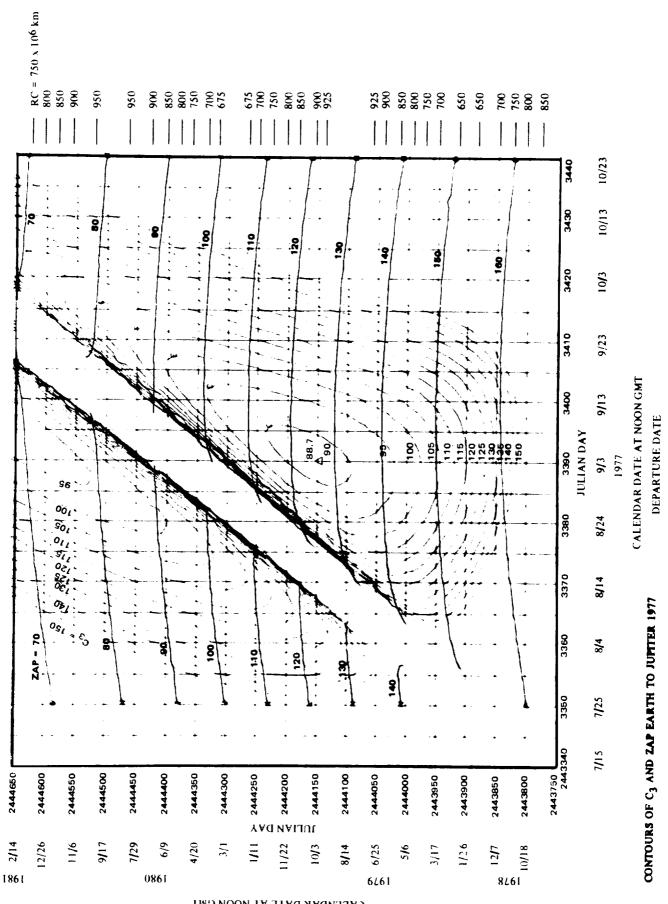
ARRIVAL DATE ("ALENDAR DATE DATE



ARRIVAL DATE CALENDAR DATE AT NOON GMT



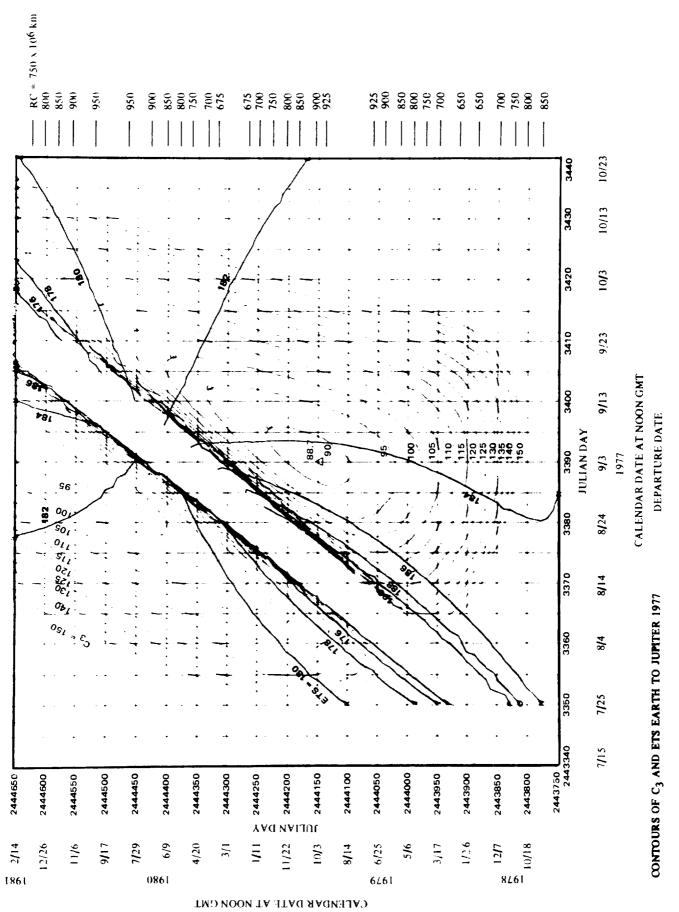
ARRIVAL DATE



ARRIVAL DATE
CALENDAR DATE AT NOON GMT

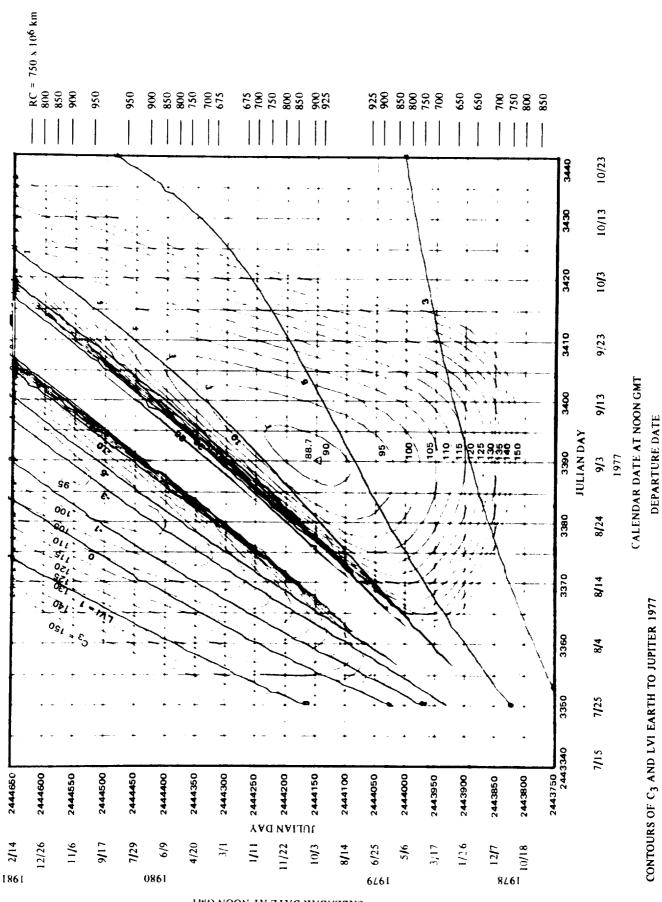
ORIGINAL PAGE IS OF POOR QUALITY

46

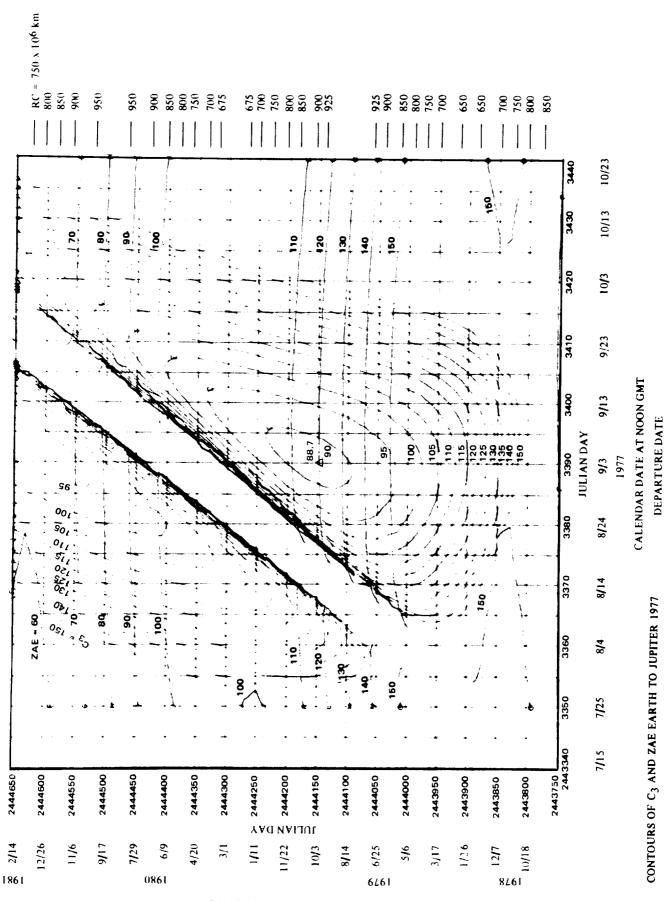


ORIGINAL PAGE IS OF POOR QUALITY

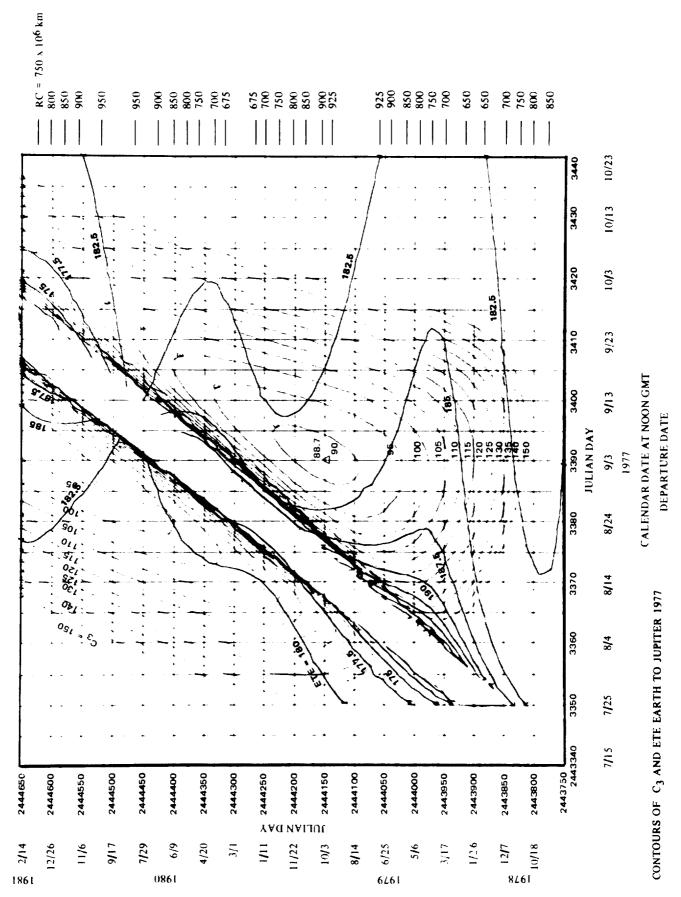
ARRIVAL DATE



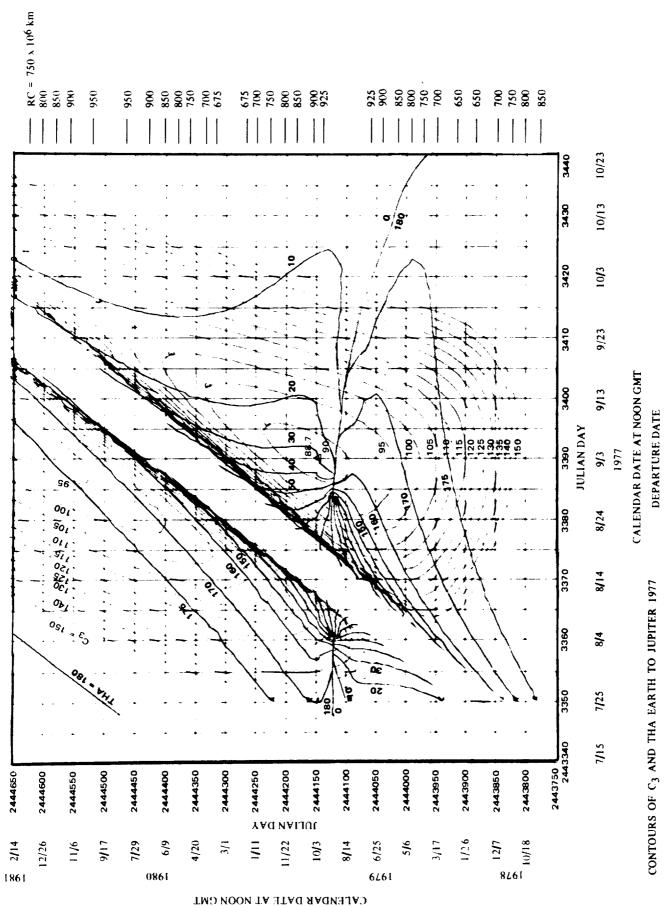
ARRIVAL DATE
CALENDAR DATE AT MOON GMT



ARRIVAL DATE
CALENDAR DATE AT NOON GMT



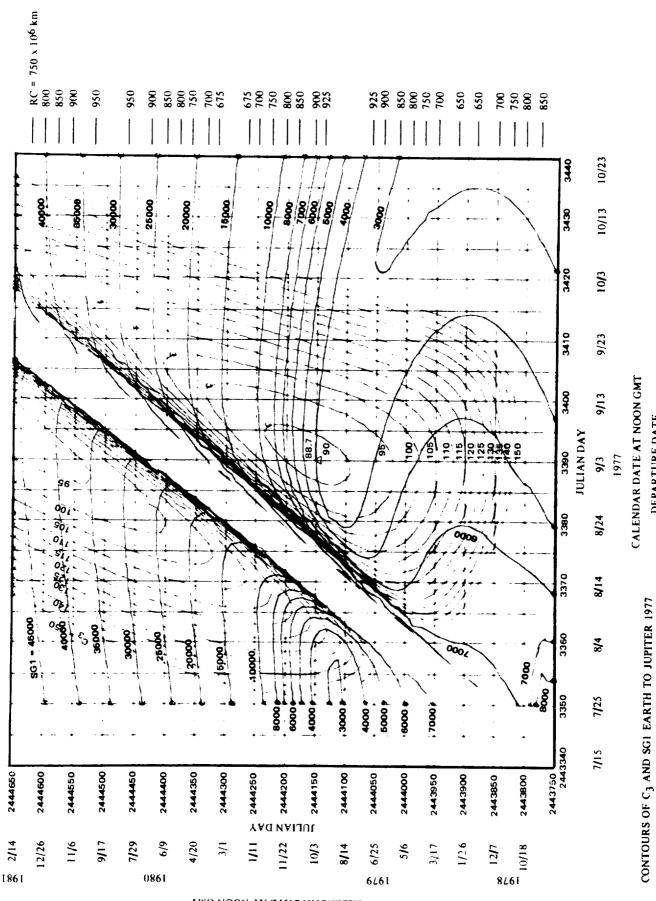
ARRIVAL DATE
CALENDAR DATE AT NOON GMT



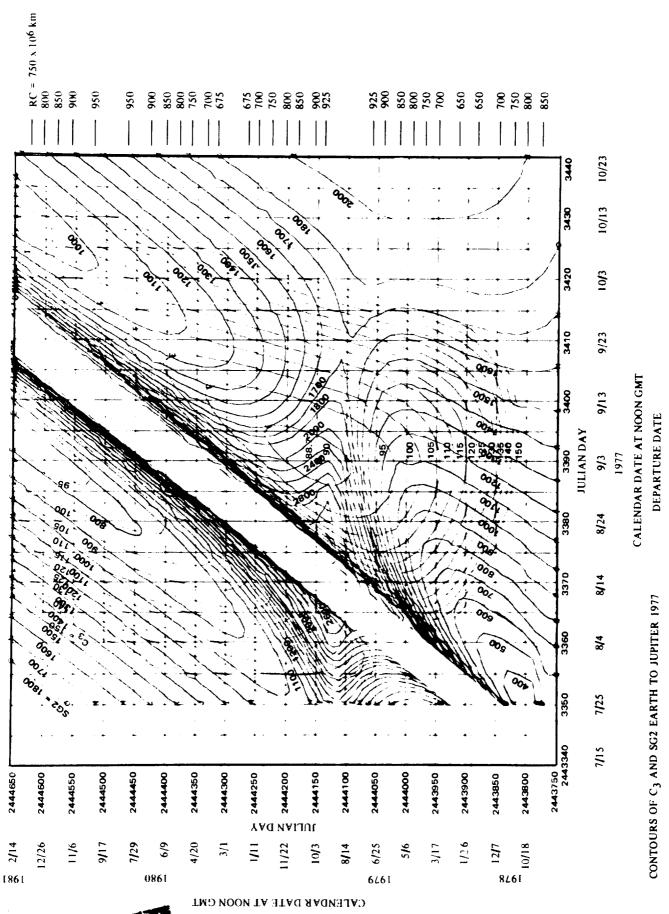
ORIGINAL PAGE IS OF POOR QUALITY

11-4 VBBIANT DATE

DEPARTURE DATE

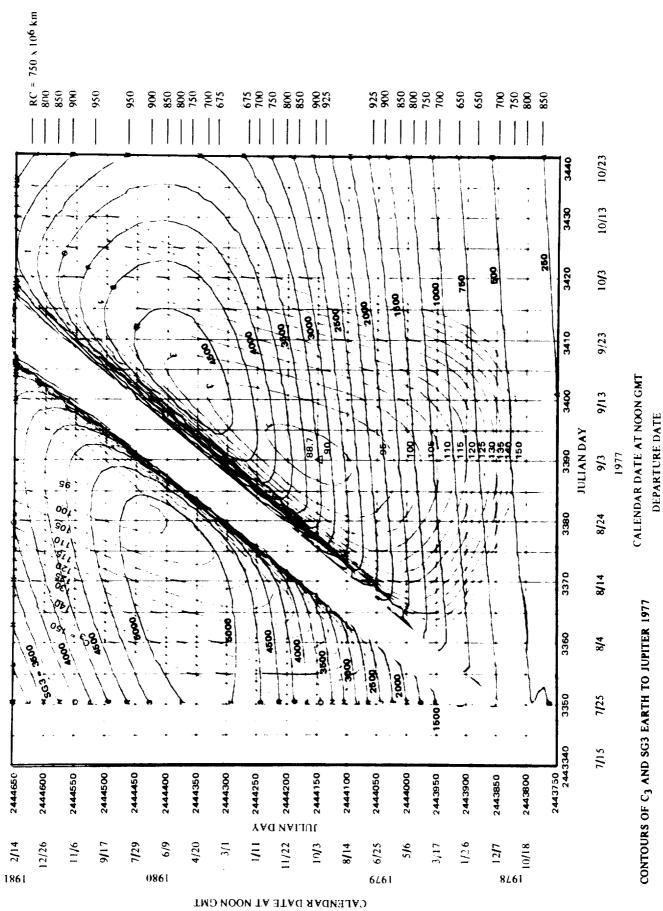


CALENDAR DATE AT NOON GMT ARRIVAL DATE 4-12

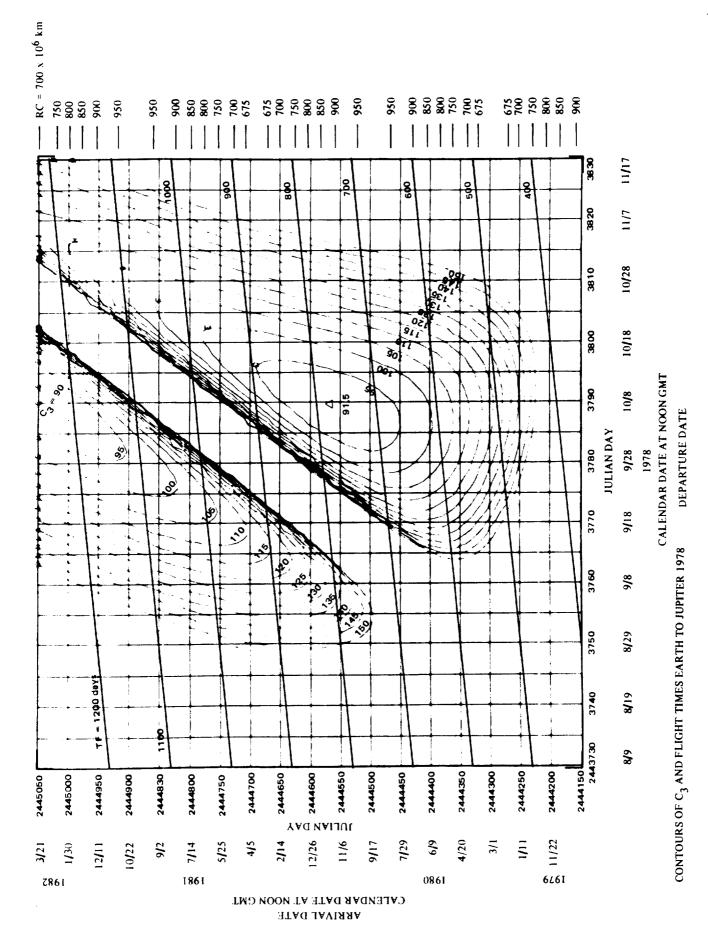


OFIGINAL PAGE IS

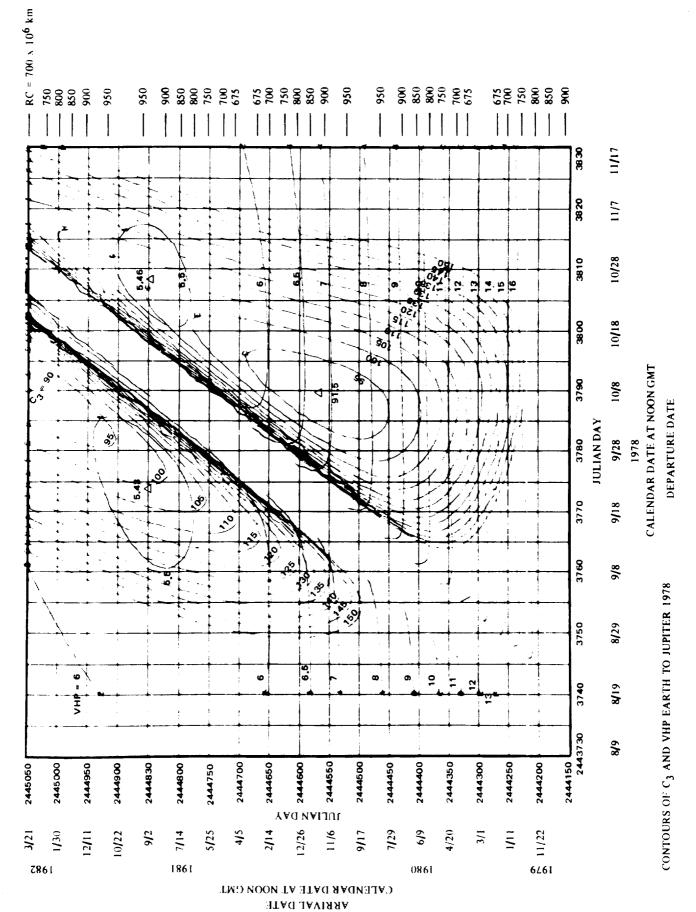
ARRIVAL DATE SALENDAR DATE TAGON C



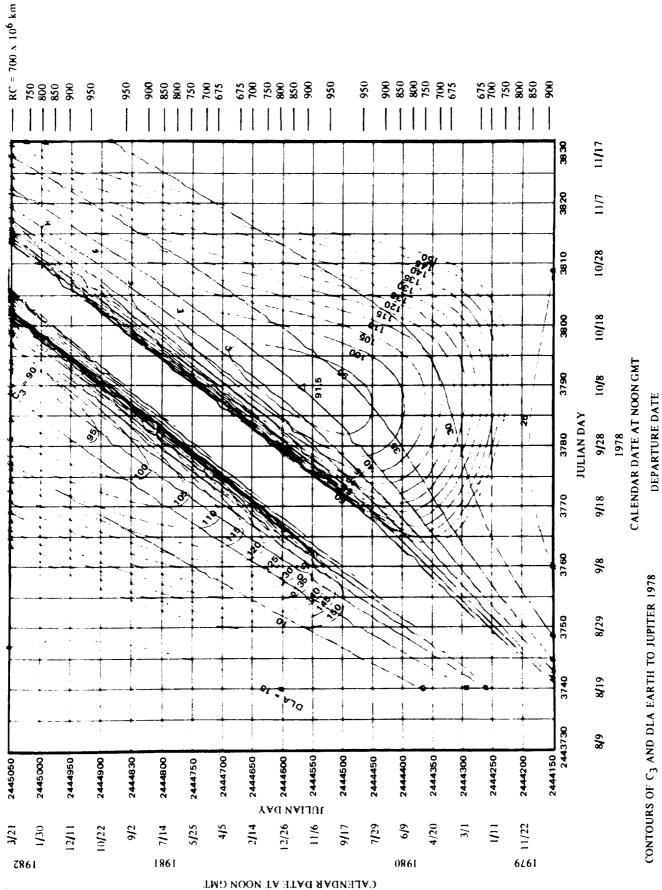
ARRIVAL DATE



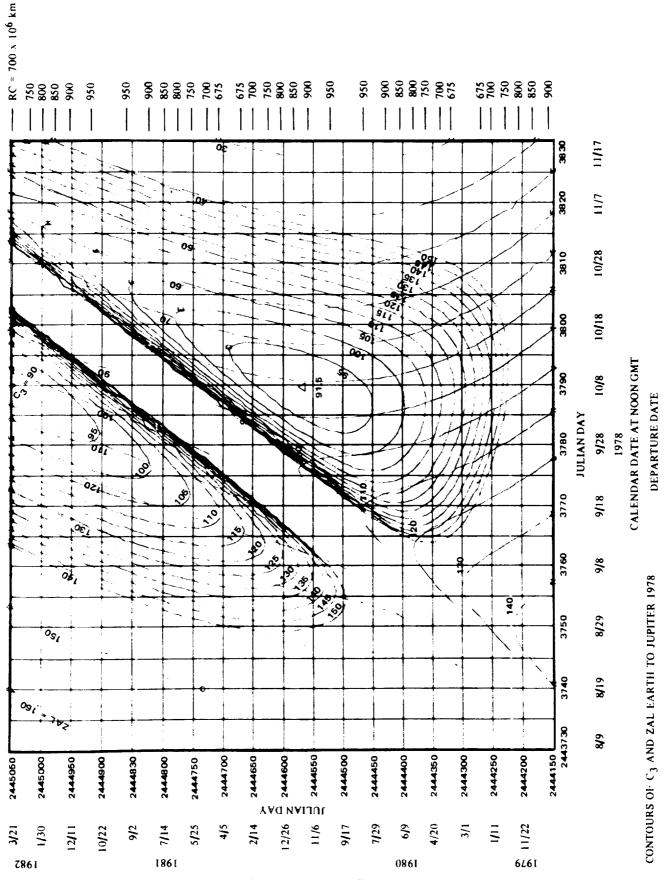
4-15



4-16

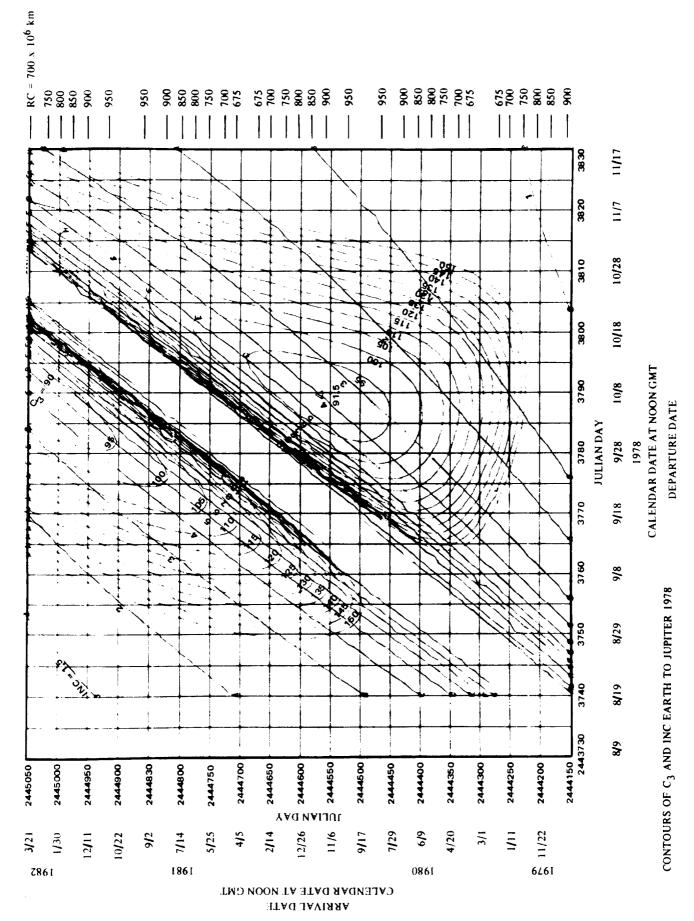


ARRIVAL DATE Cal budar date at uoon gmi

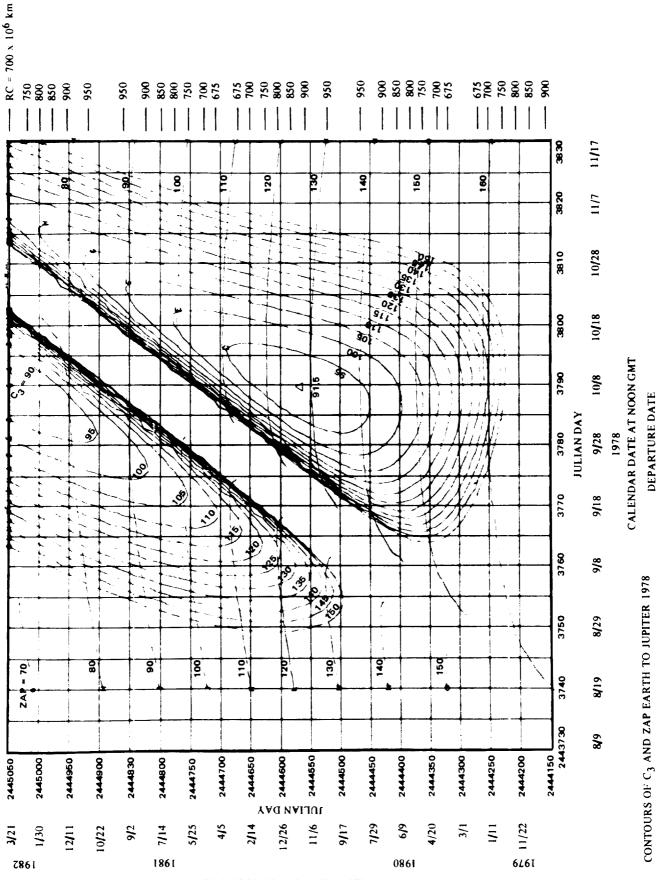


ARRIVAL DATE

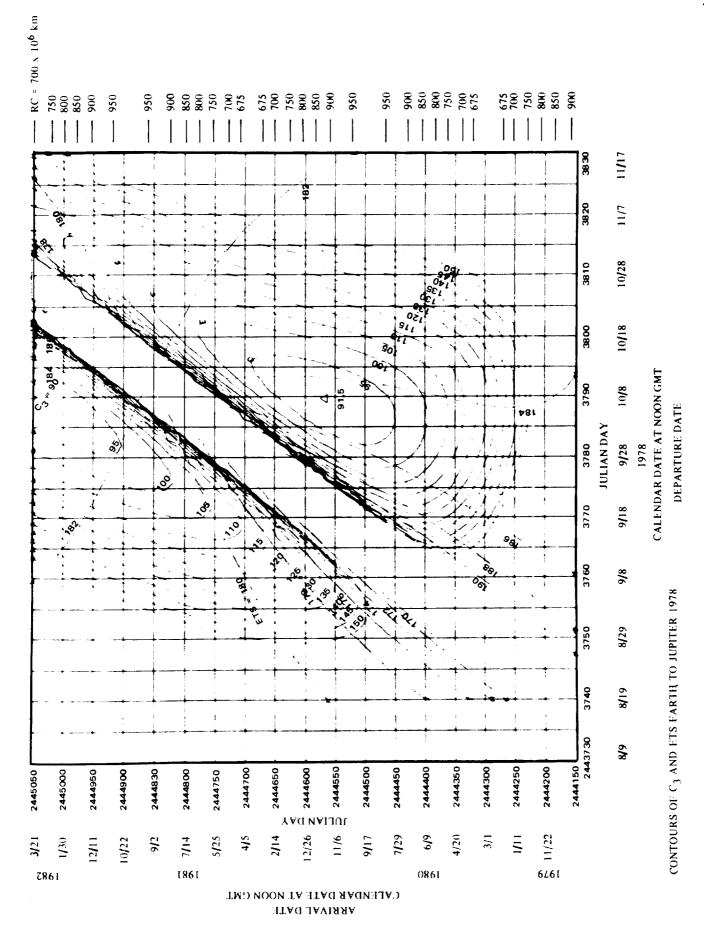
CALENDAR DATE AT NOON GMT

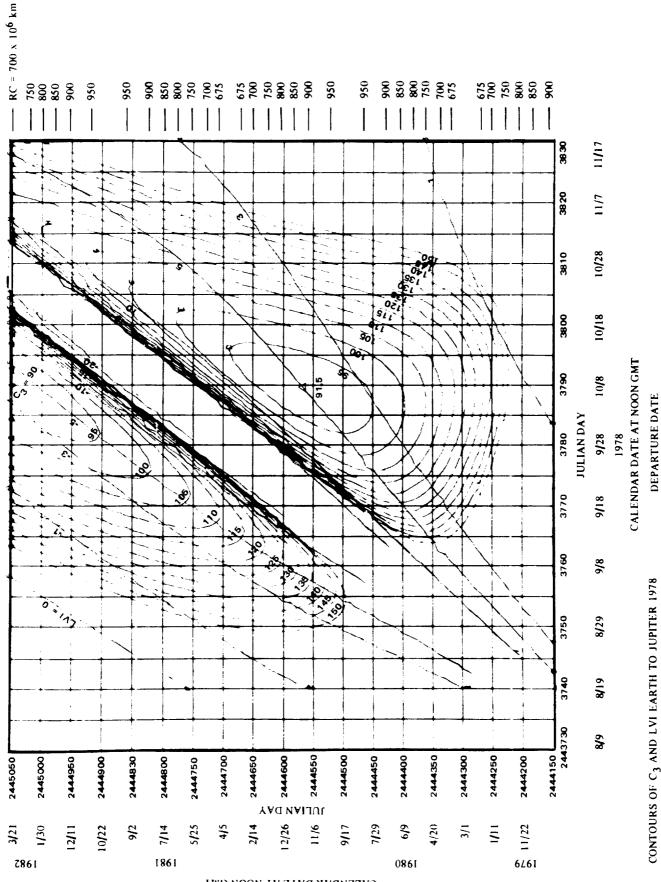


4-19

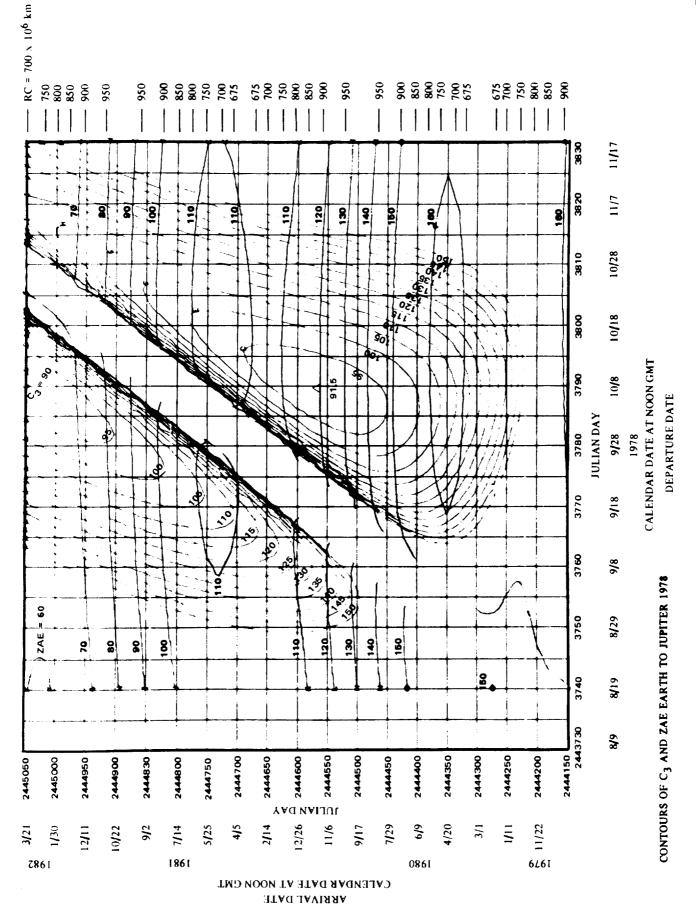


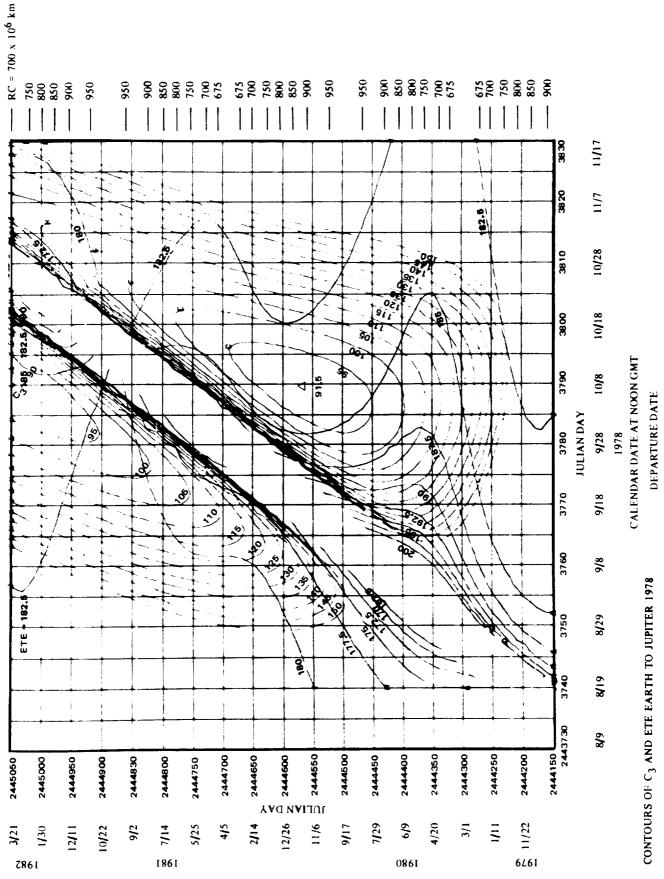
ARRIVAL DATE CALENDAR DATE AT NOON GMT





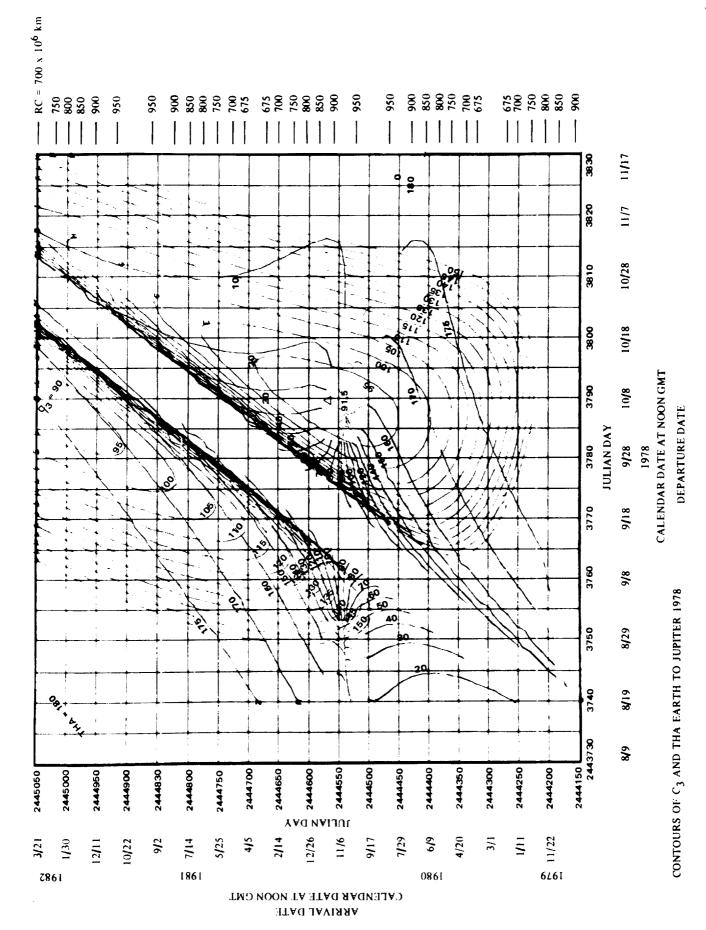
ARRIVAL DATE CALENDAR DATE

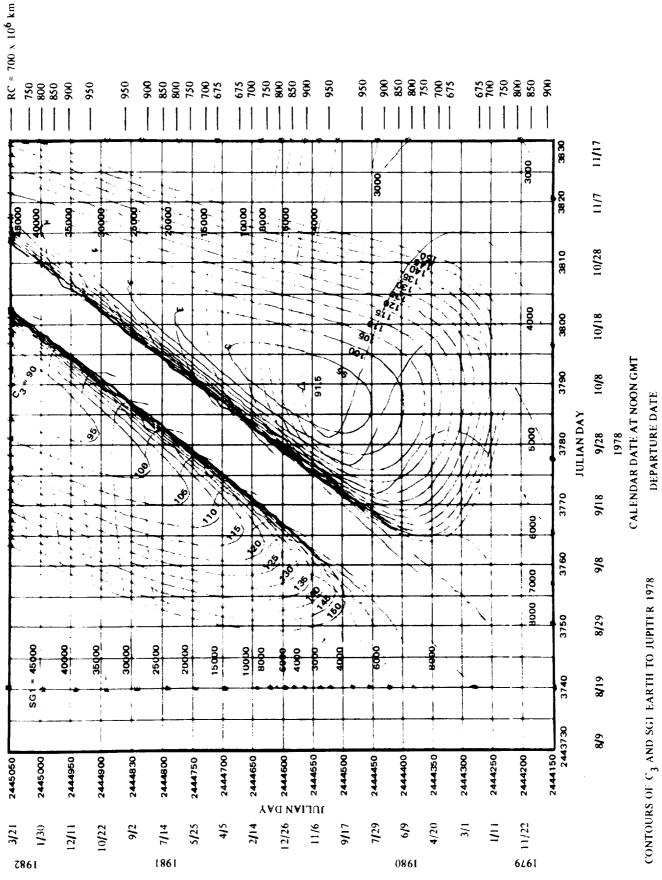




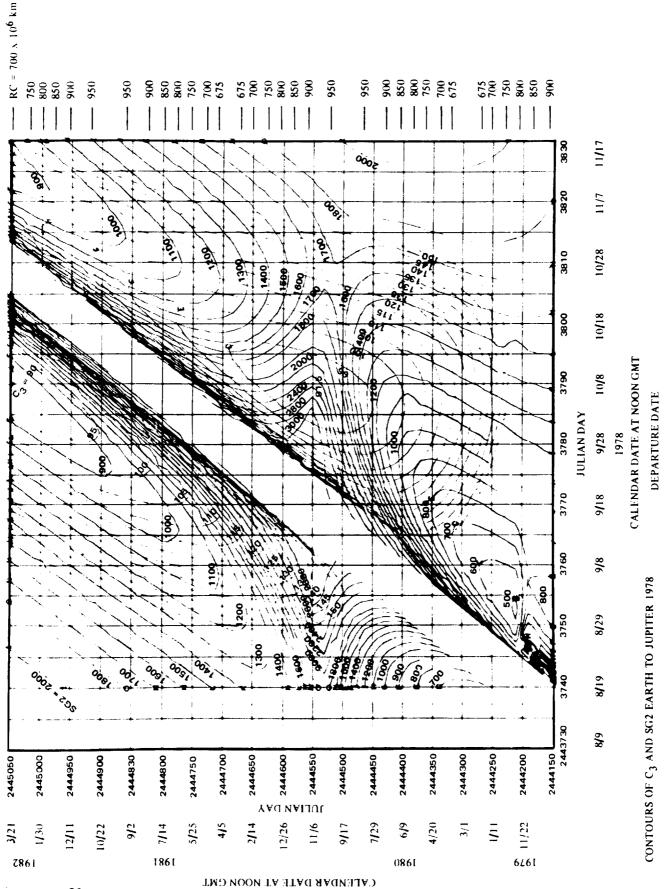
ARRIVAL DATE

CALENDAR DATE AT NOON GMT





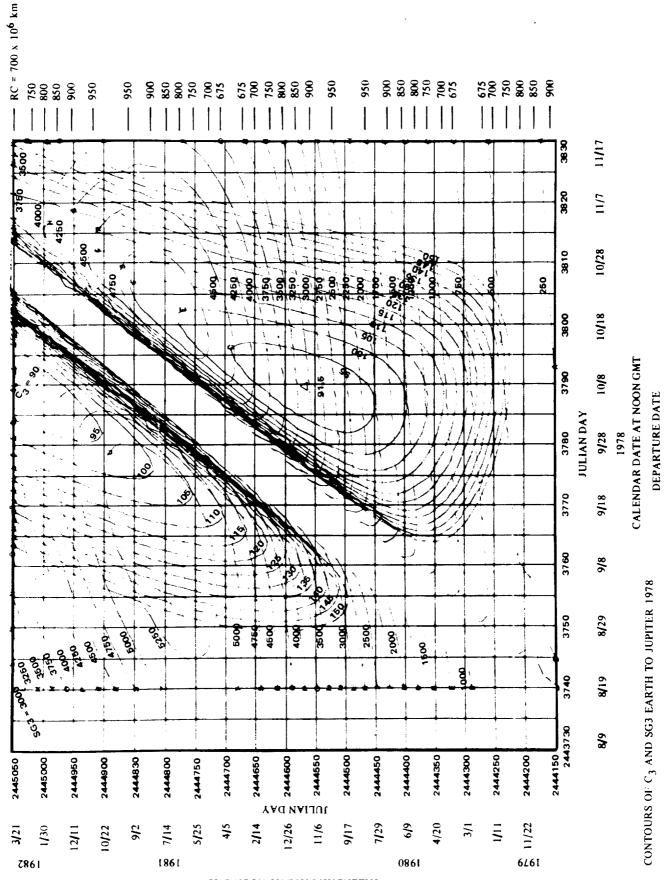
ARRIVAL DATE CALENDAR DATE AT NOON GMT



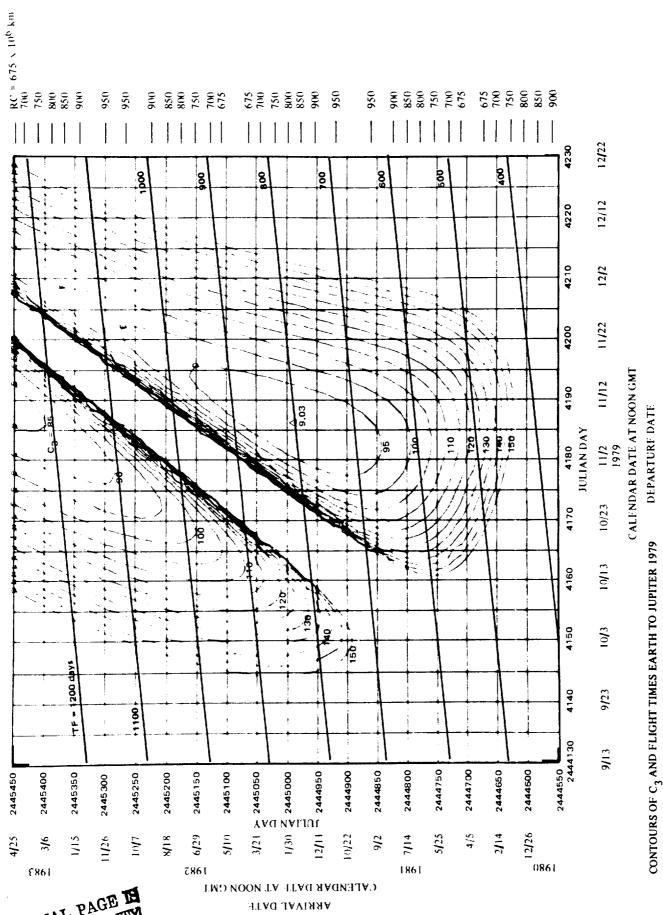
PARIVAL DATE: THE TANK TO THE

ORIGINAL PAGE IN OF POOR QUALITY

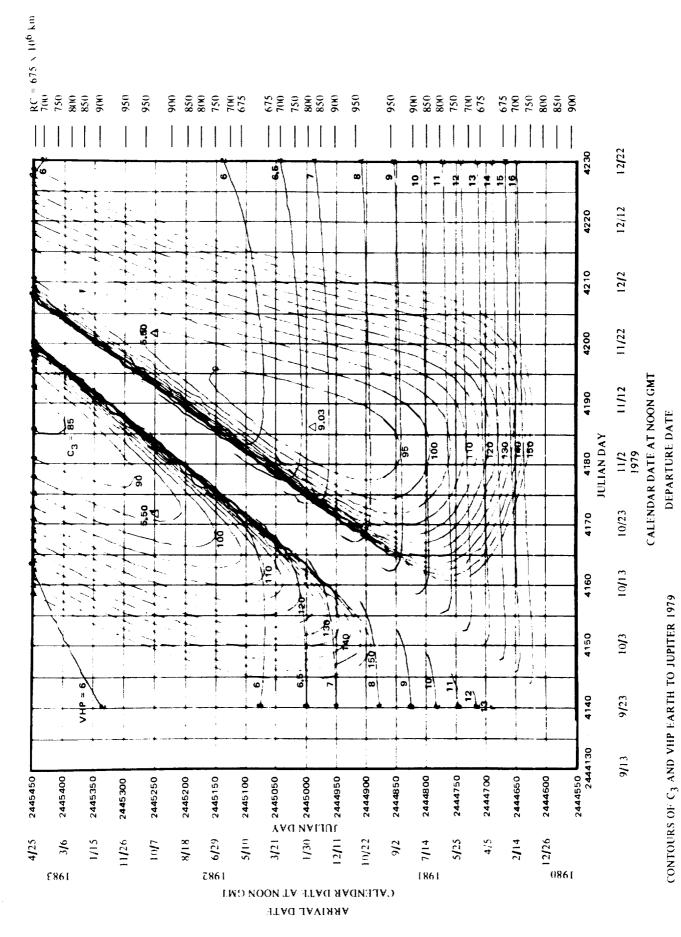
4-27



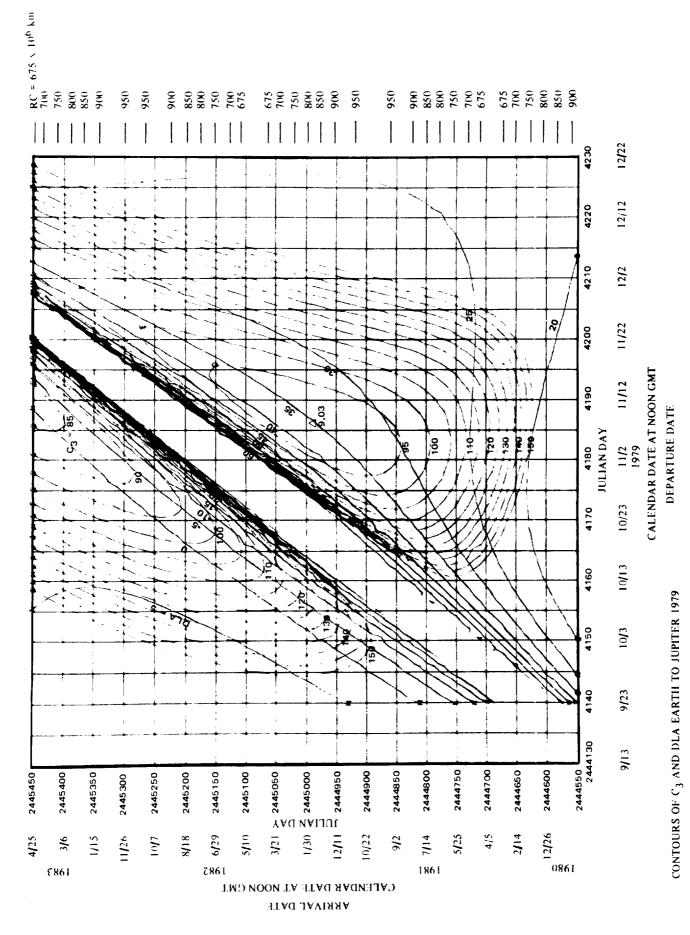
ARRIVAL DATE
CALENDAR DATE AT NOON GMT



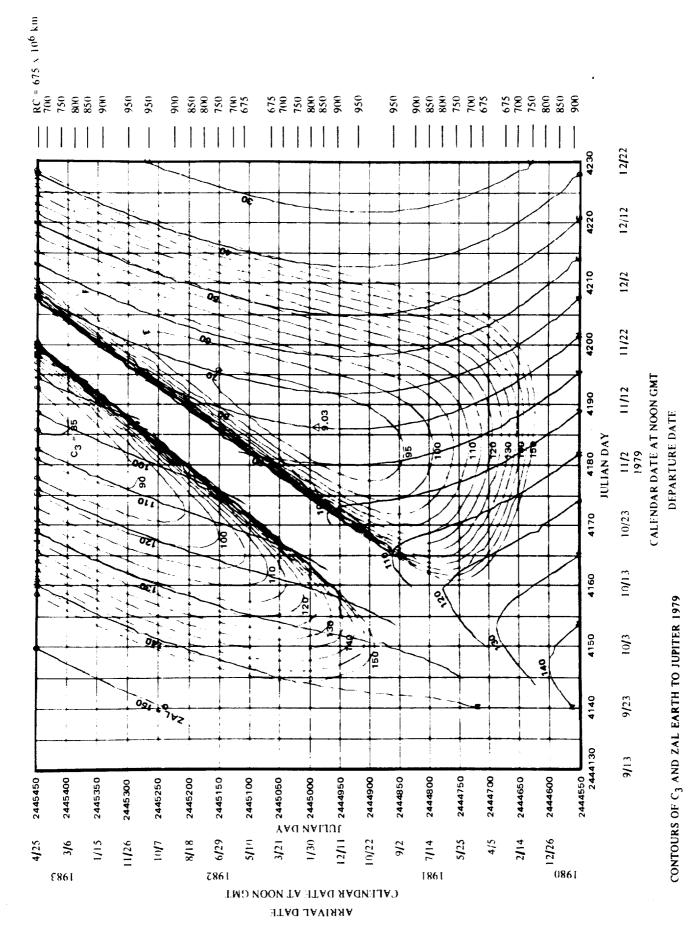
ORIGINAL PAGE IS



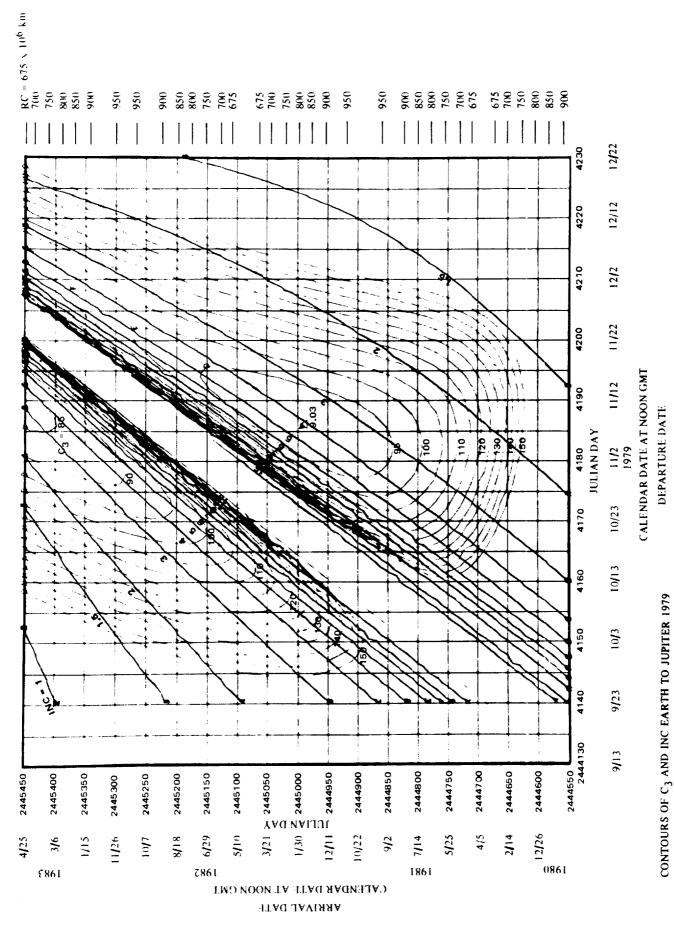
4-30



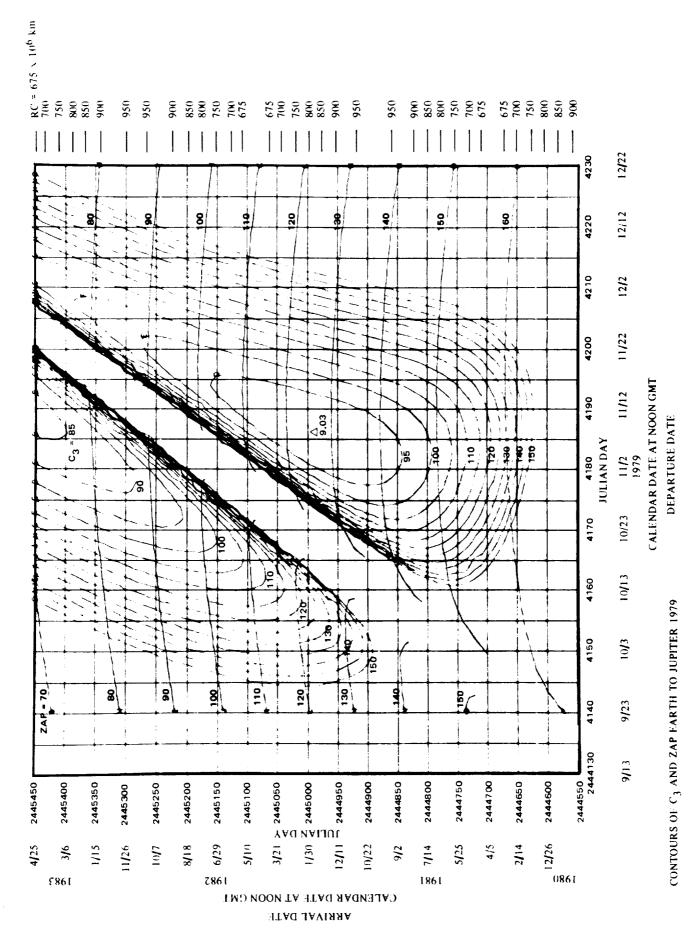
4-31



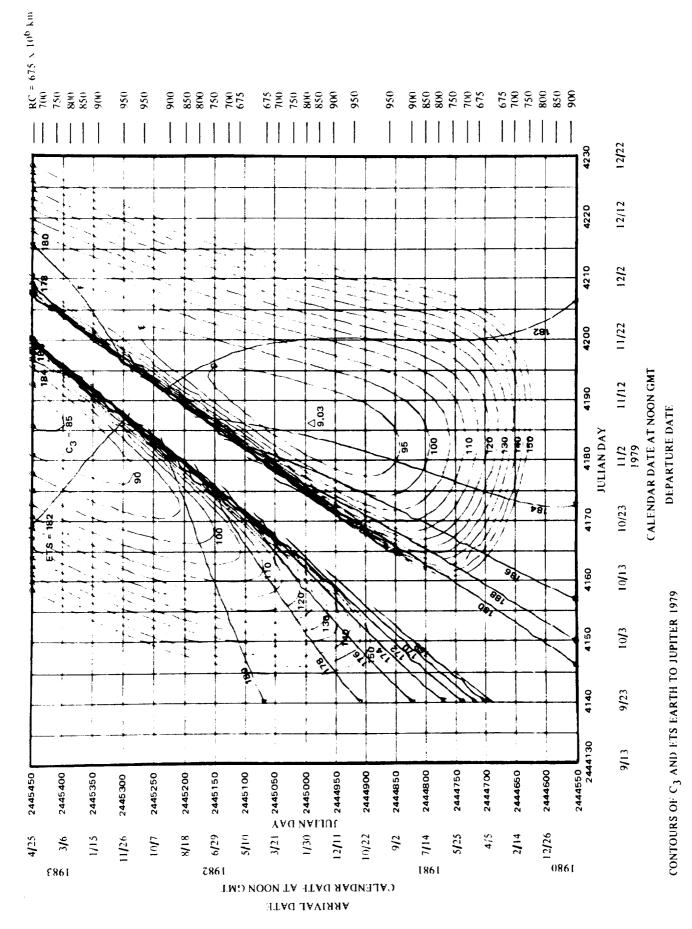
4-32



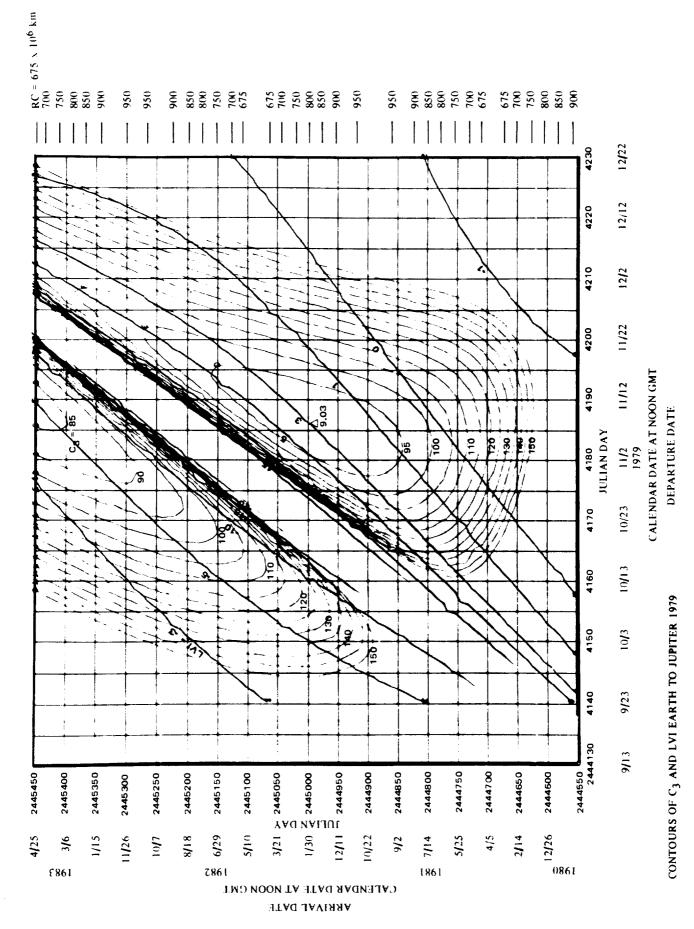
4-33

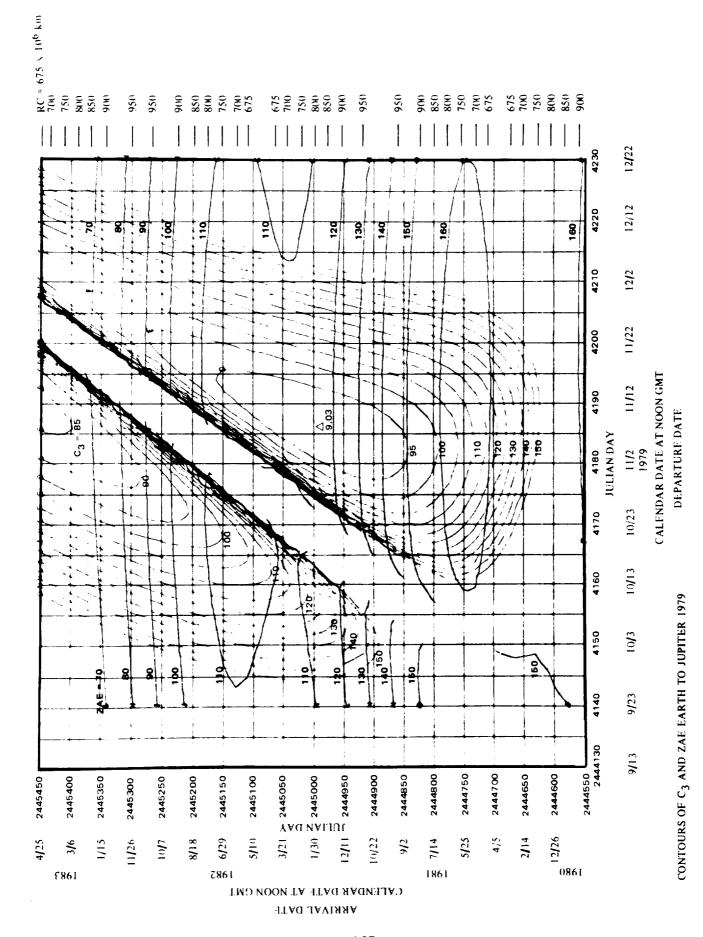


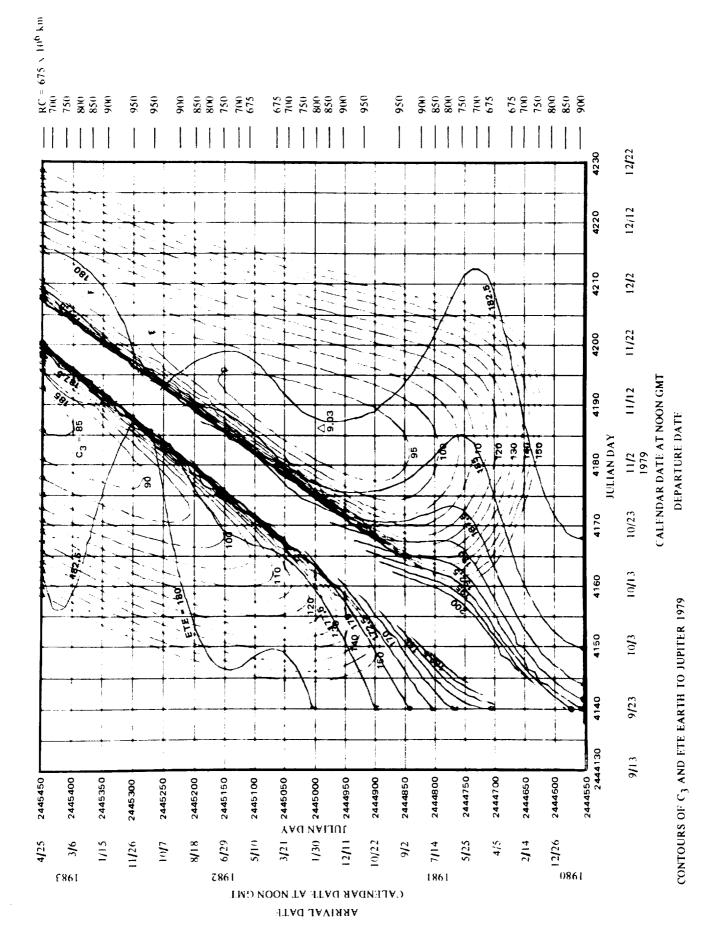
4-34



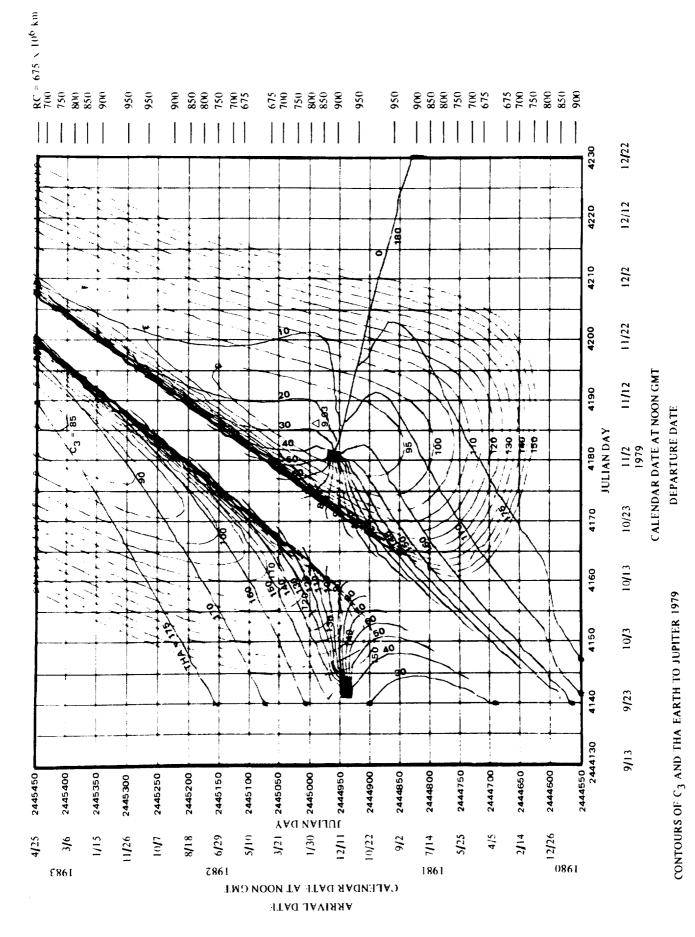
4-35

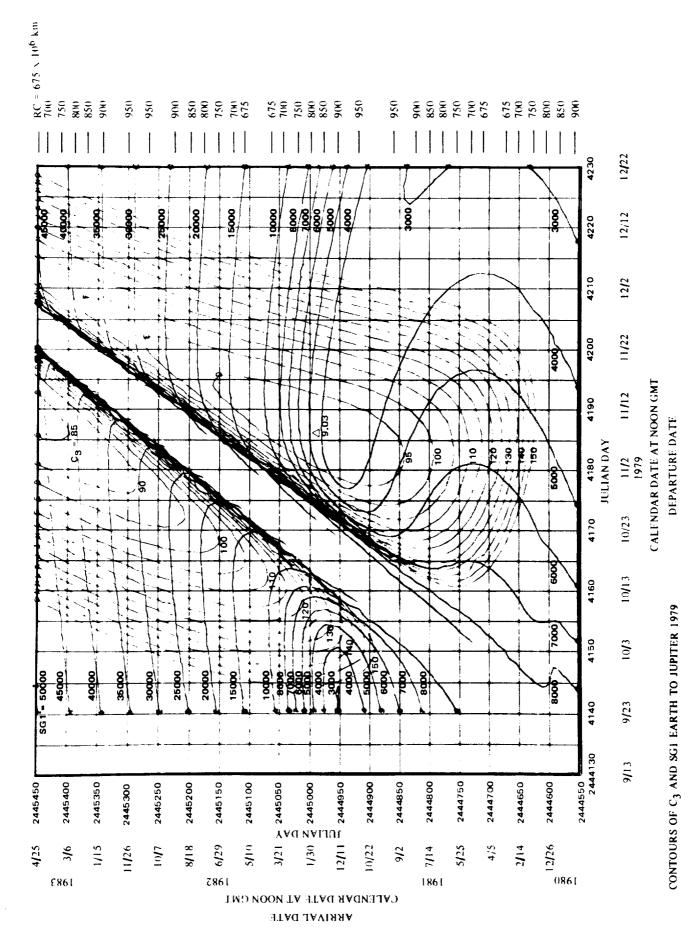




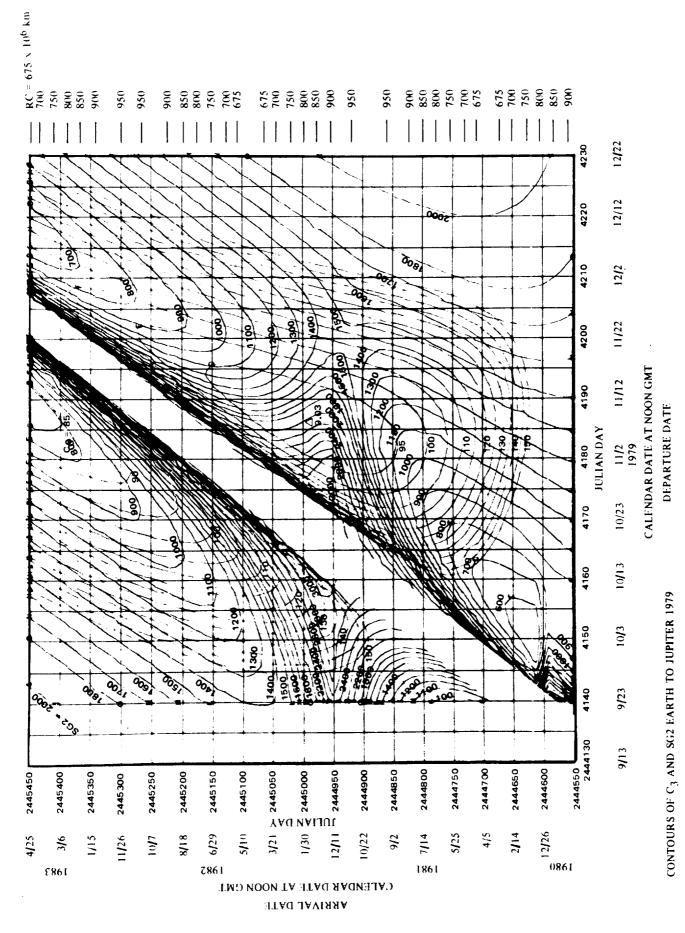


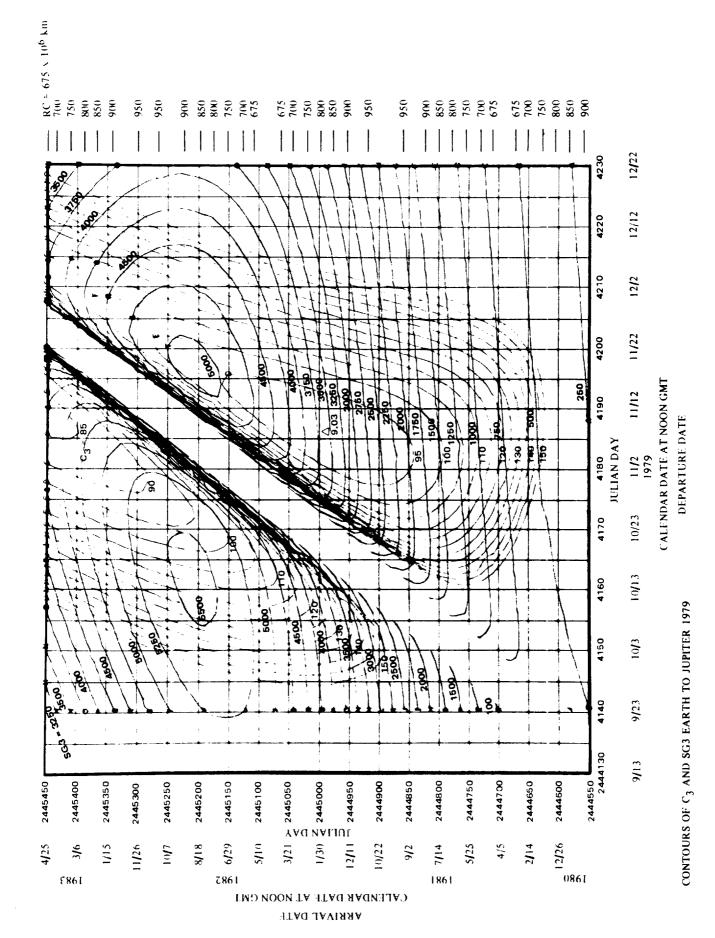
4-38



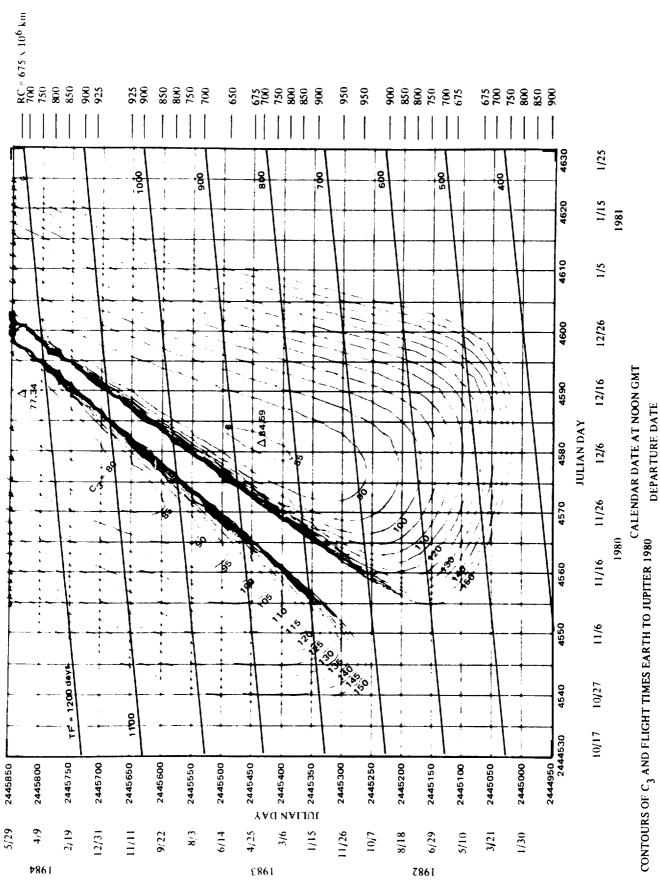


4-40



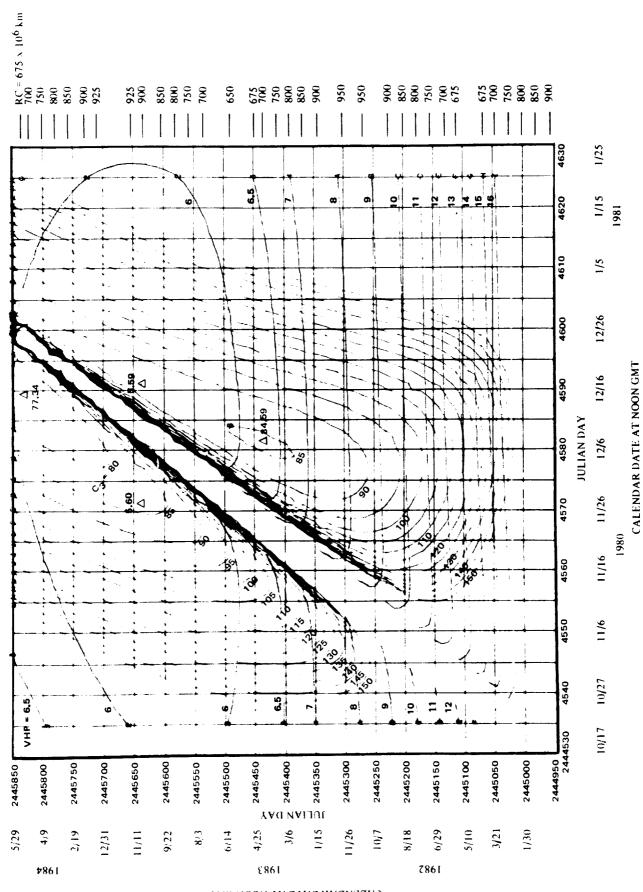


4-42



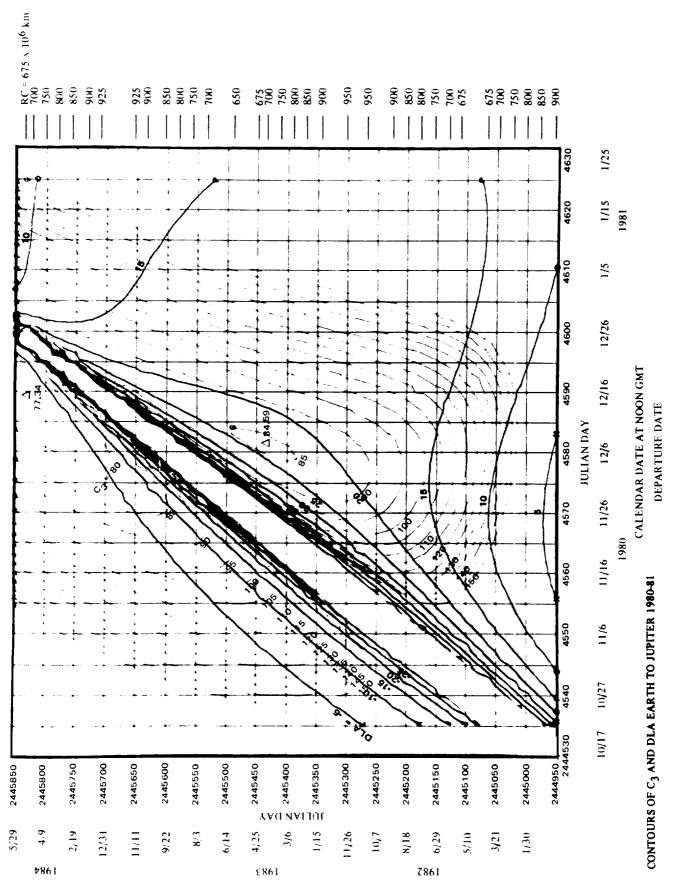
CALENDAR DATE AT NOON GMT

CONTOURS OF C_3 AND VHP EARTH TO JUPITER 1980-81



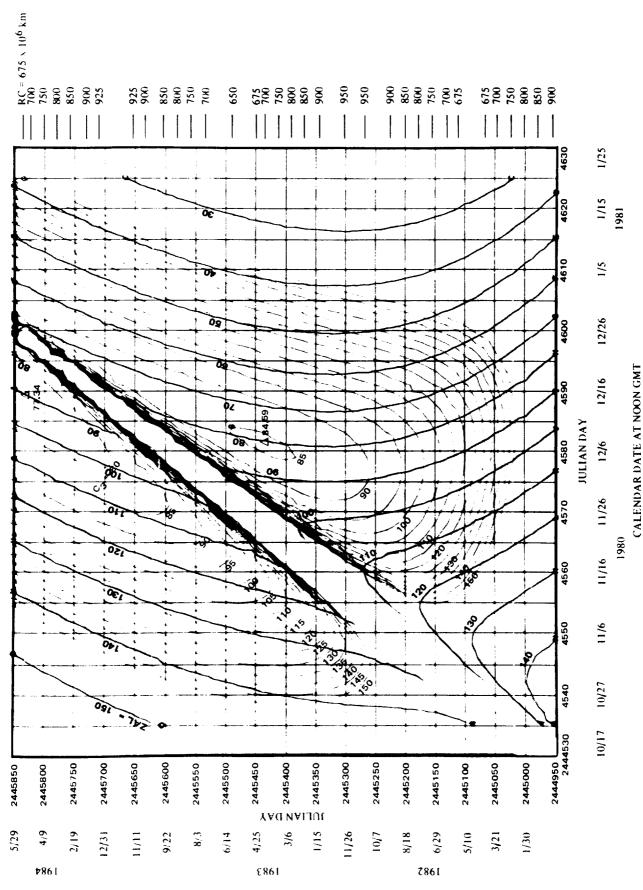
ARRIVAL DATE

CALENDAR DATE AT NOON GMT

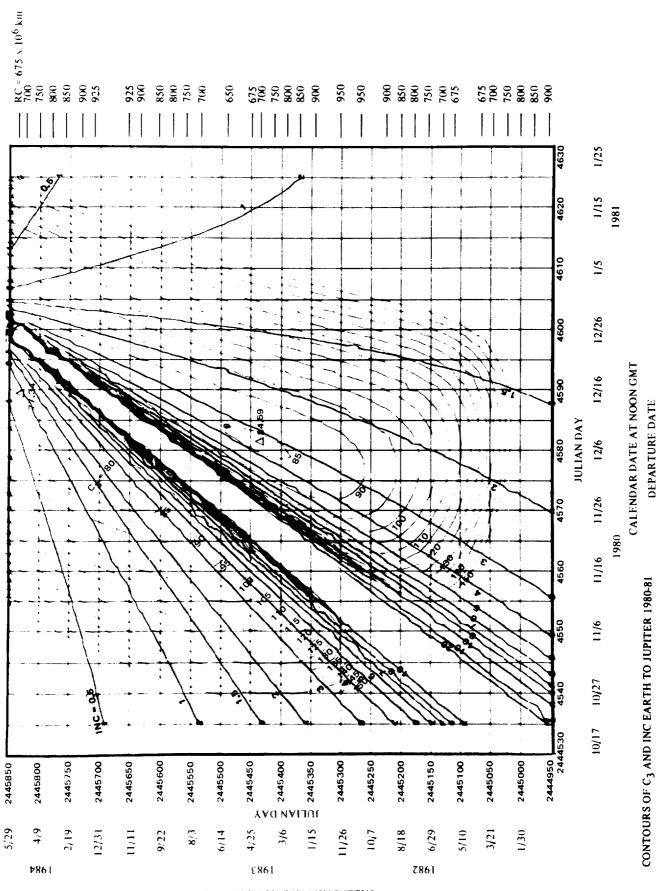


CALENDAR DATE AT NOON GMT

CONTOURS OF C₃ AND ZAL EARTH TO JUPITER 1980-81



CALENDAR DATE
ARRIVAL DATE
OPENDAR DATE

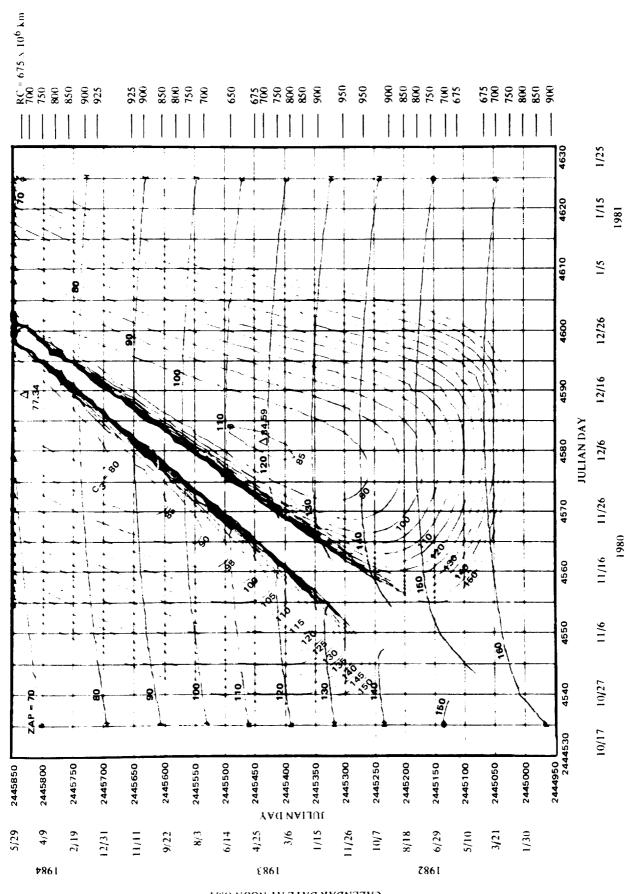


CALENDAR DATE AT NOON GMT

CALENDAR DATE AT NOON GMT

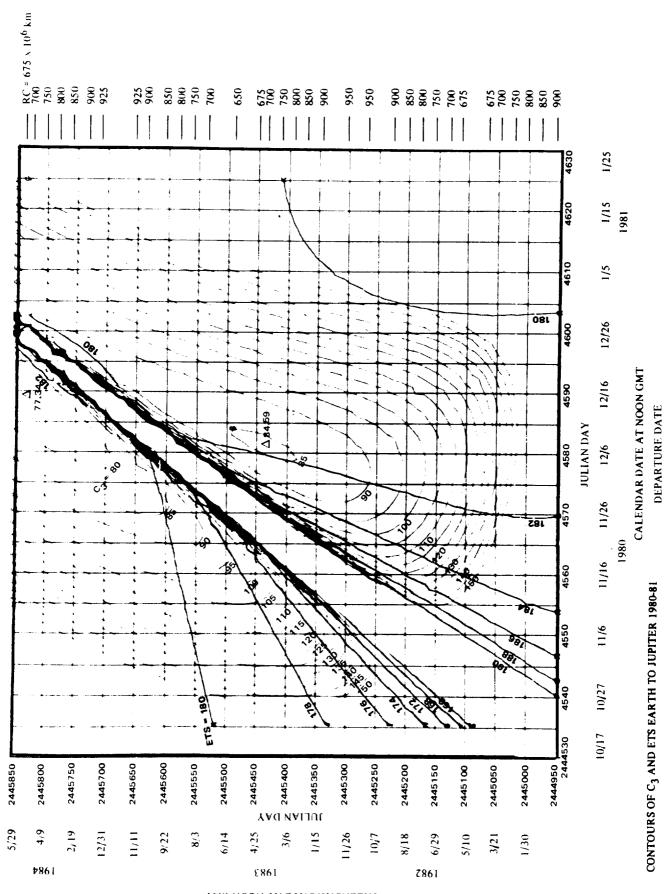
DEPARTURE DATE

CONTOURS OF C_3 and Zap earth to Jupiter 1980-81



ARRIVAL DATE

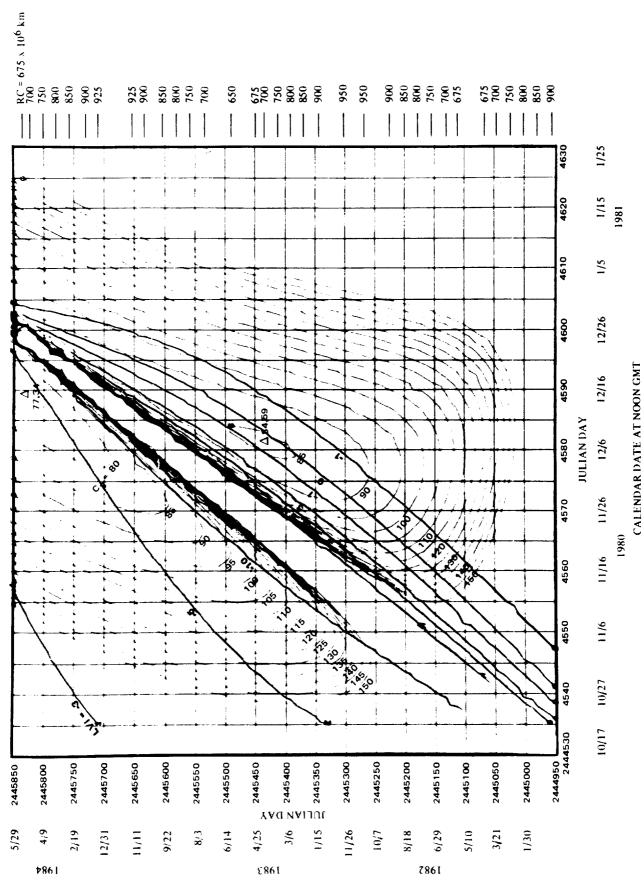
CALENDAR DATE AT NOON GMT



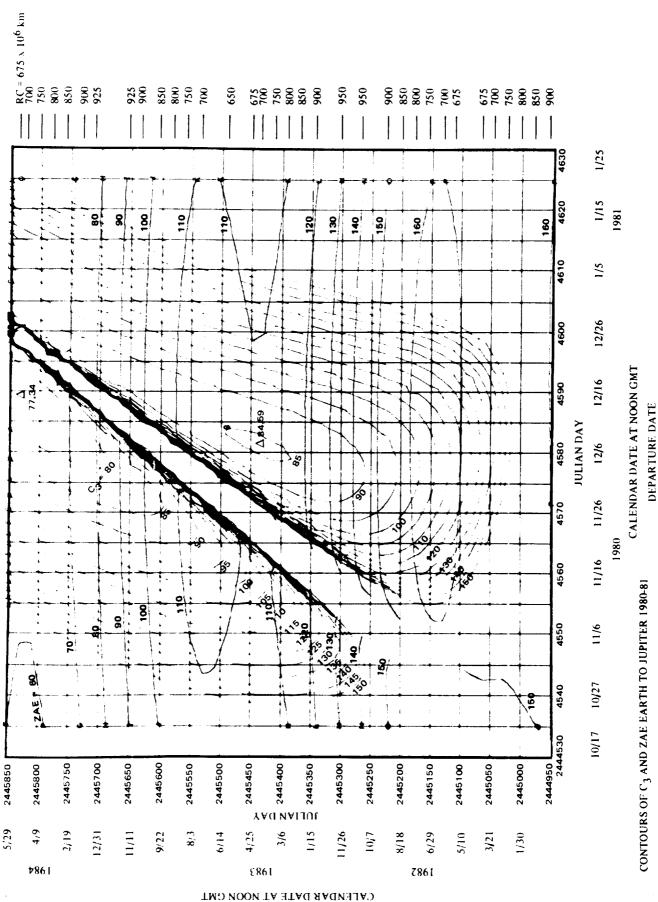
ARRIVAL DATE

CALENDAR DATE AT NOON GMT

CONTOURS OF C₃ AND LVI EARTH TO JUPITER 1980-81

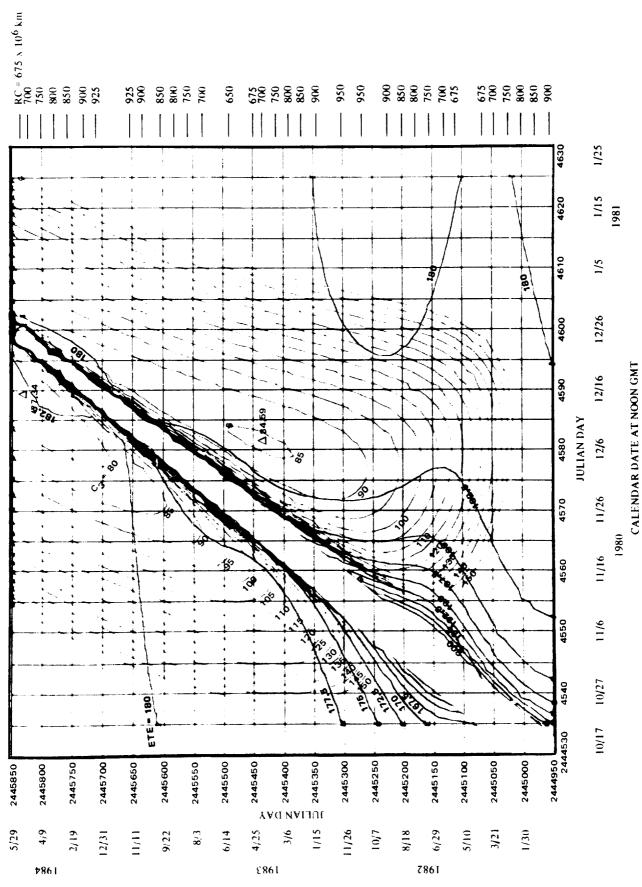


CALENDAR DATE
ARRIVAL DATE
05-4

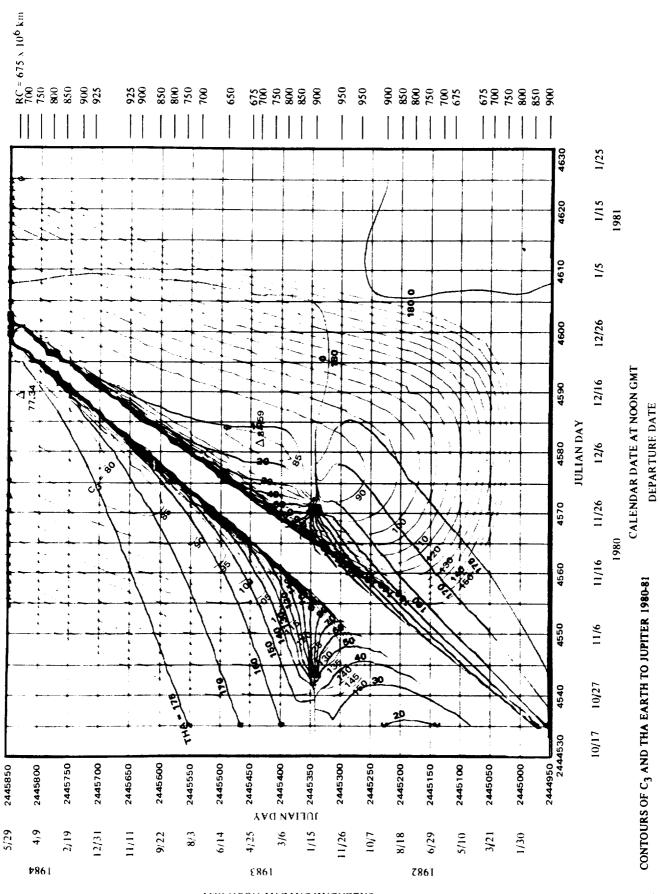


CALENDAR DATE AT NOON GMT

CONTOURS OF C₃ AND ETE EARTH TO JUPITER 1980-81

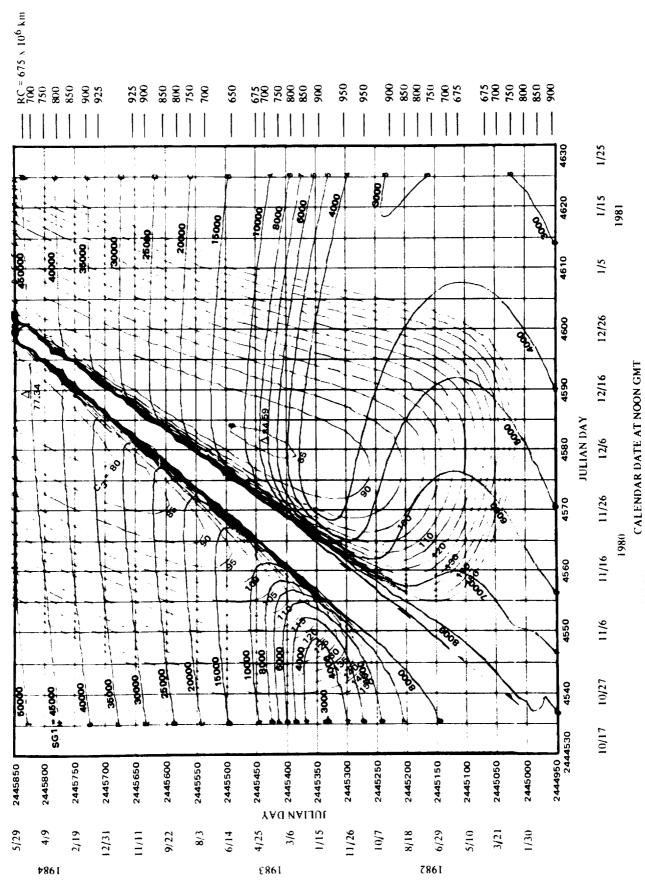


CALENDAR DATE AT NOON GMT

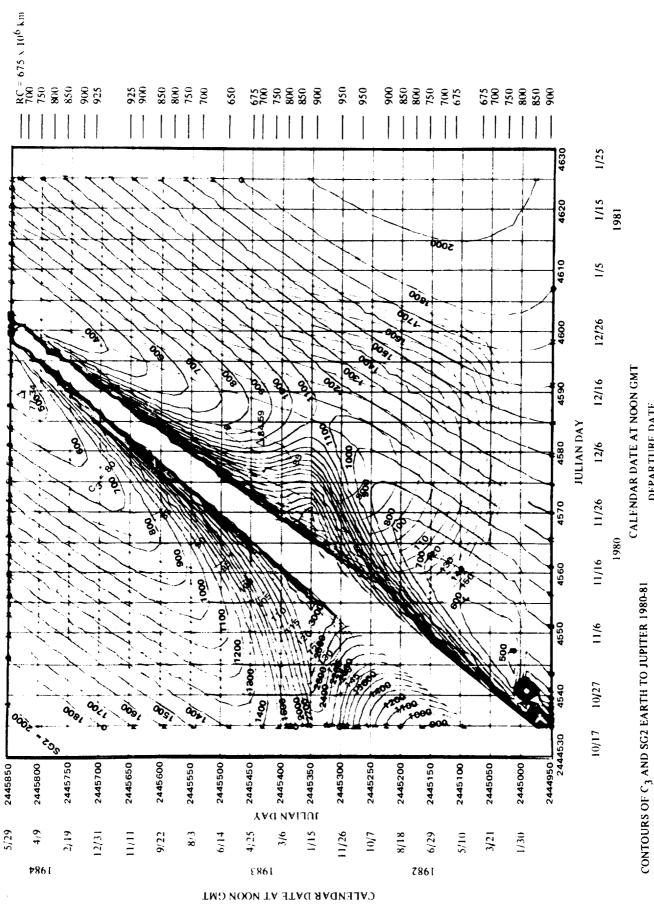


ARRIVAL DATE AT NOON GMT

CONTOURS OF C_3 AND SG1 EARTH TO JUPITER 1980-81

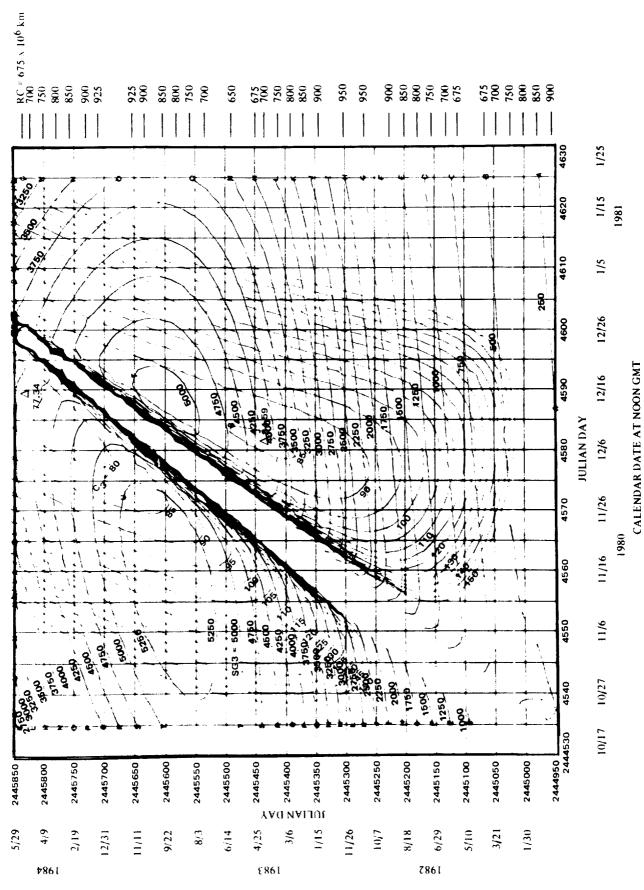


CALENDAR DATE AT NOON GMT

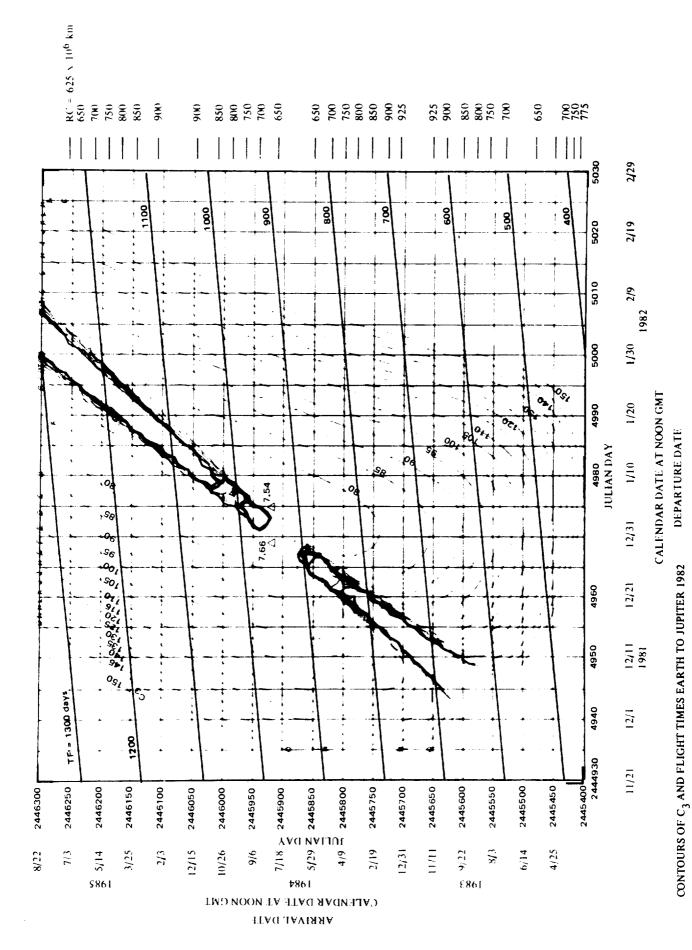


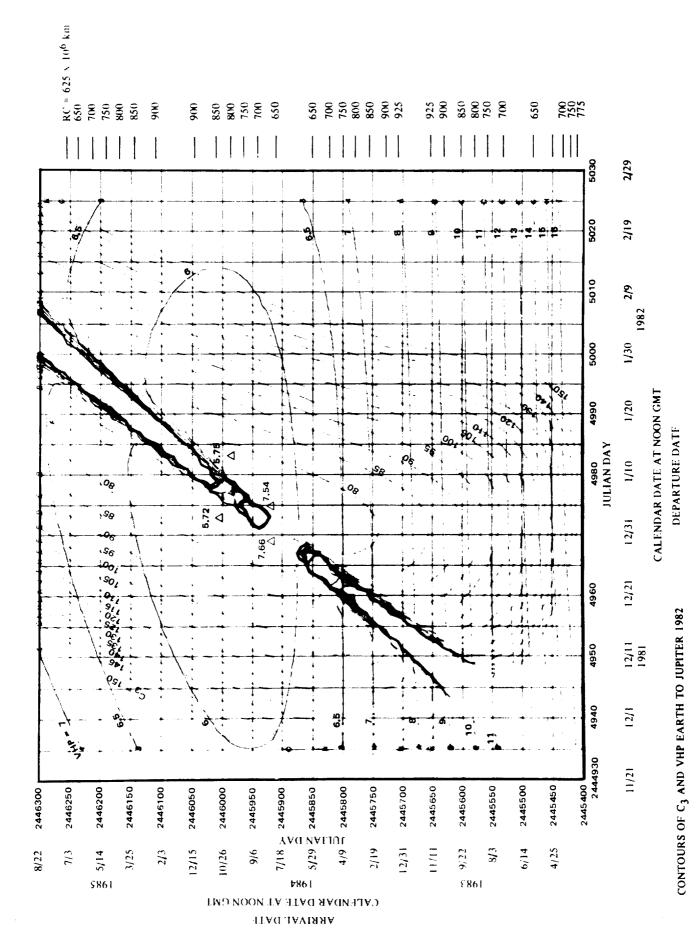
ARRIVAL DATE

CONTOURS OF $\mathbf{C_3}$ AND SG3 EARTH TO JUPITER 1980-81

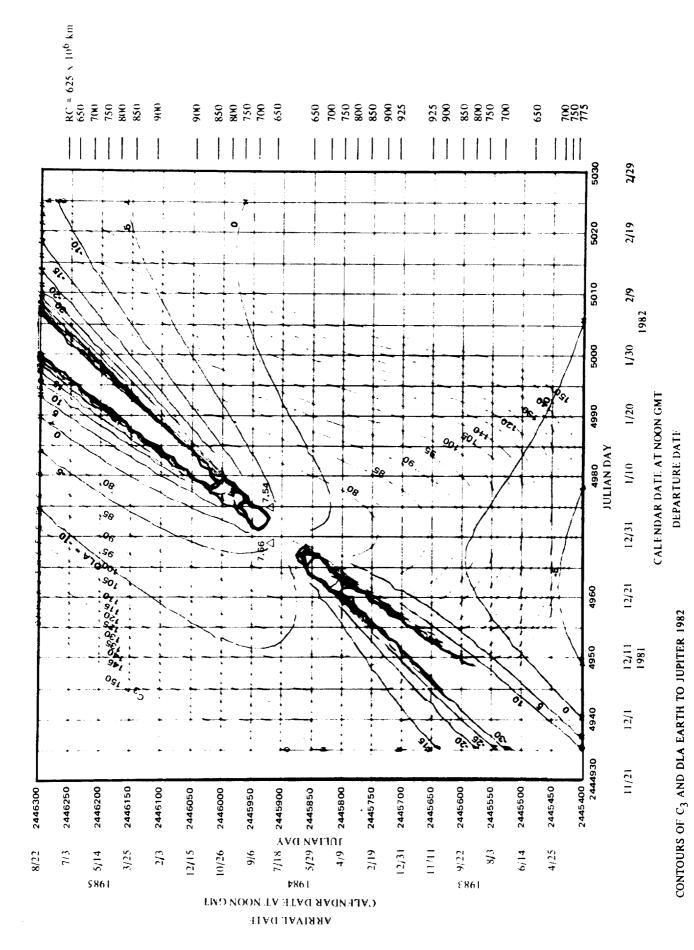


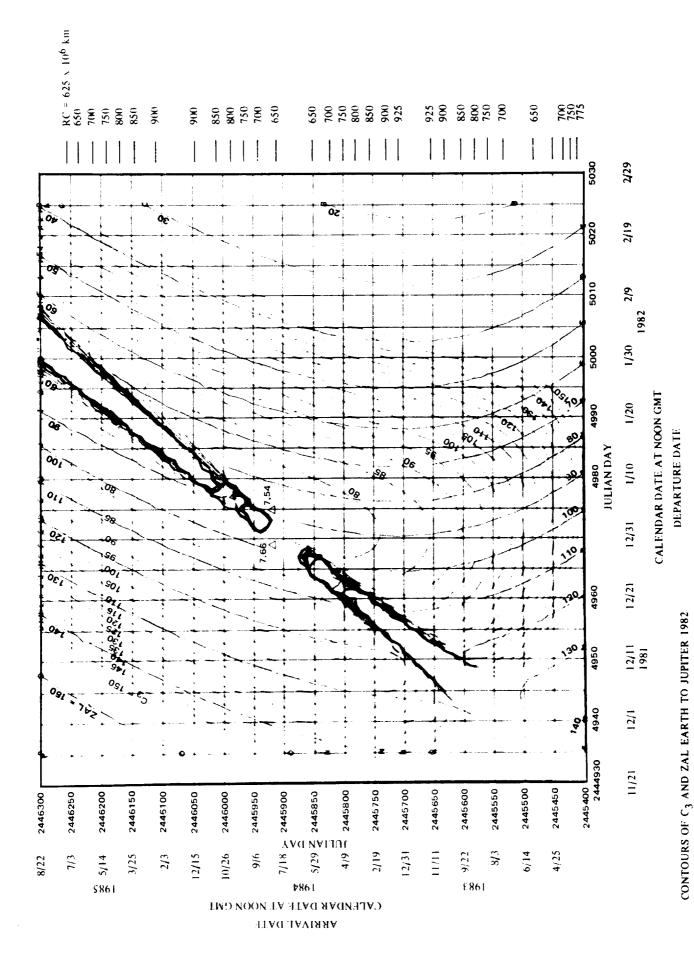
CALEUDAR DATE ARRIVAL DATE 44-7



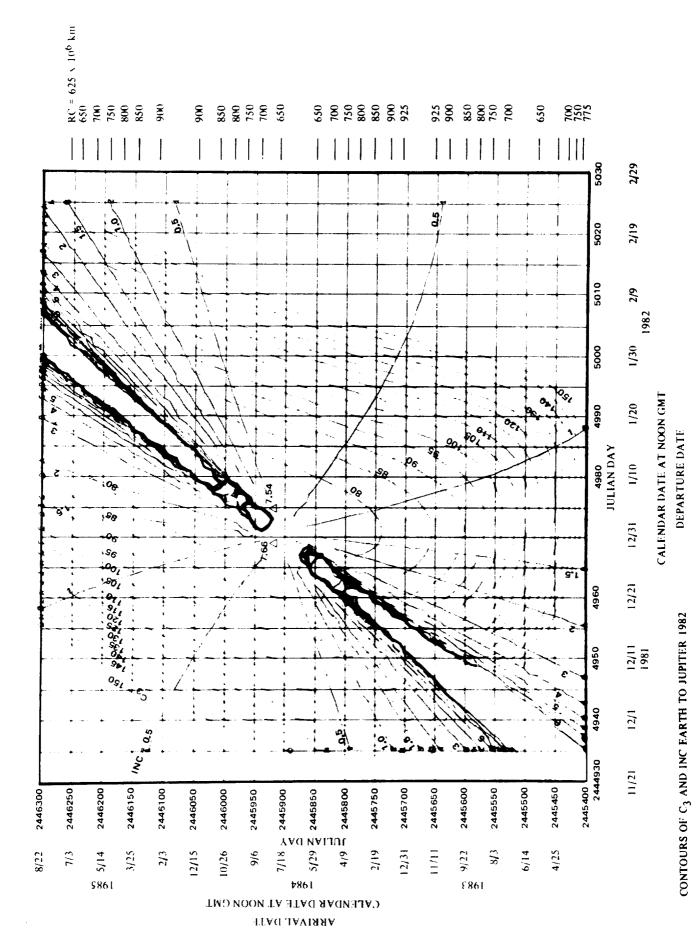


4-58

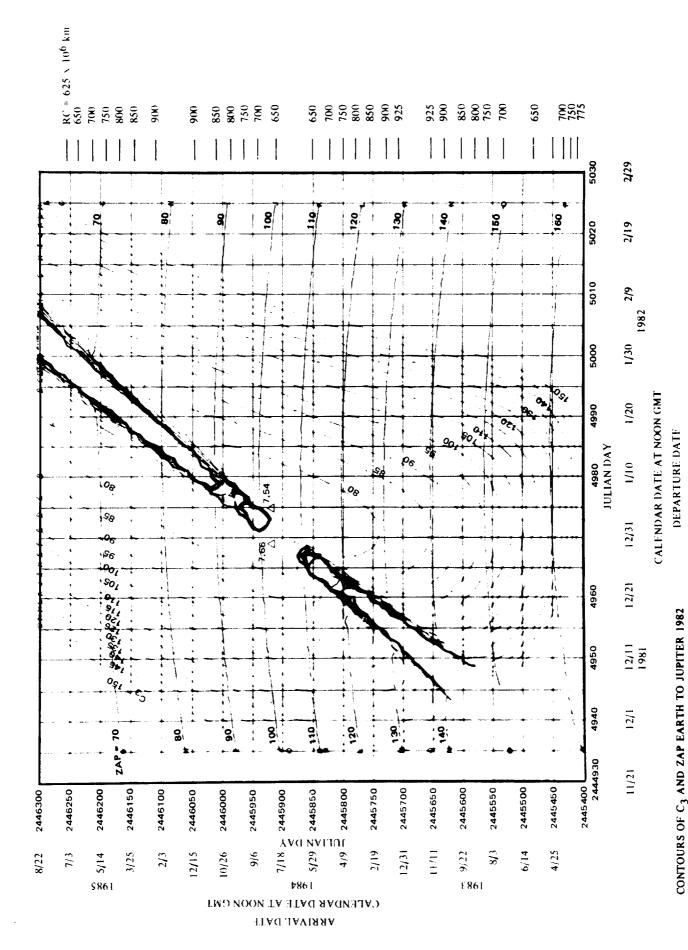




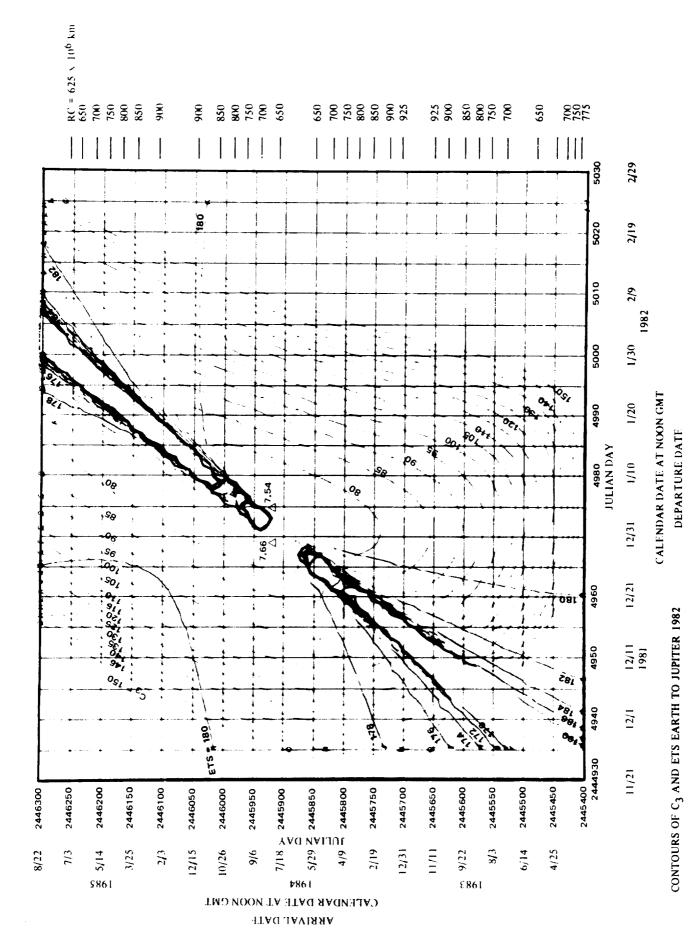
4-60



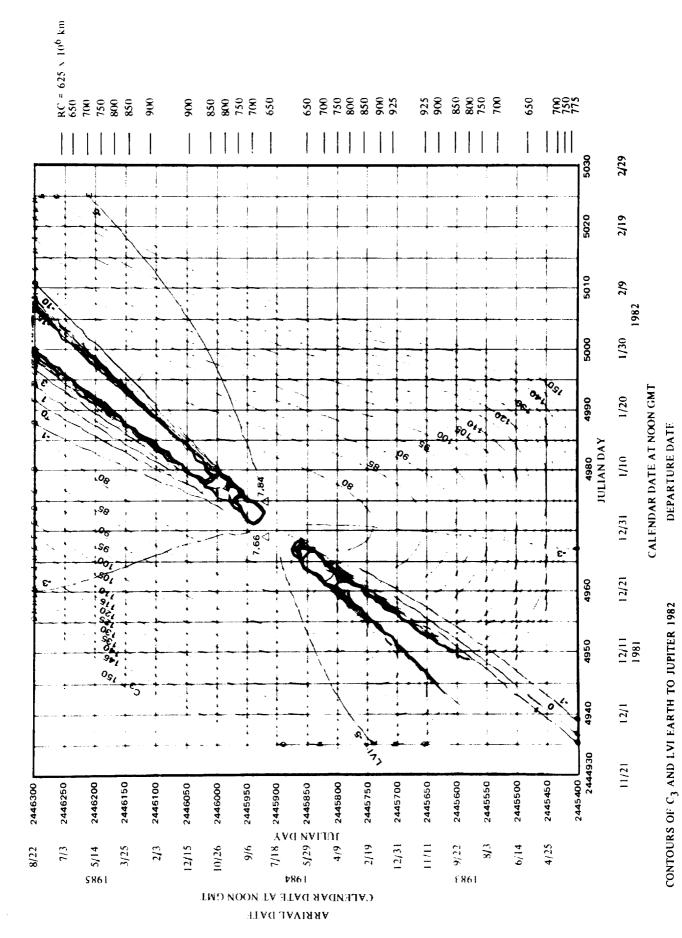
4-61



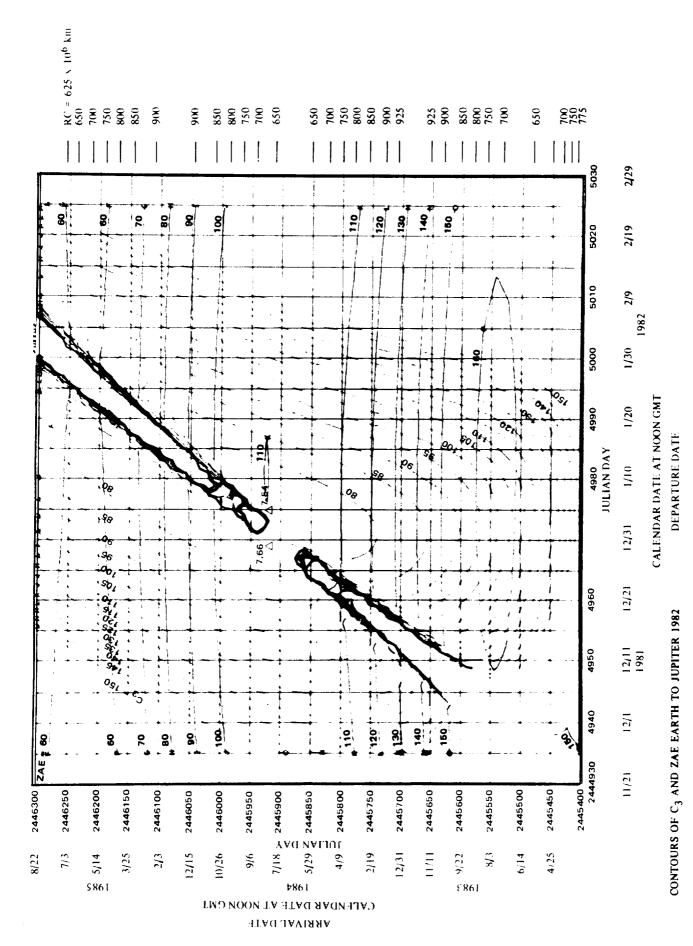
4-62



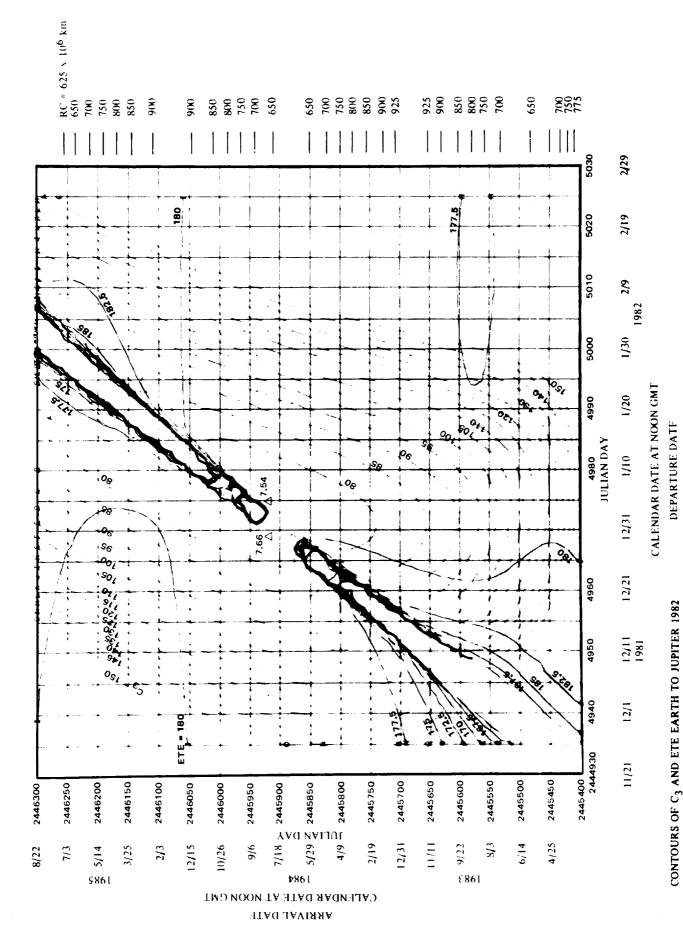
4-63



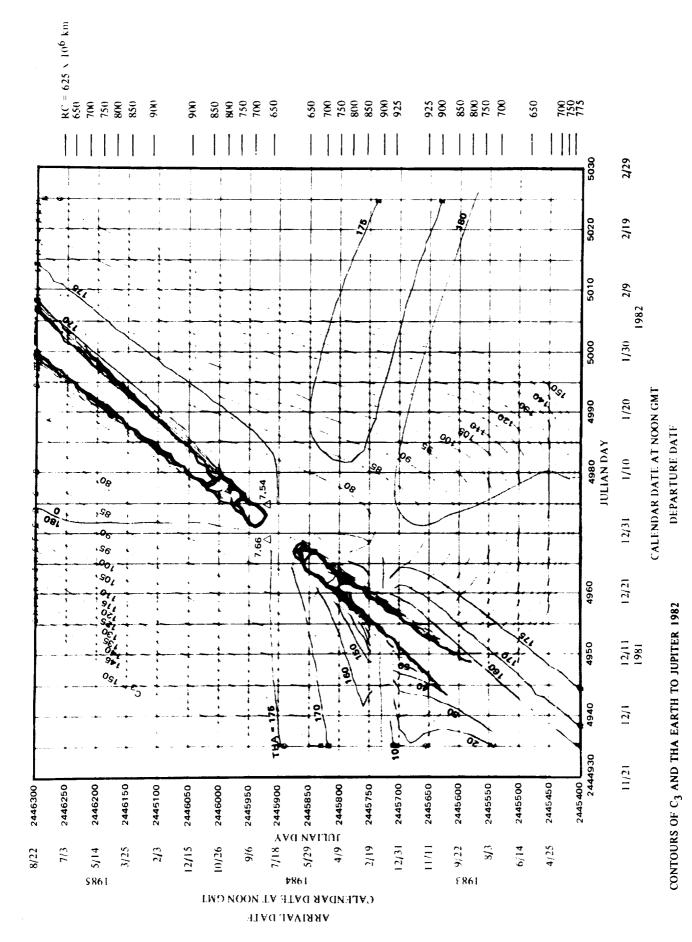
4-64

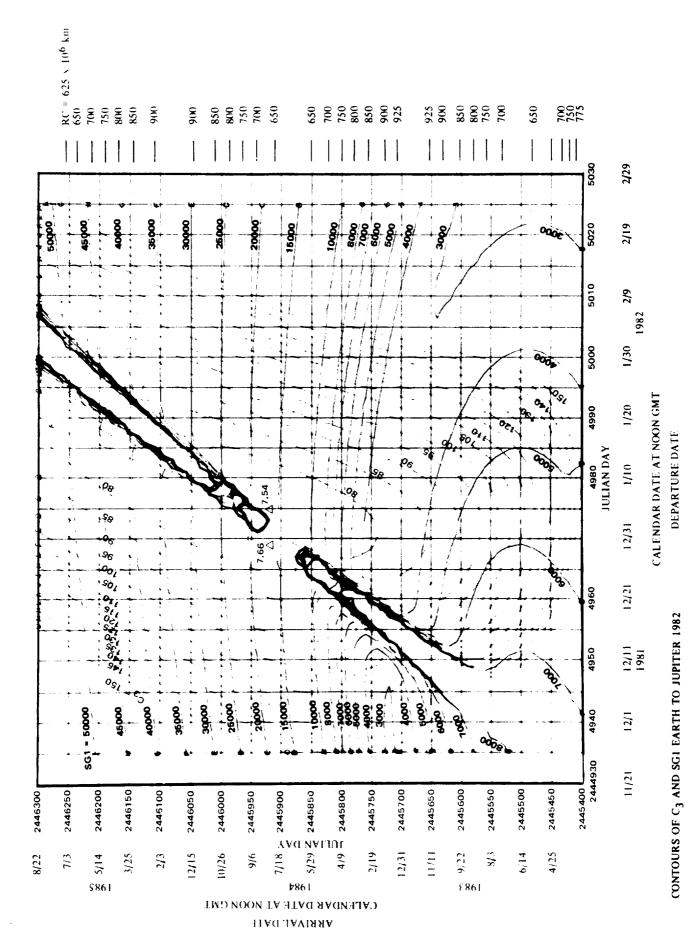


4-65

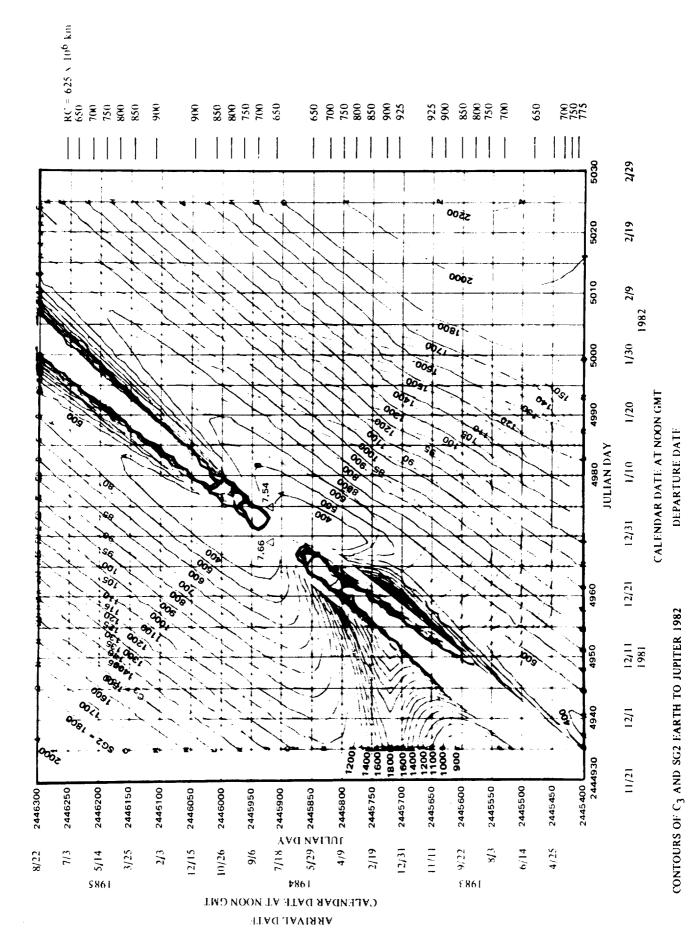


4-66

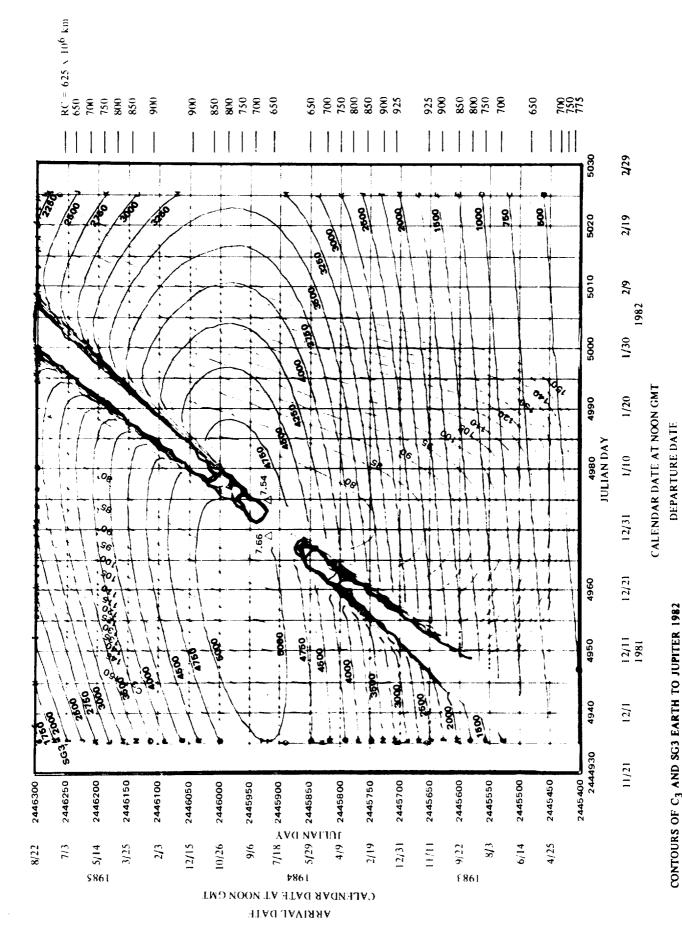




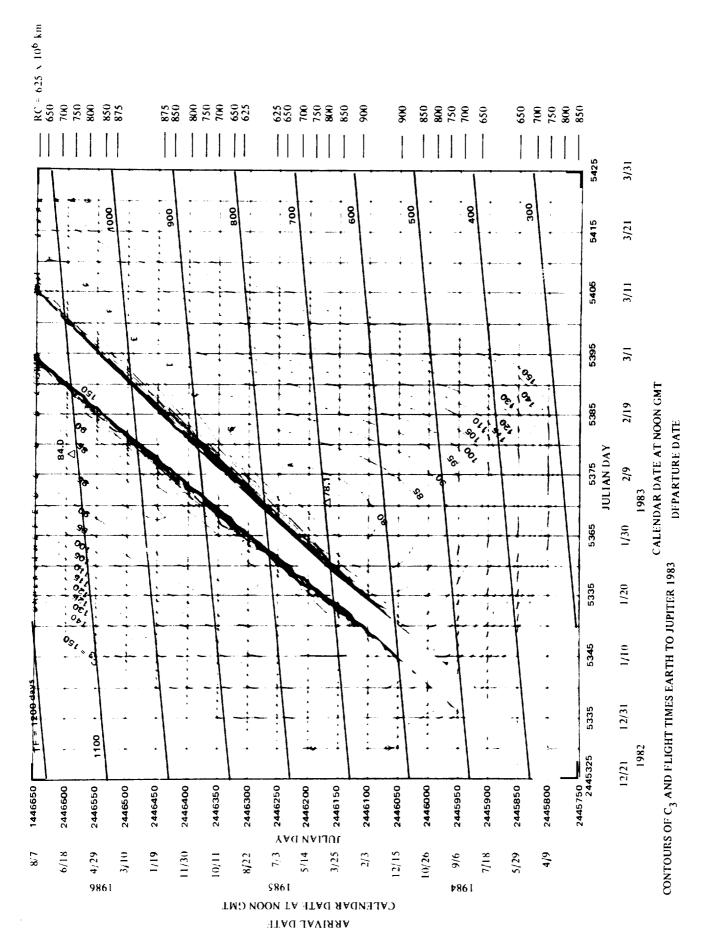
4-68



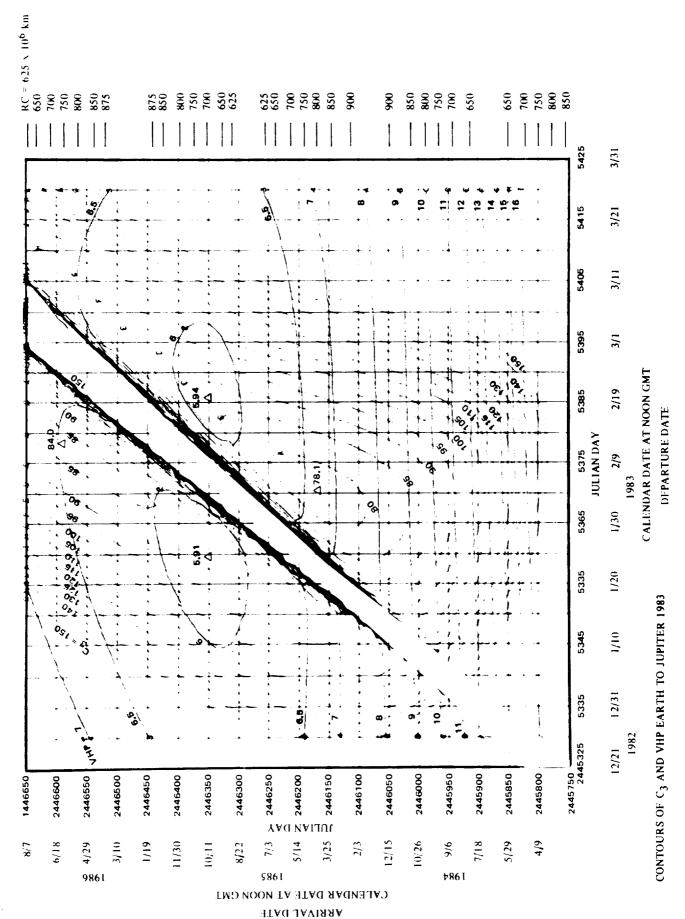
4-69

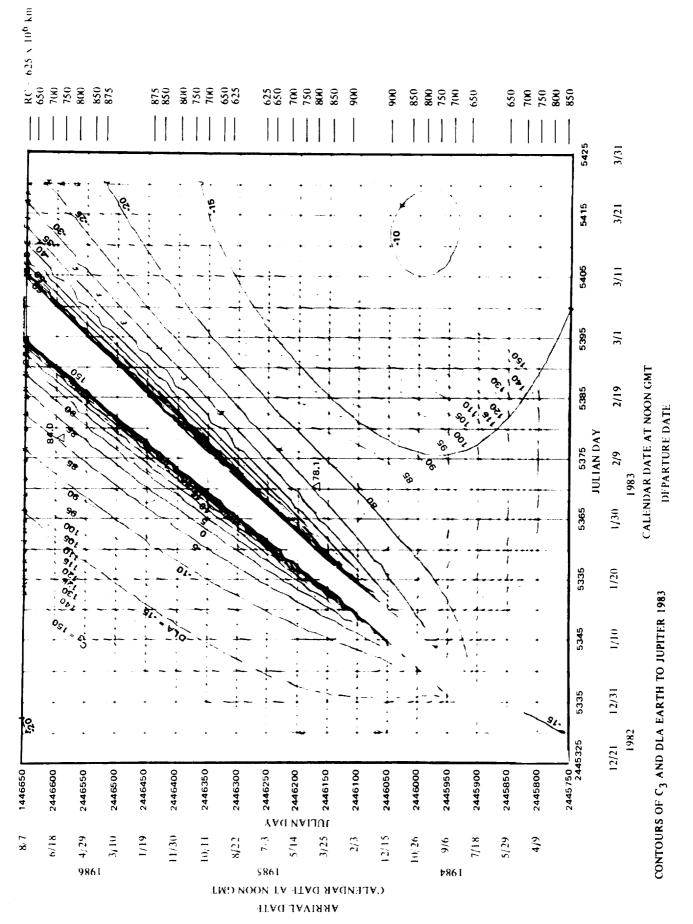


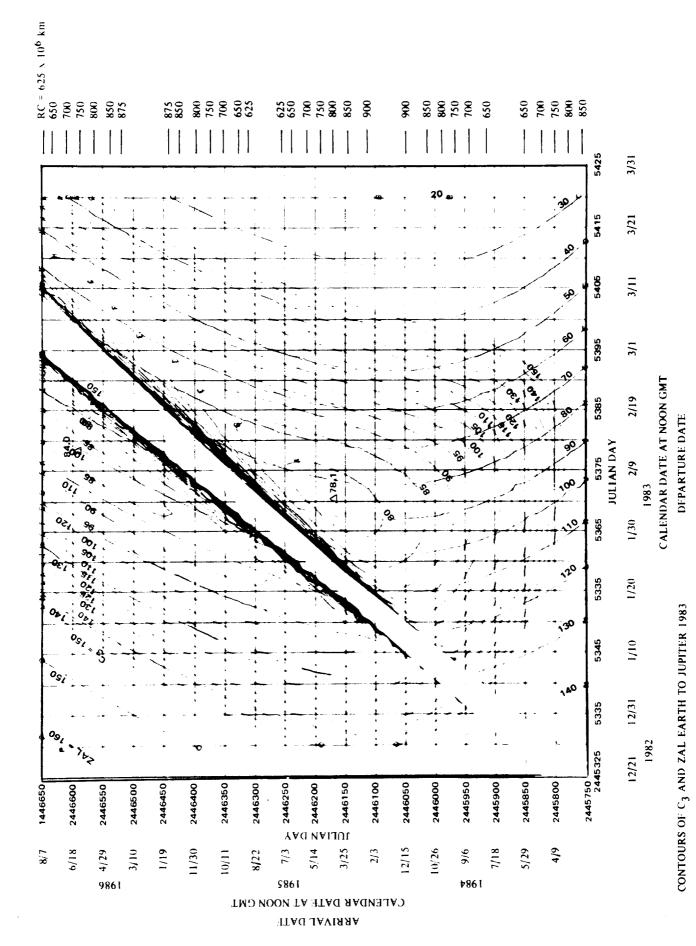
4-70



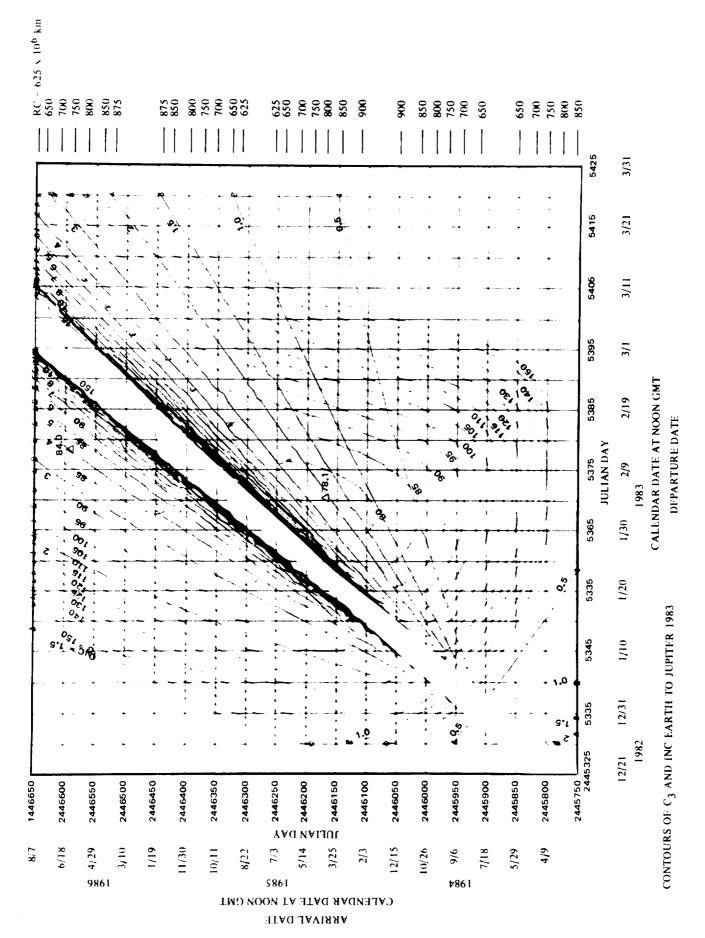
4-71



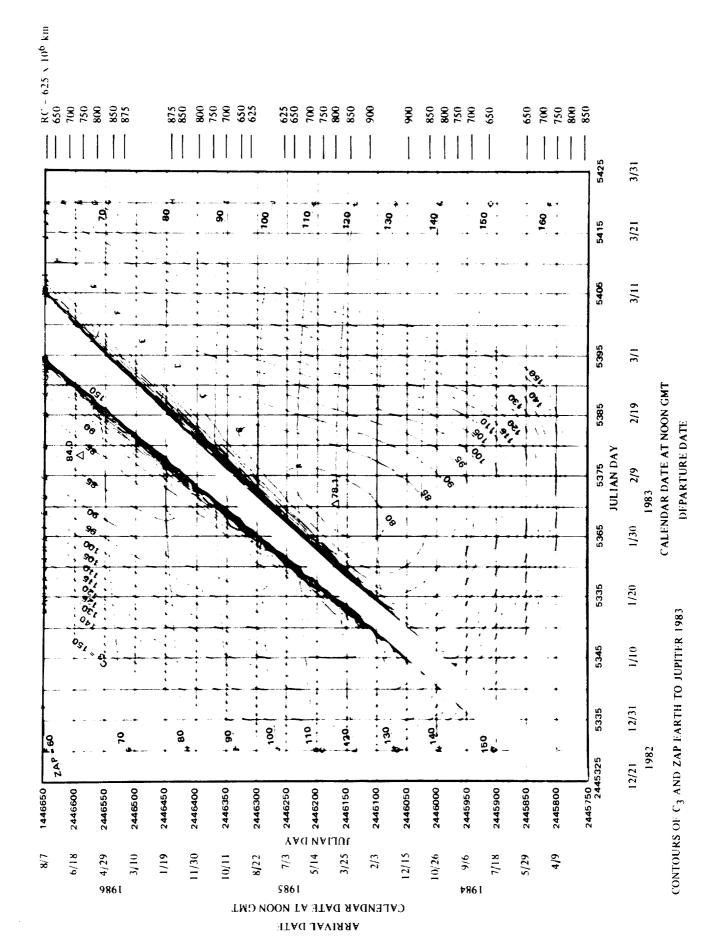




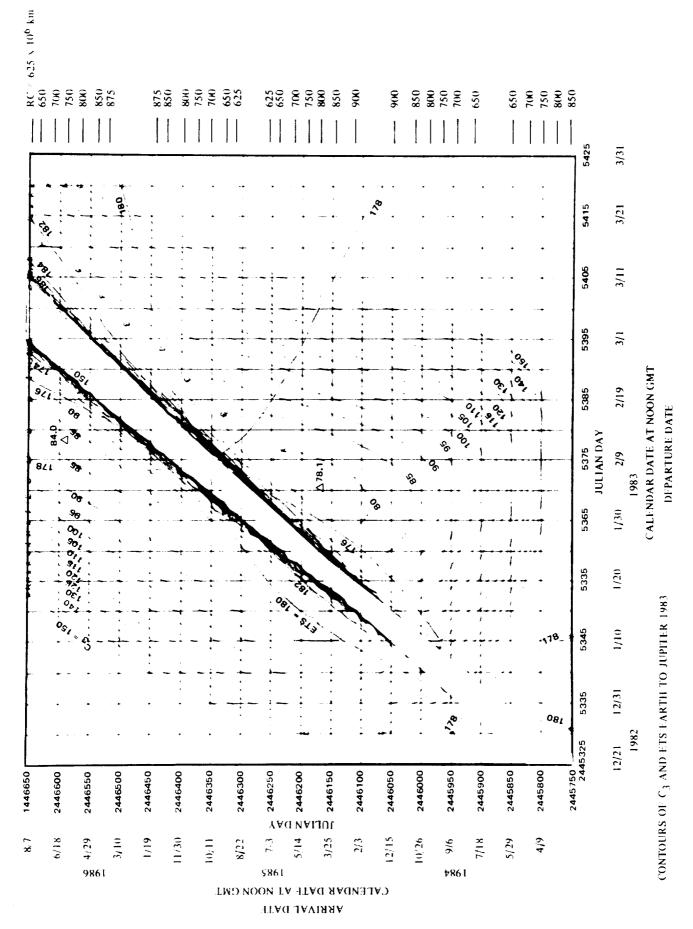
4-74



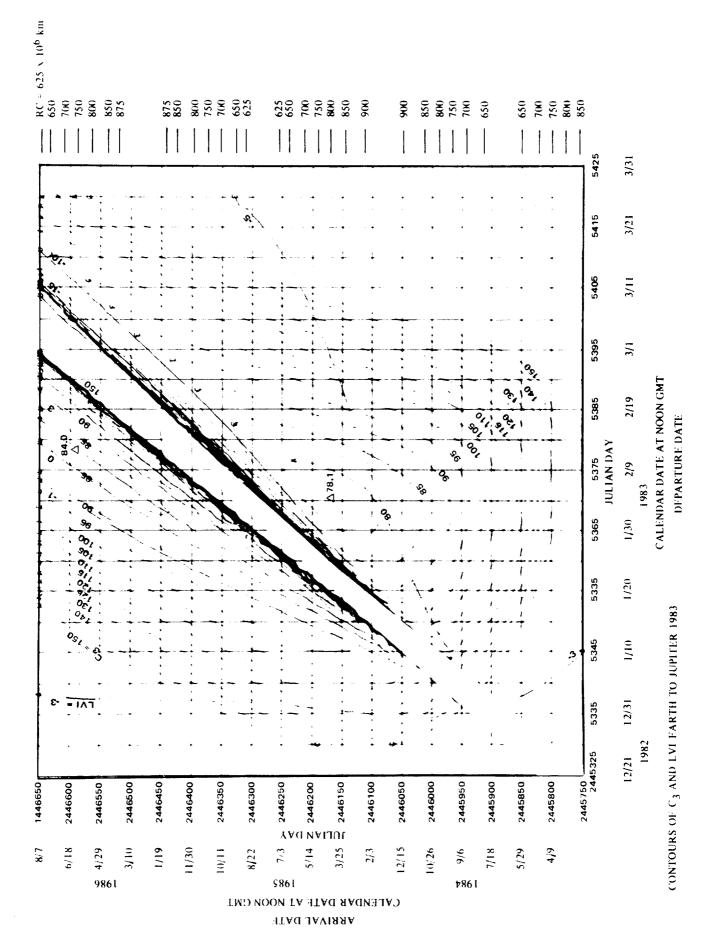
4-75



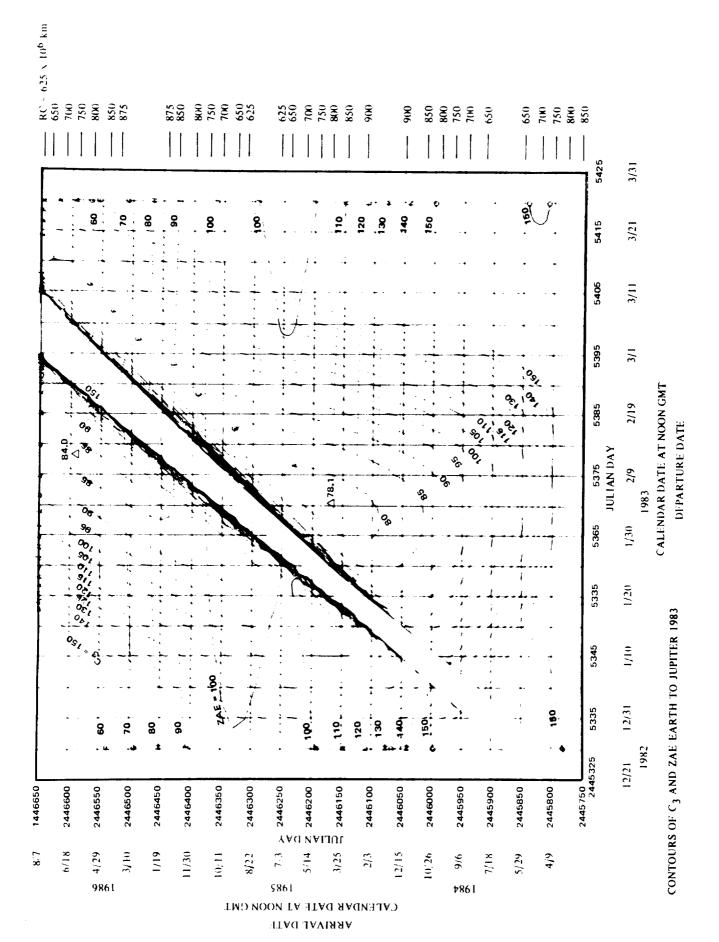
4-76



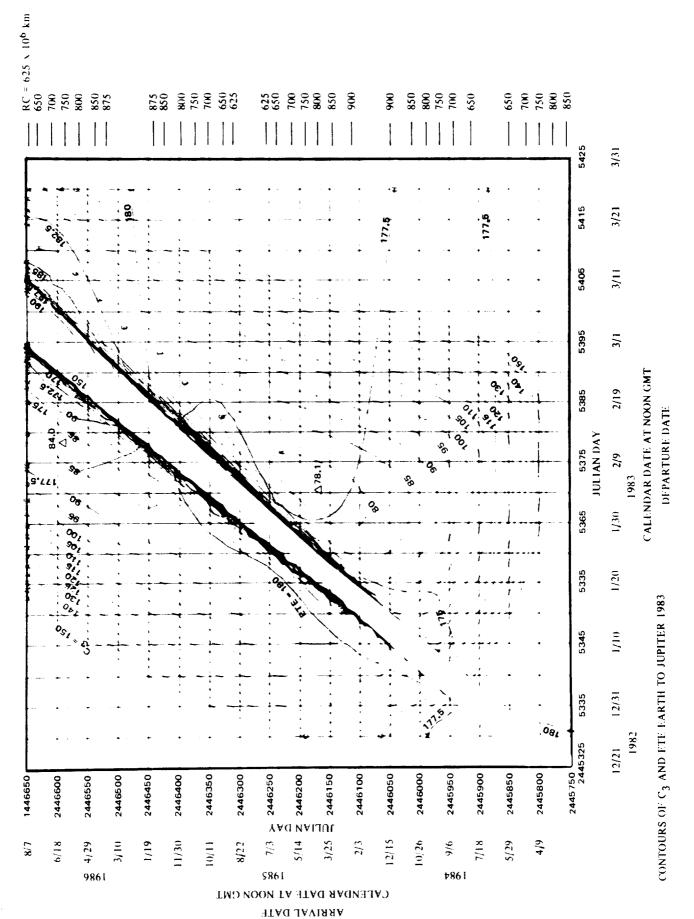
4-77



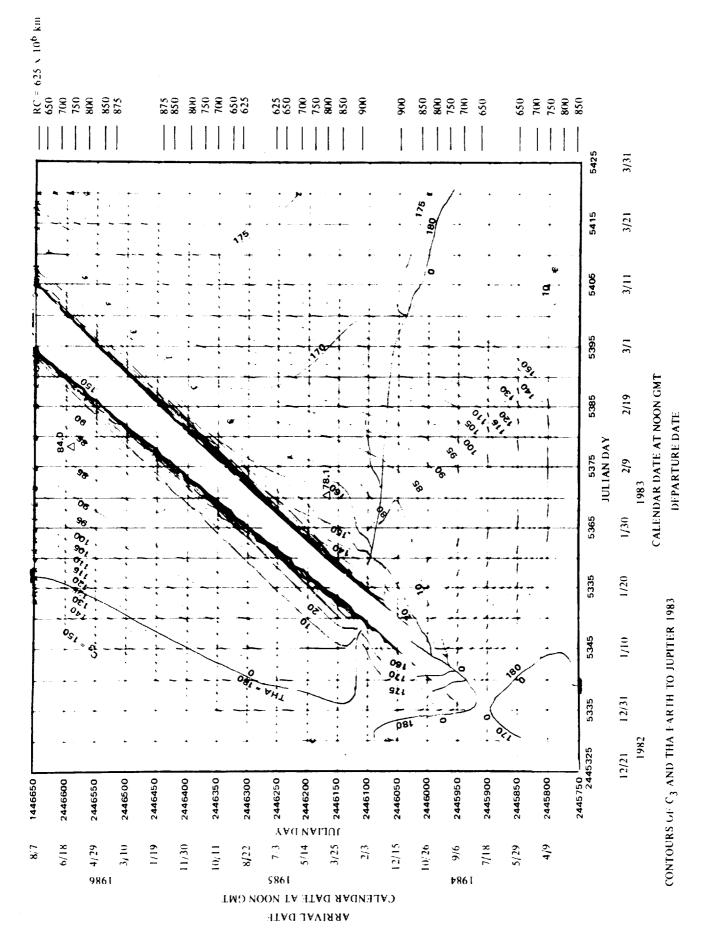
4-78



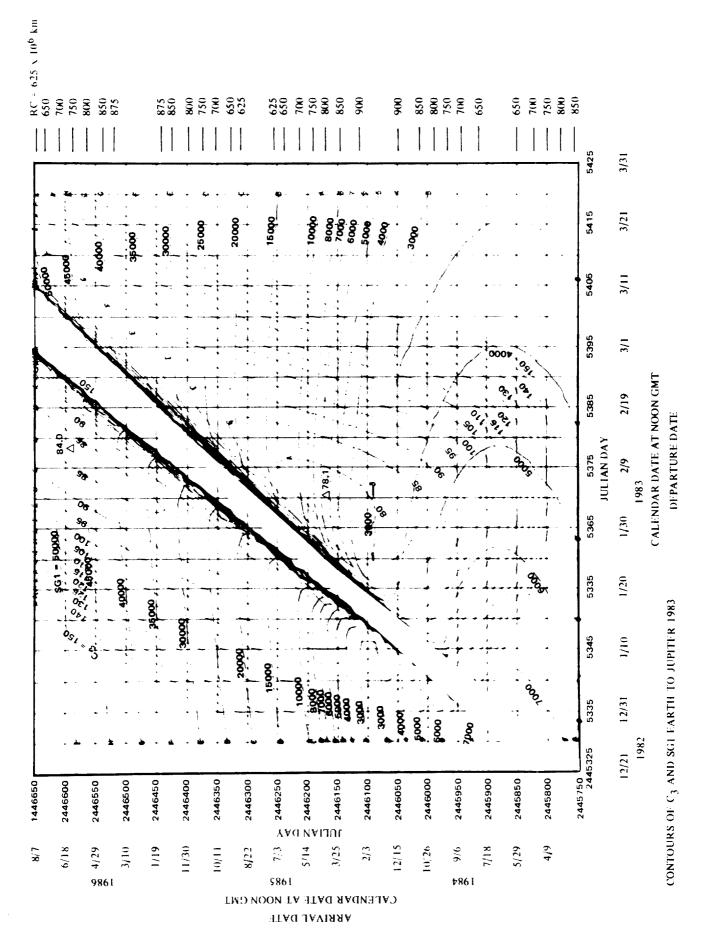
4-79



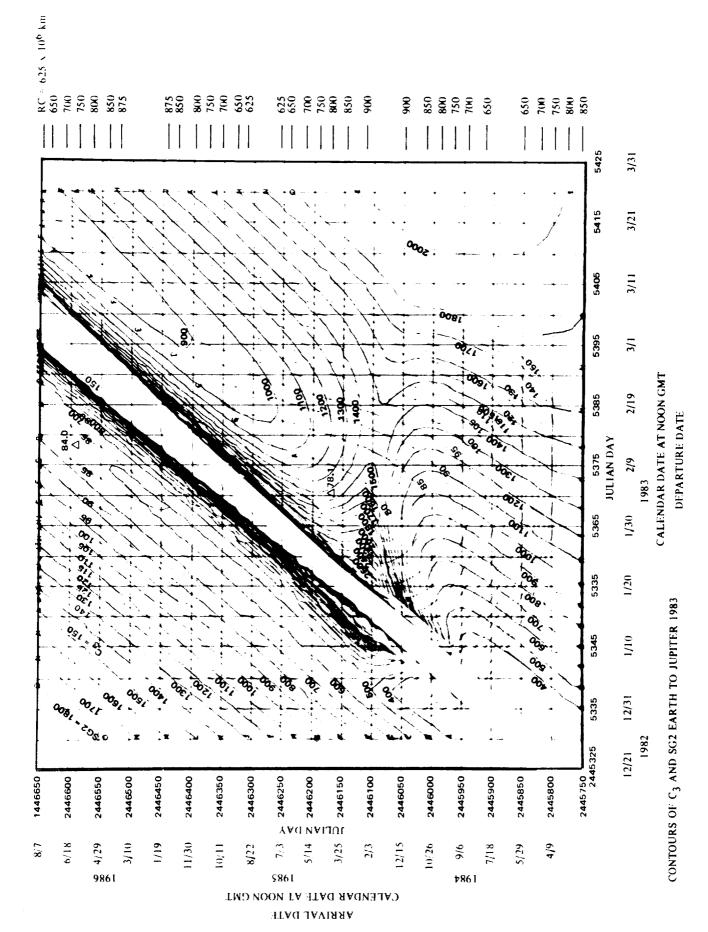
4-80



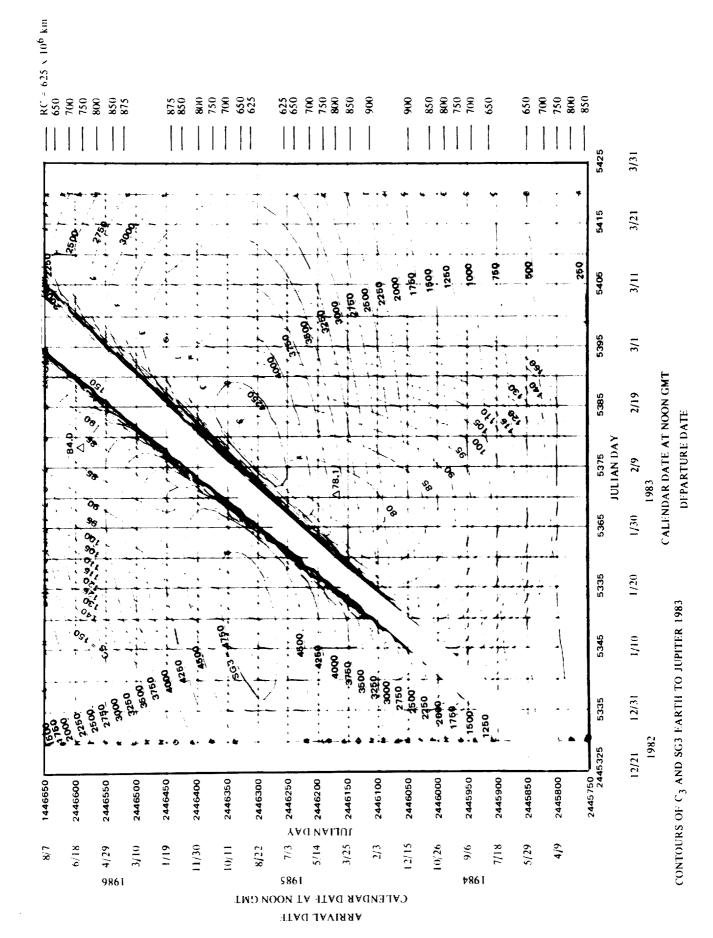
4-81



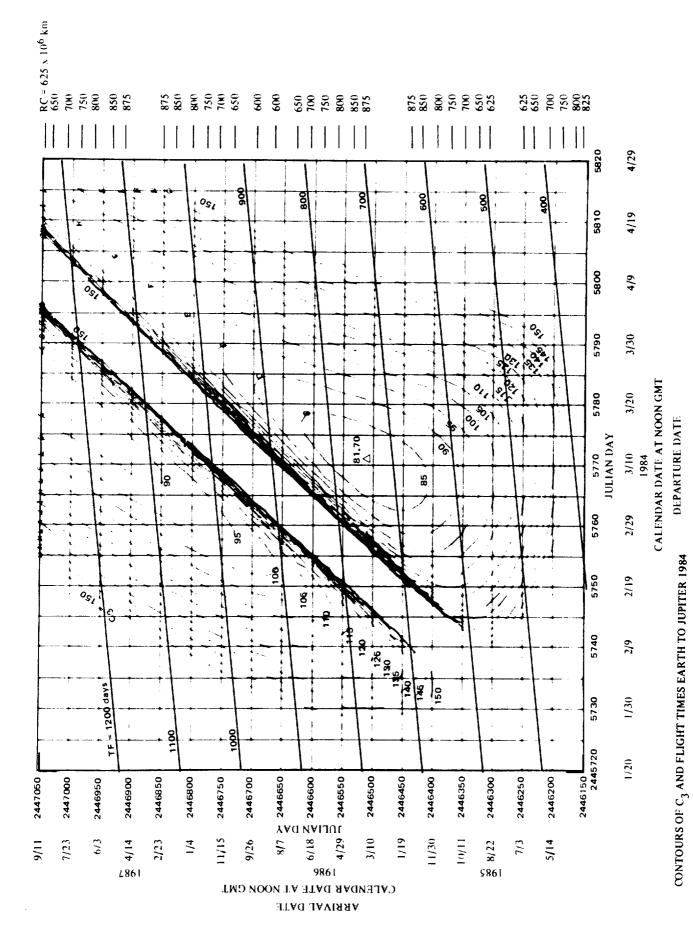
4-82

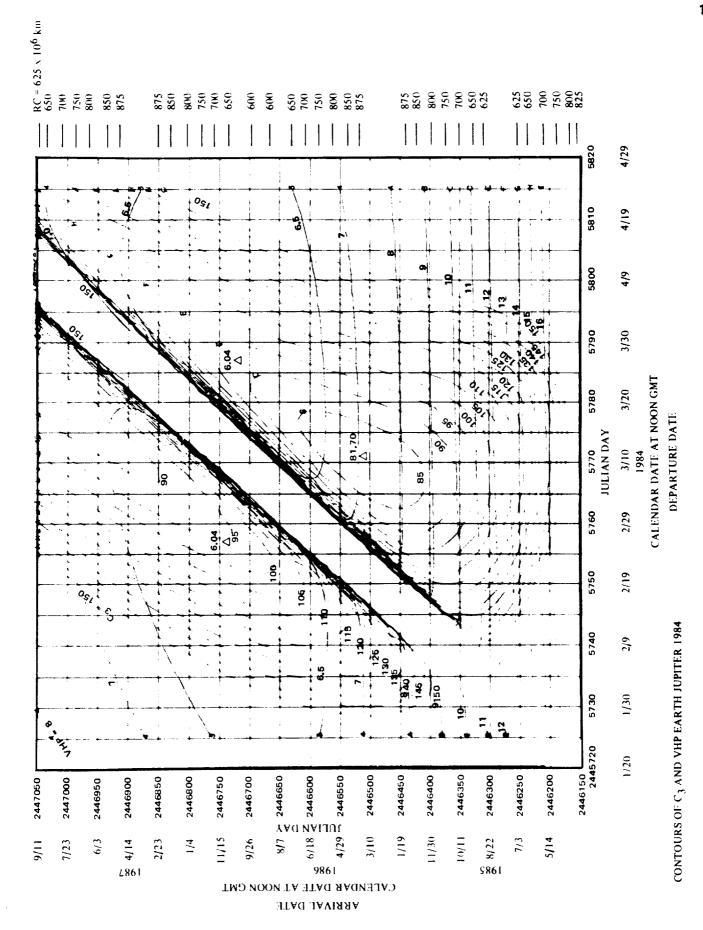


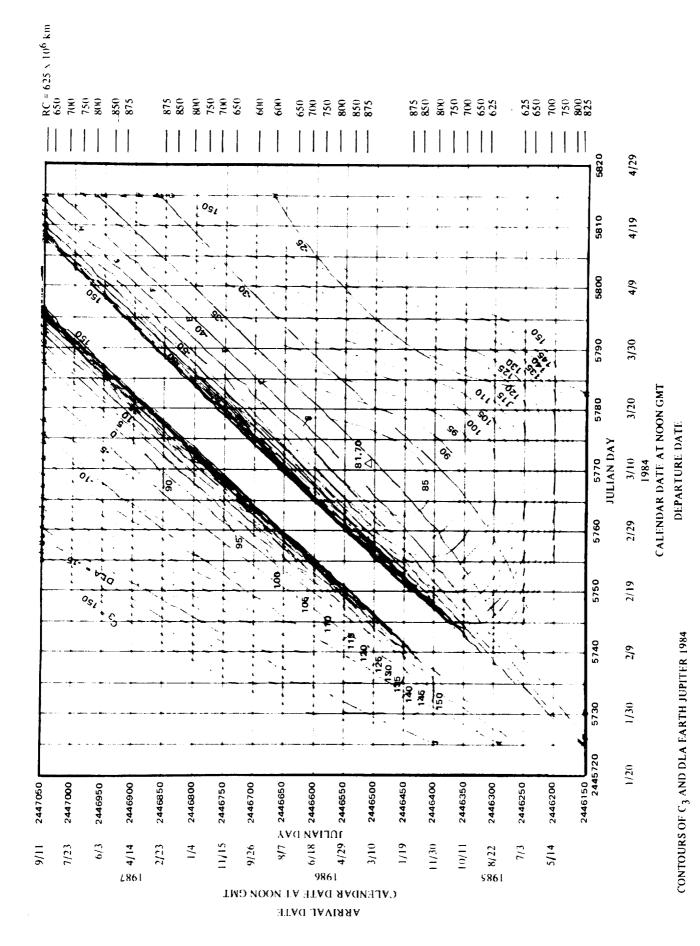
4-83

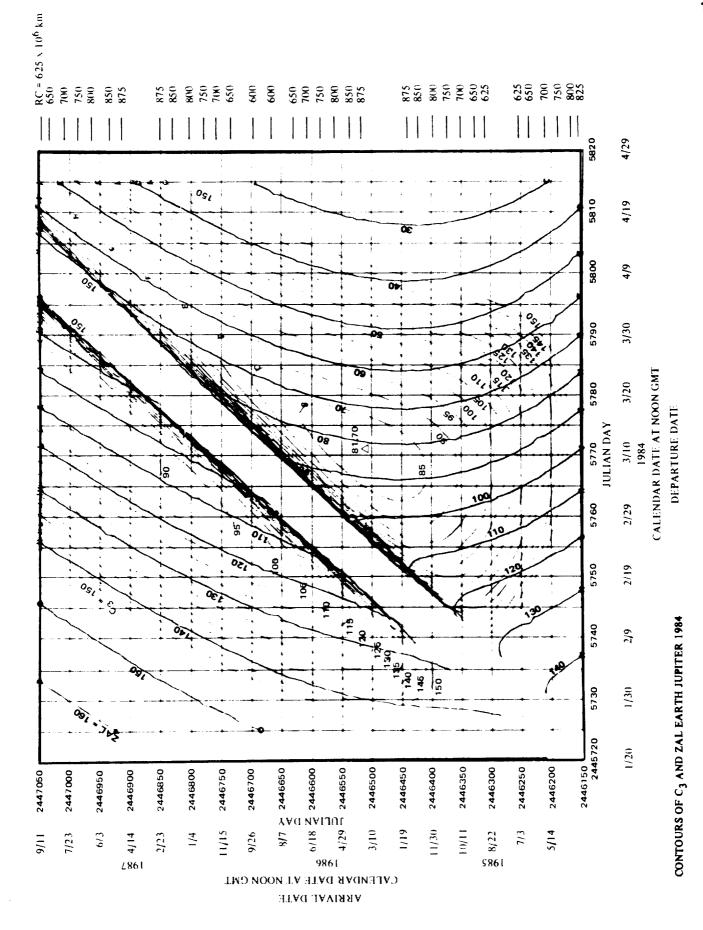


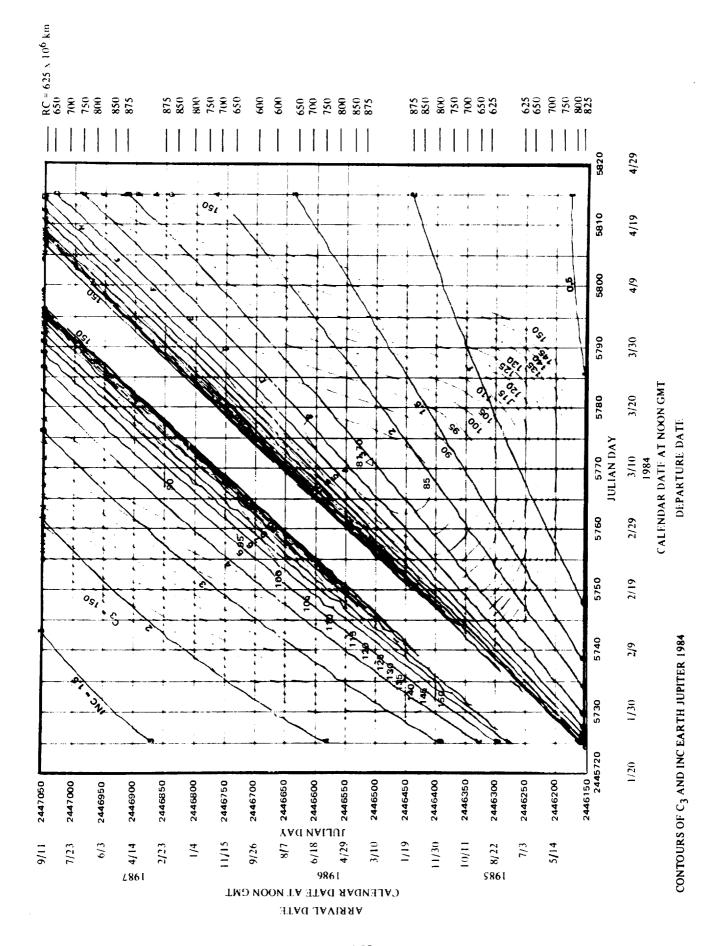
4-84





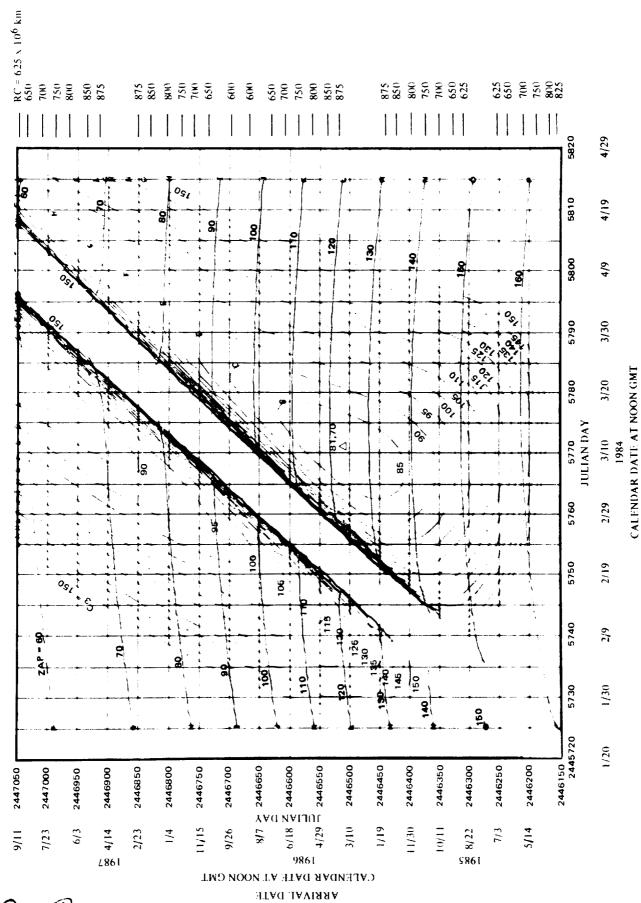


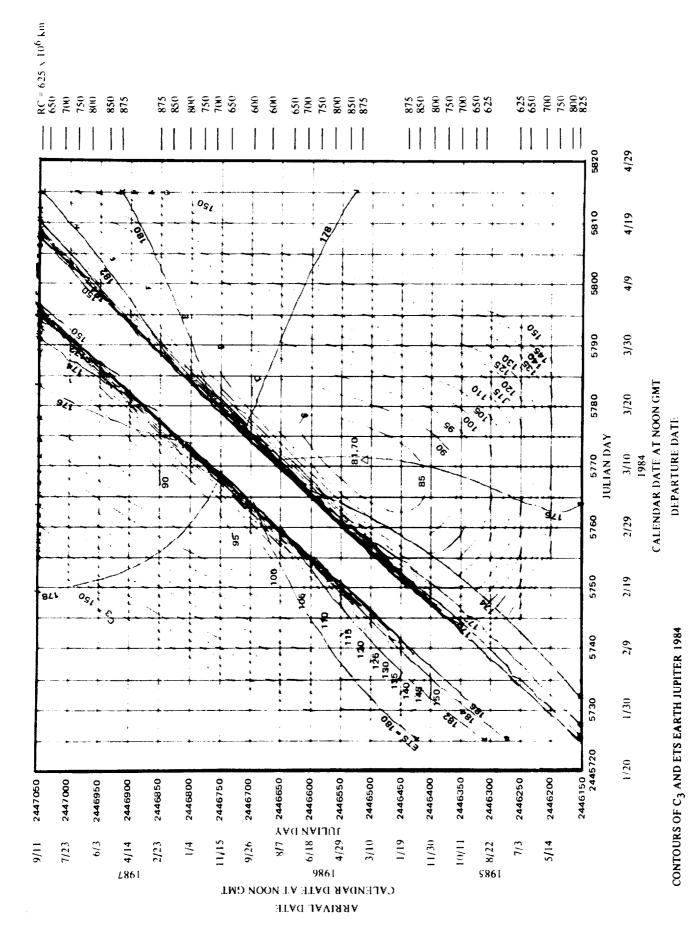


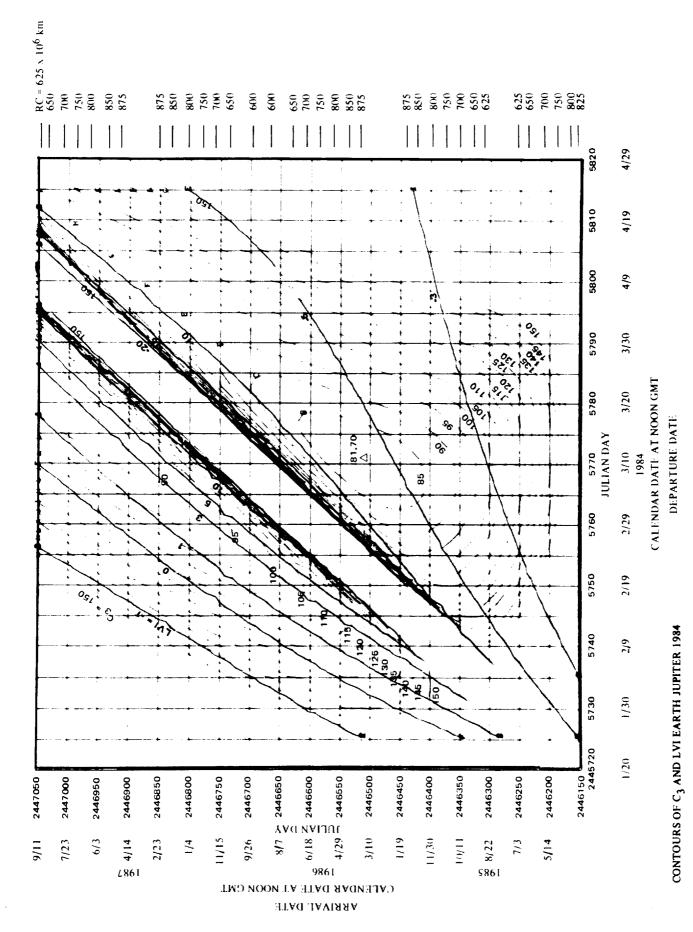


DEPARTURE DATE

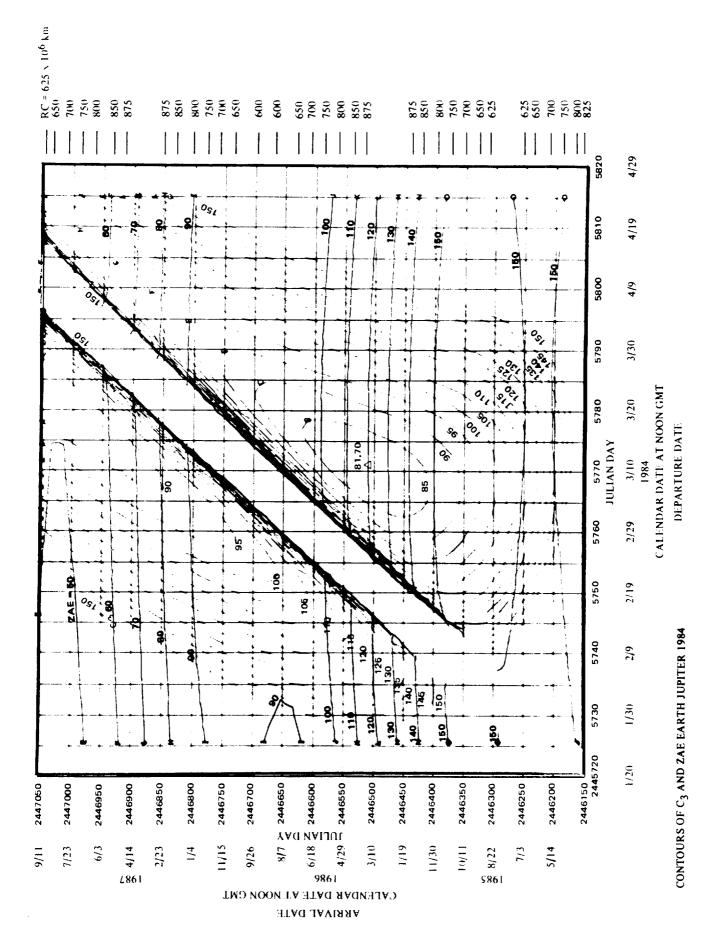
CONTOURS OF C_3 AND ZAP EARTH JUPITER 1984

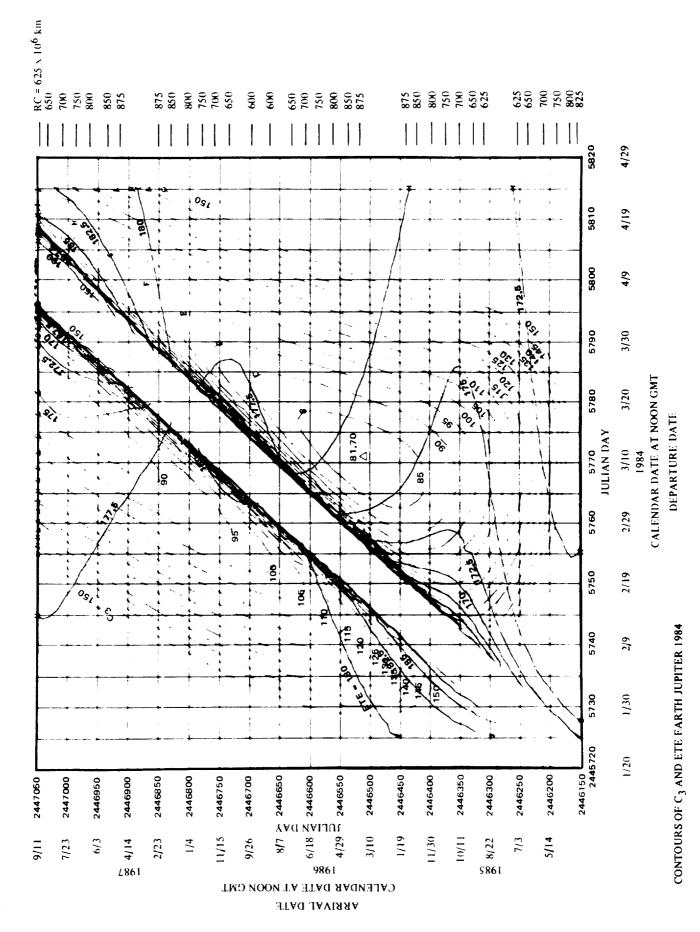


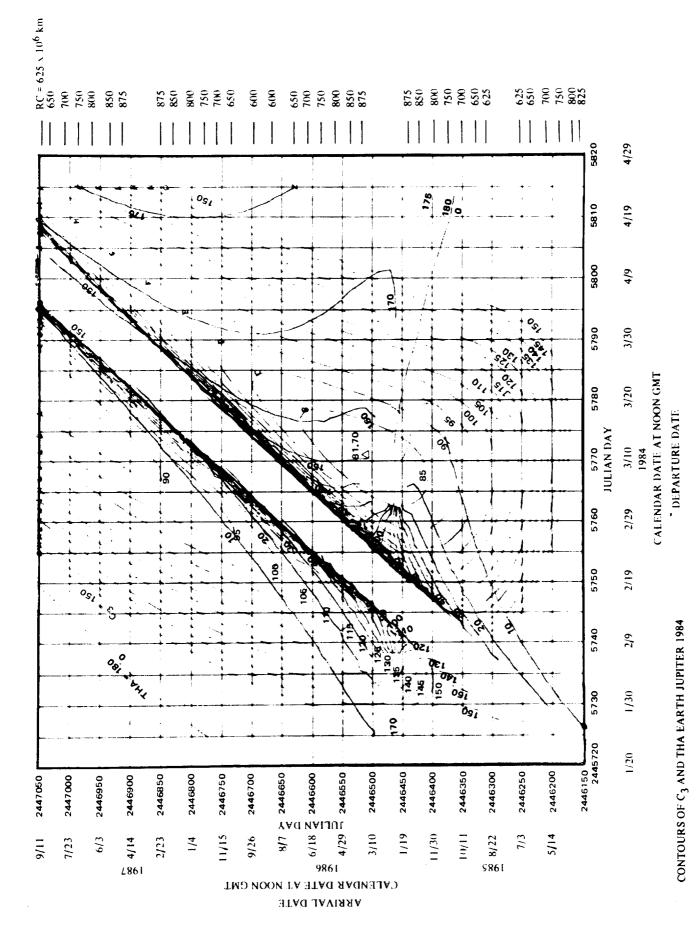


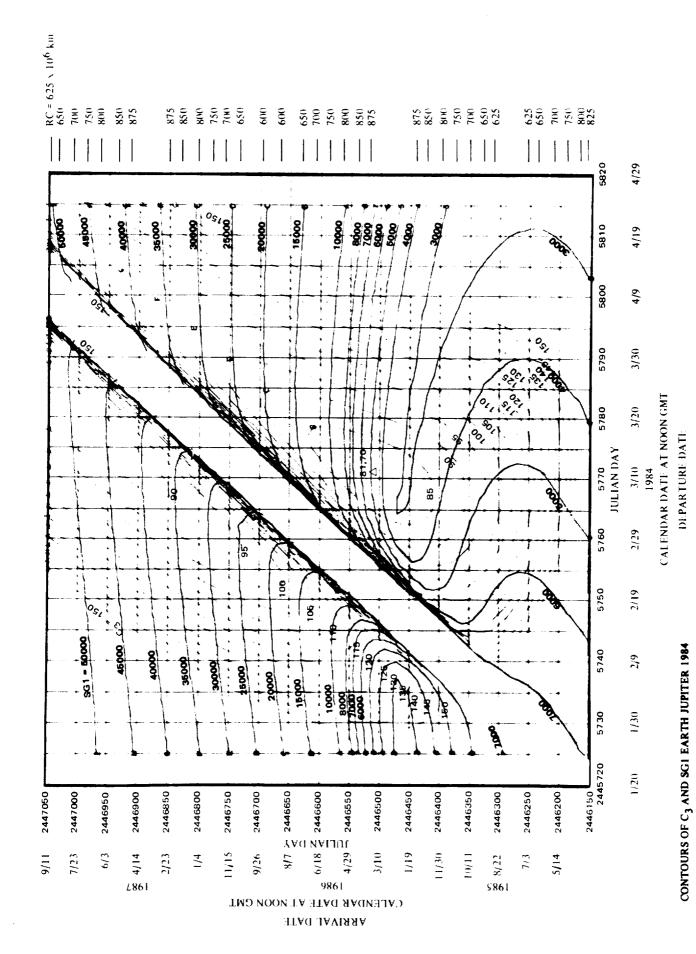


4-92

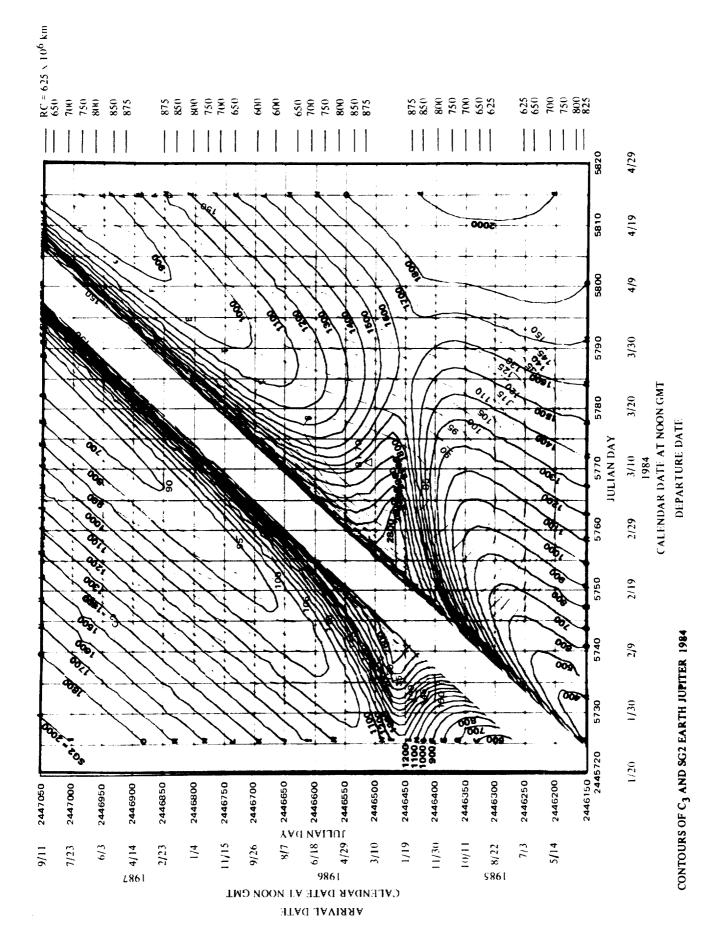


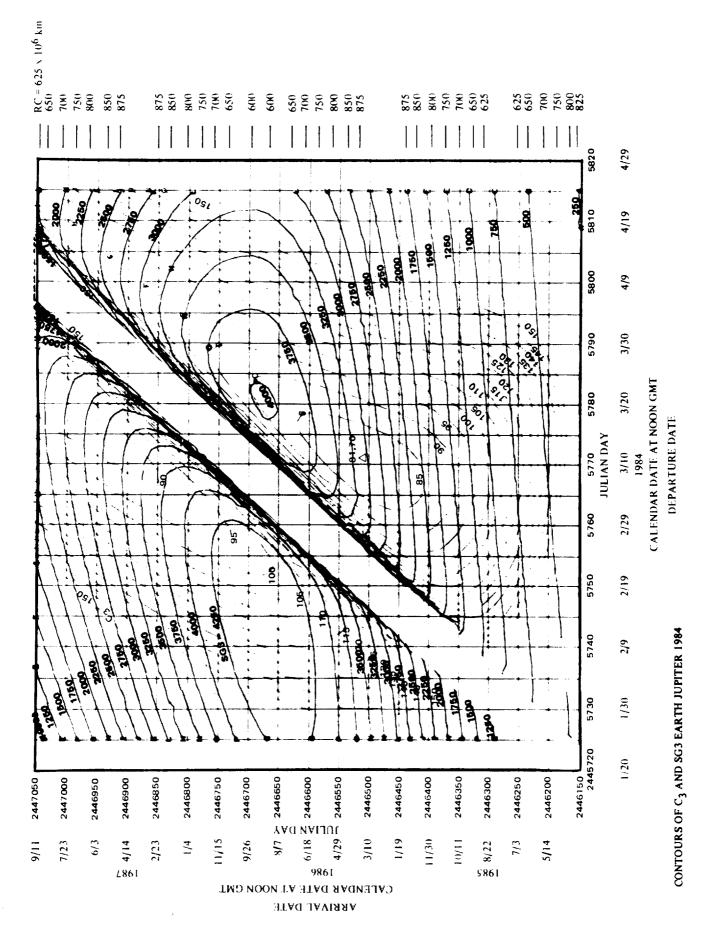




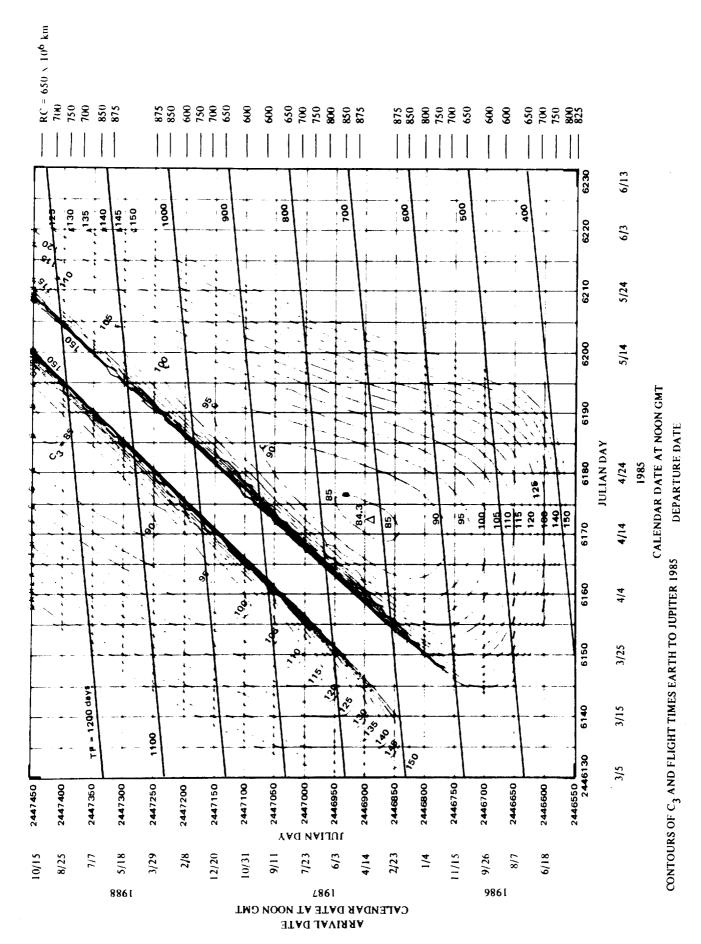


4-96

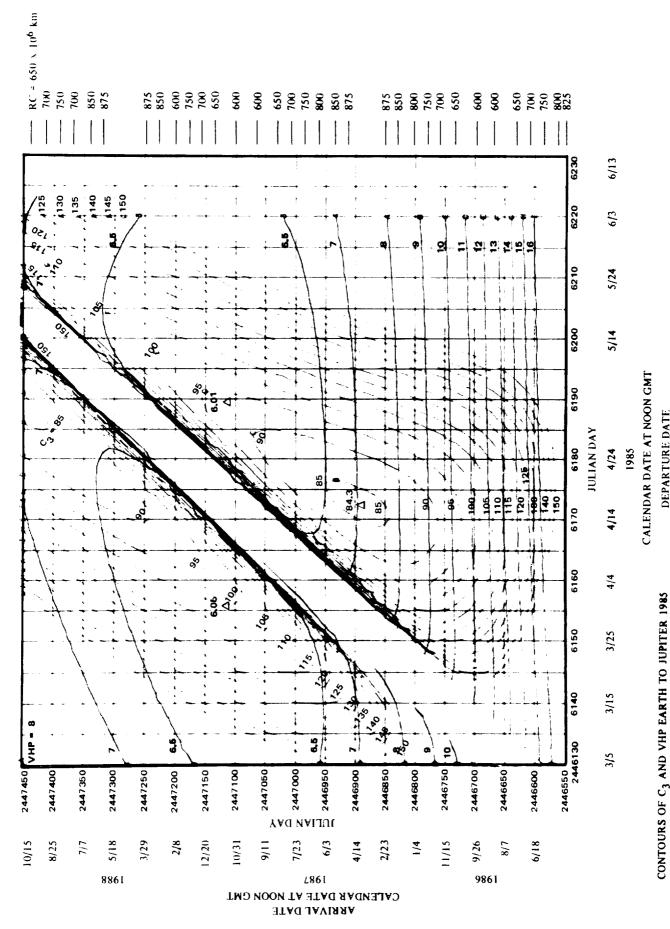




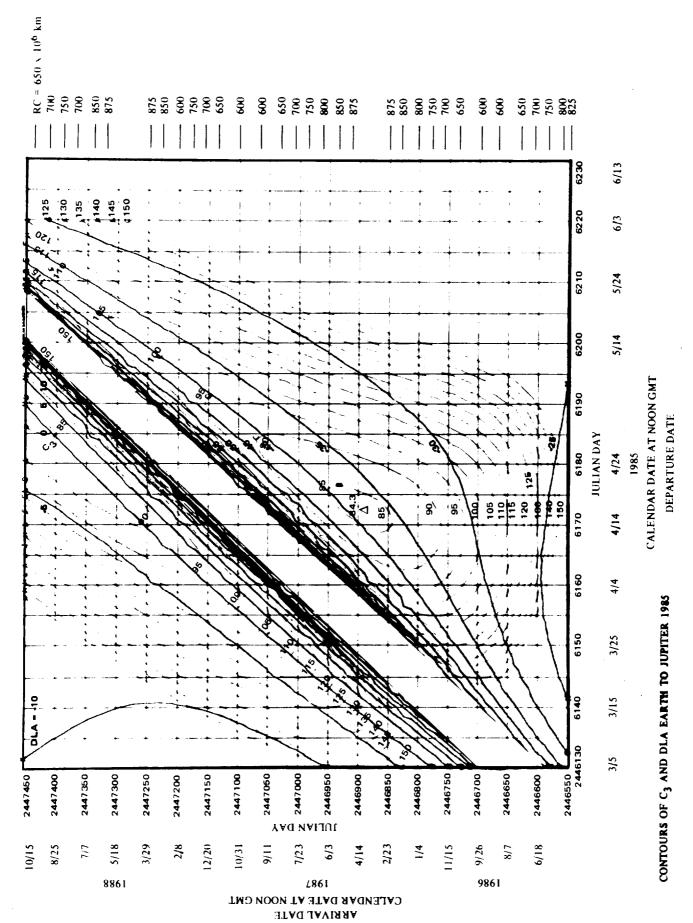
4-98



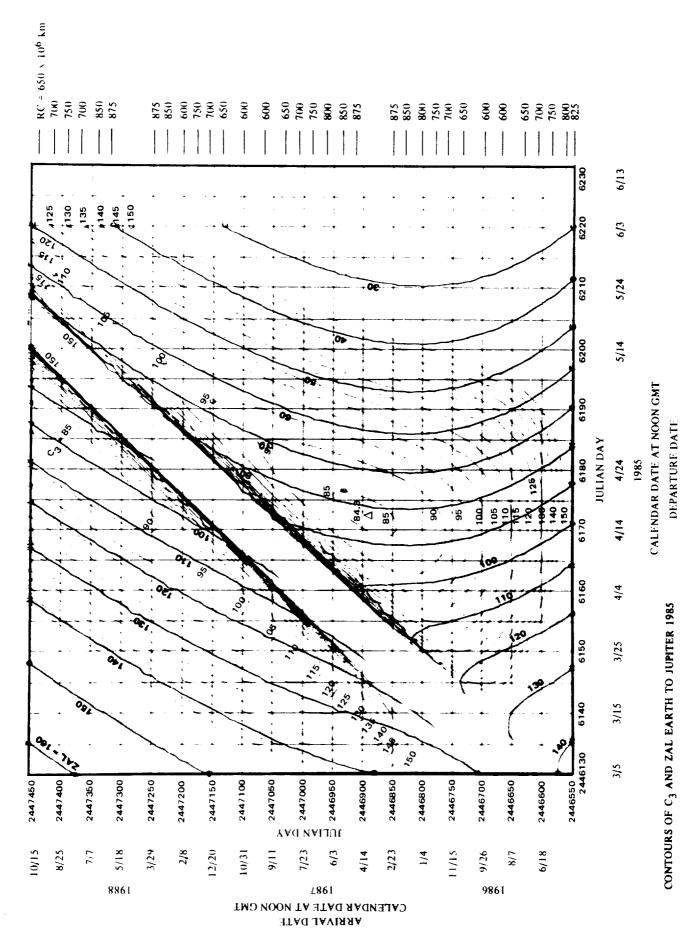
DEPARTURE DATE



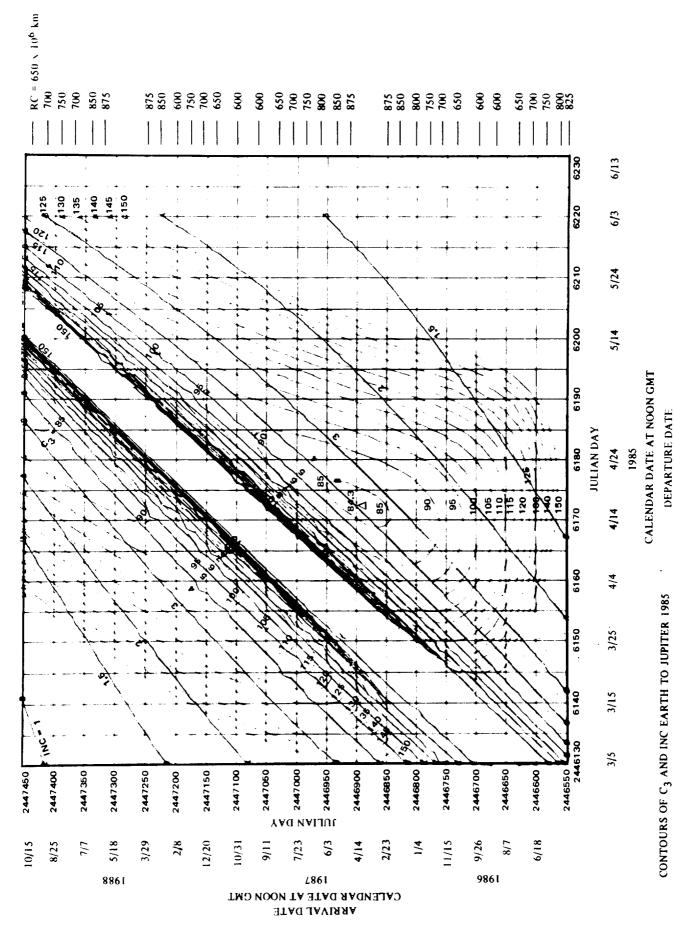
4-100



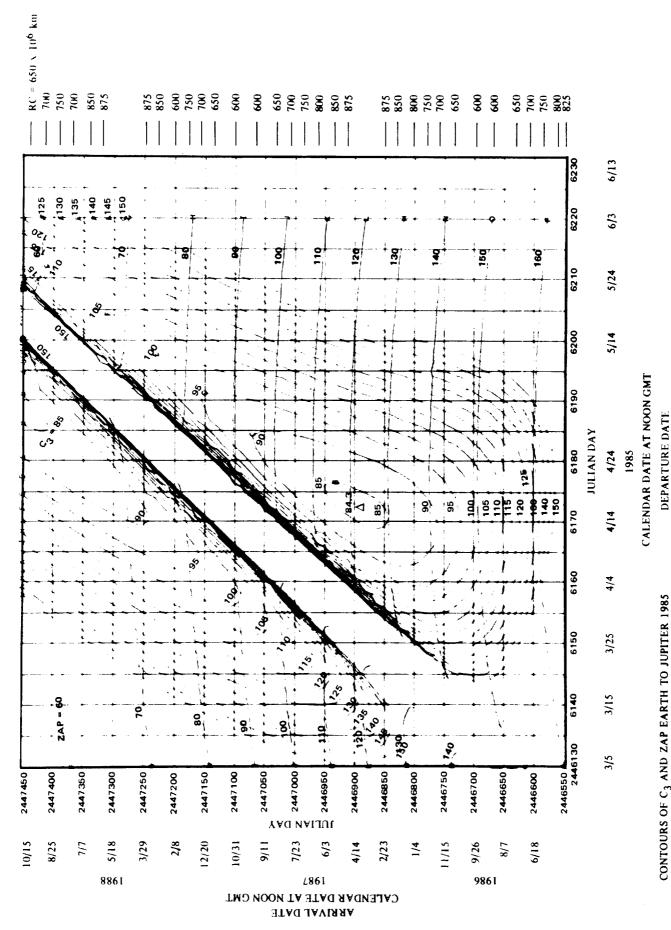
4-101



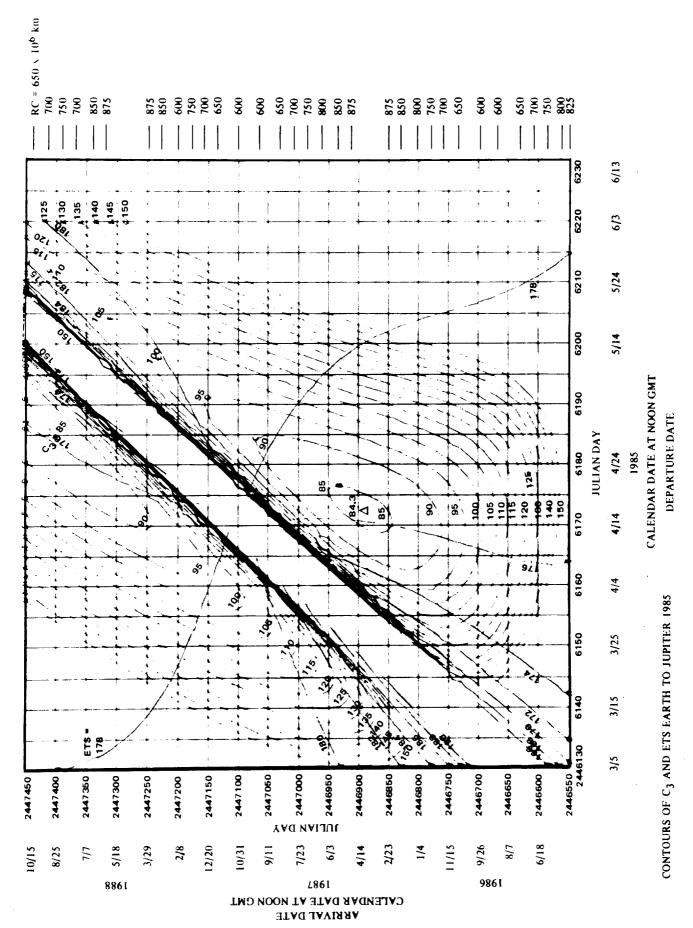
4-102

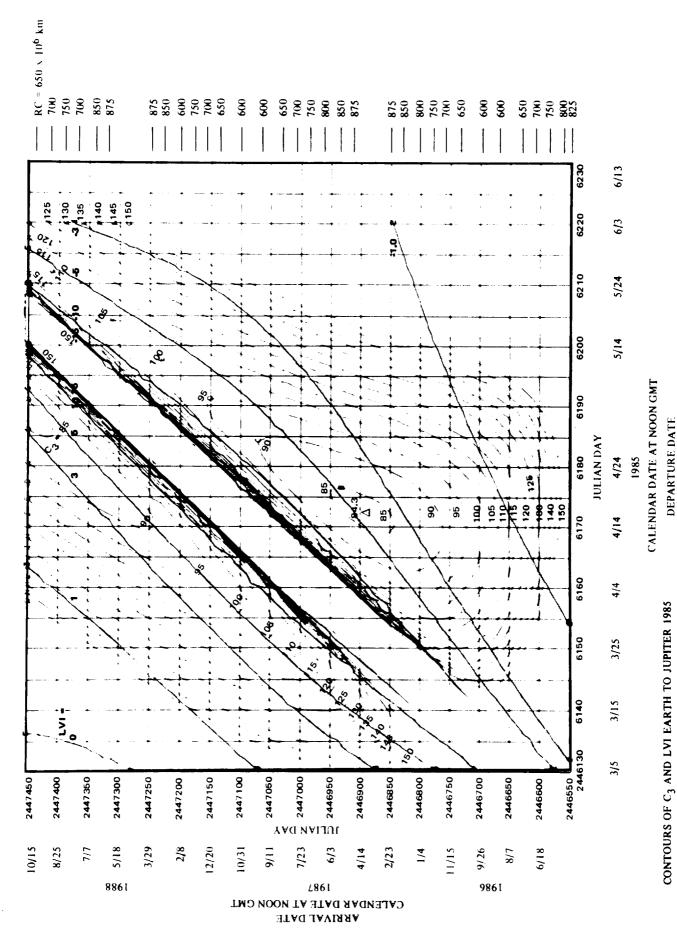


4-103

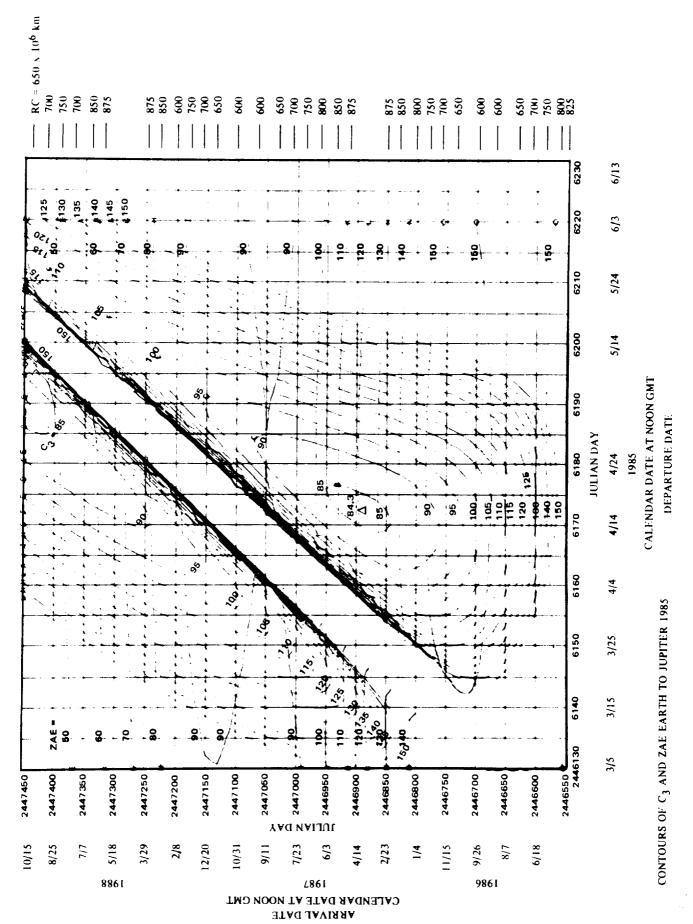


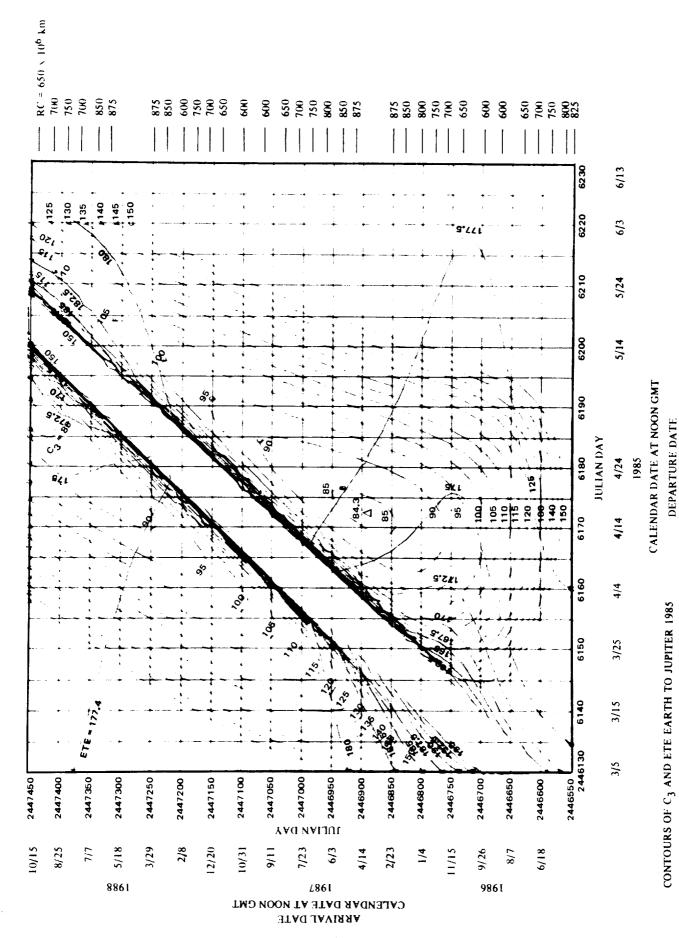
4-104



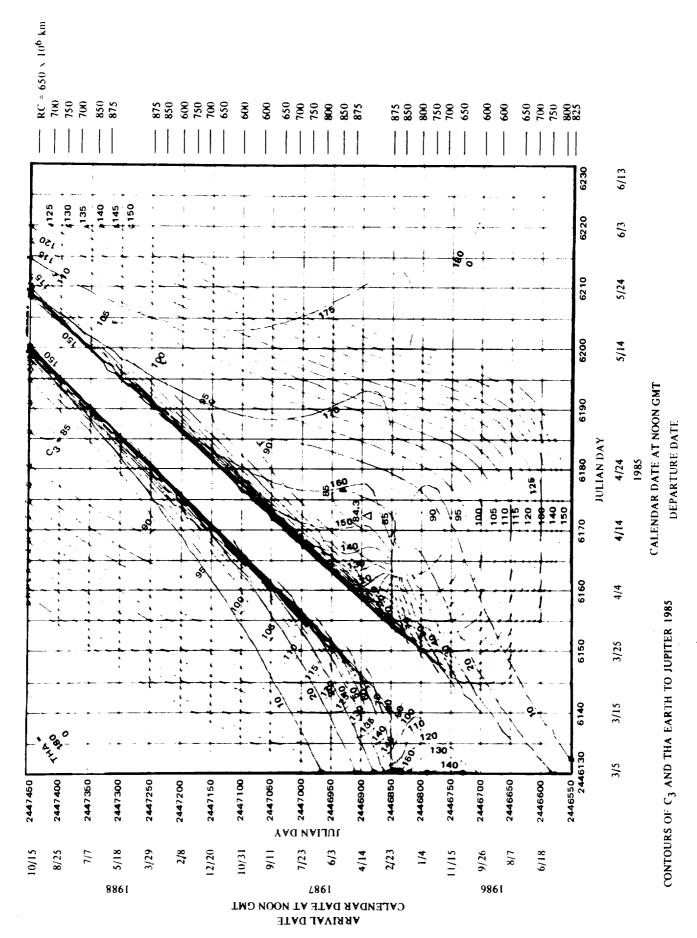


4-106

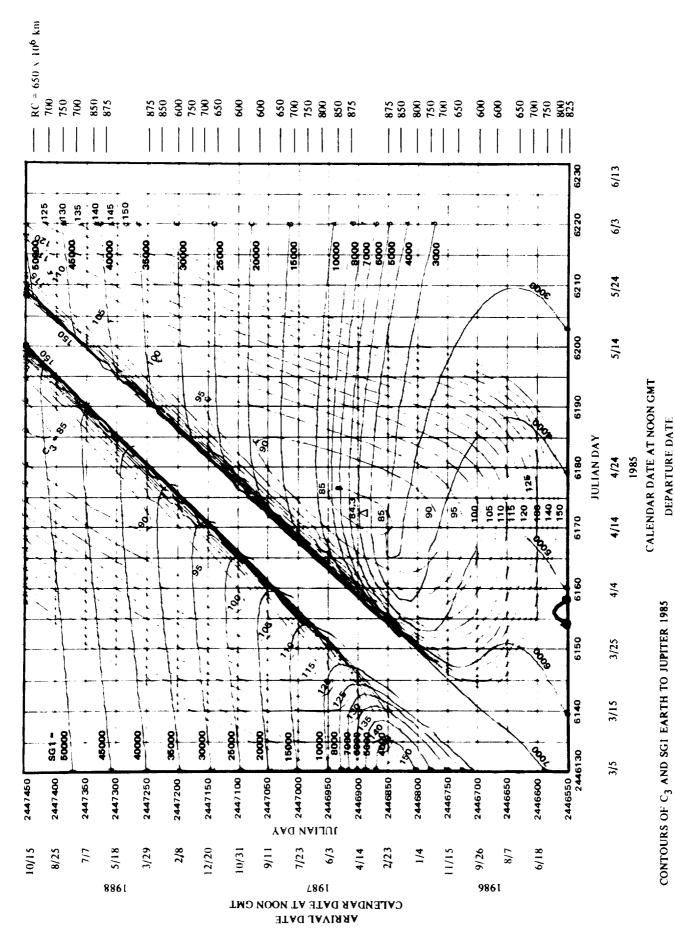


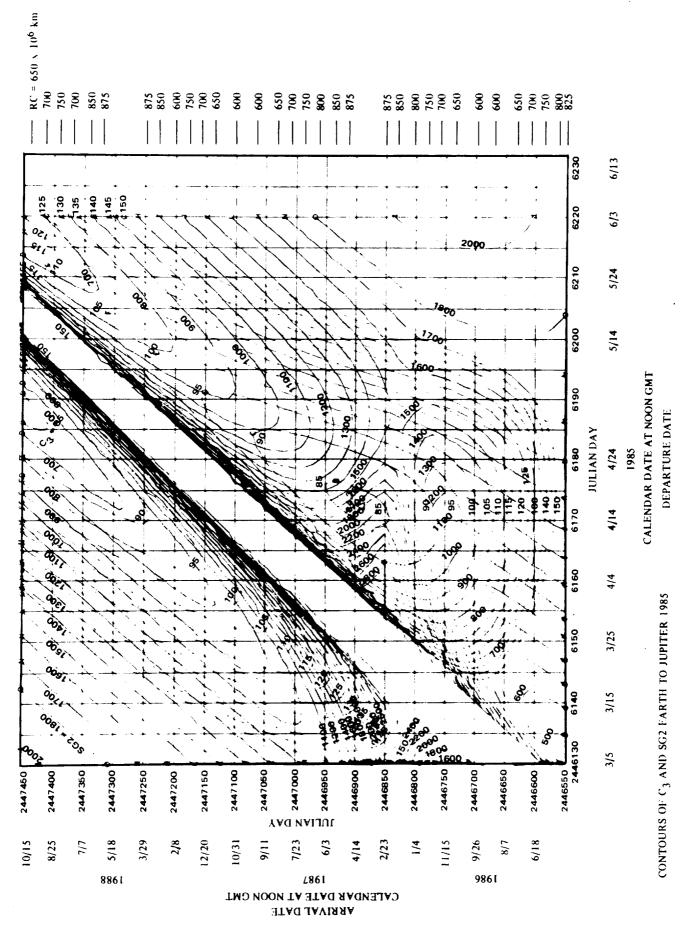


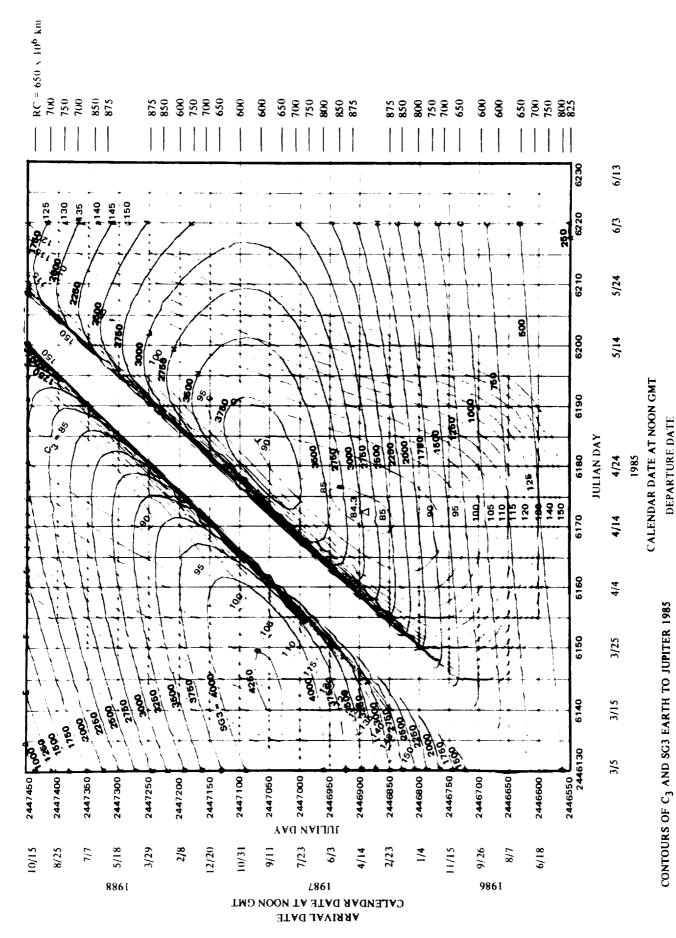
4-108



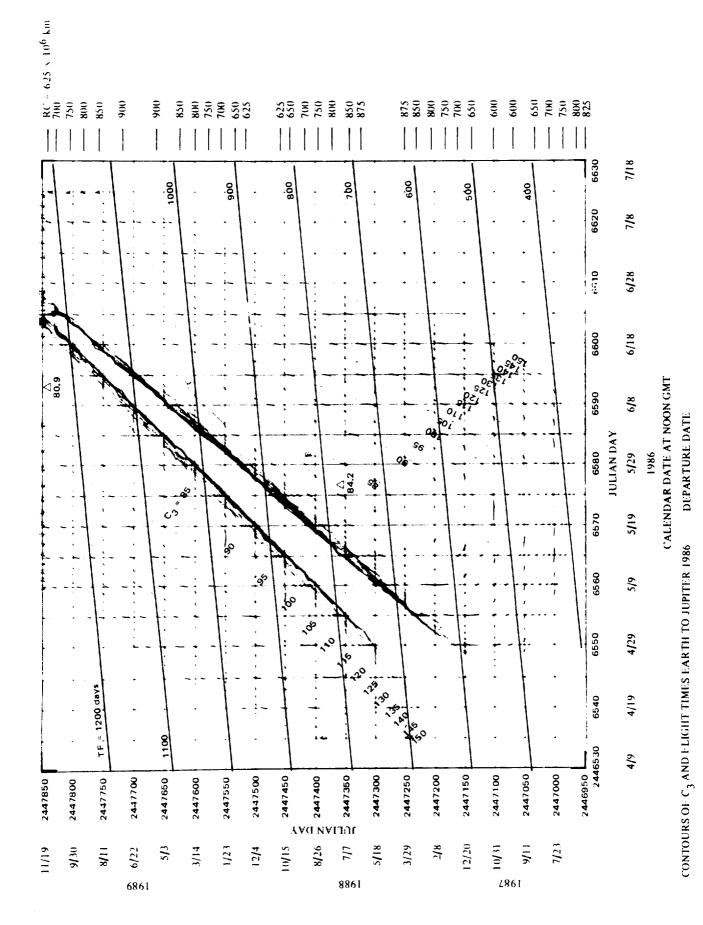
4-109



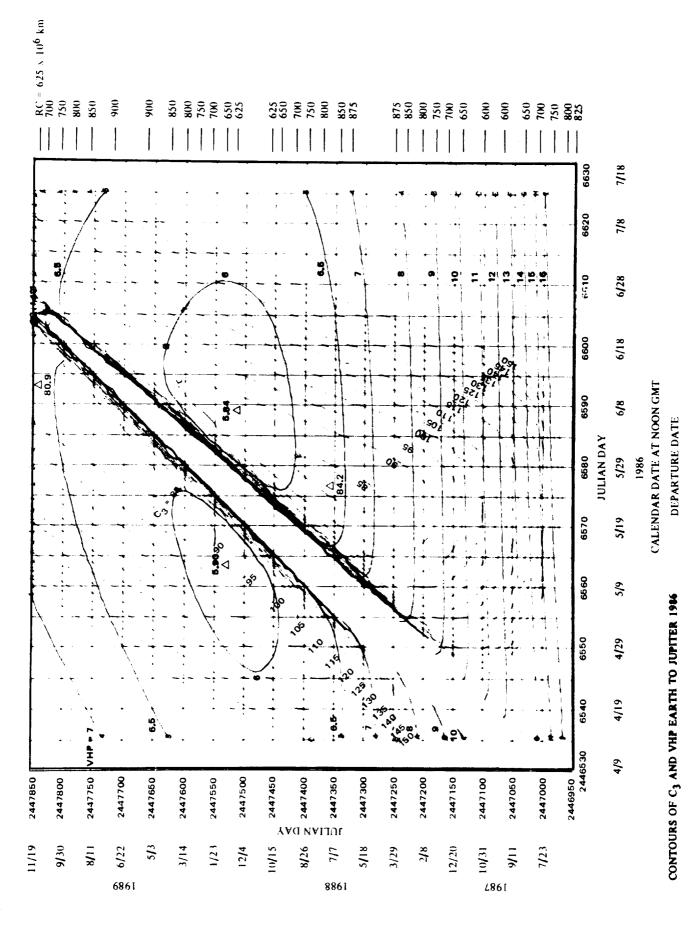




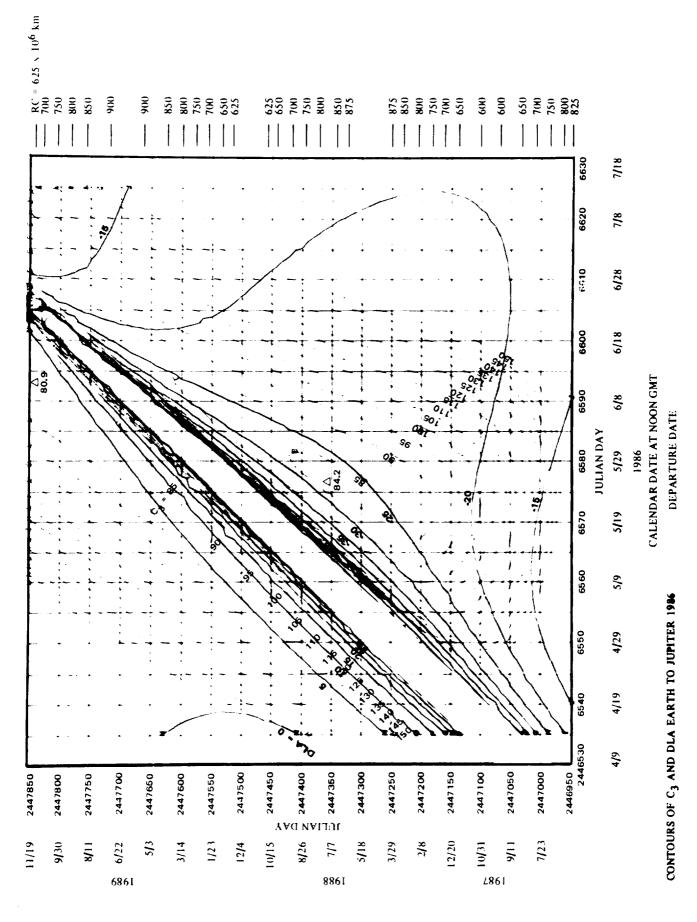
4-112

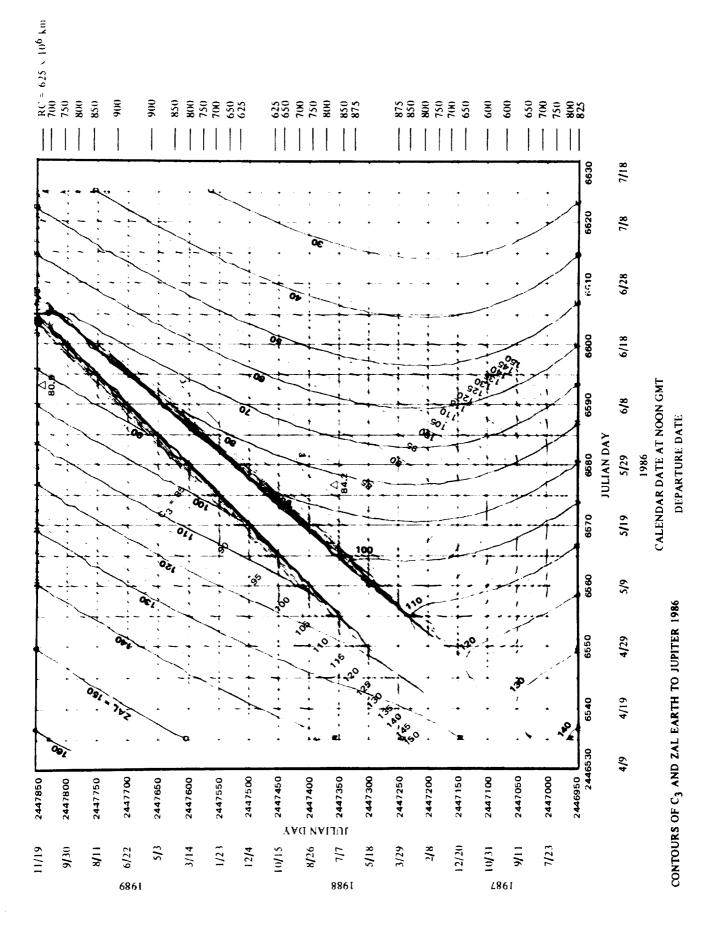


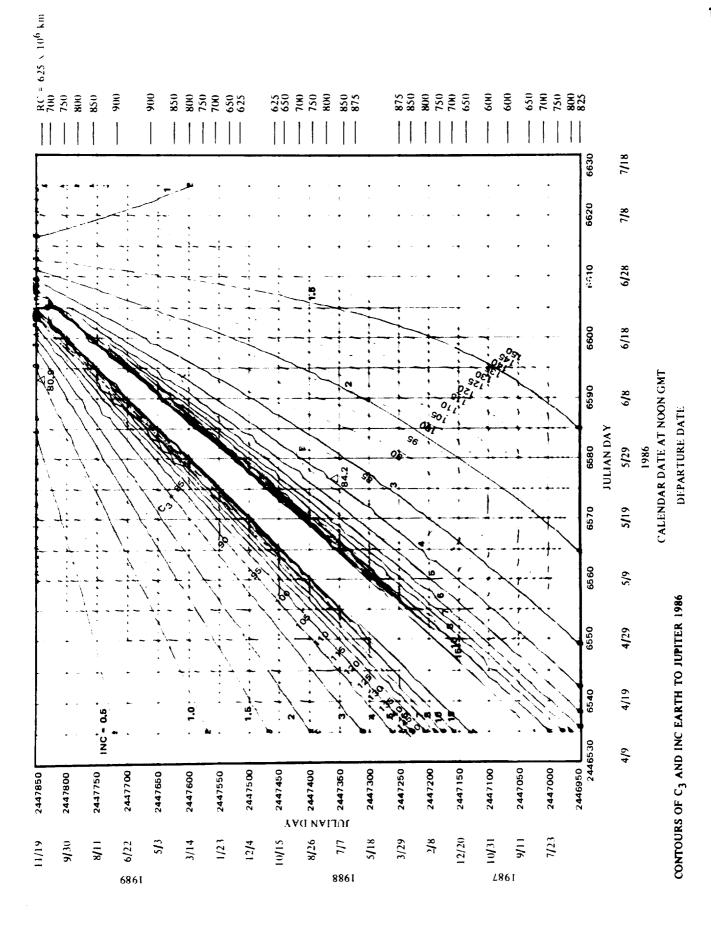
4-113

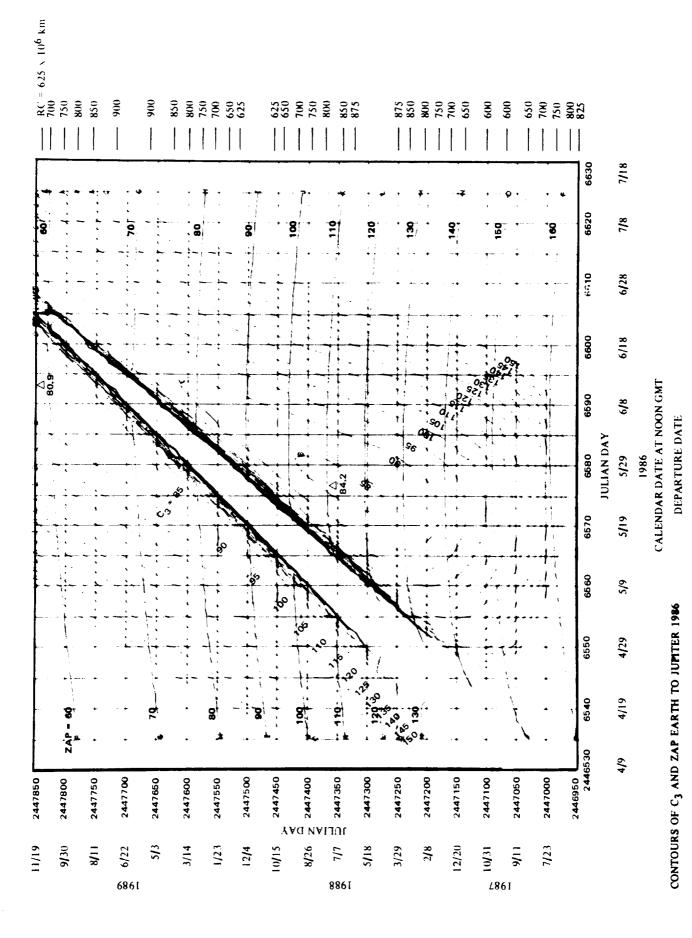


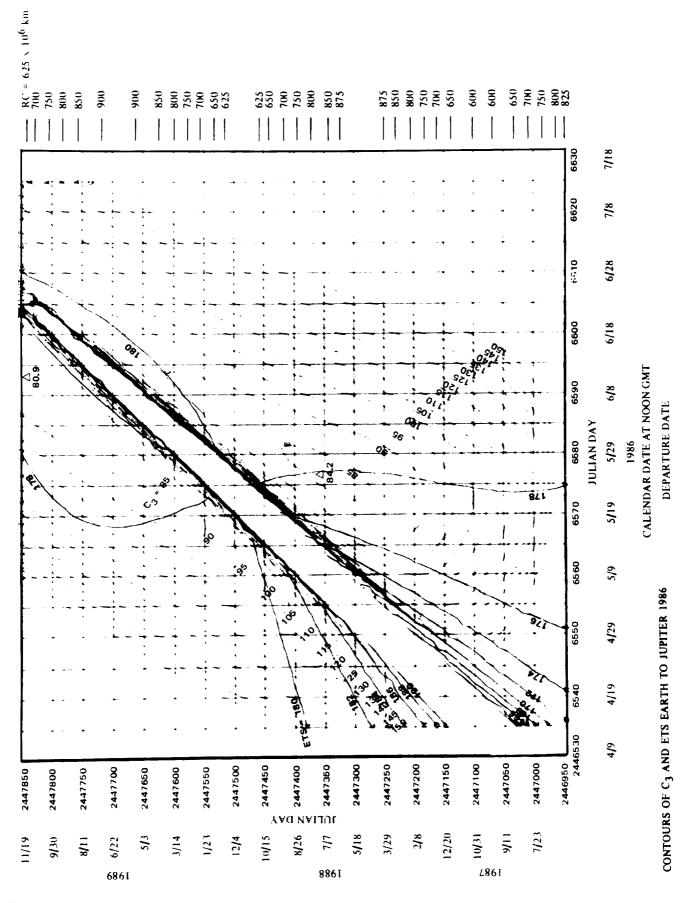
4-114

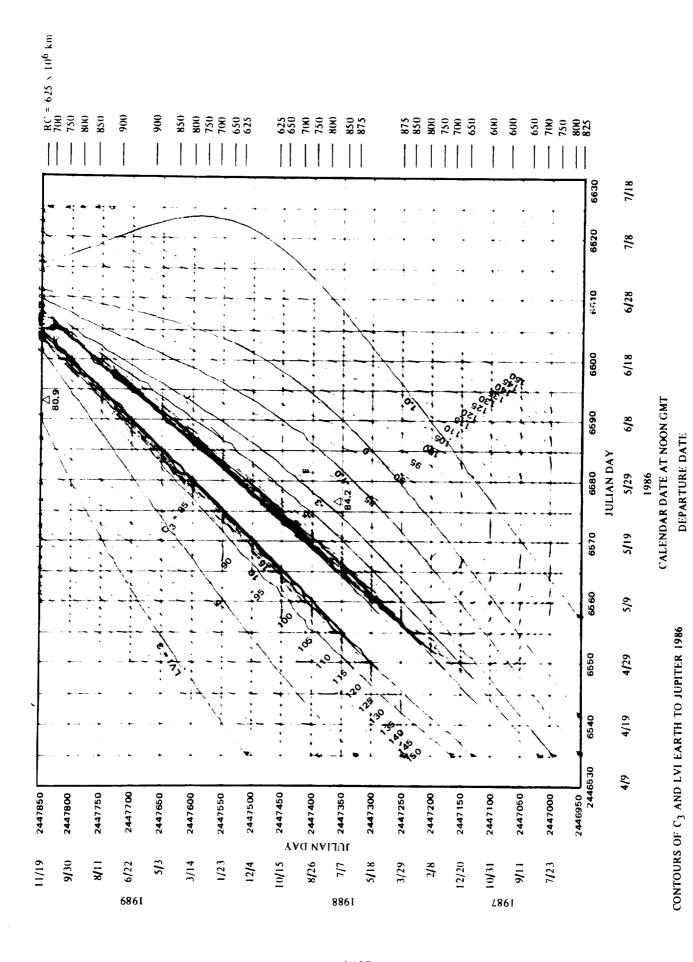


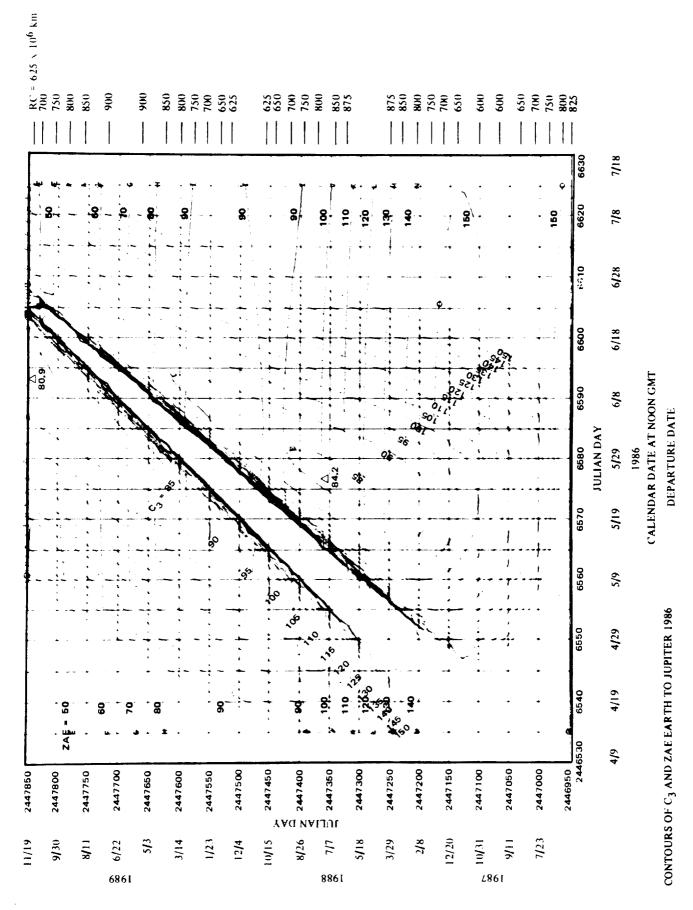


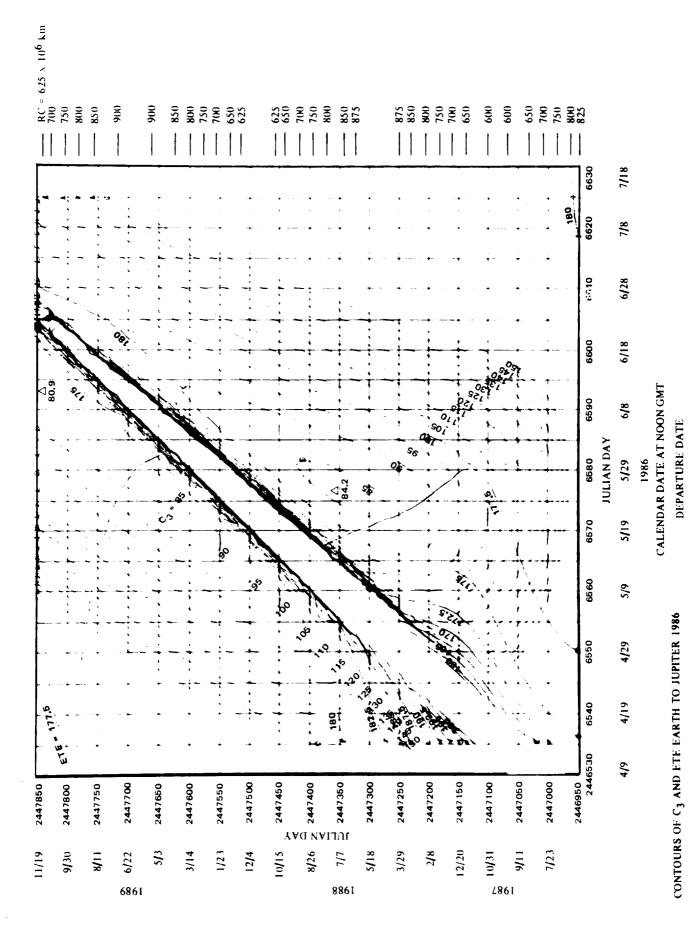


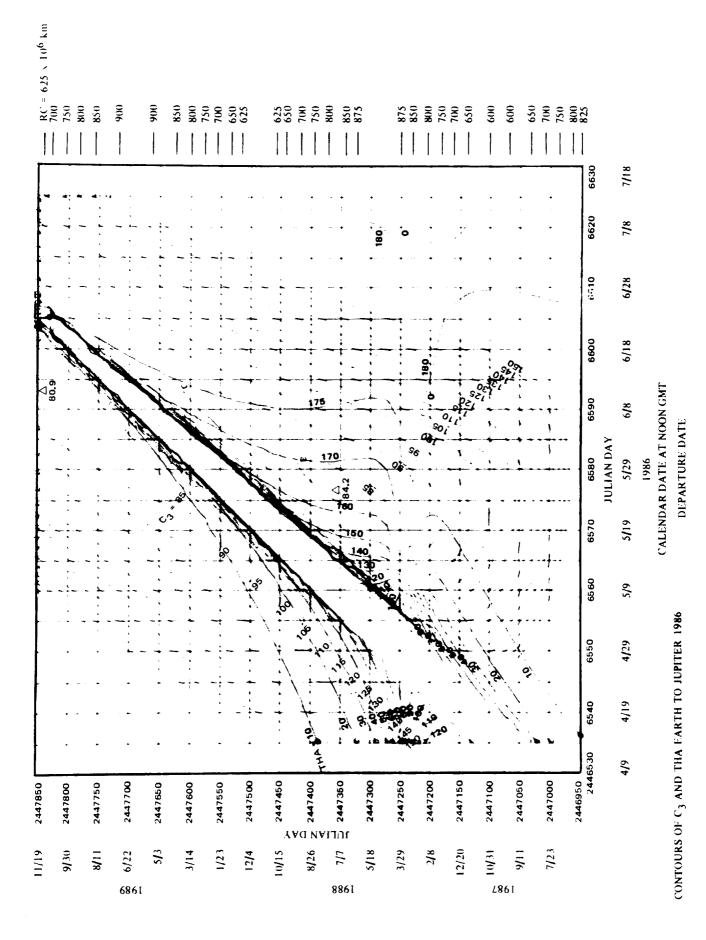


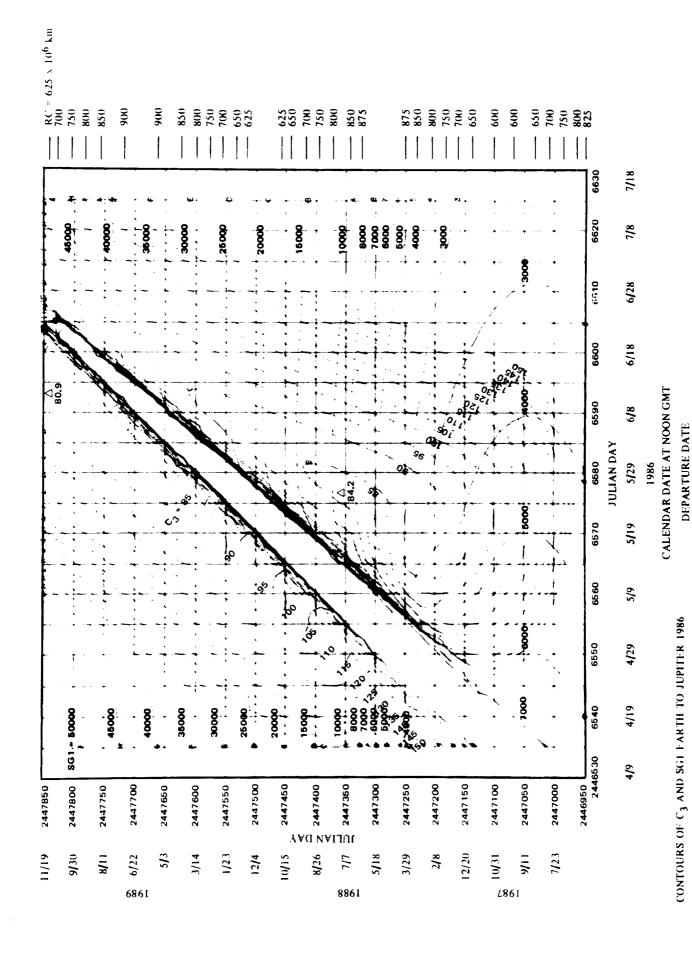




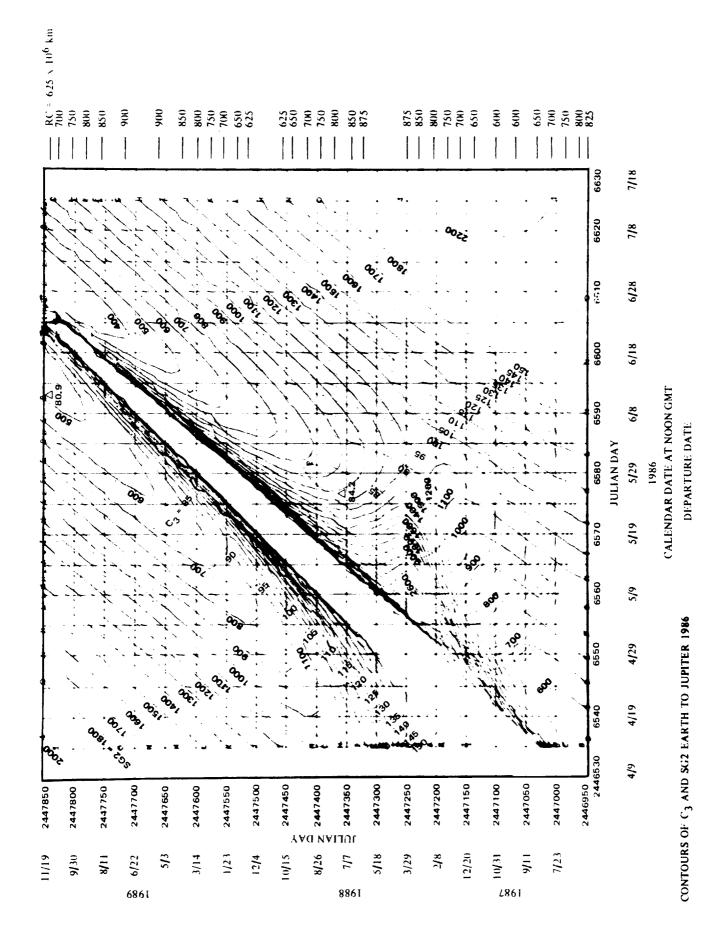


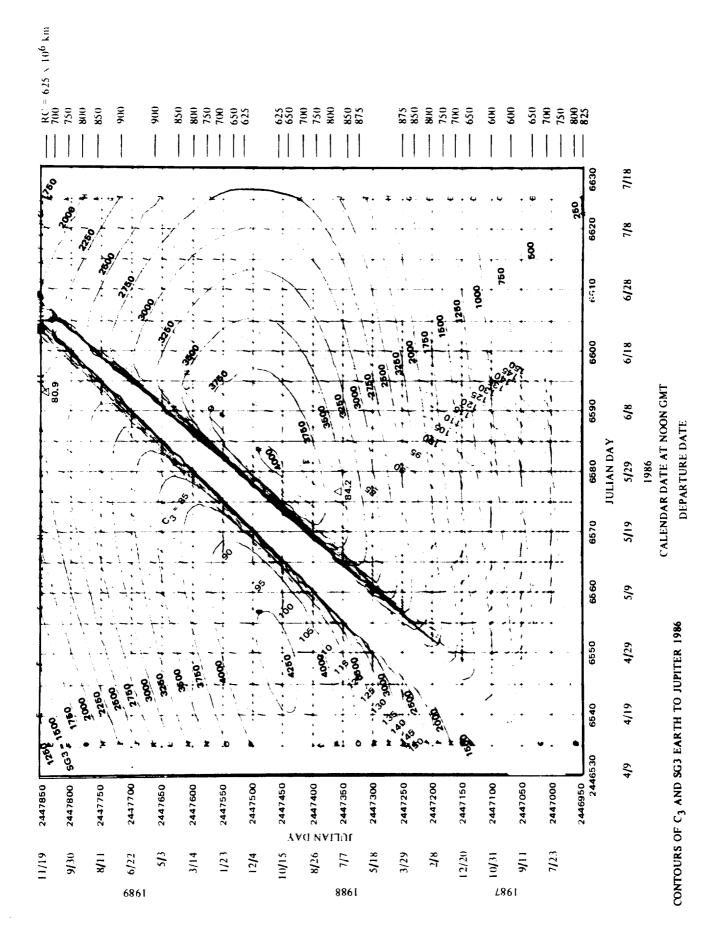


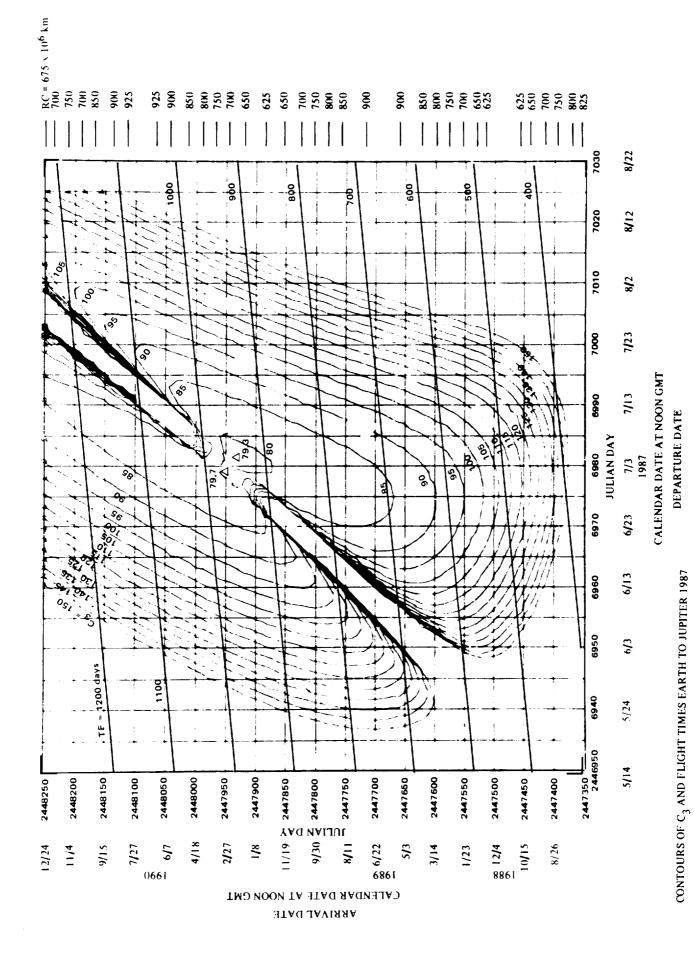


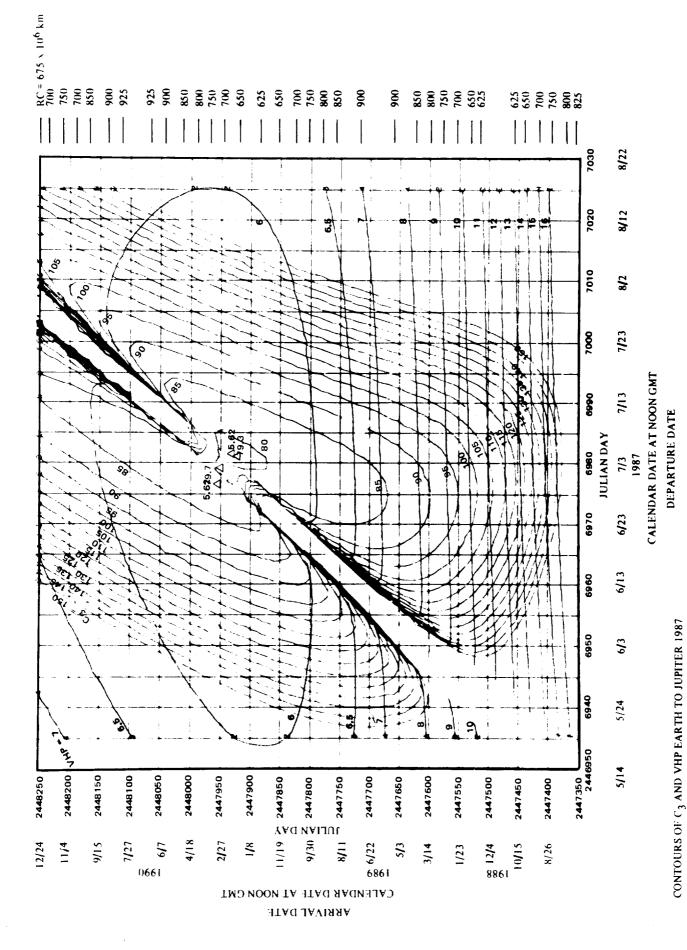


4-124

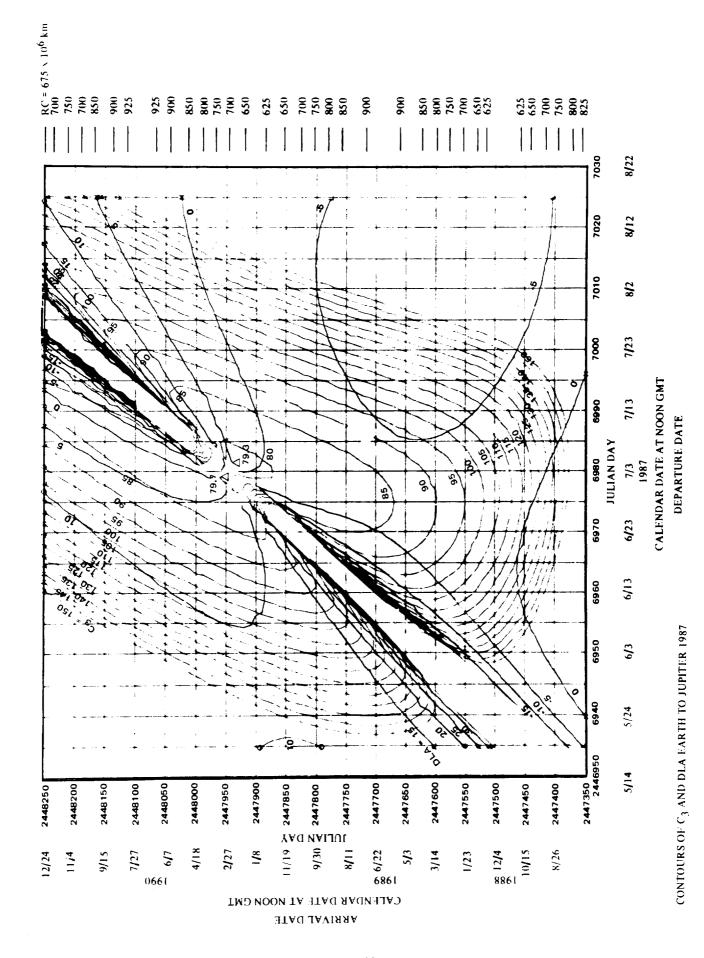




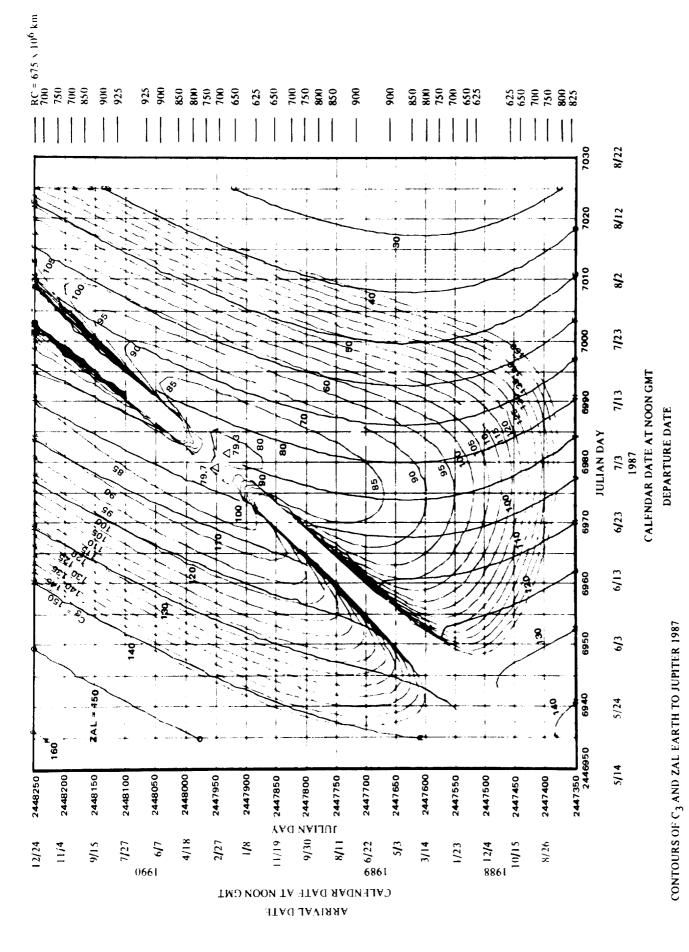


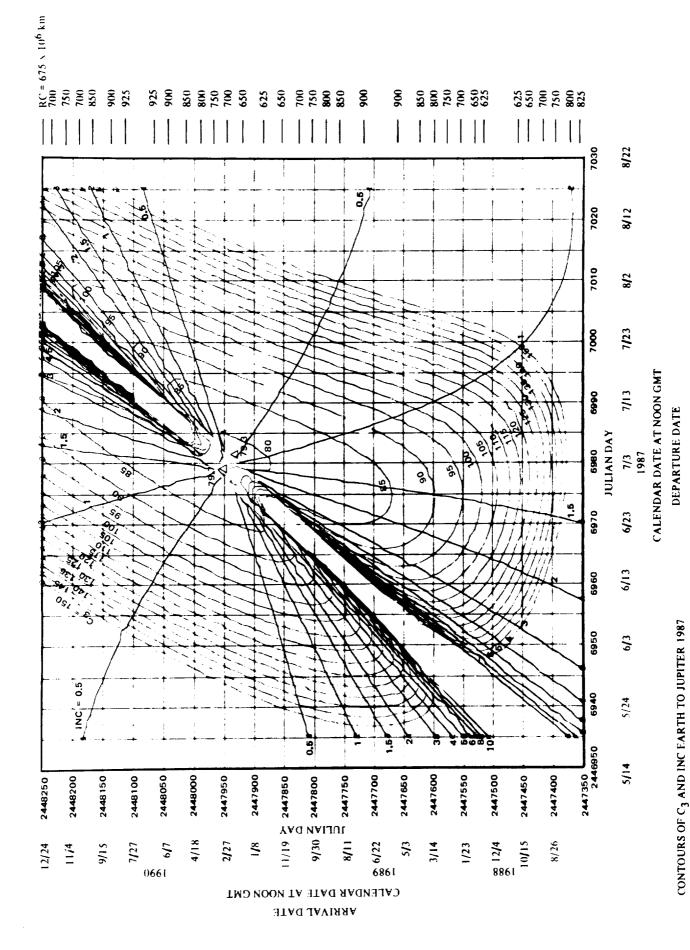


4-128

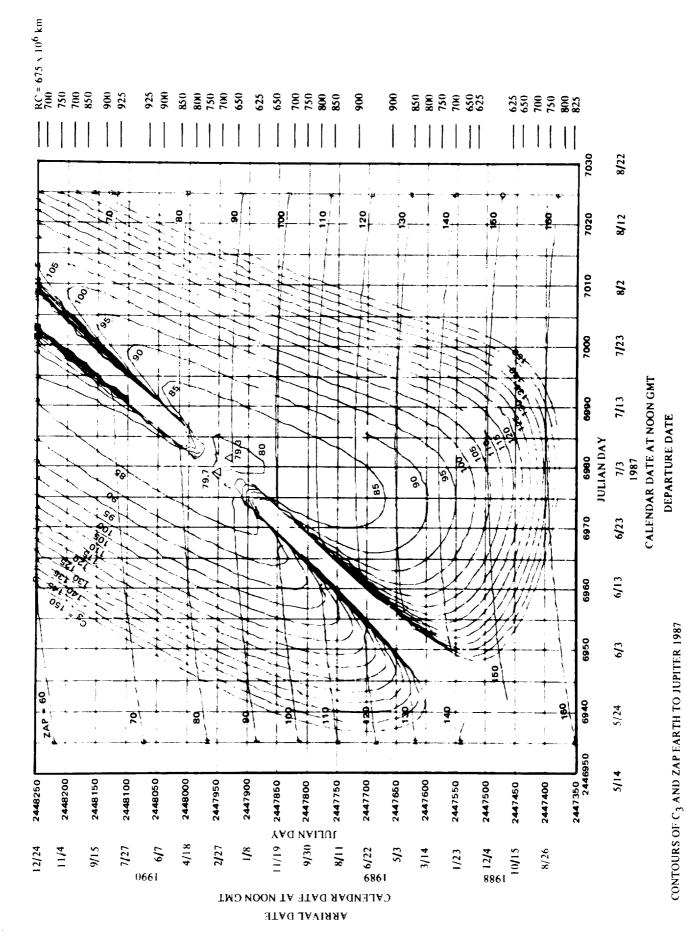


4-129

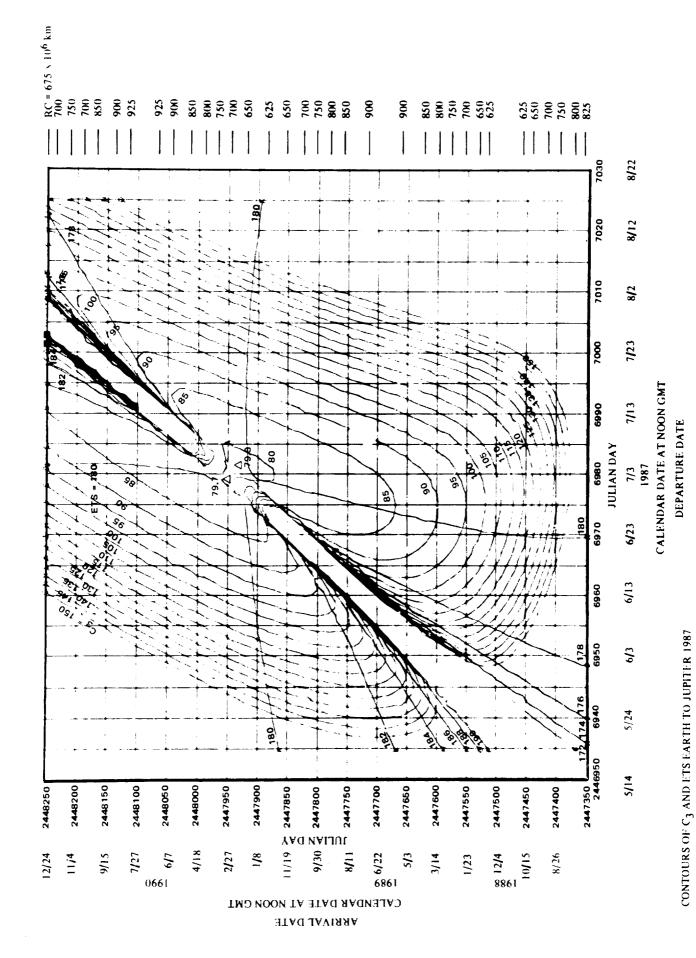




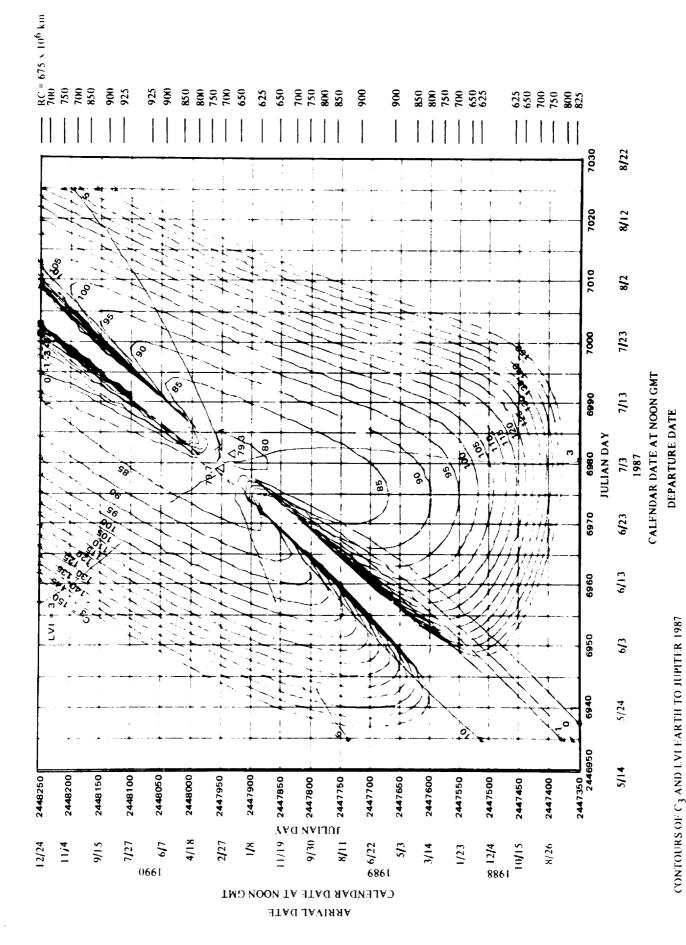
4-131



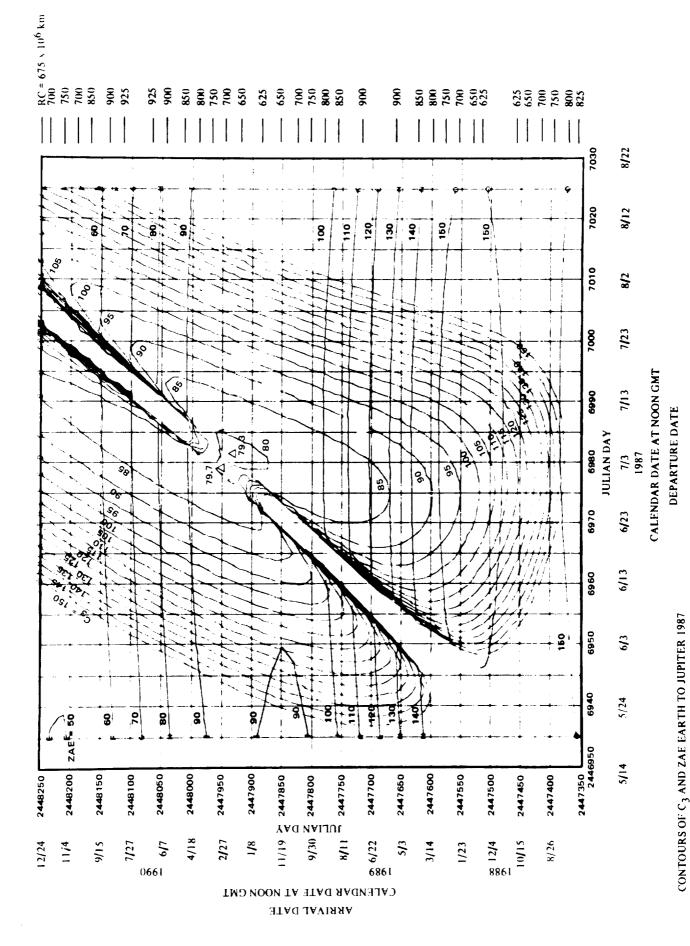
4-132



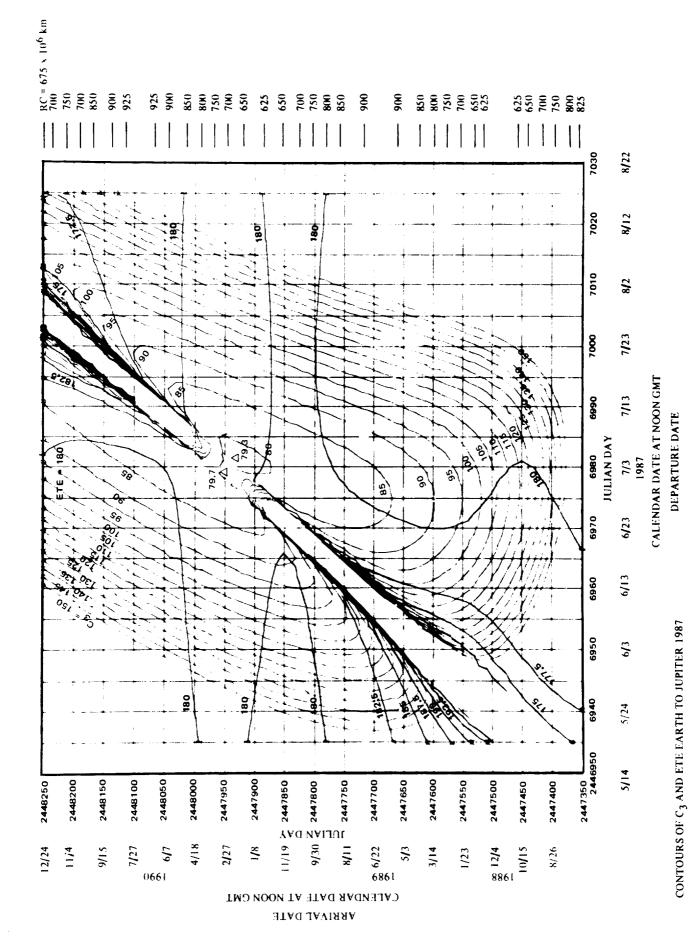
4-133



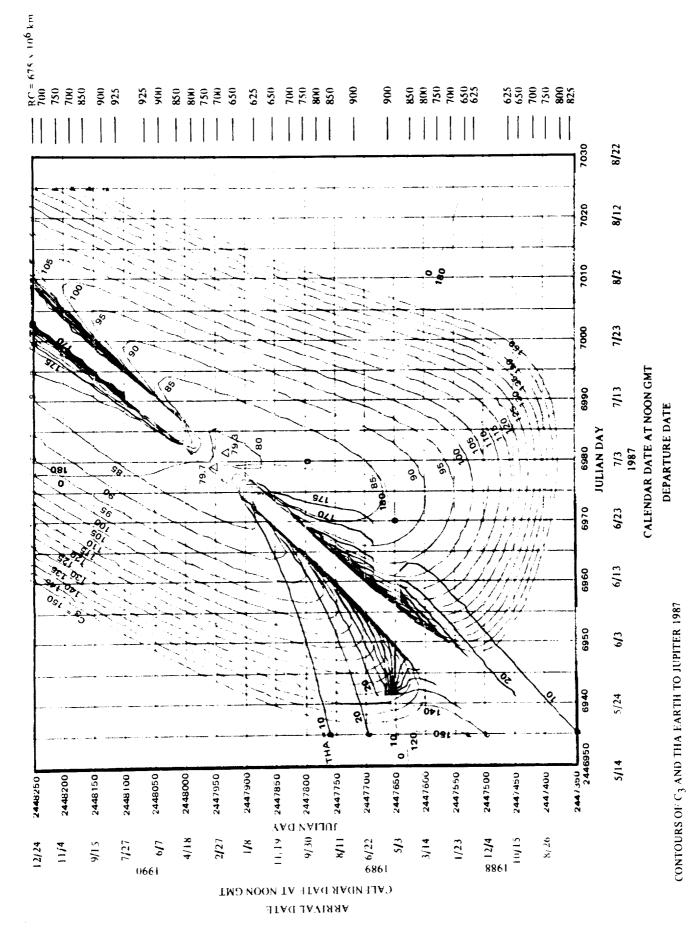
4-134

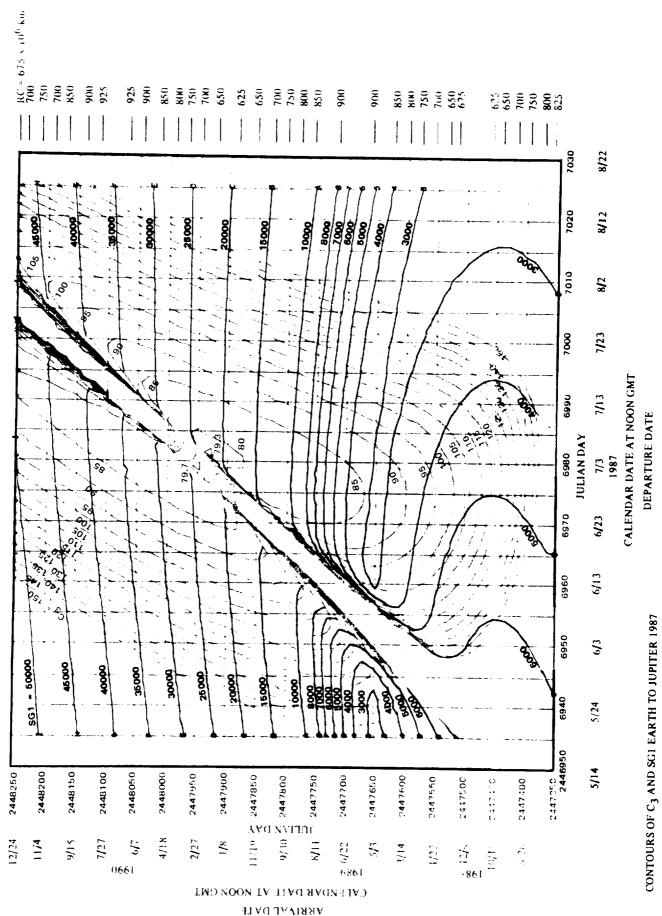


4-135

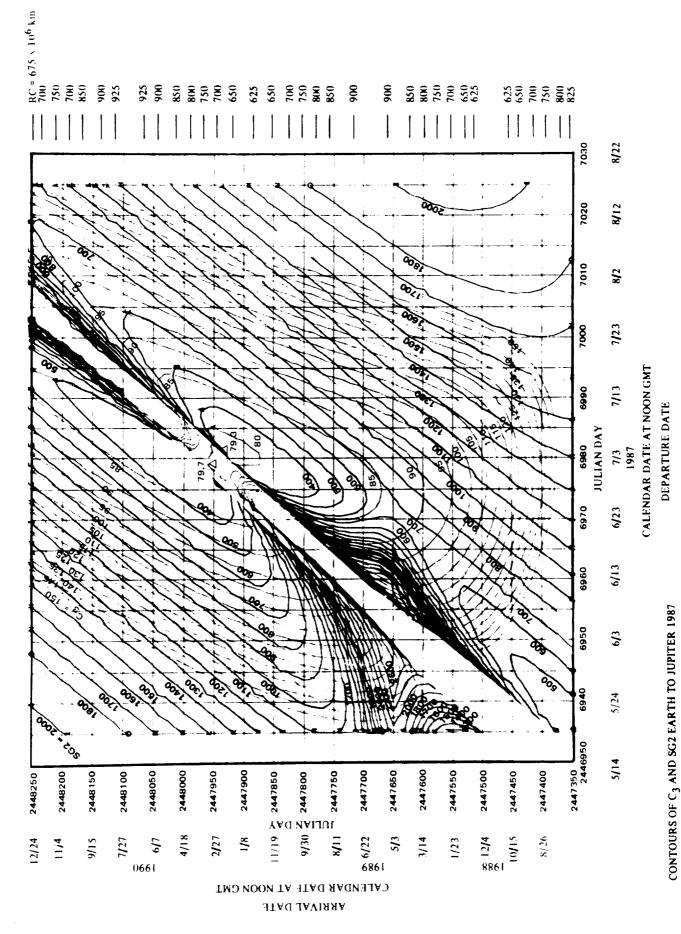


4-136

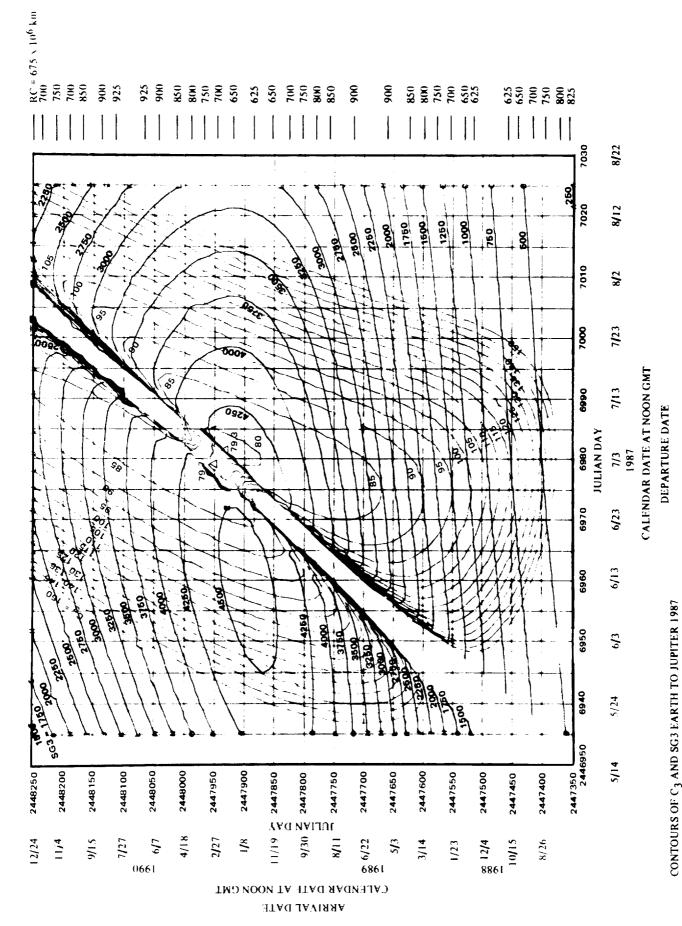




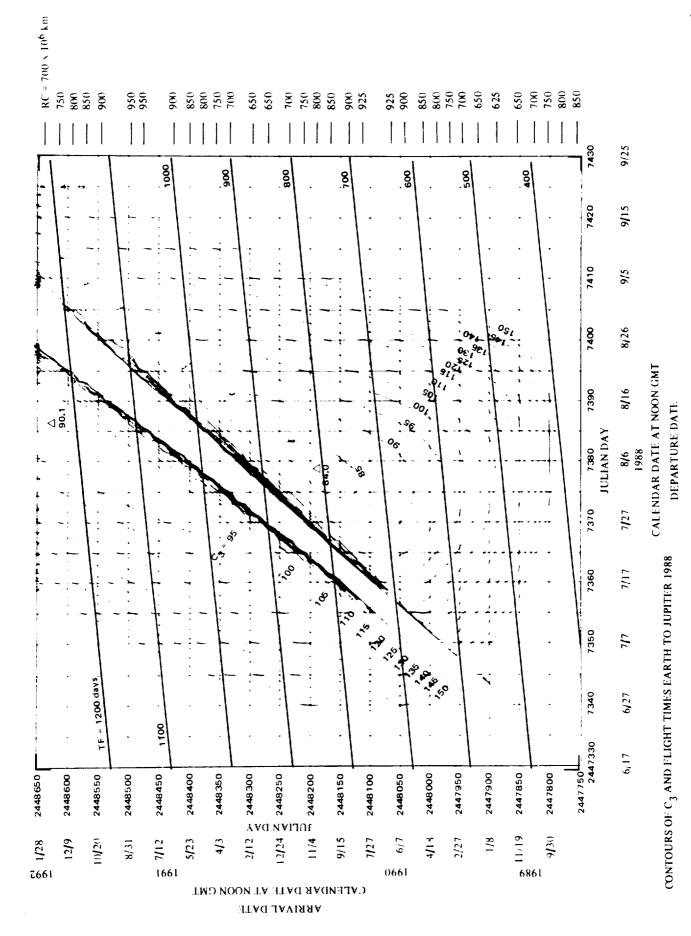
iid (i v

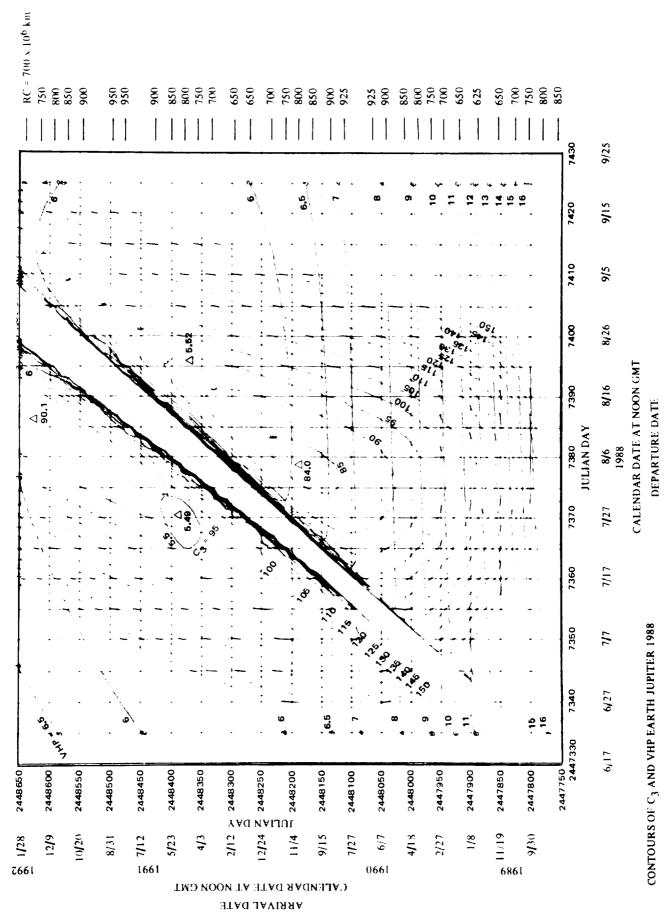


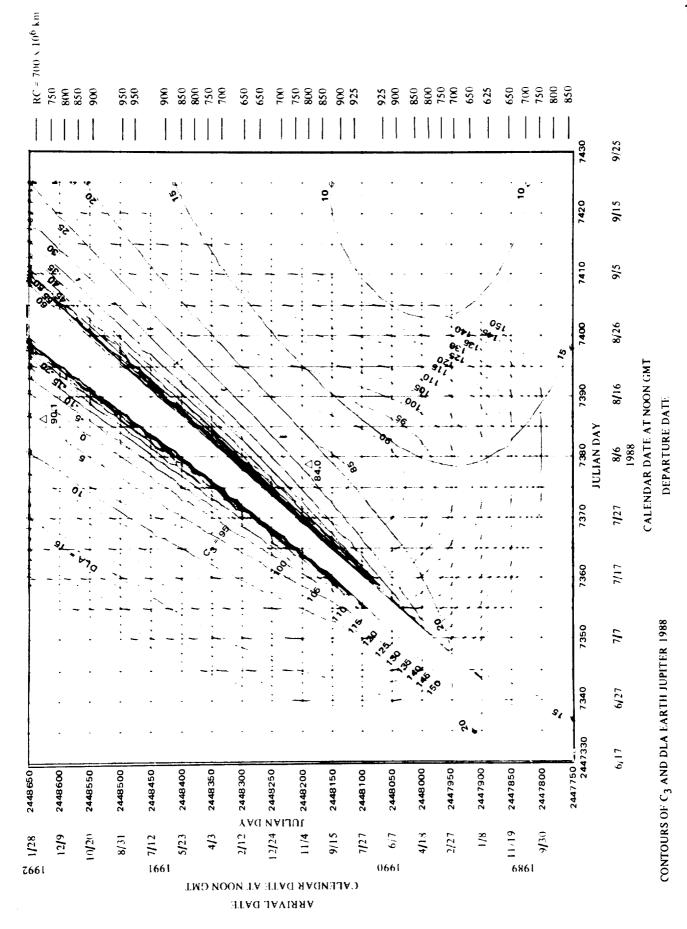
4-139

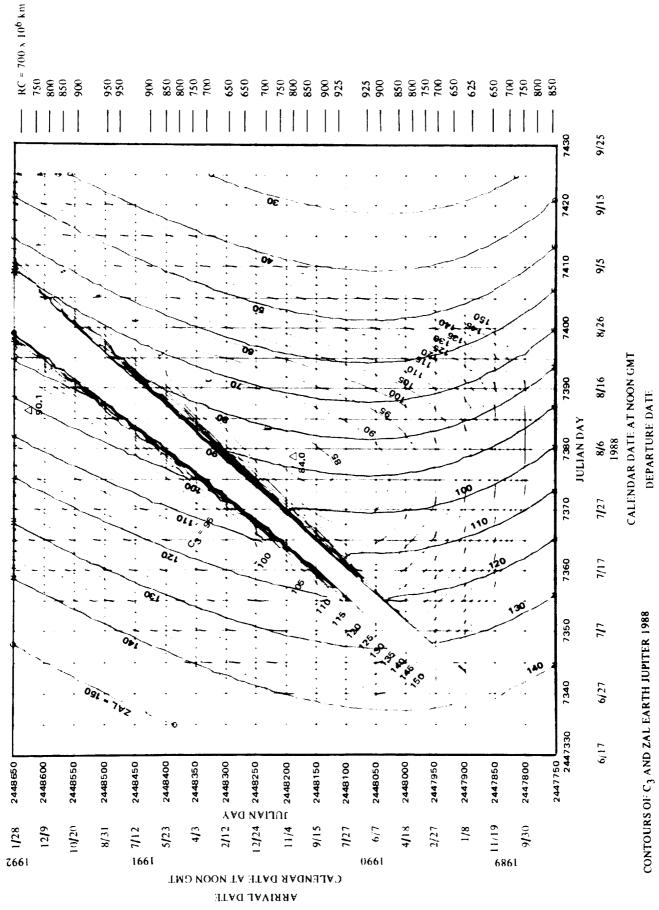


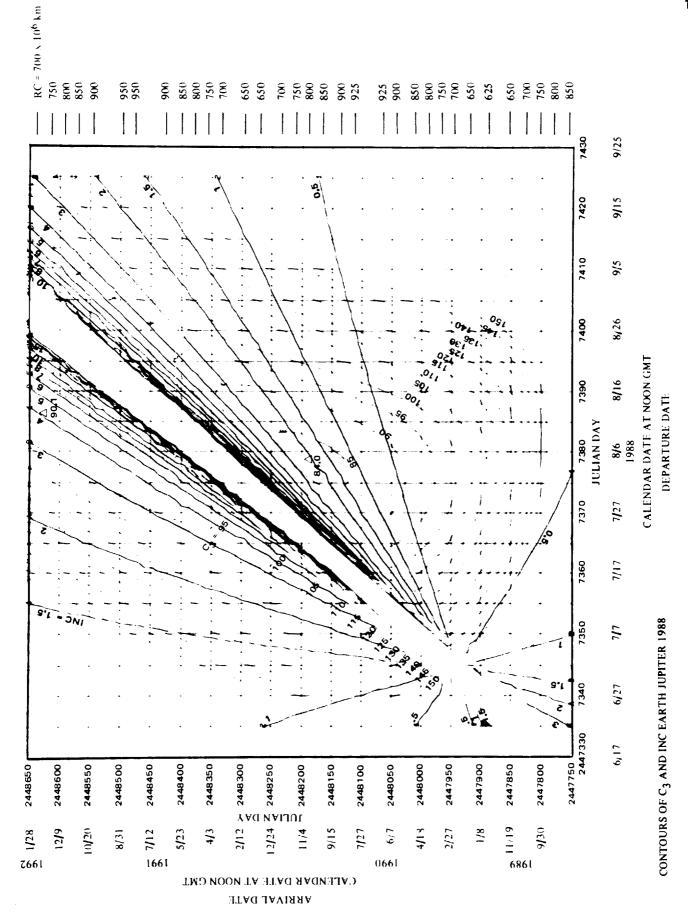
4-140

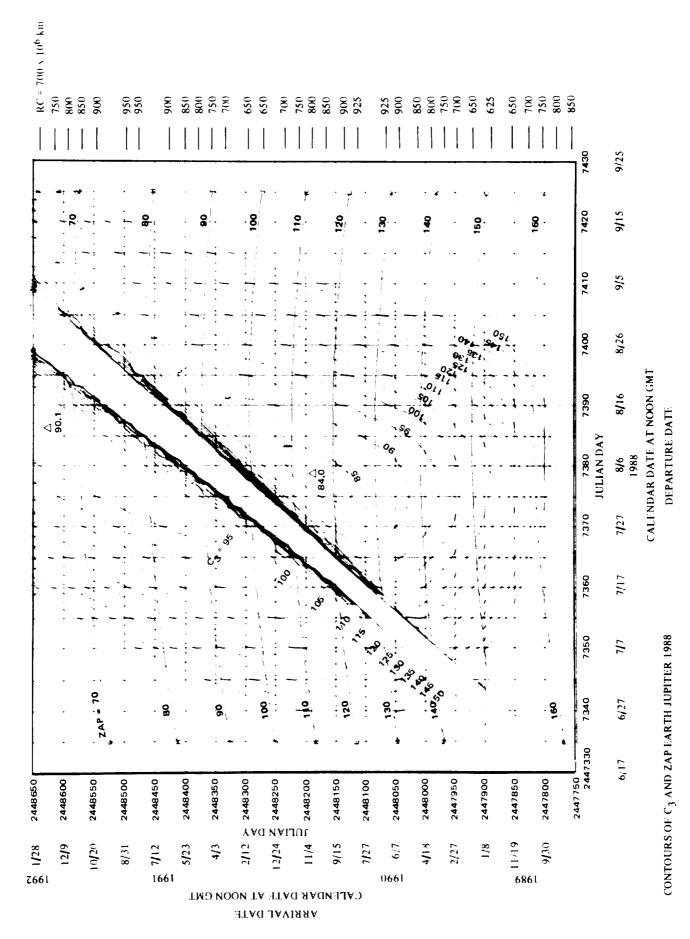


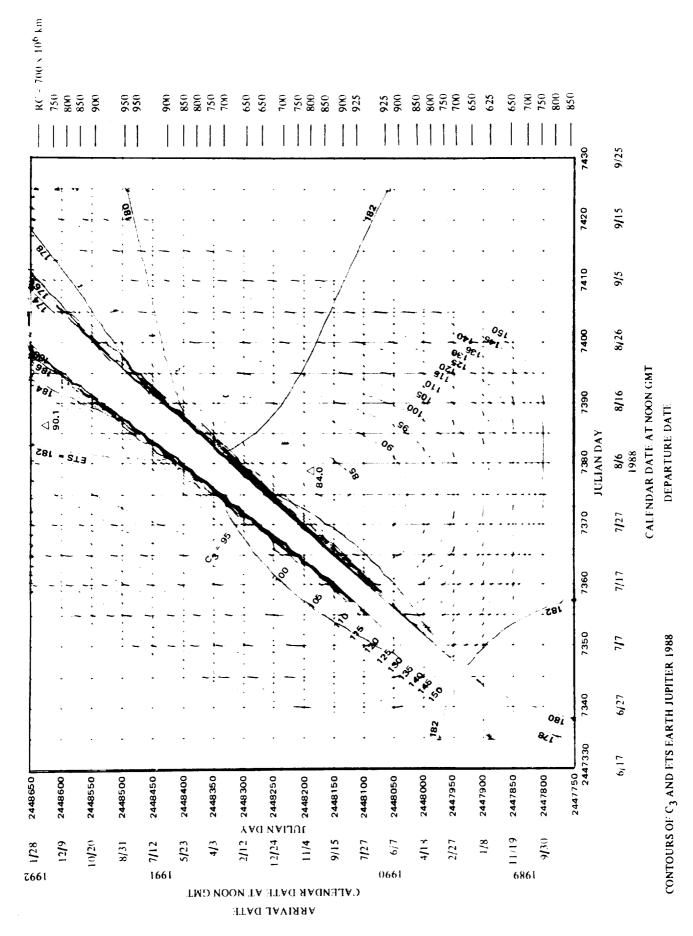


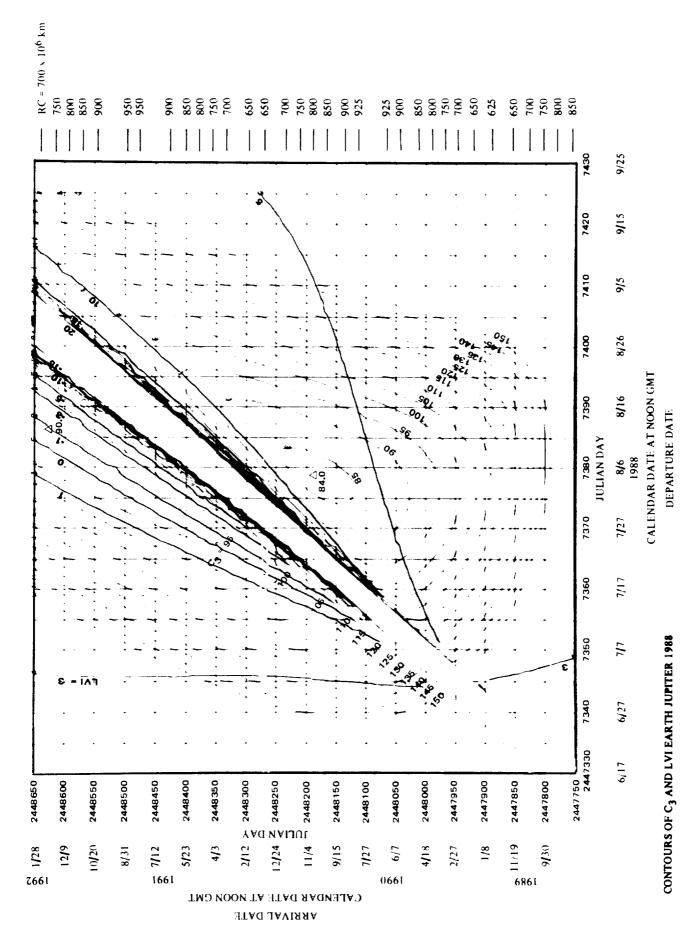




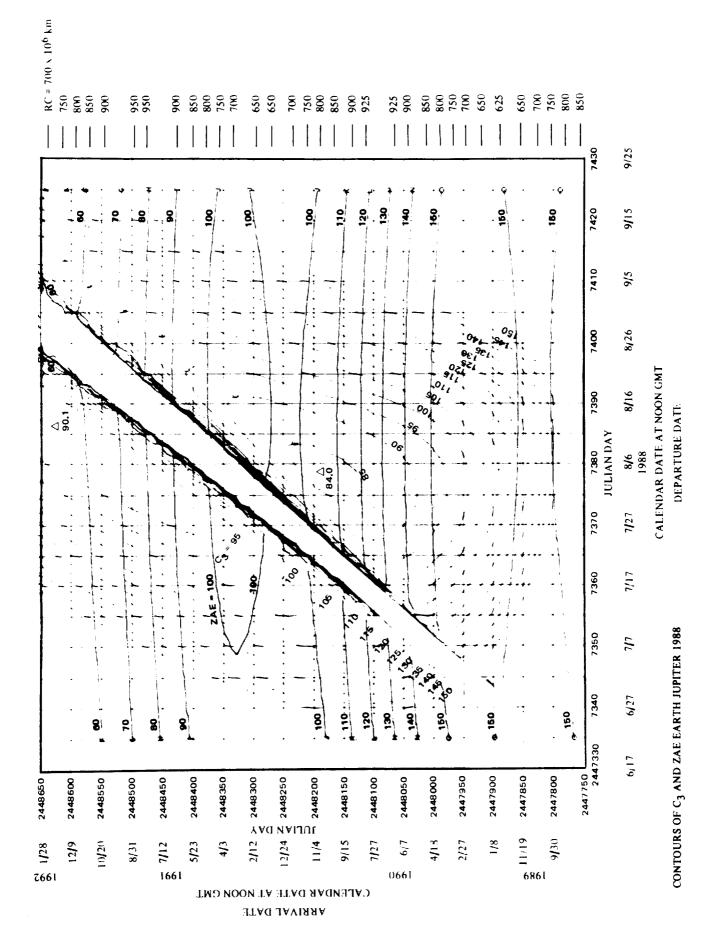




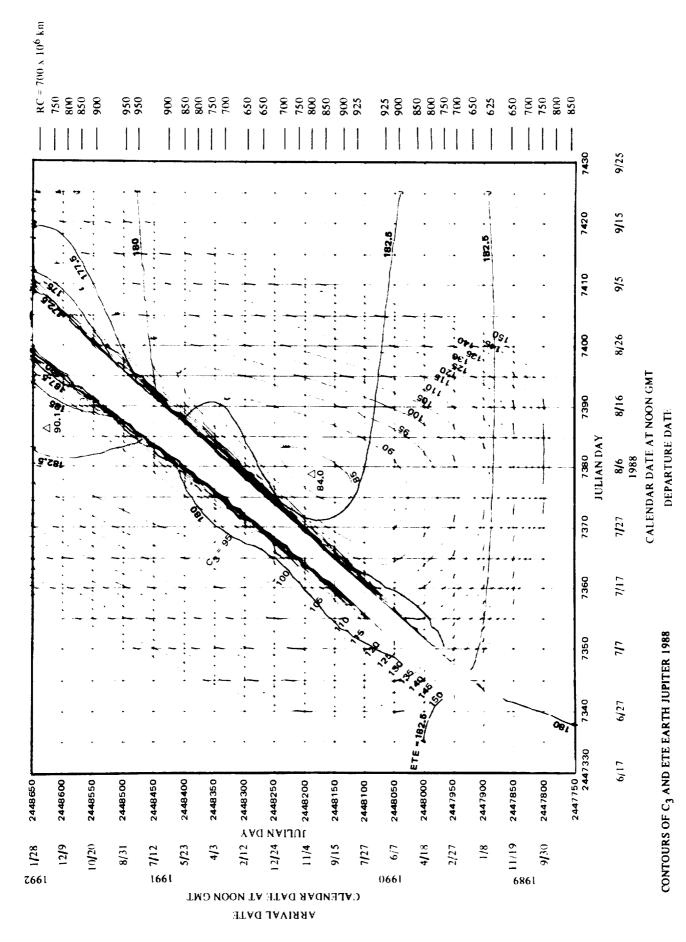


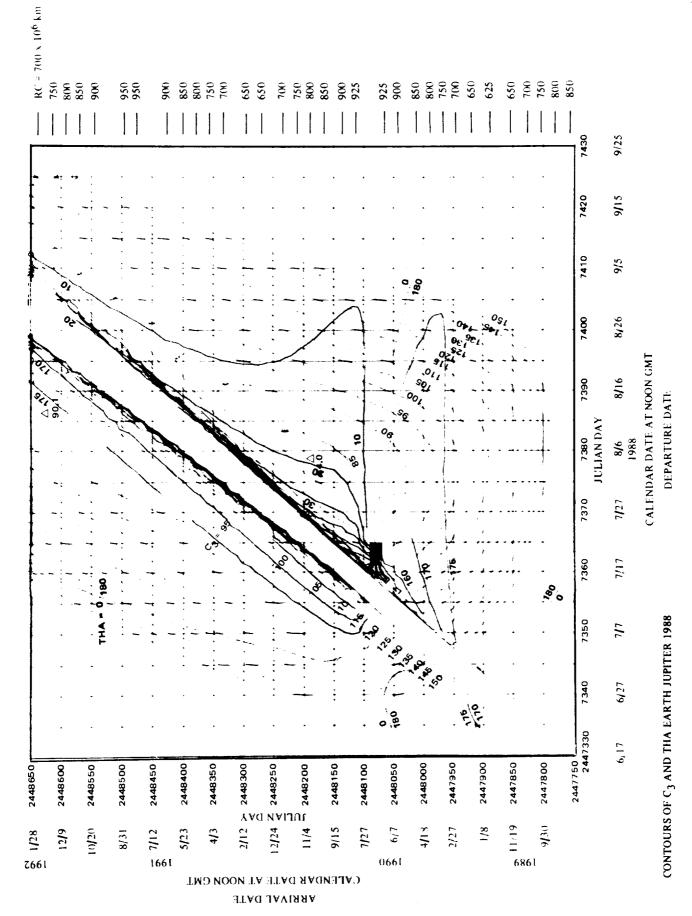


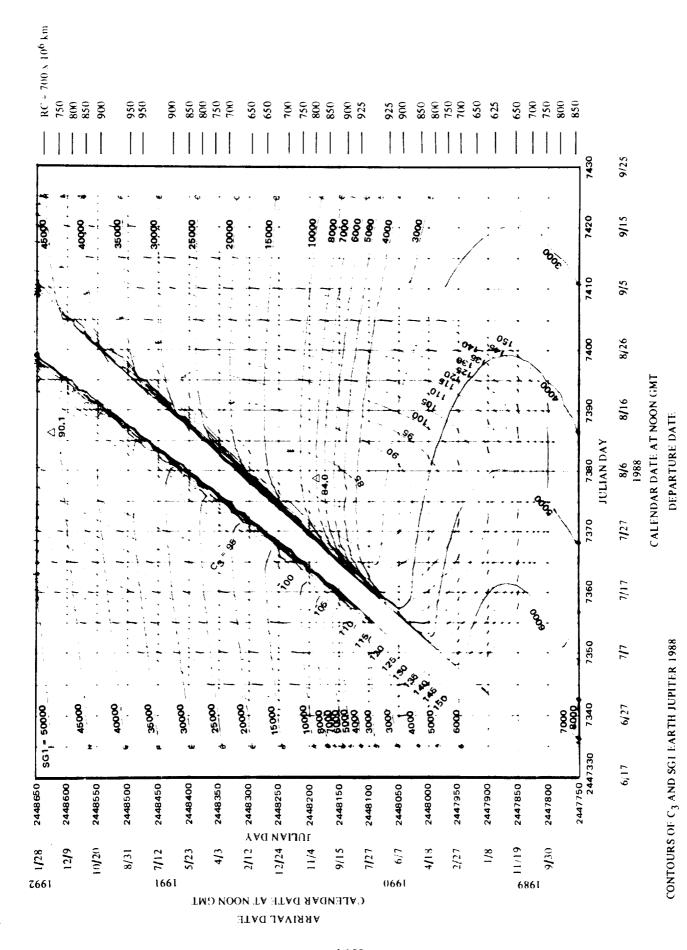
4-148



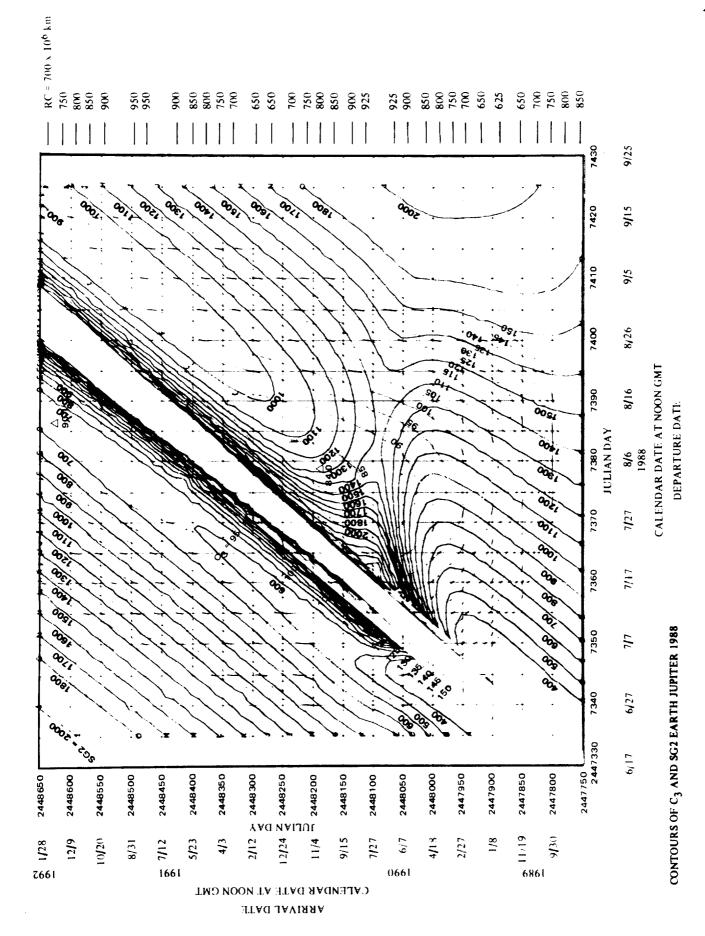
4-149

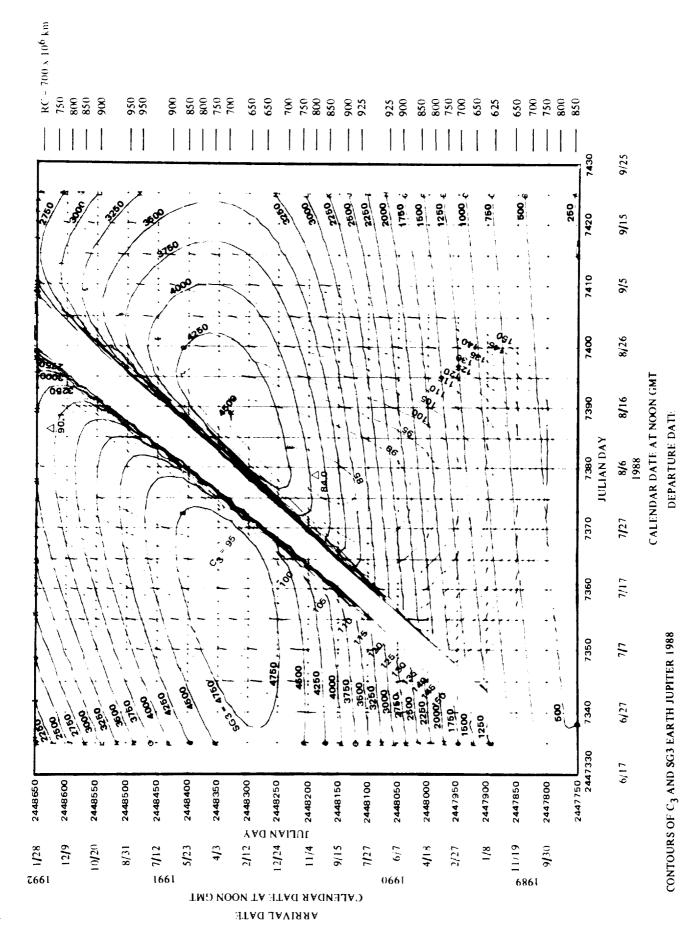




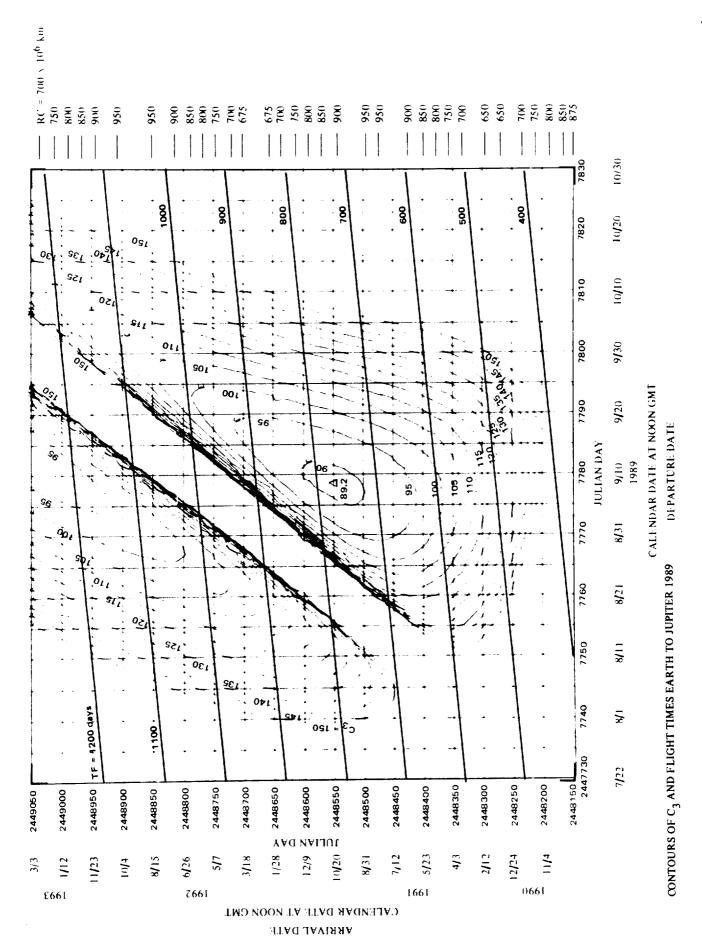


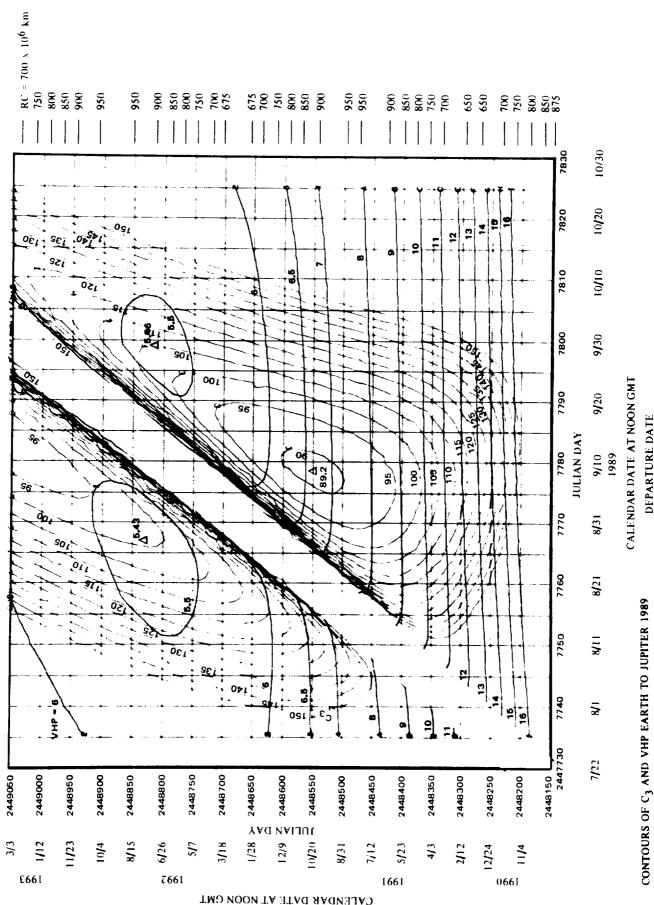
4-152





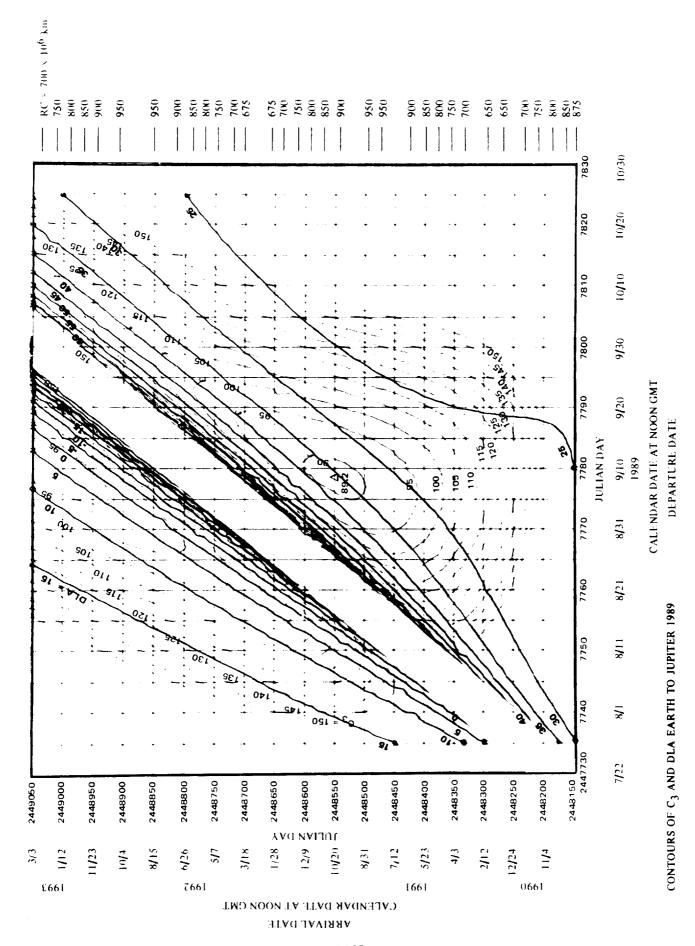
4-154





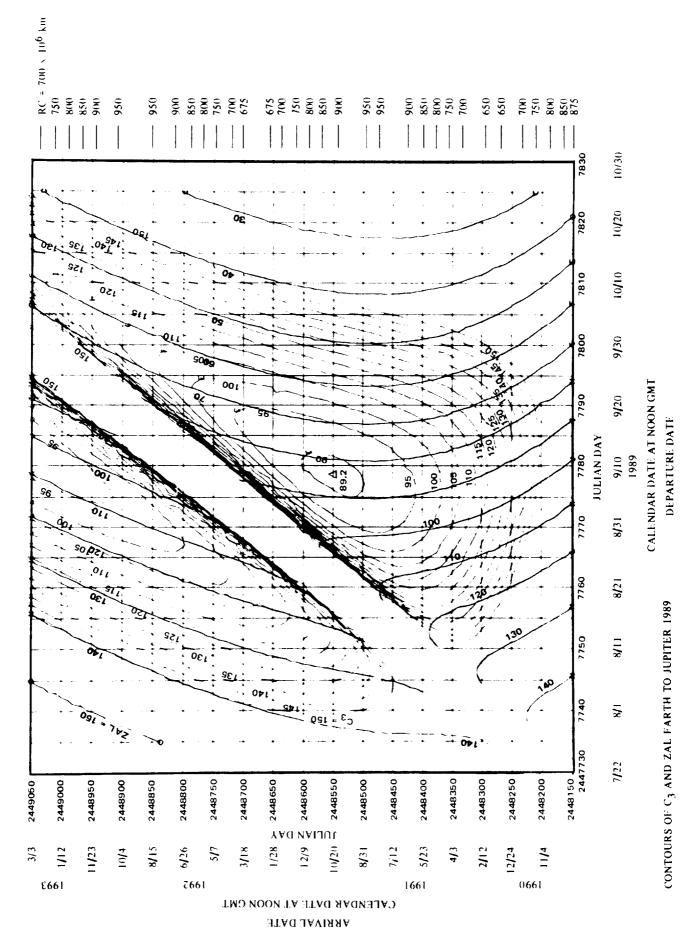
DEPARTURE DATE

CALENDAR DATE AT NOON GMT ARRIVAL DATE

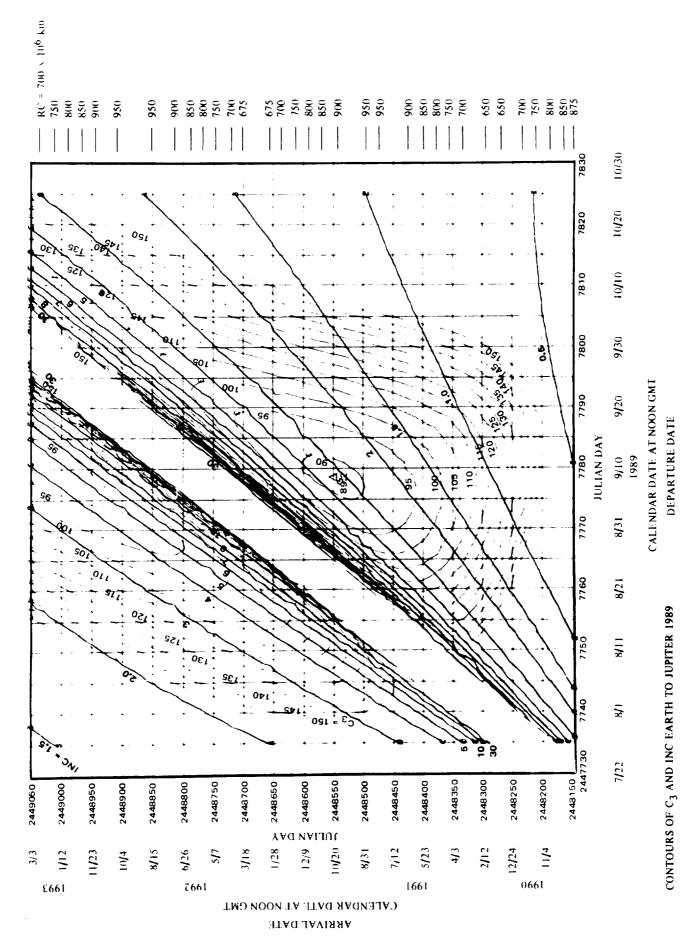


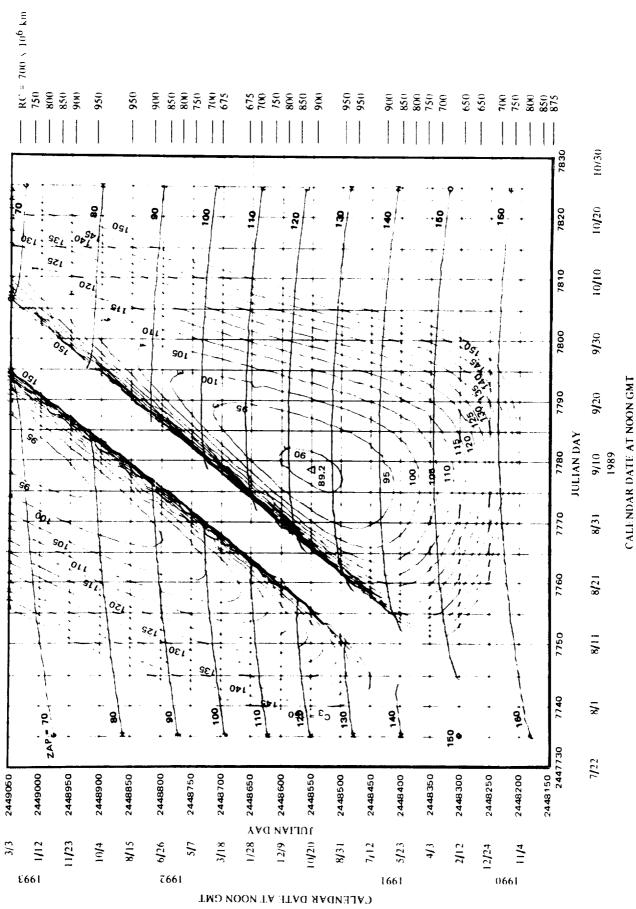


1



4-158

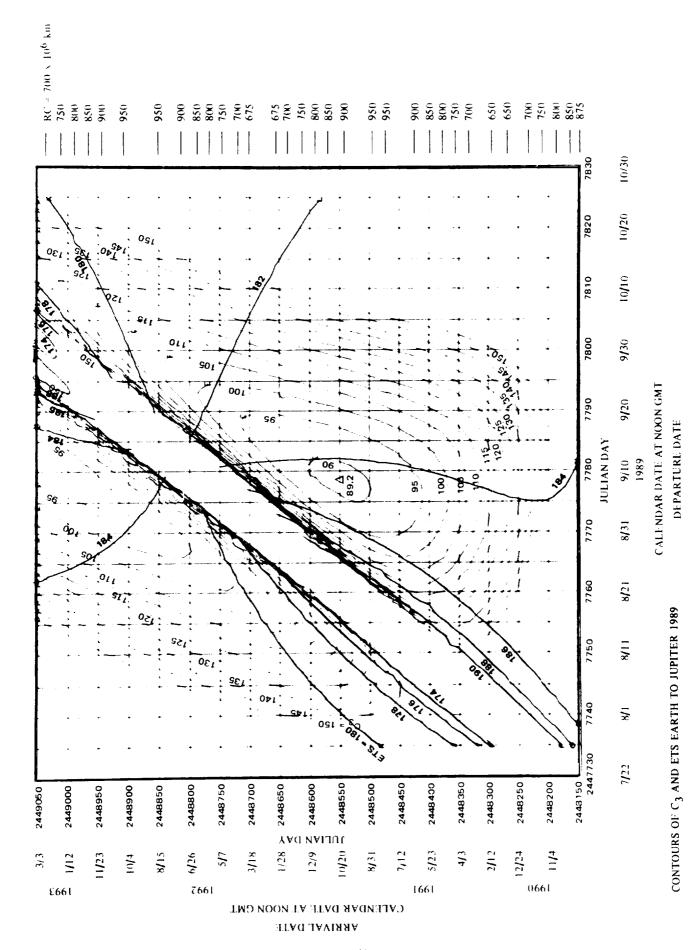




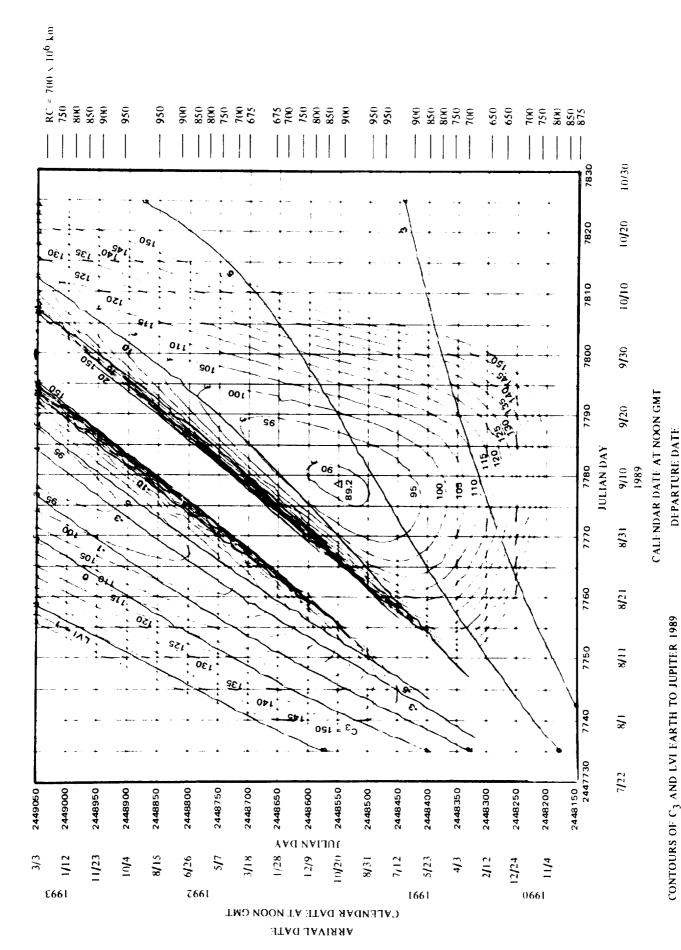
DEPARTURE DATE

CONTOURS OF C₃ AND ZAP EARTH TO JUPITER 1989

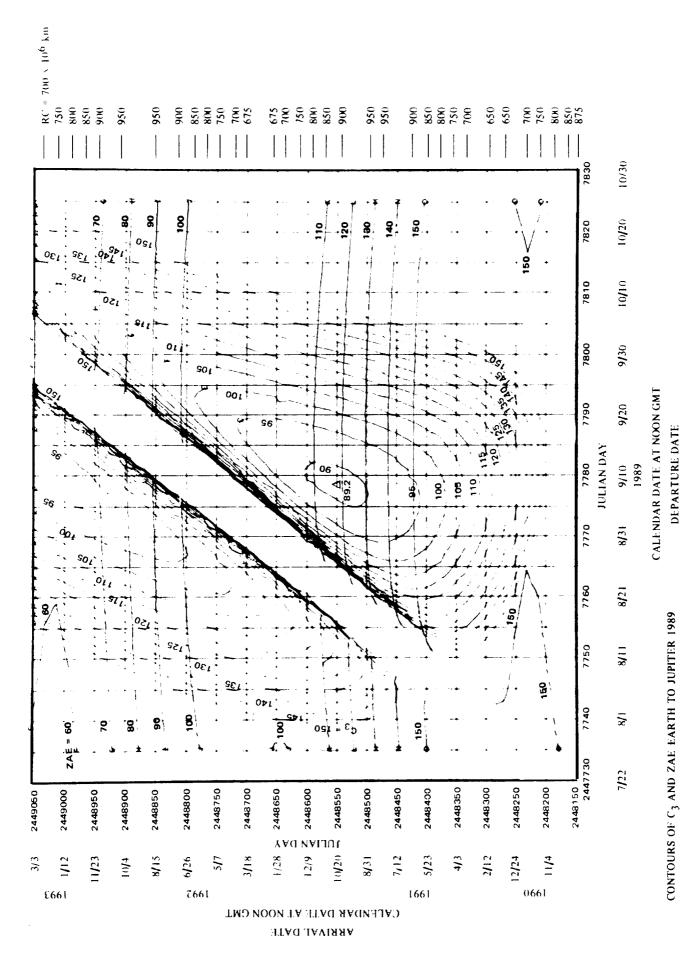
ARRIVAL DATE



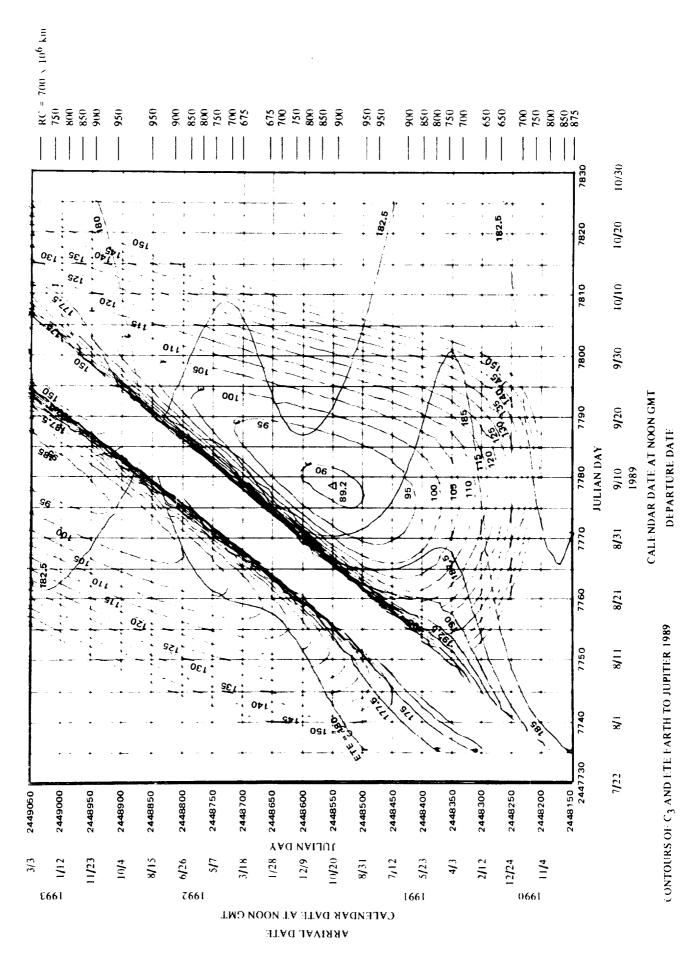
4-161



4-162



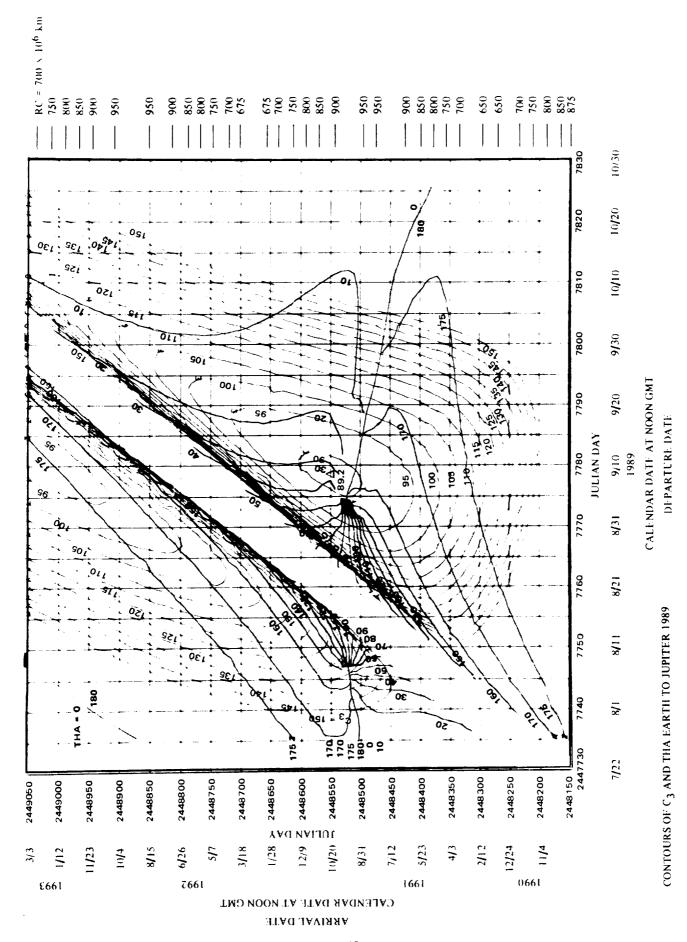
4-163



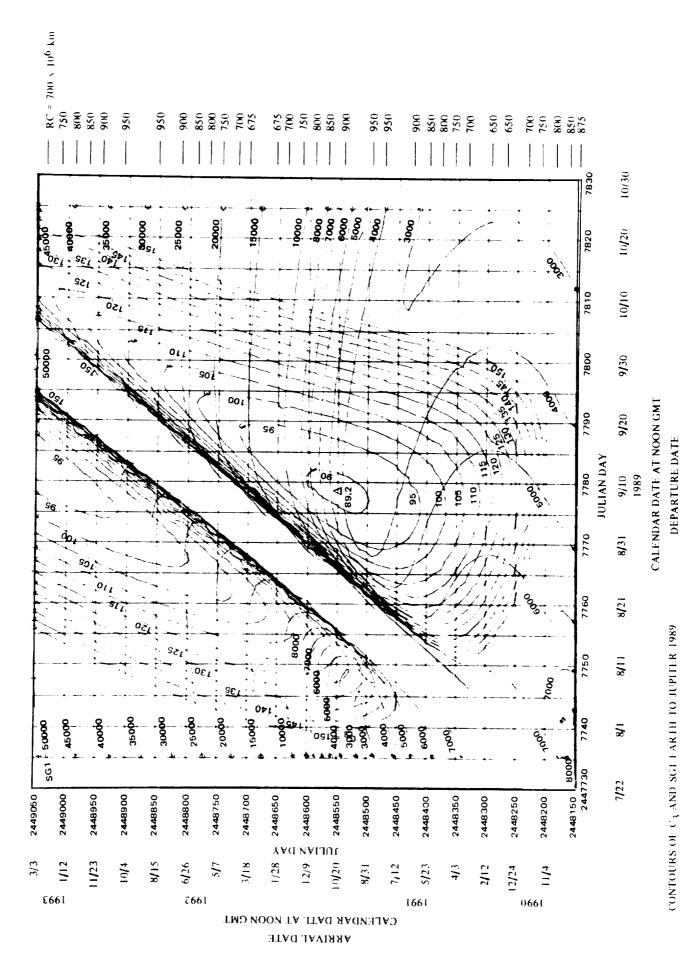
E

1

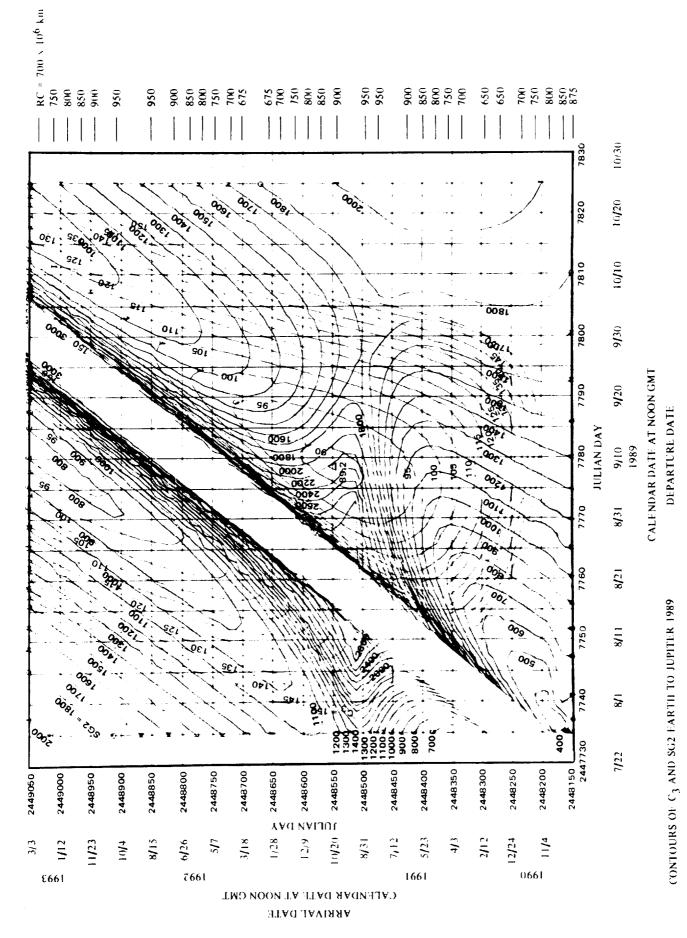
4-164

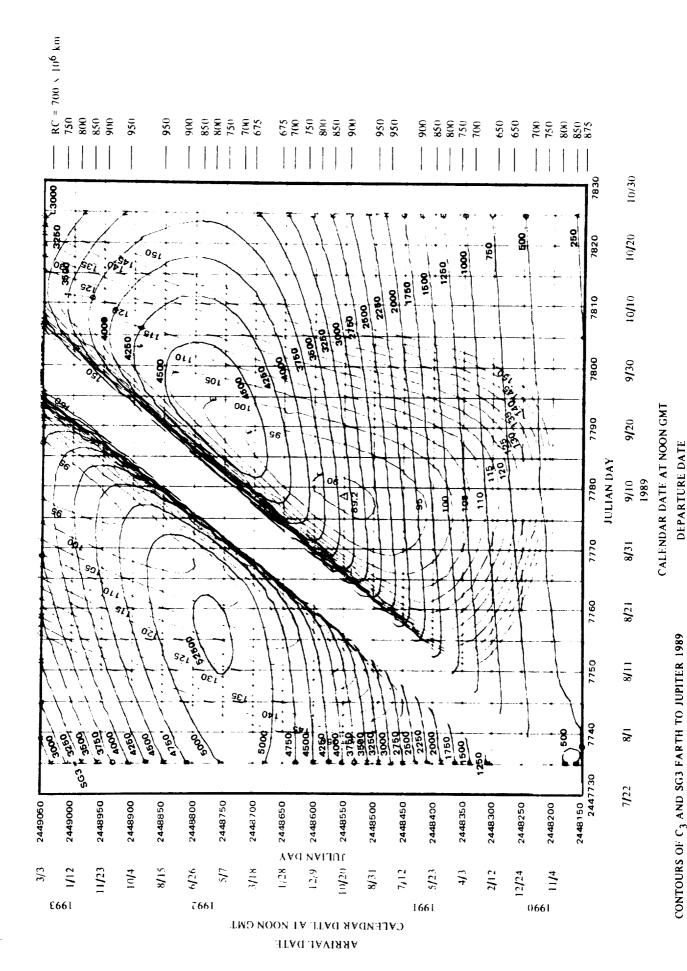


4-165

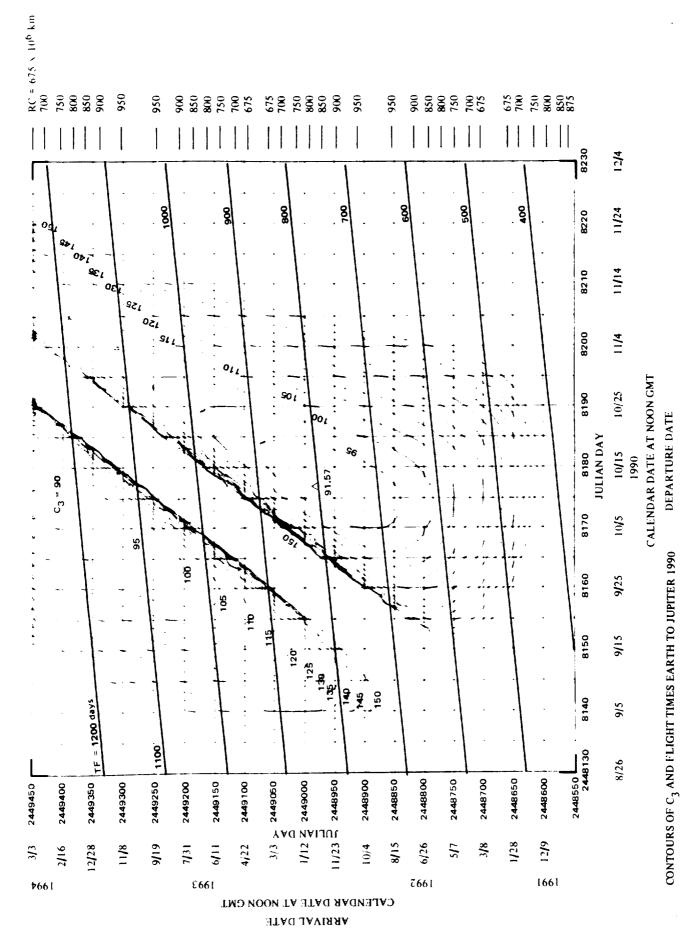


4-166

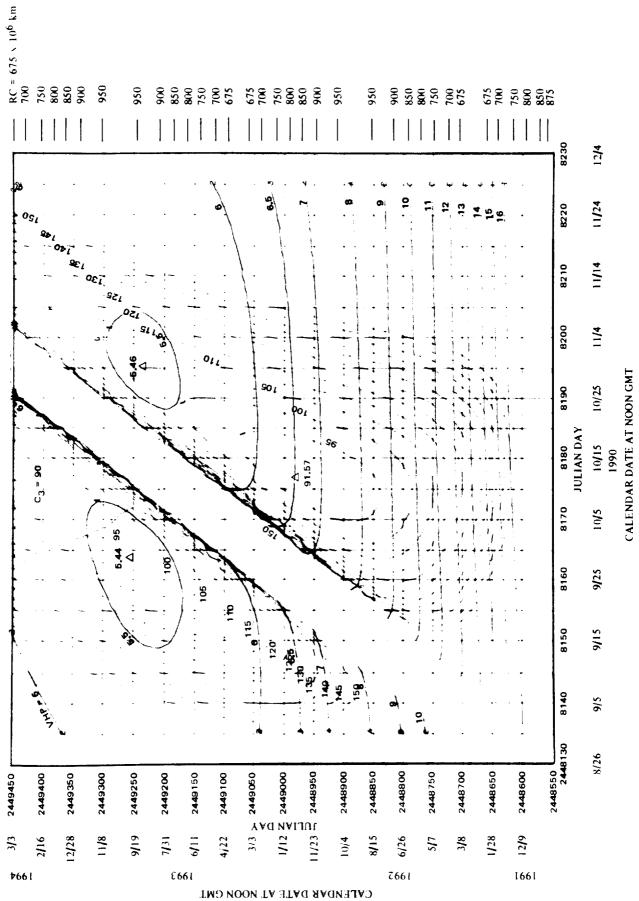




4-168



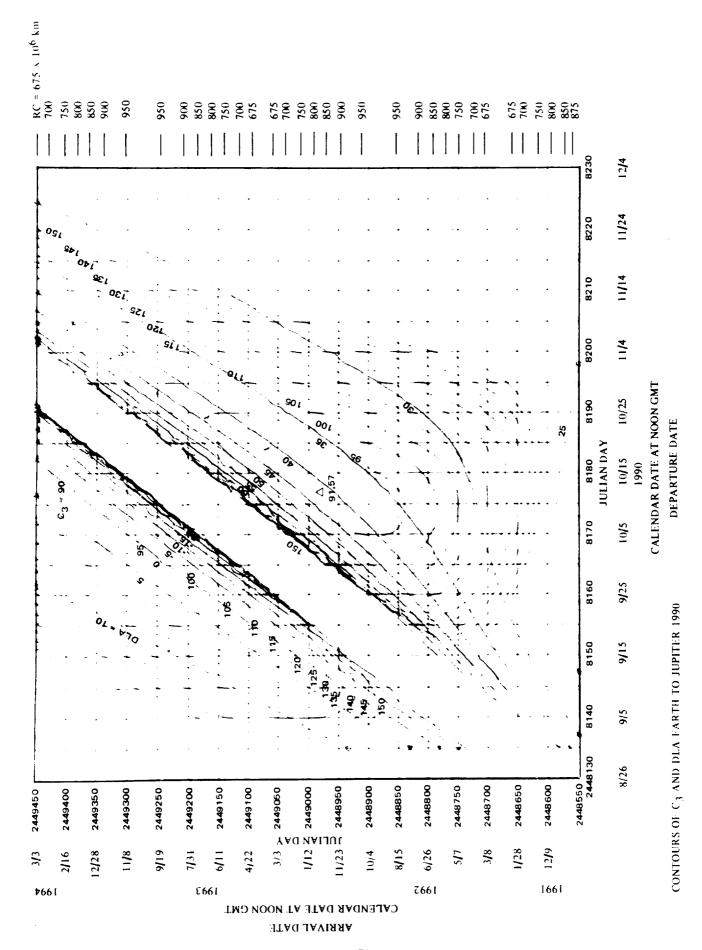
4-169



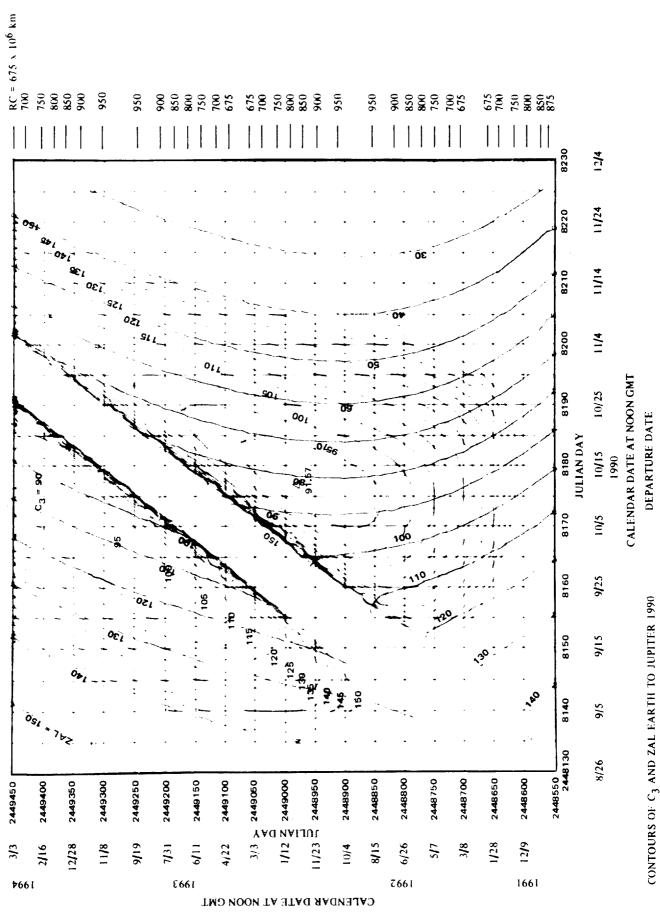
DEPARTURE DATE

CONTOURS OF C₃ AND VHP EARTH TO JUPITER 1990

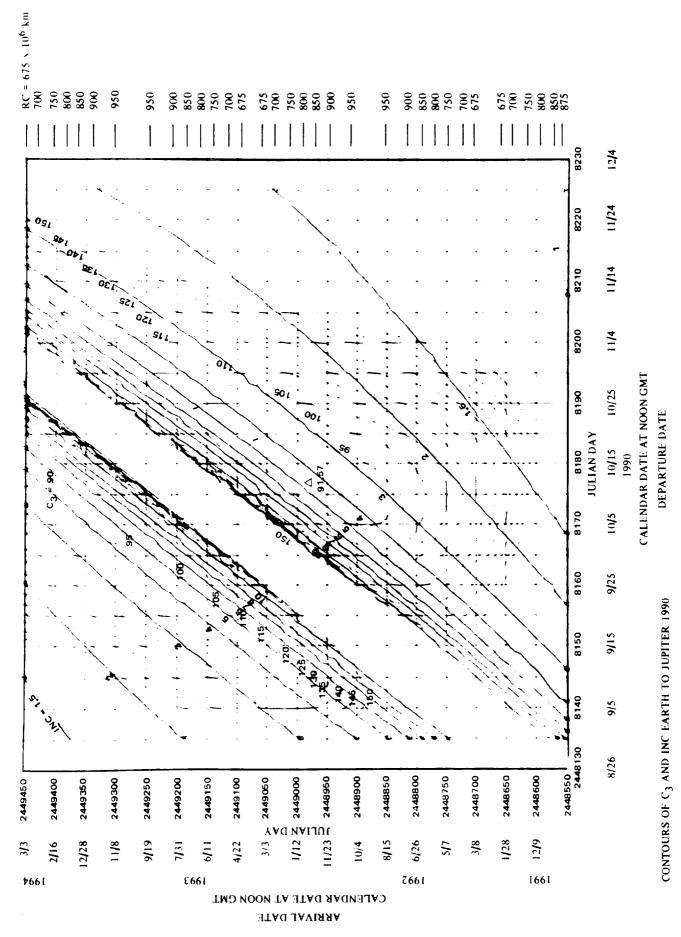
ARRIVAL DATE



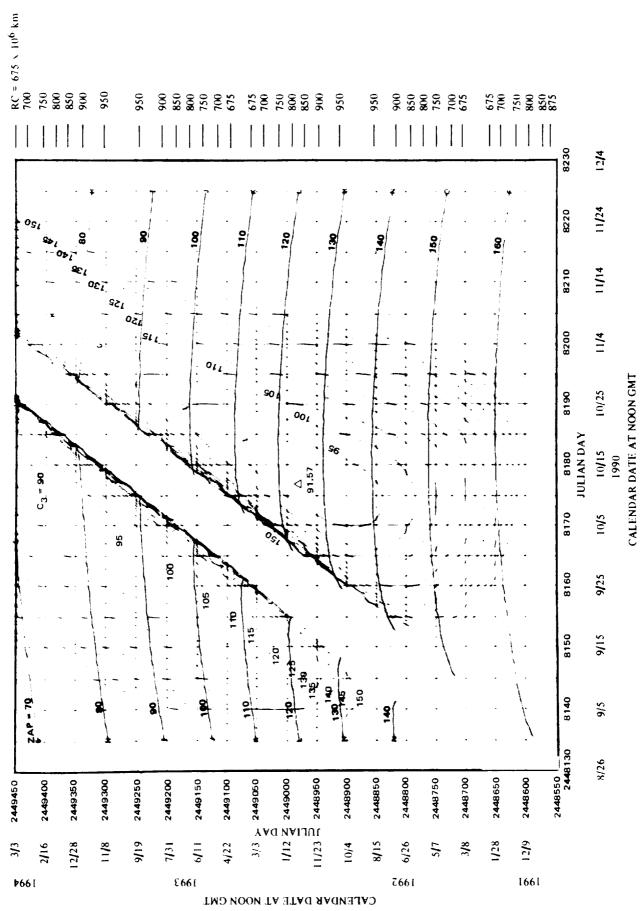
4-171



ARRIVAL DATE



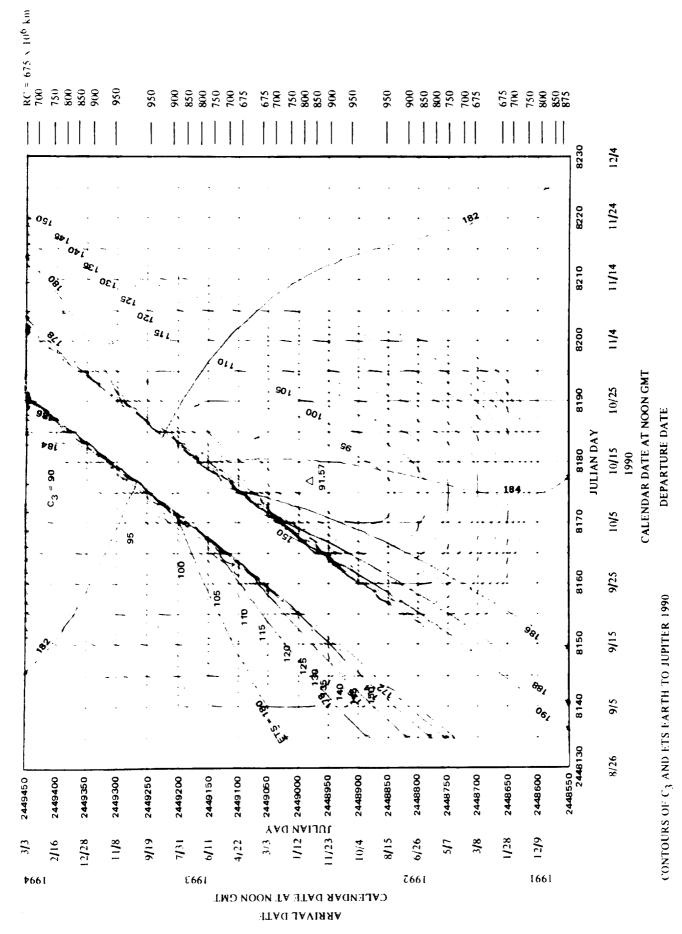
4-173



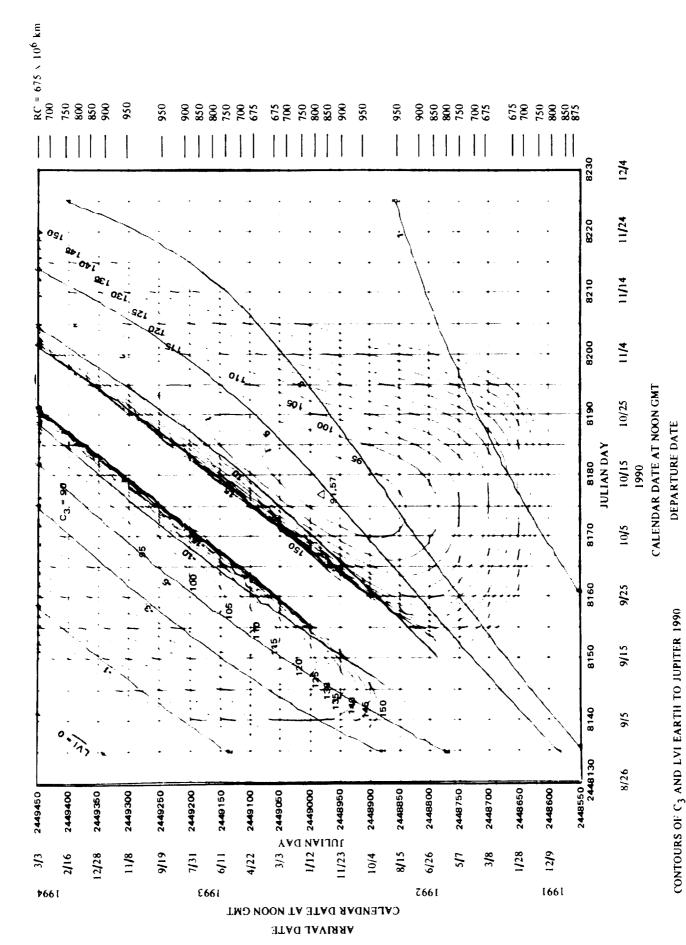
DEPARTURE DATE

CONTOURS OF C3 AND ZAP EARTH TO JUPITER 1990

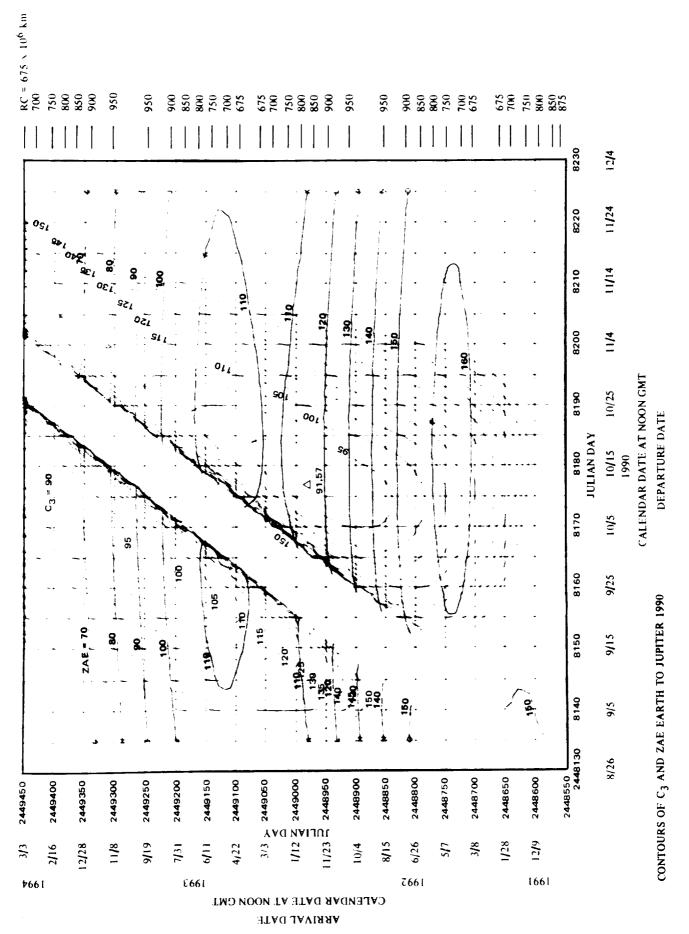
ARRIVAL DATE



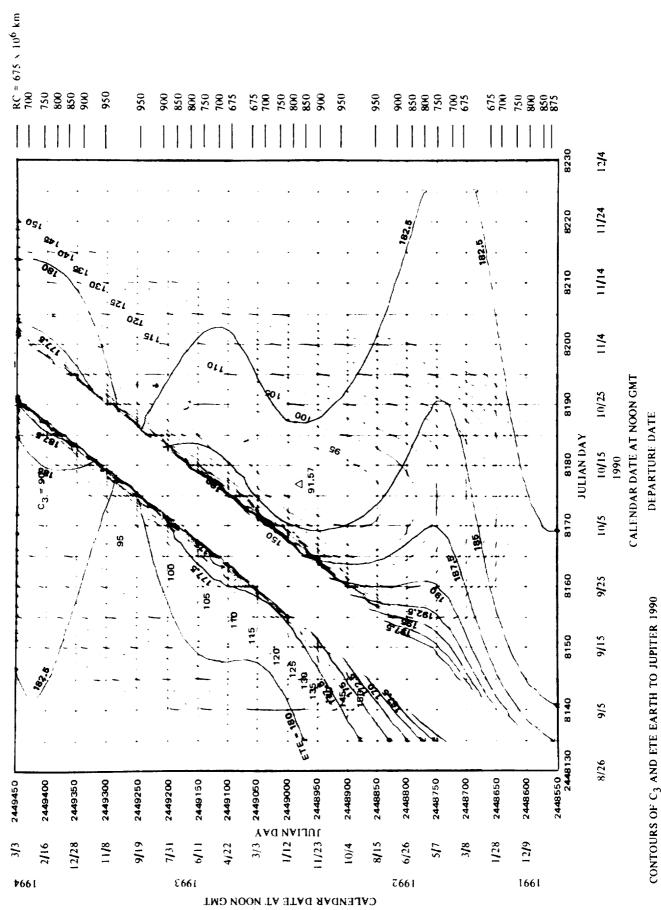
4-175



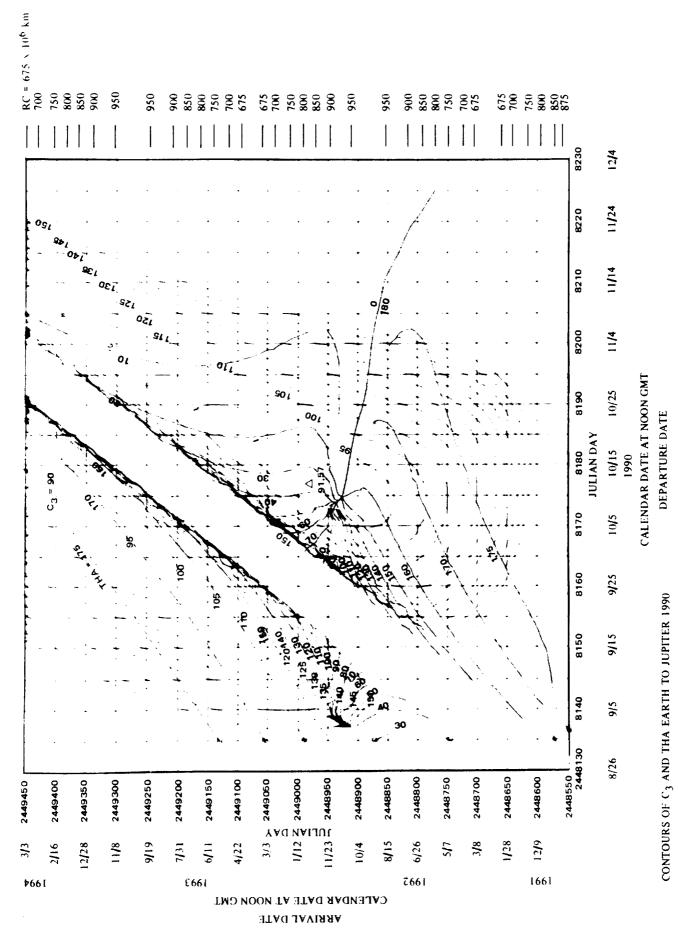
4-176



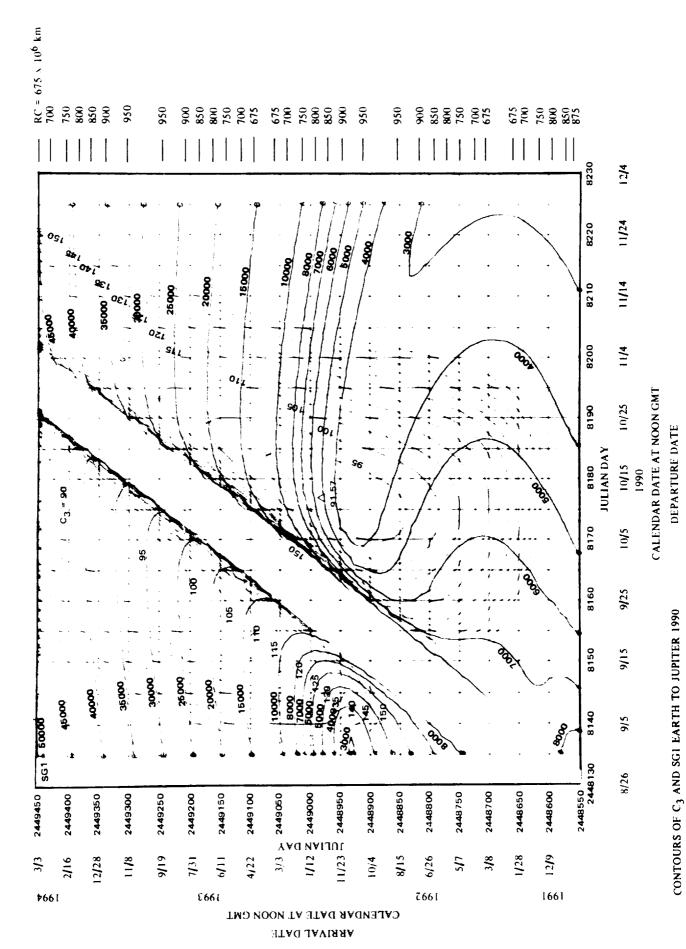
4-177



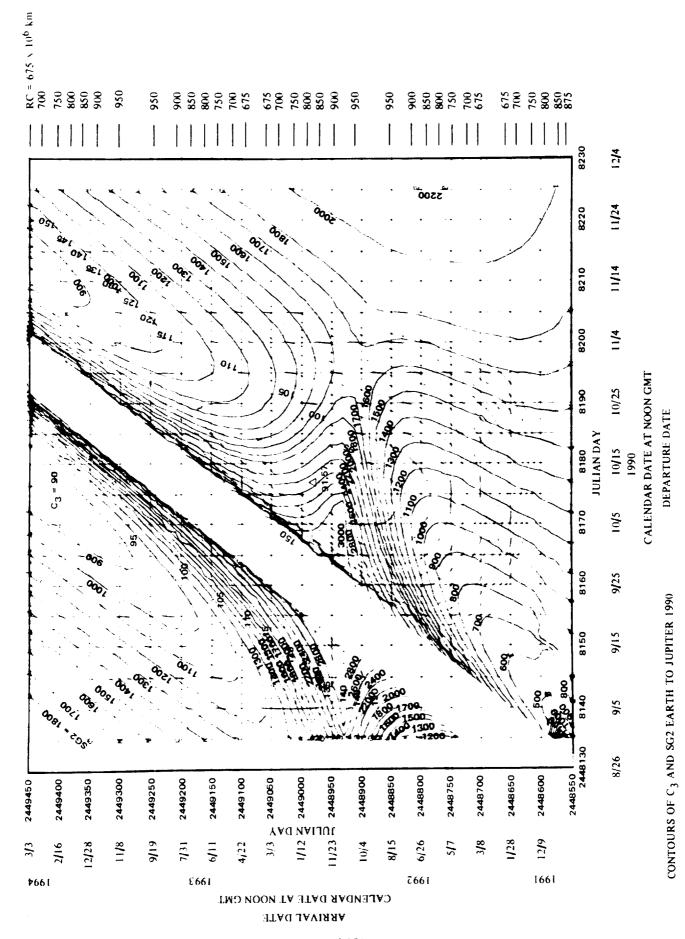
ARRIVAL DATE



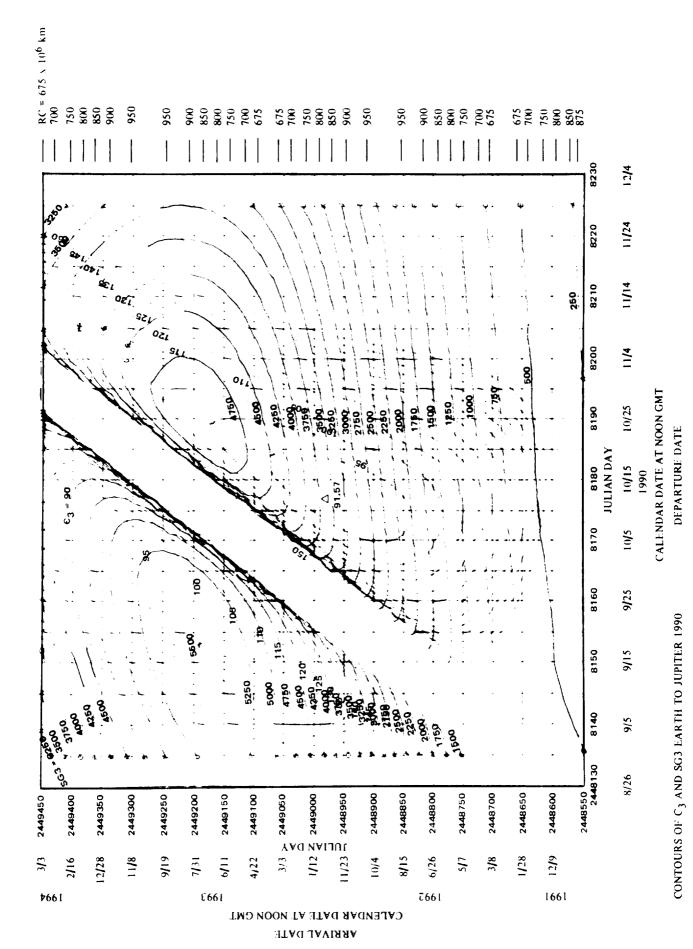
4-179



4-180

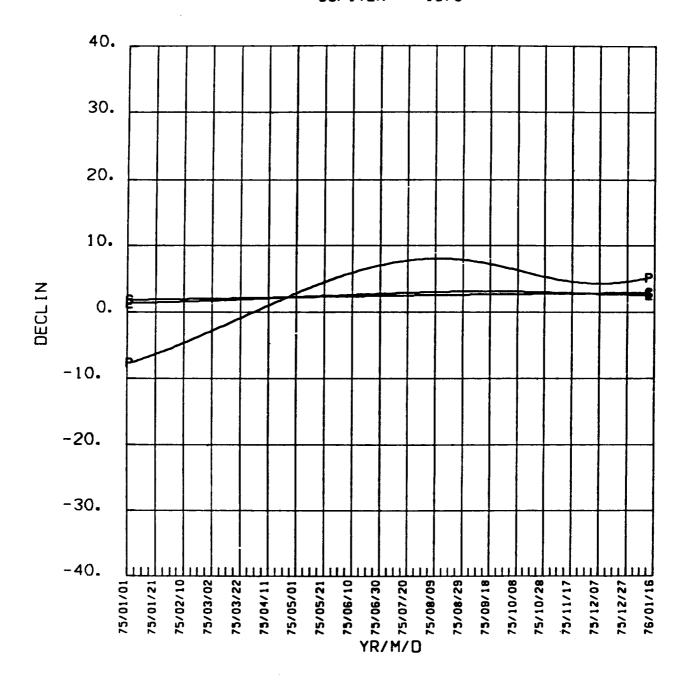


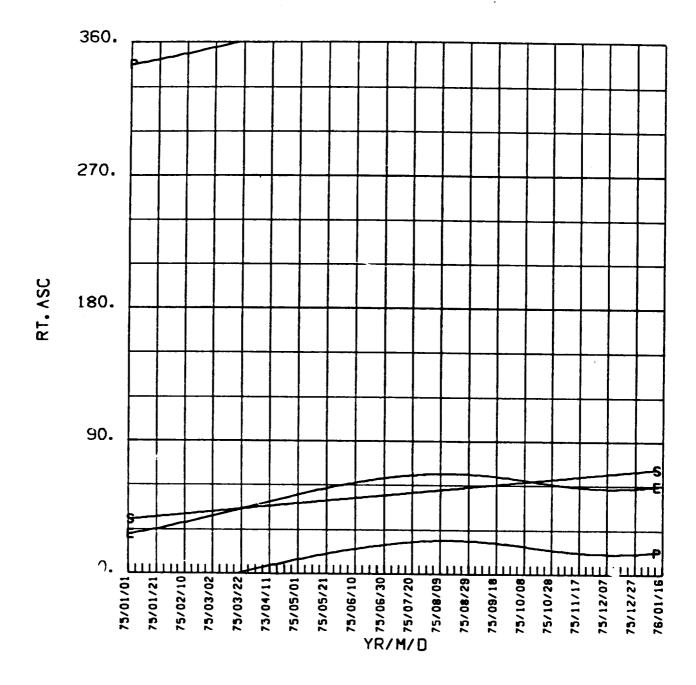
4-181

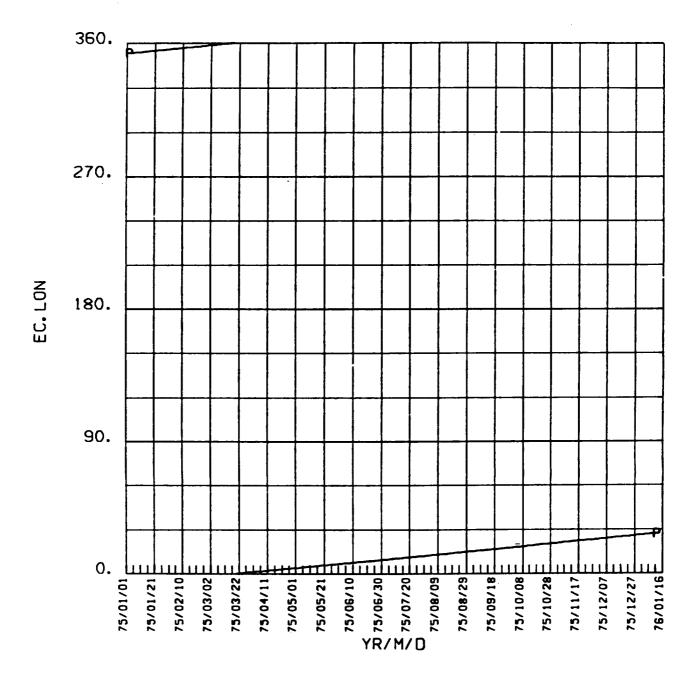


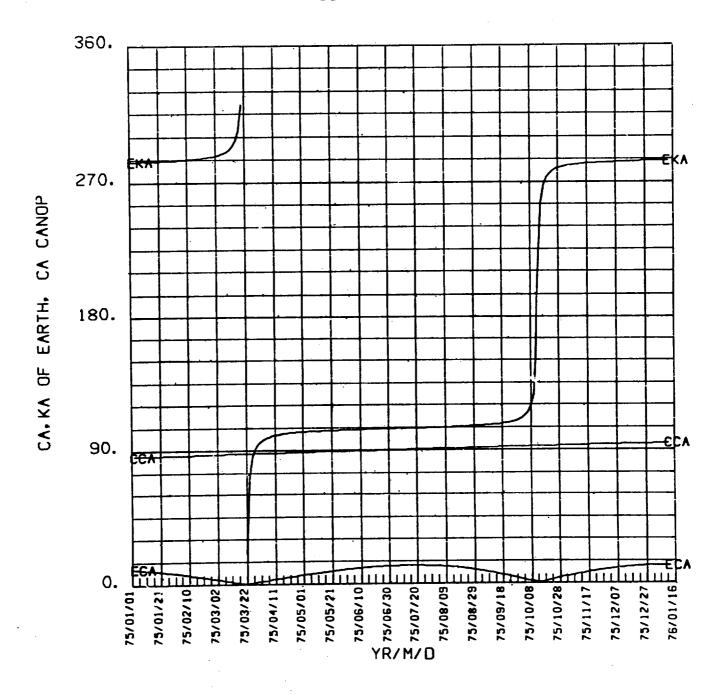
4-182



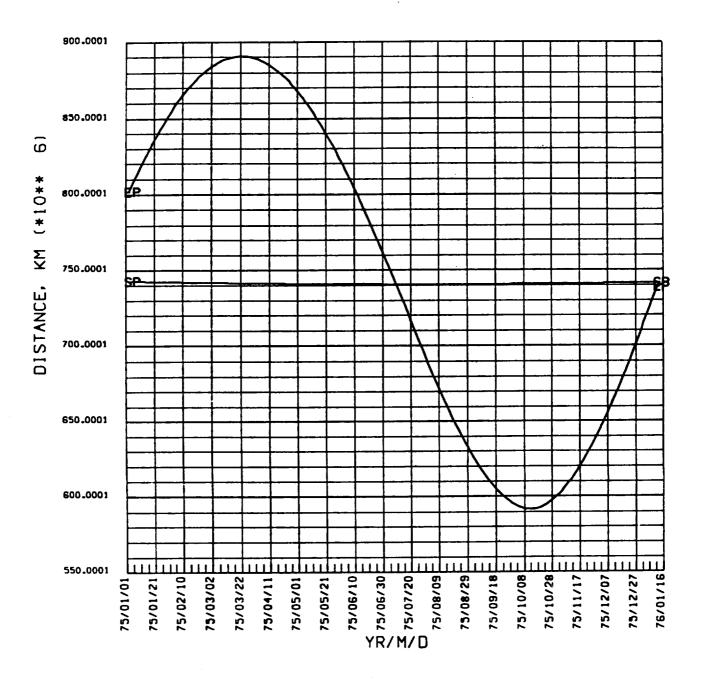




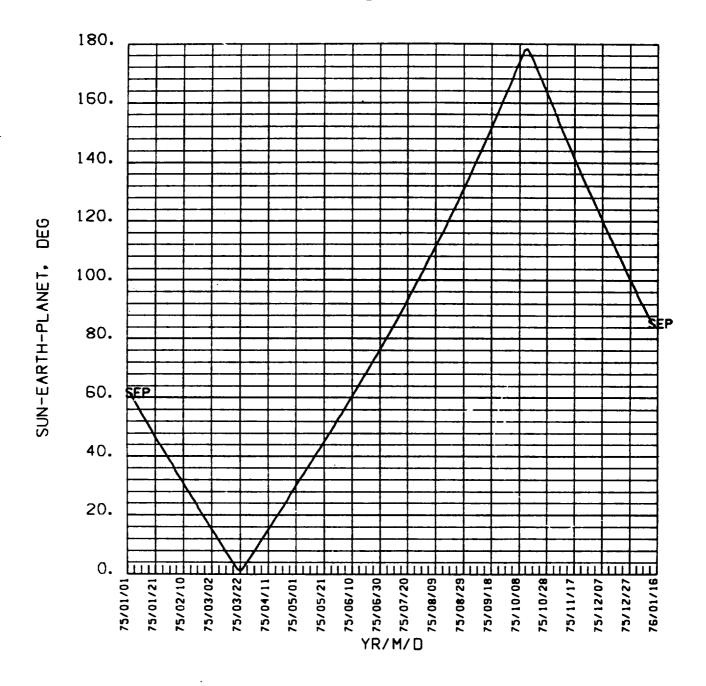


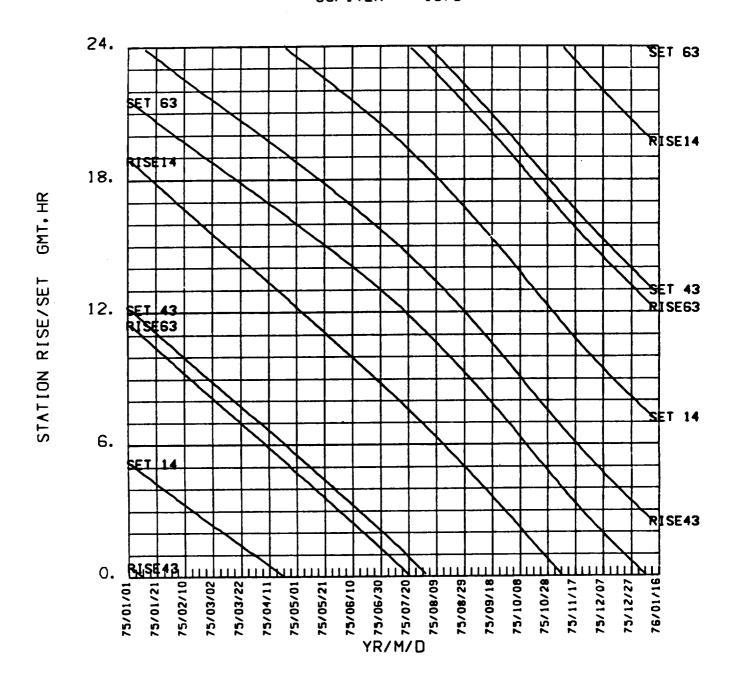


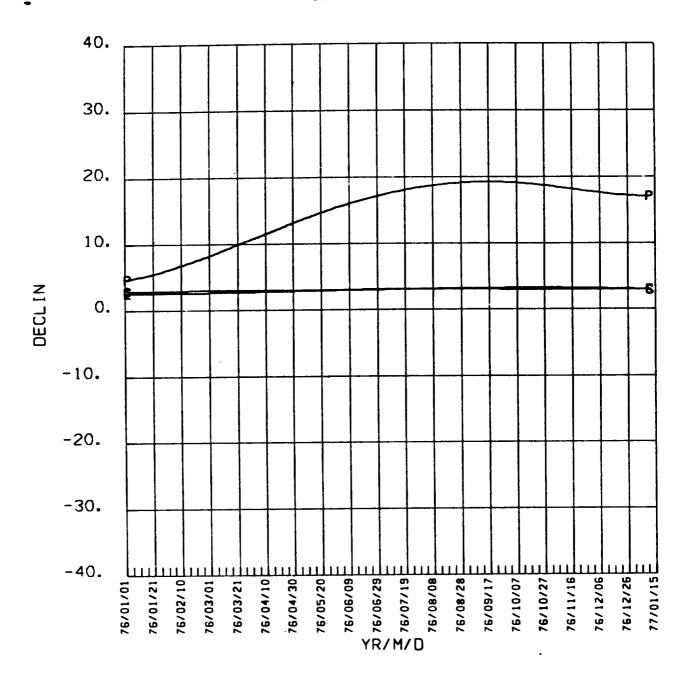


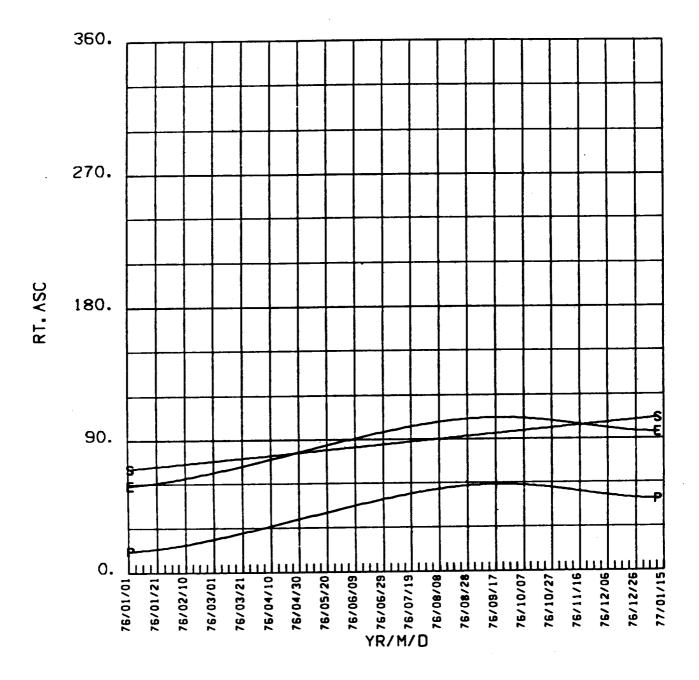


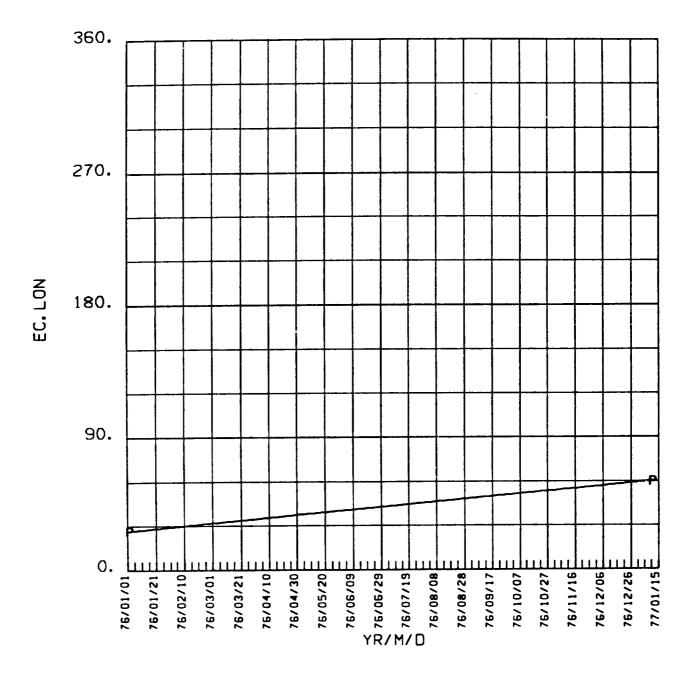
C 3

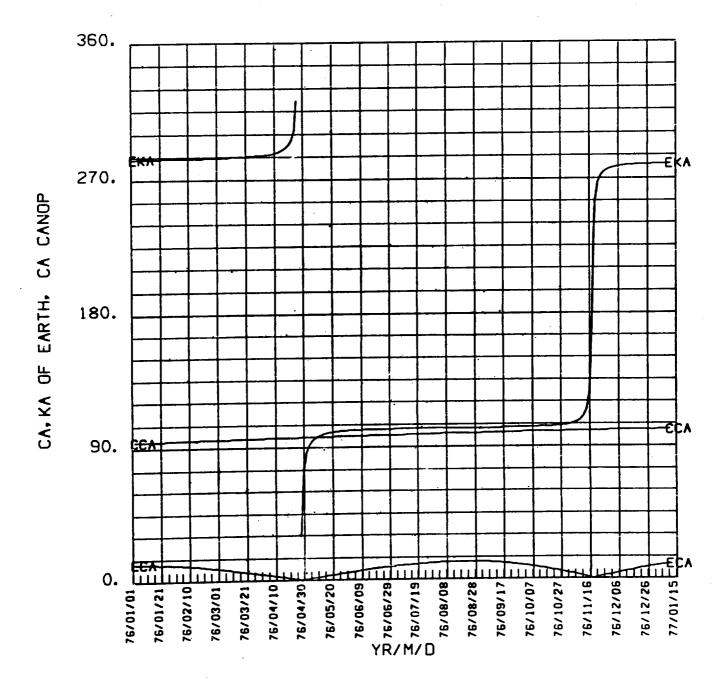


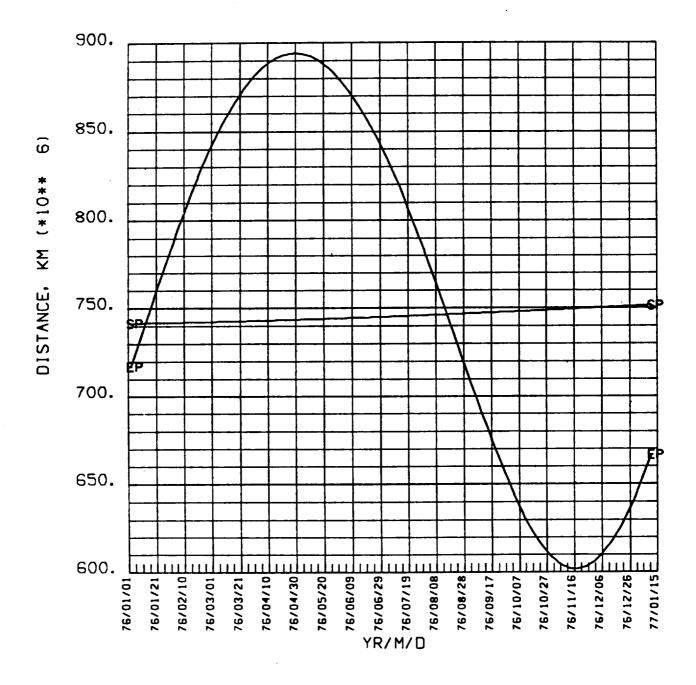




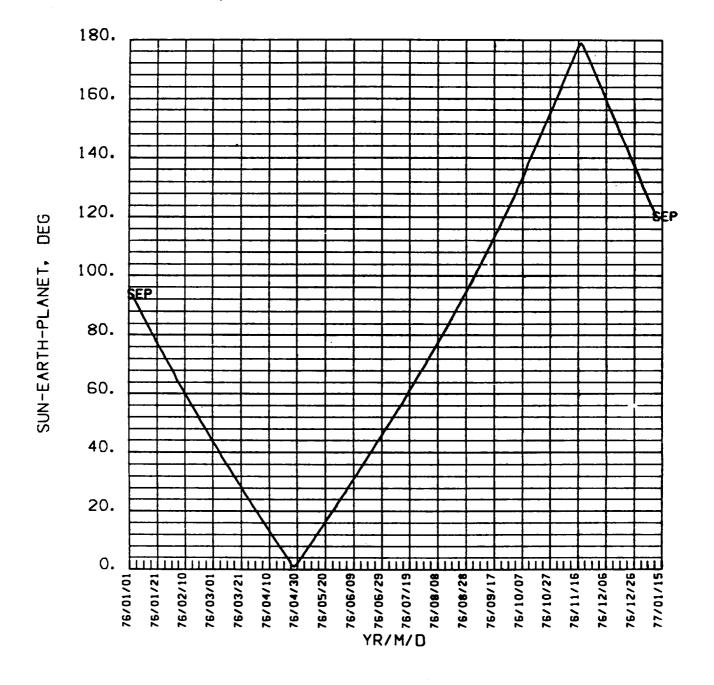


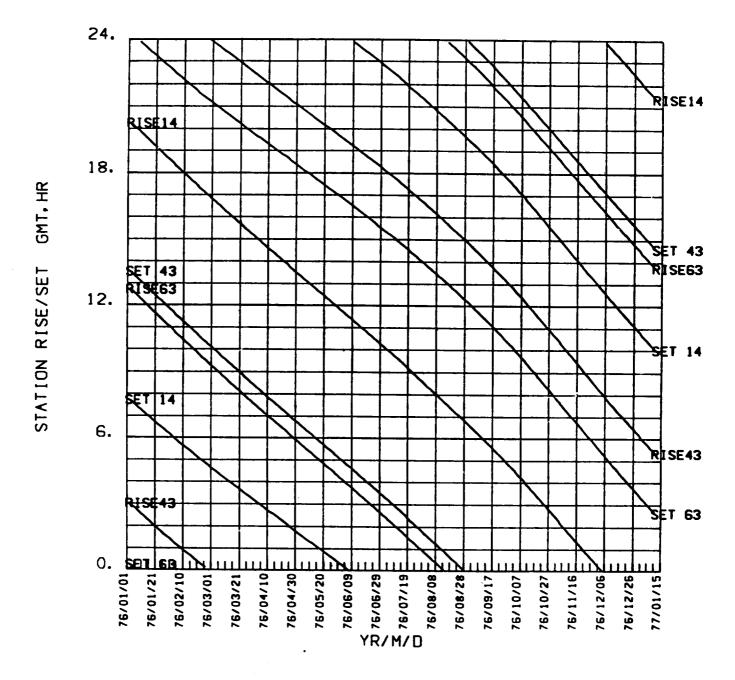




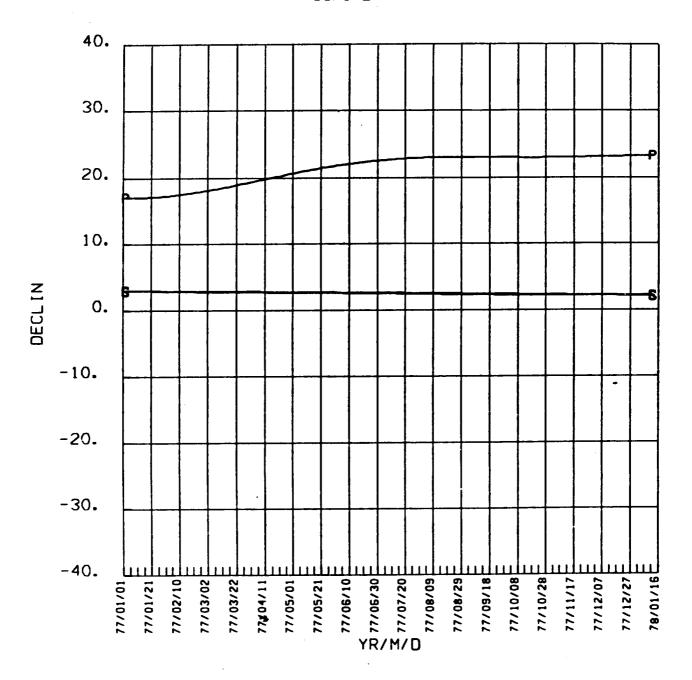


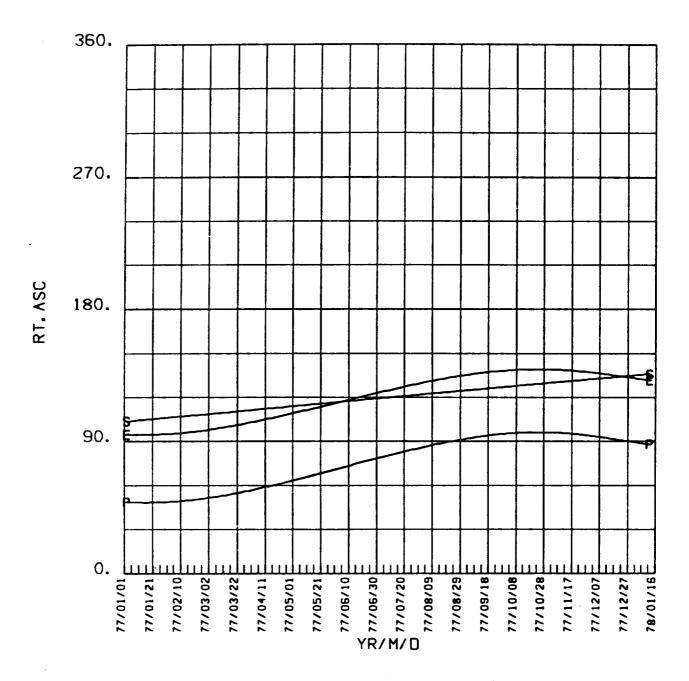
JUPITER 1976

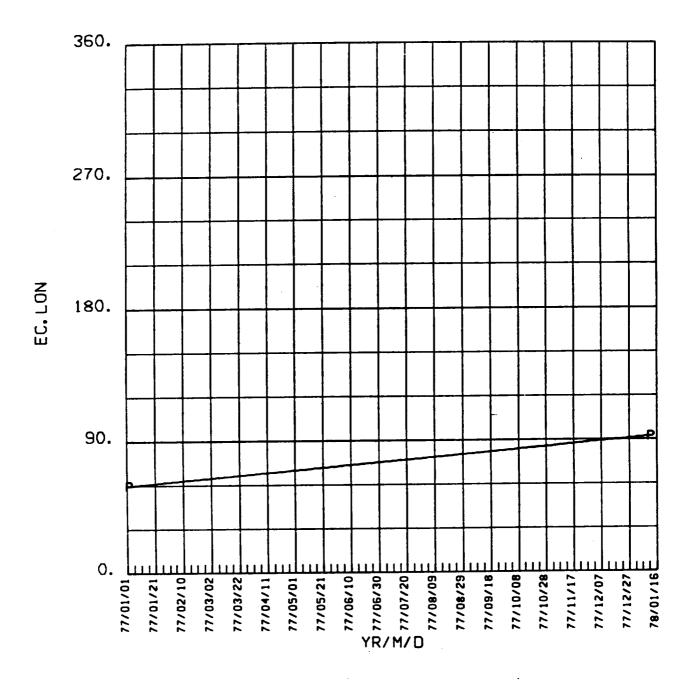


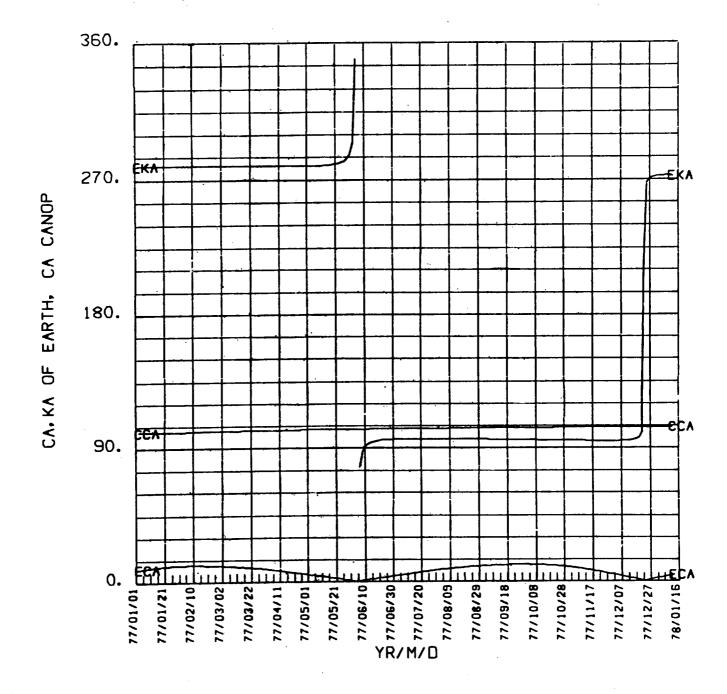


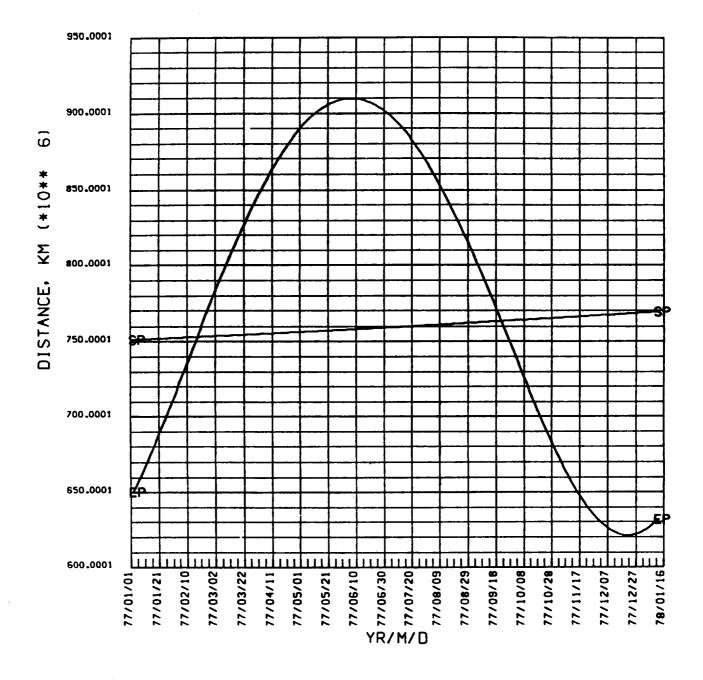


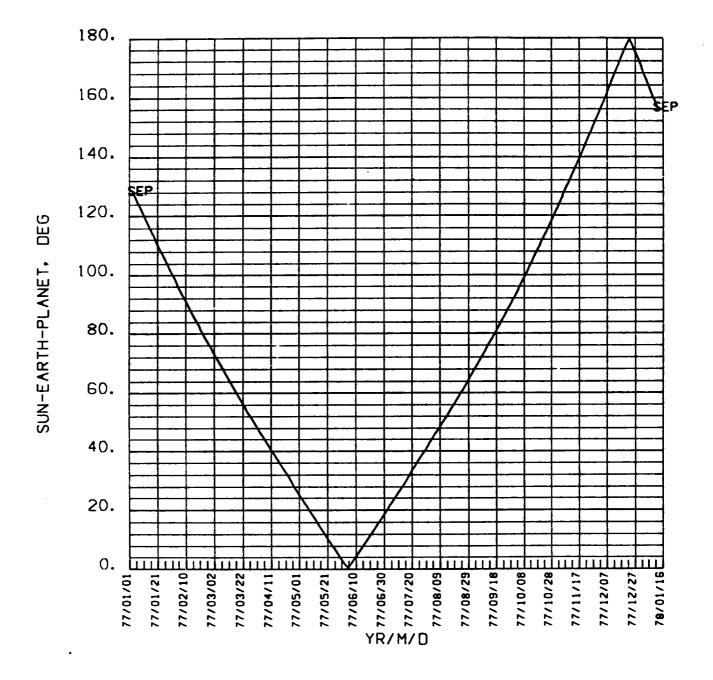


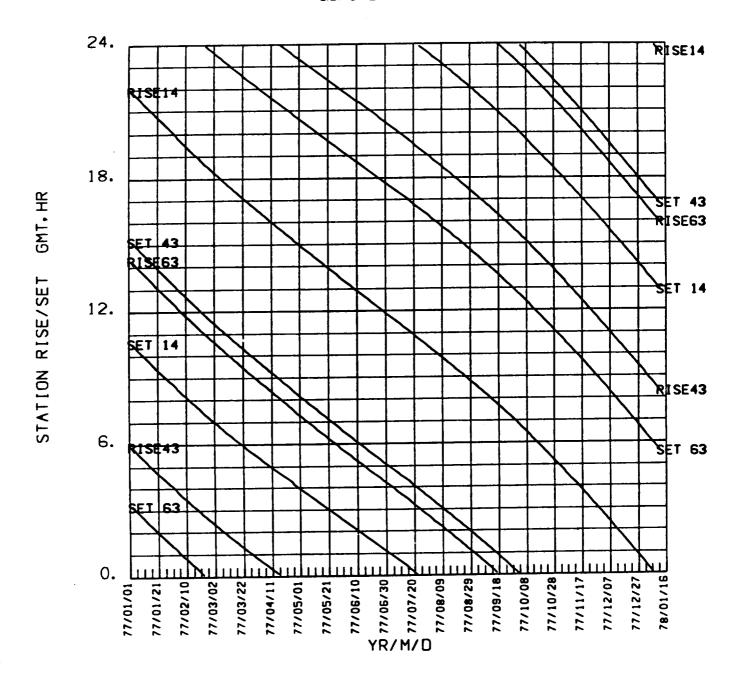




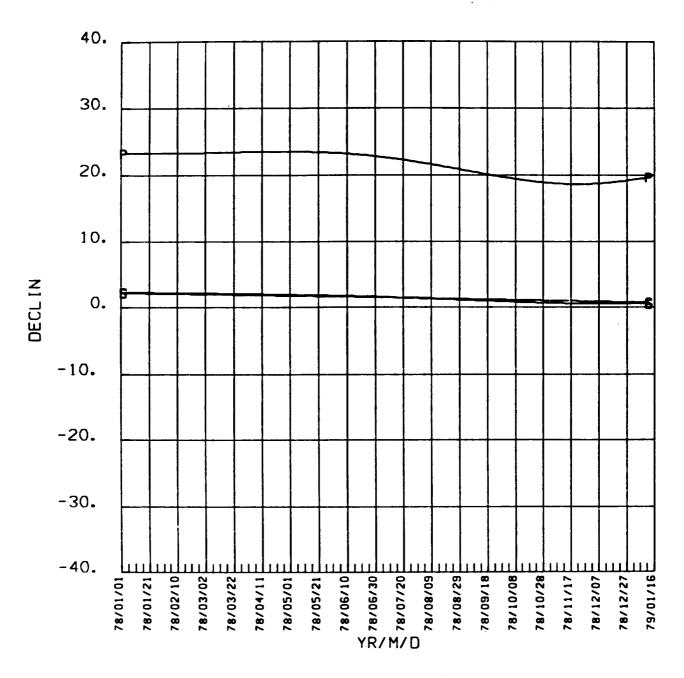


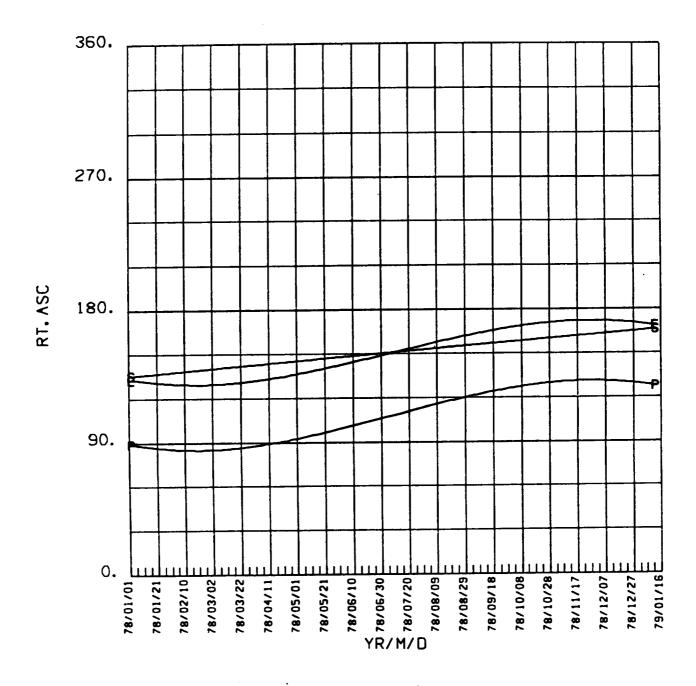




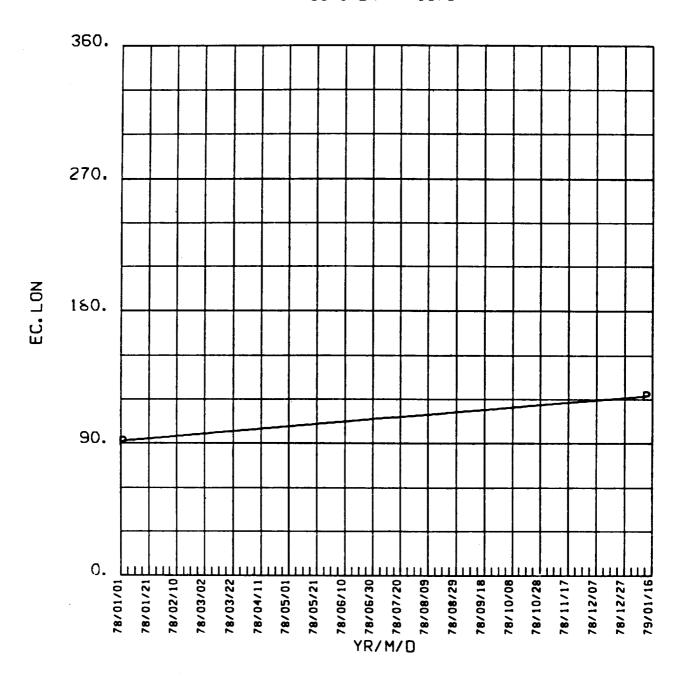


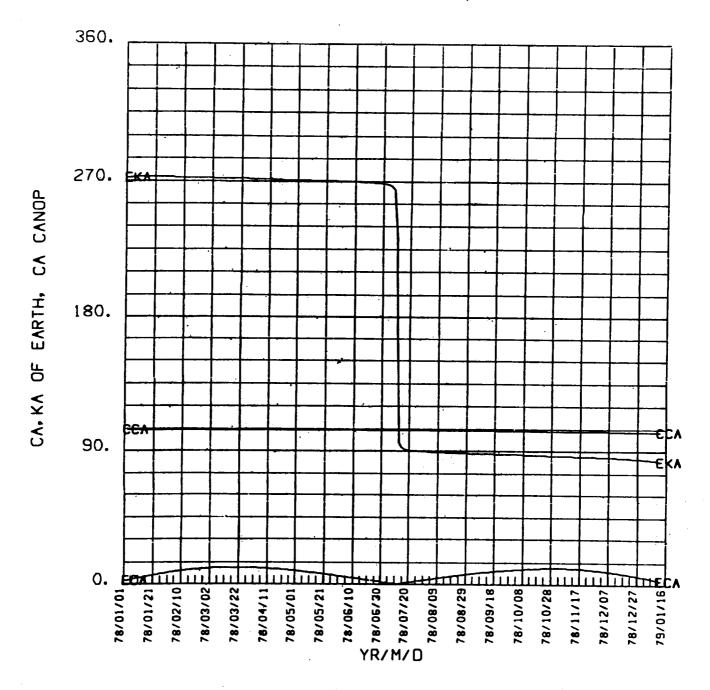


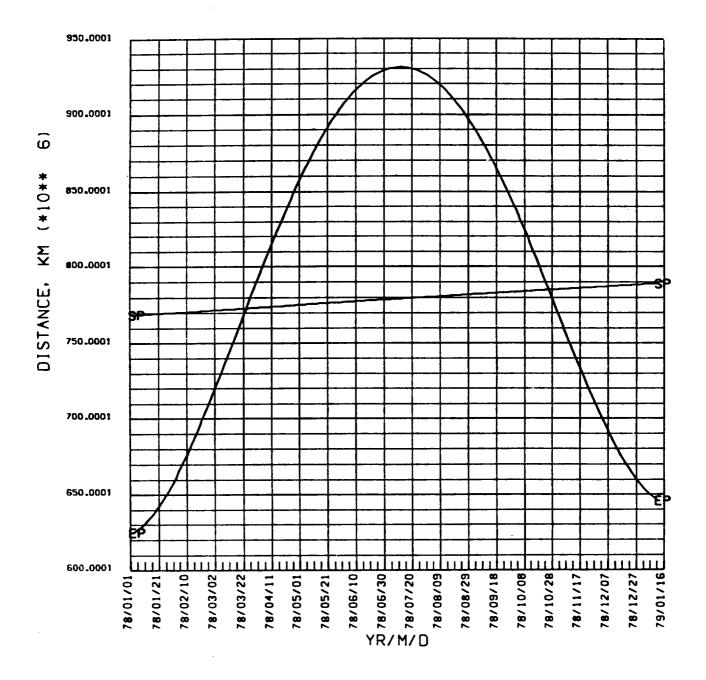




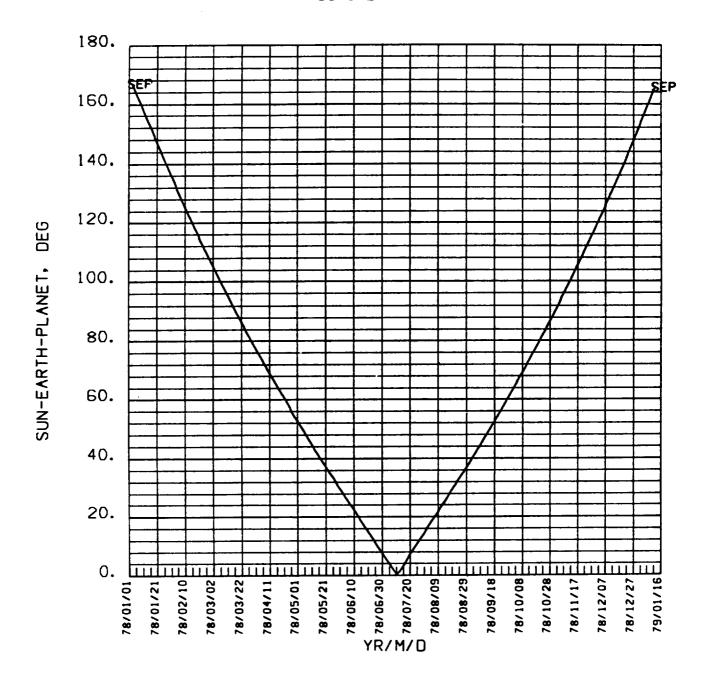
JUPITER 1978

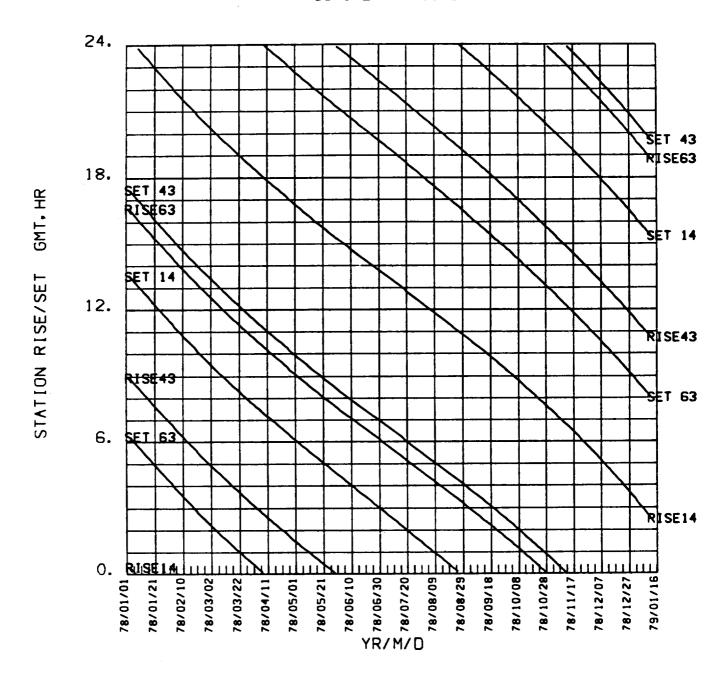




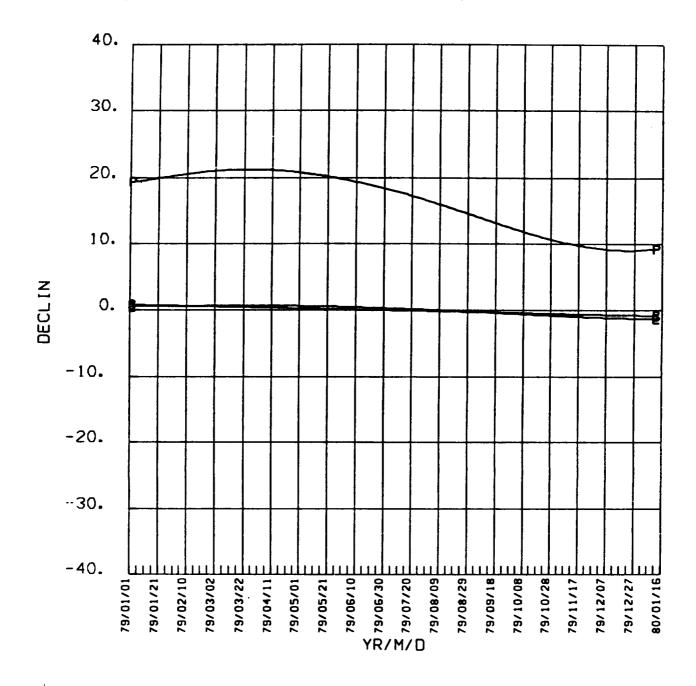


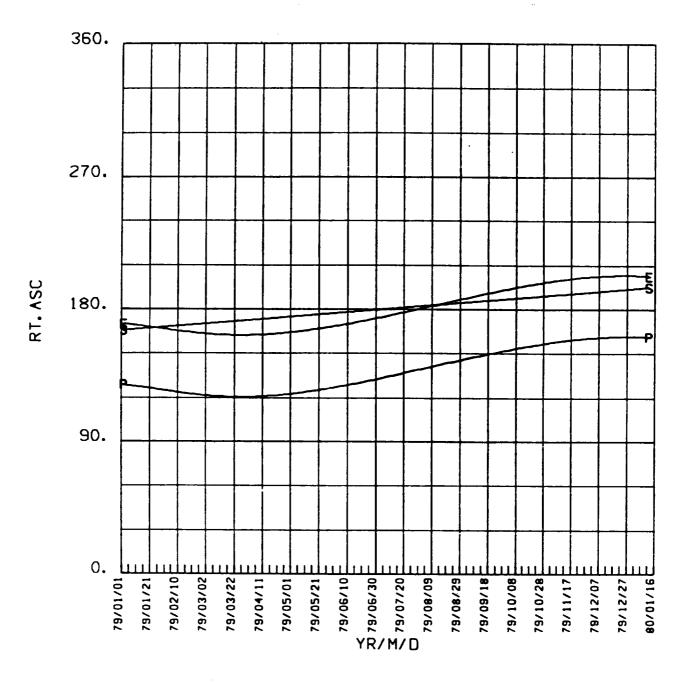
JUPITER 1978



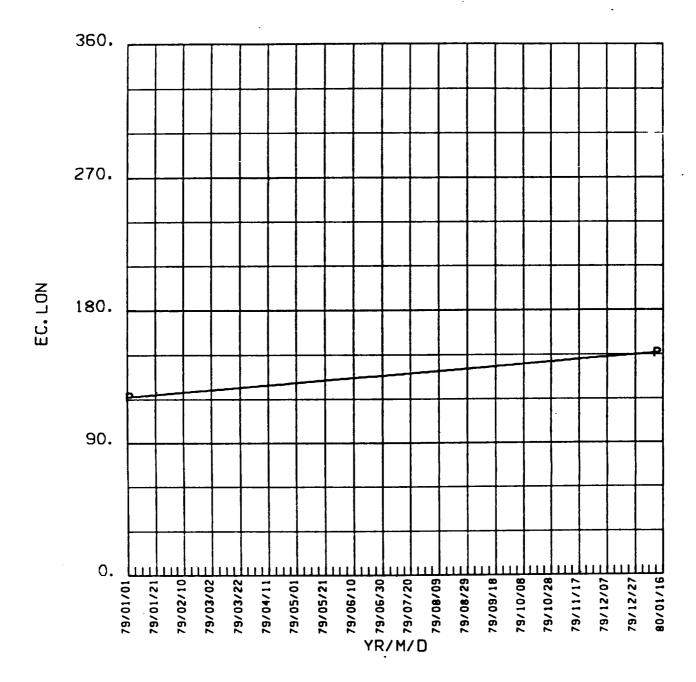


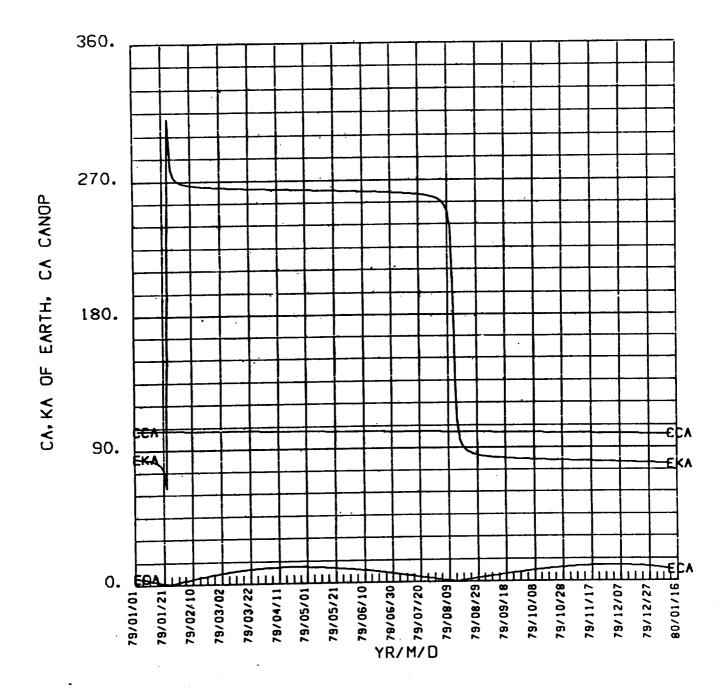




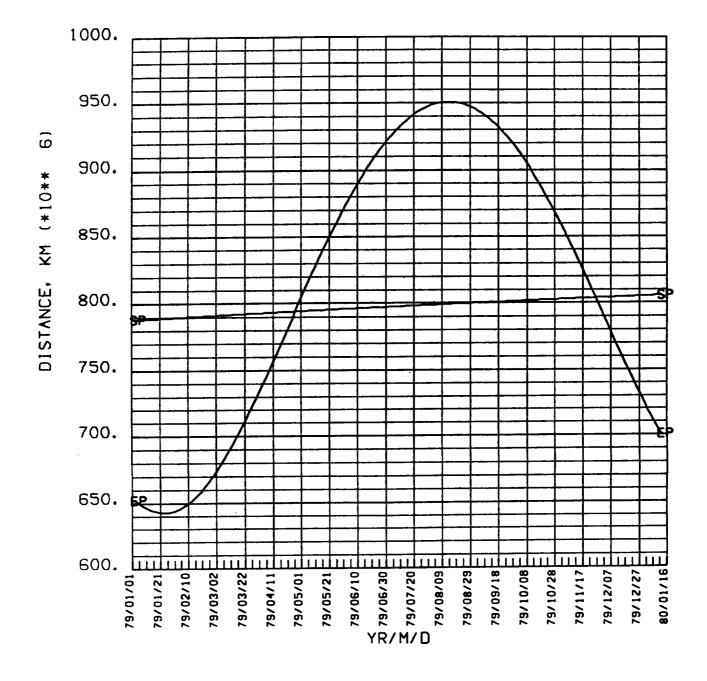


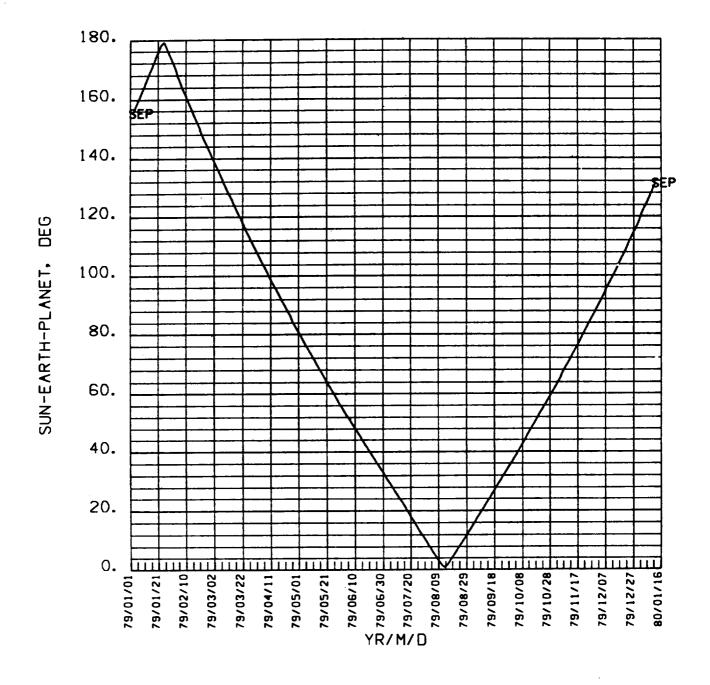


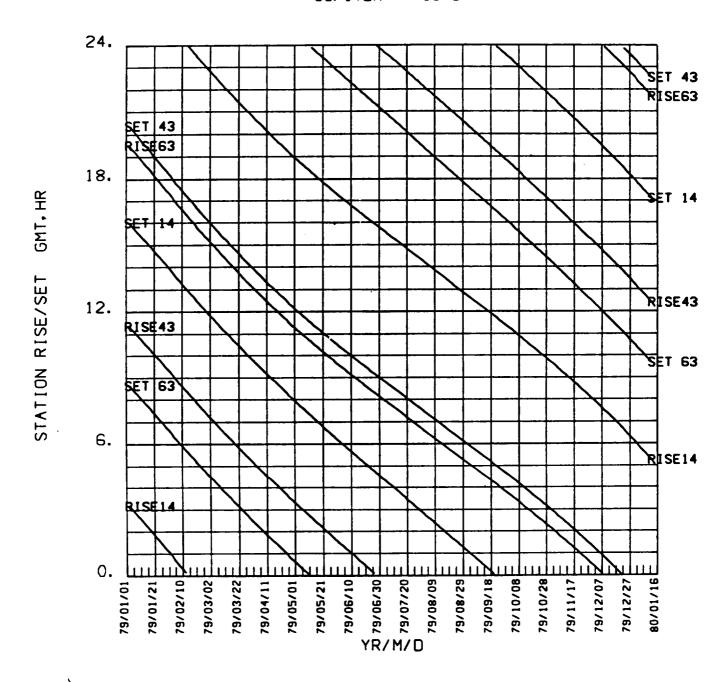




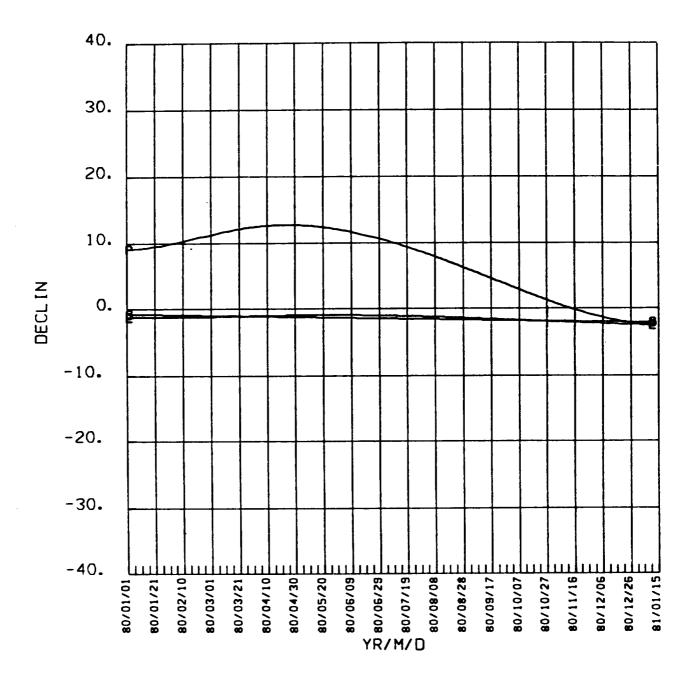
JUPITER 1979

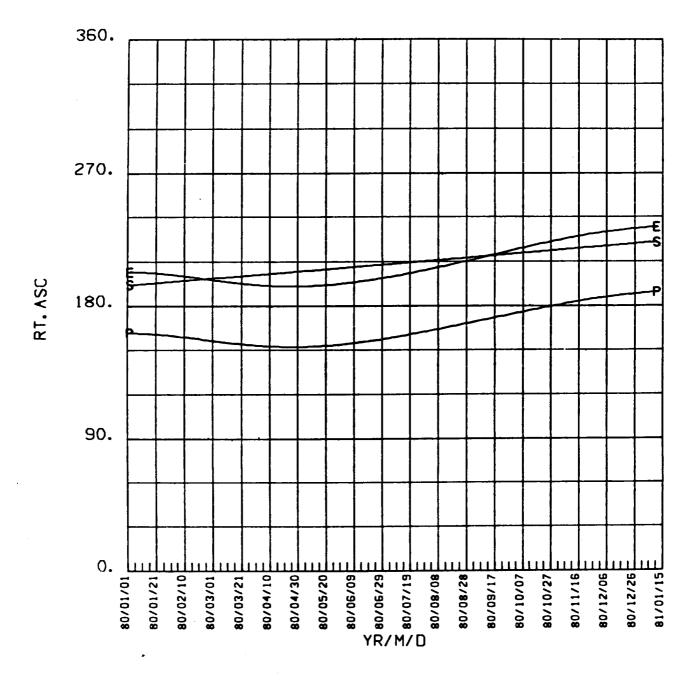


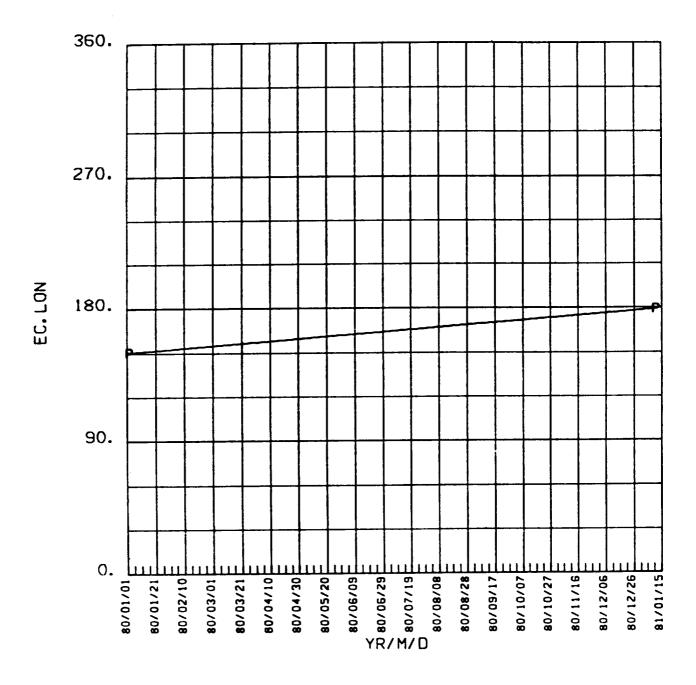


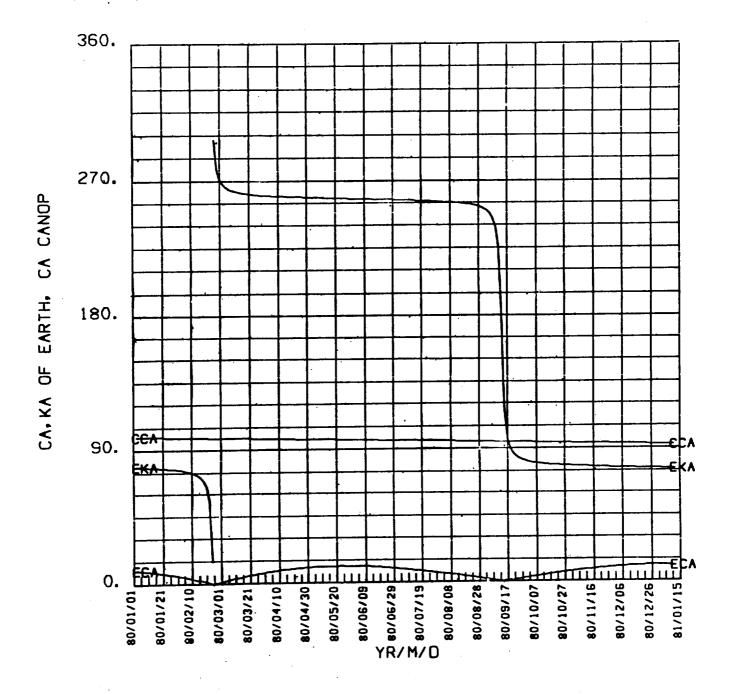




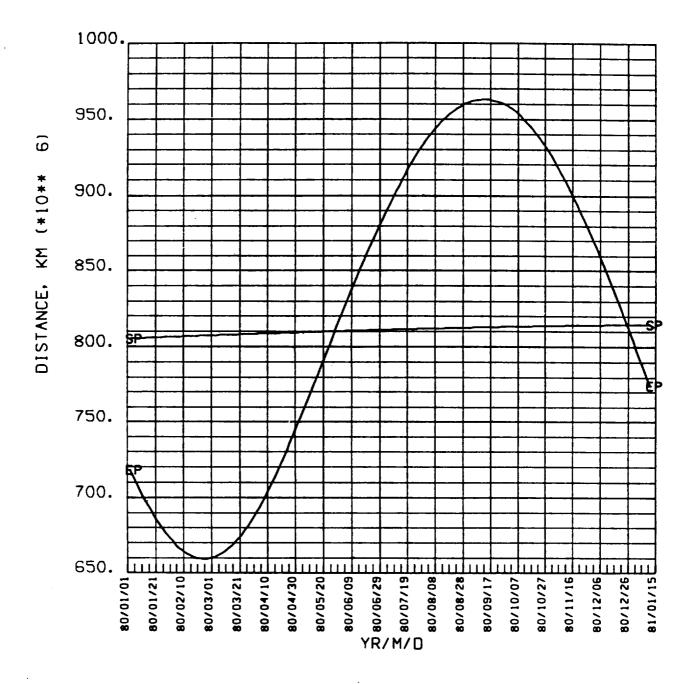


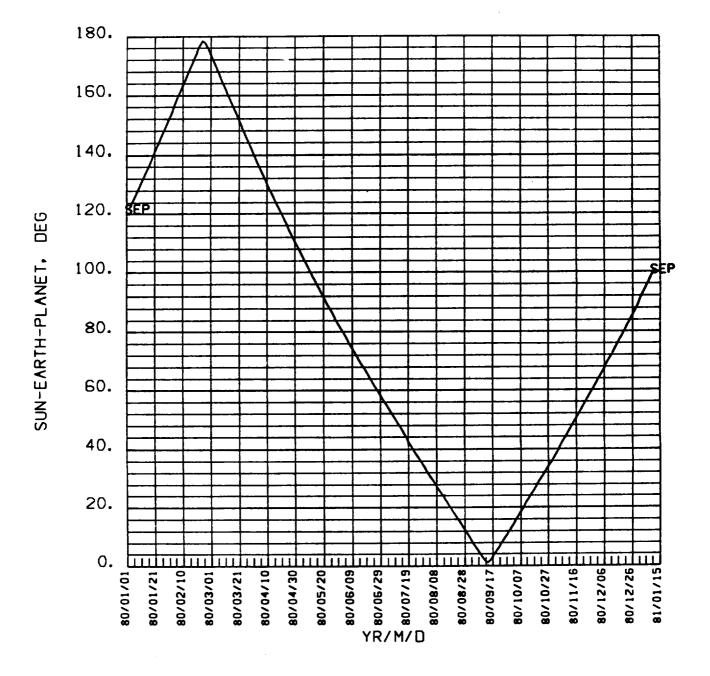


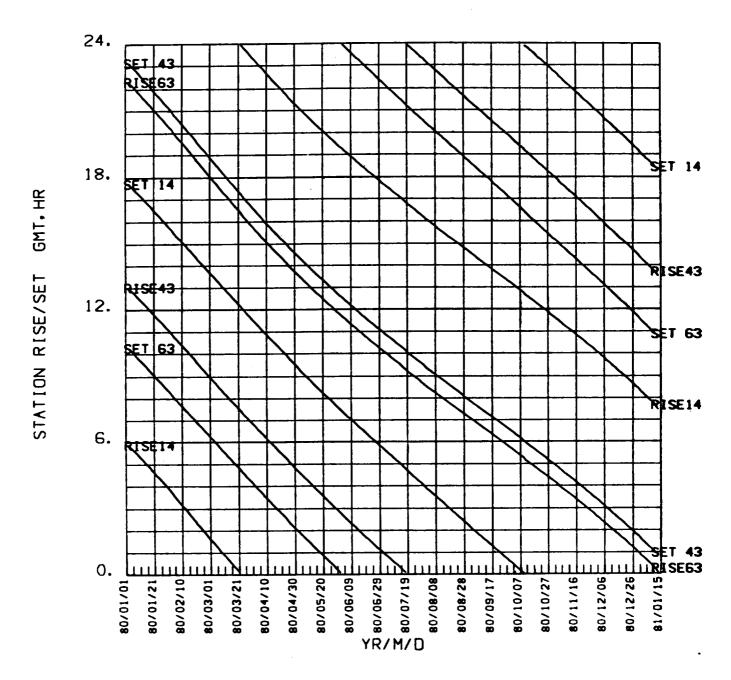




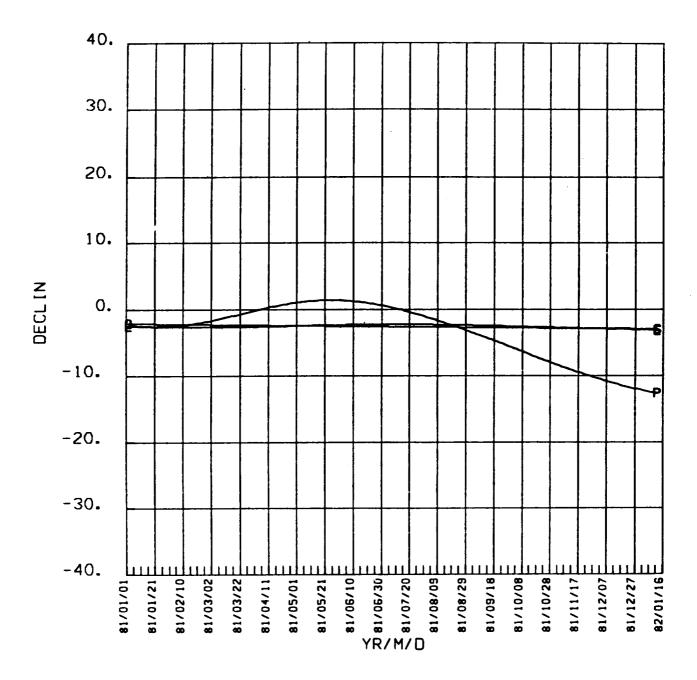


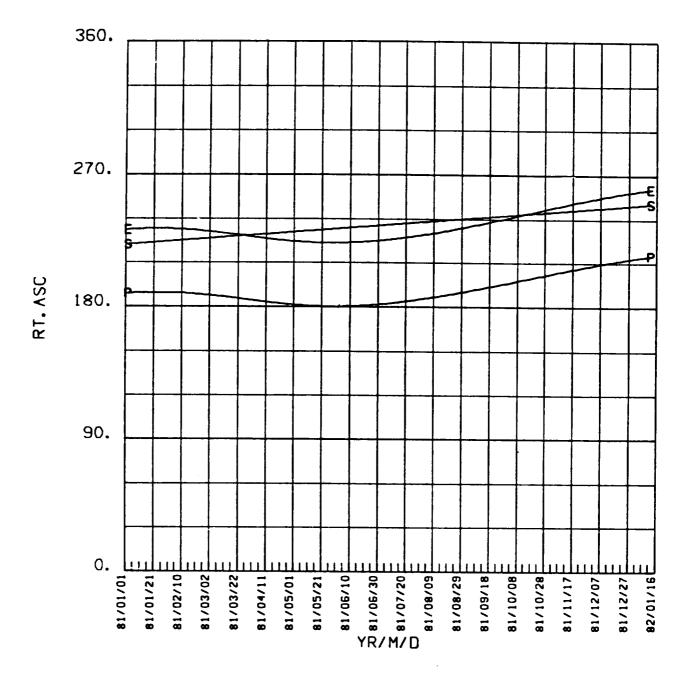




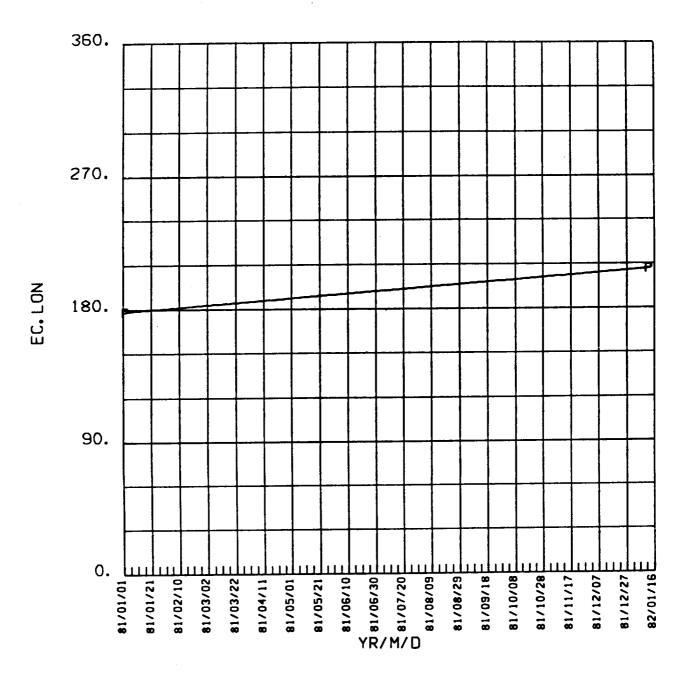


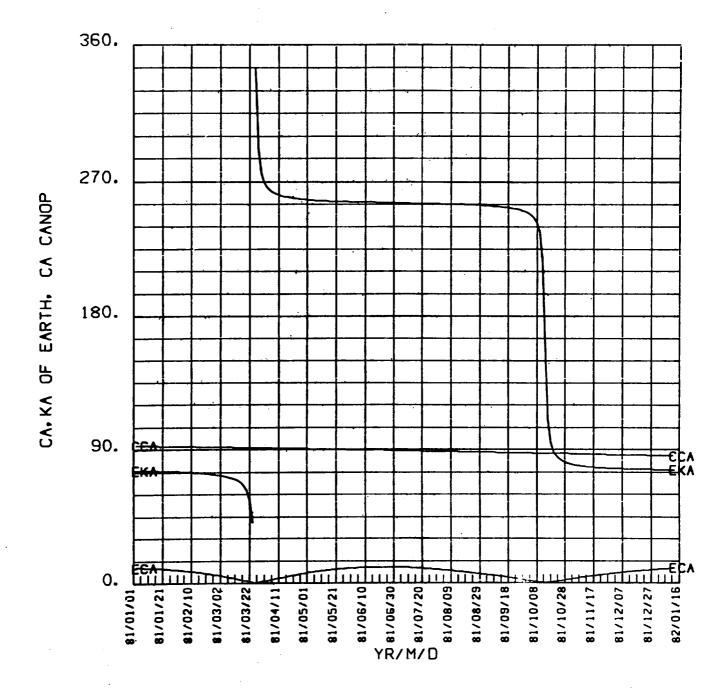




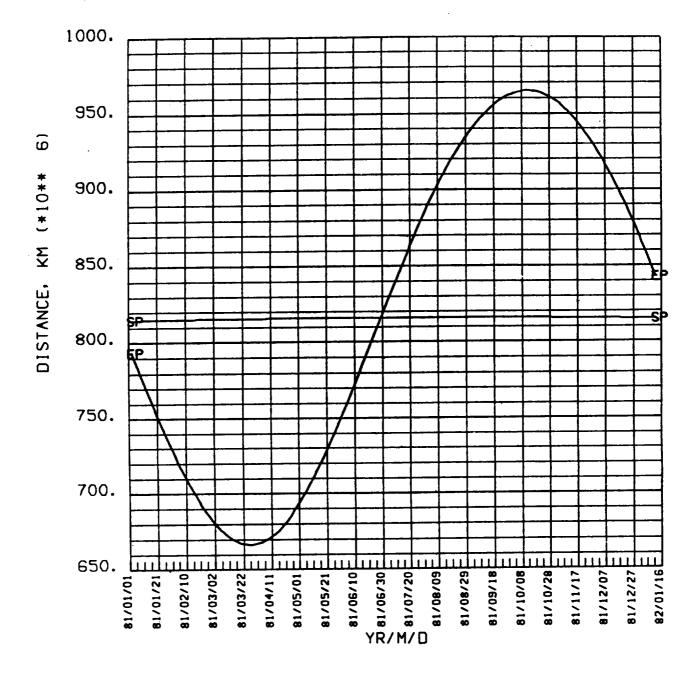


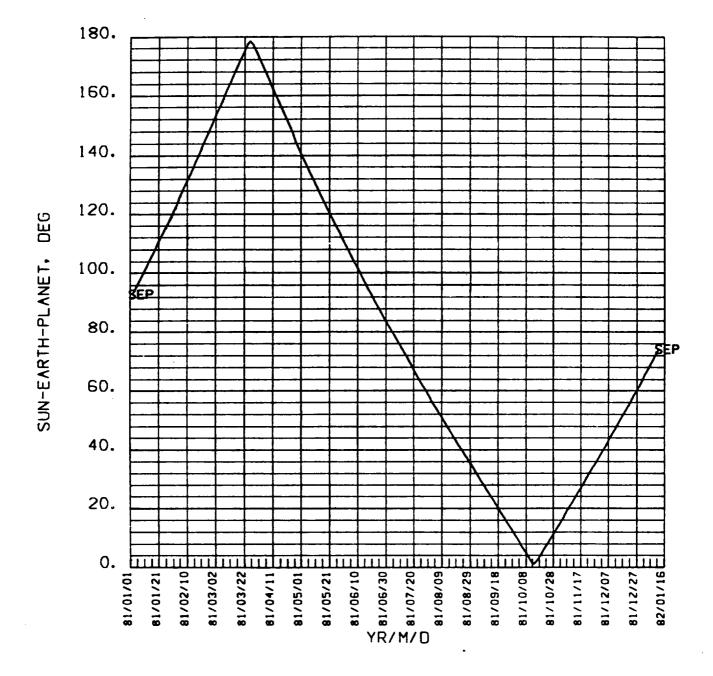


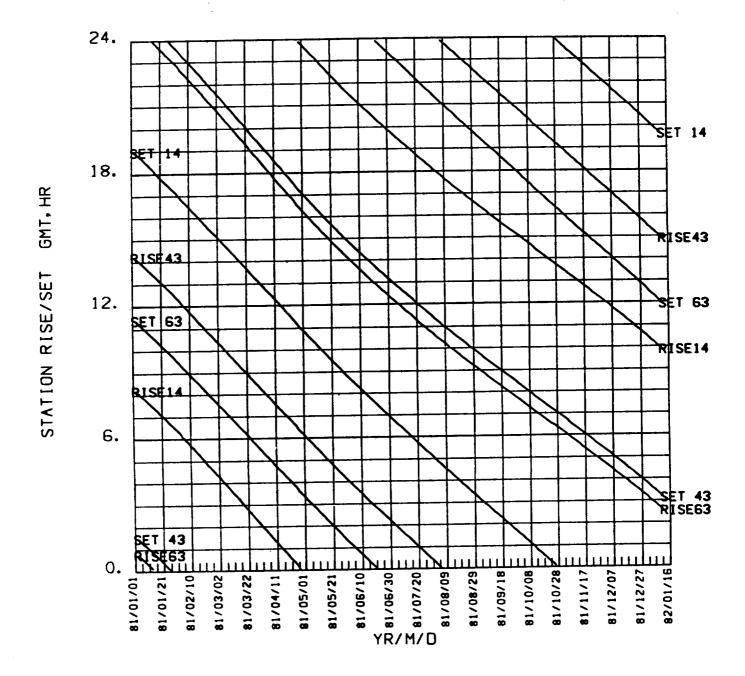




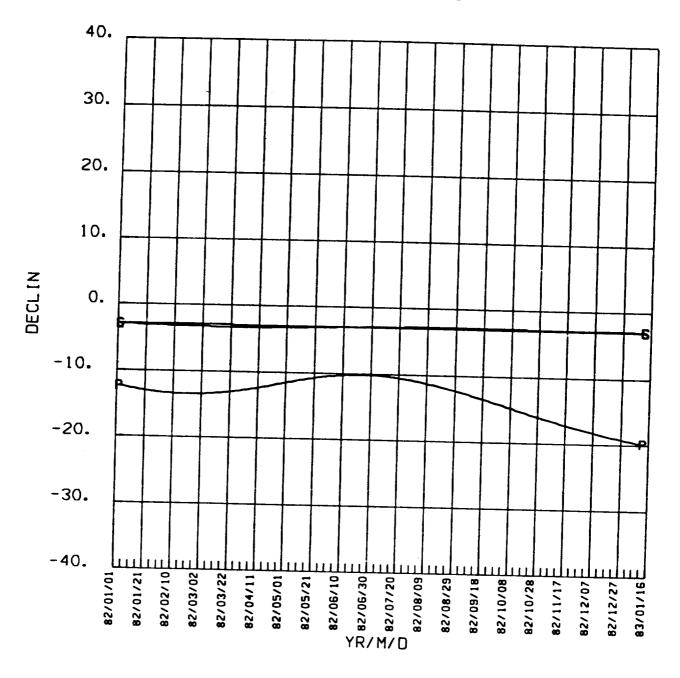
JUPITER 1981

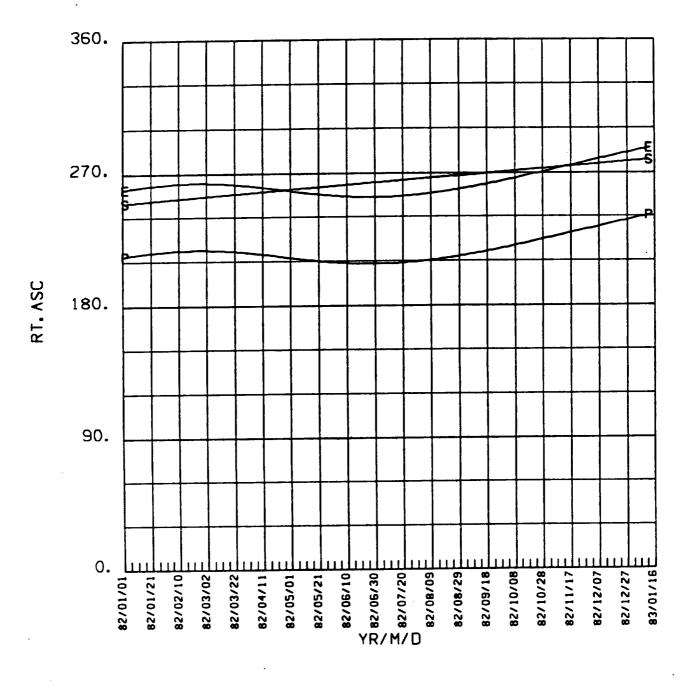


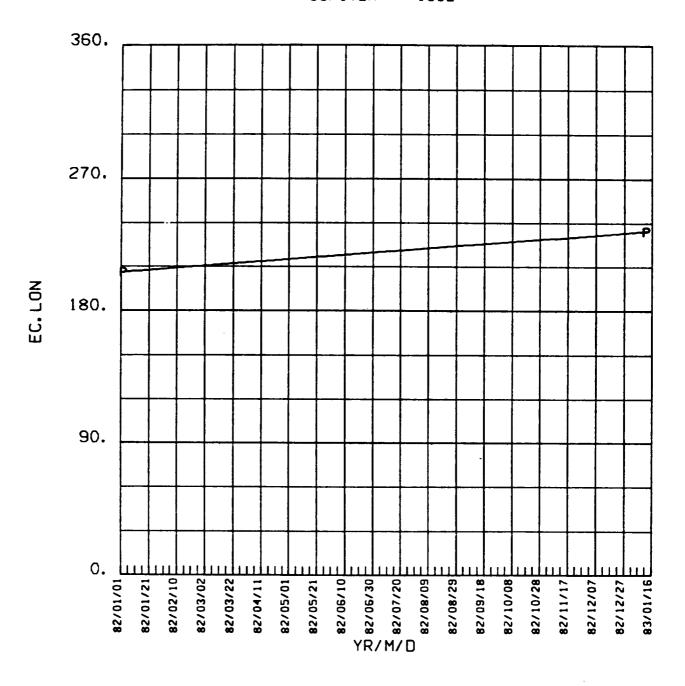


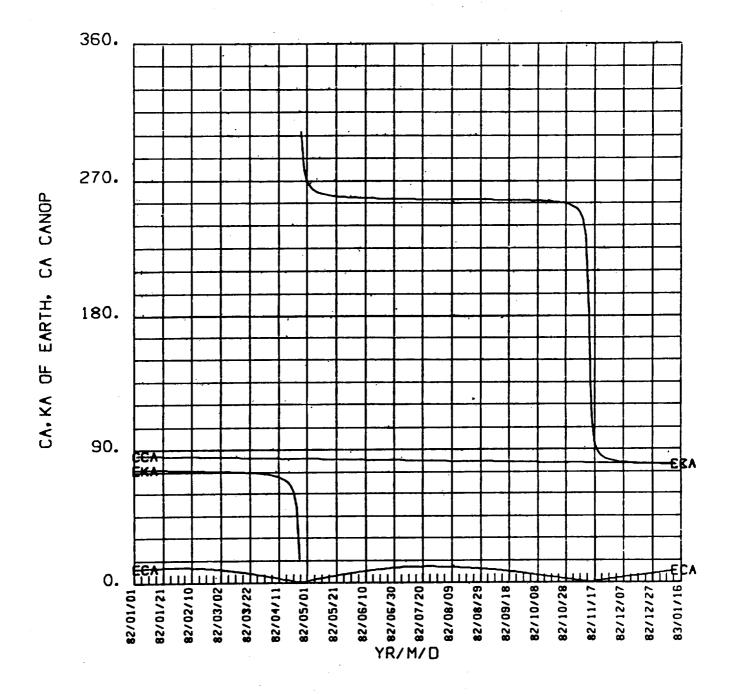


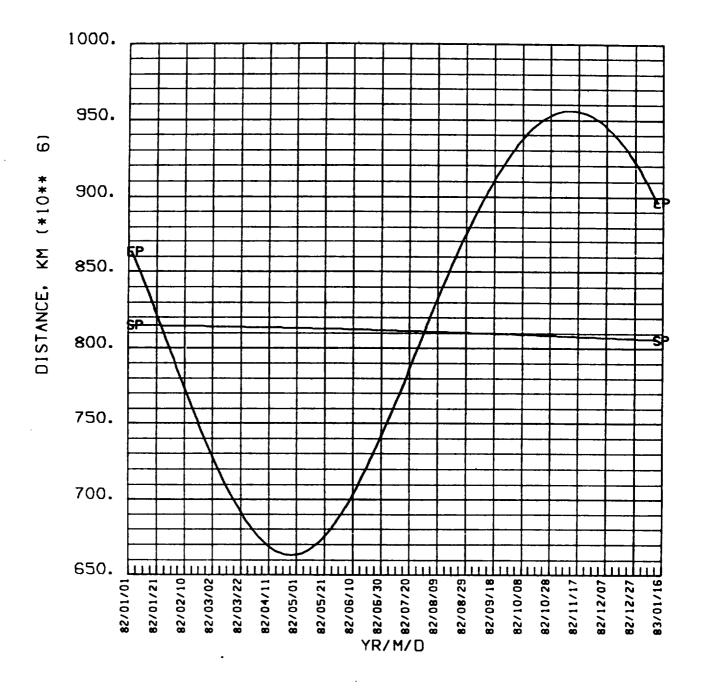


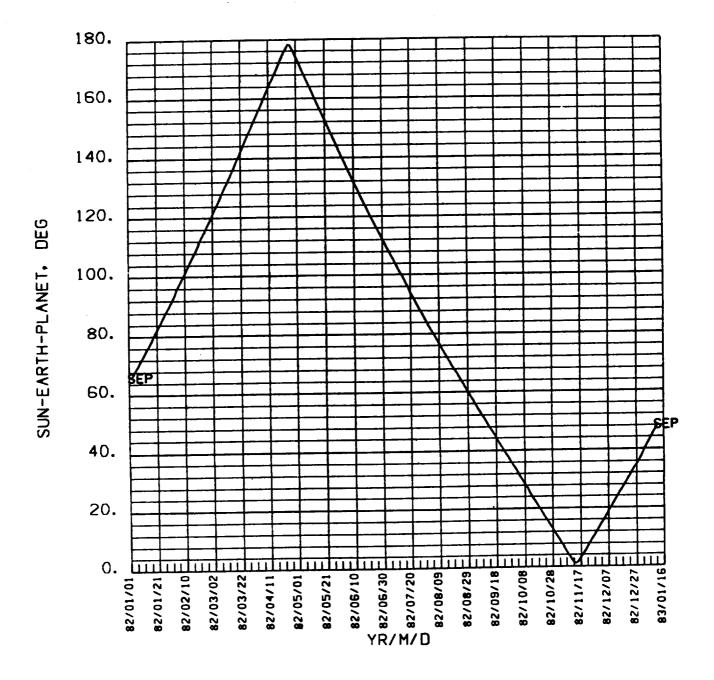




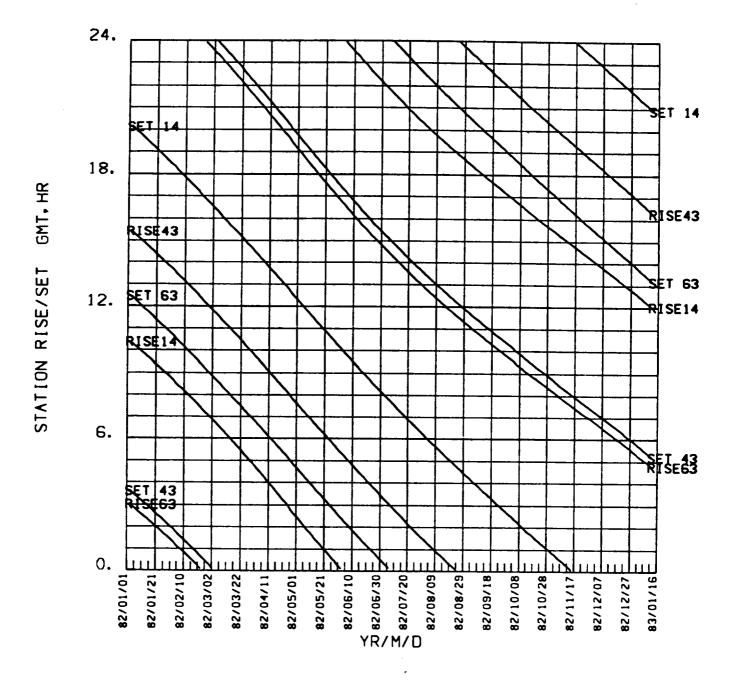




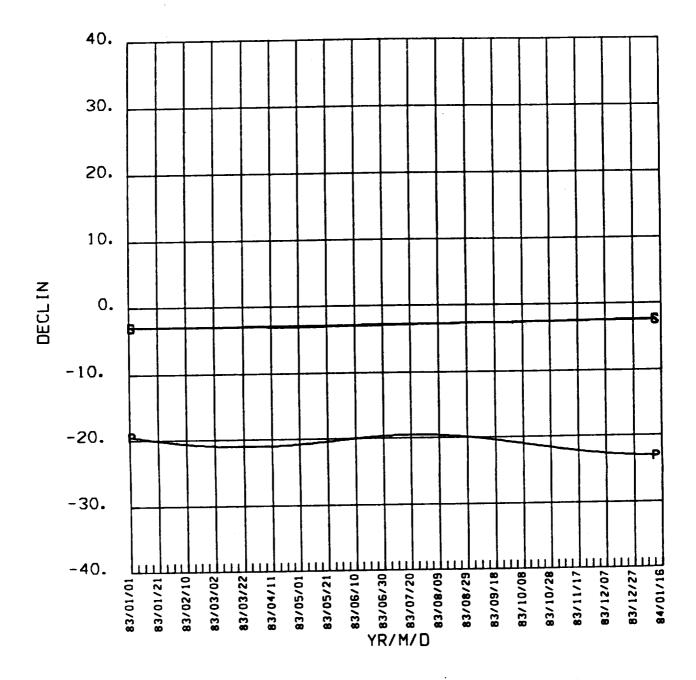


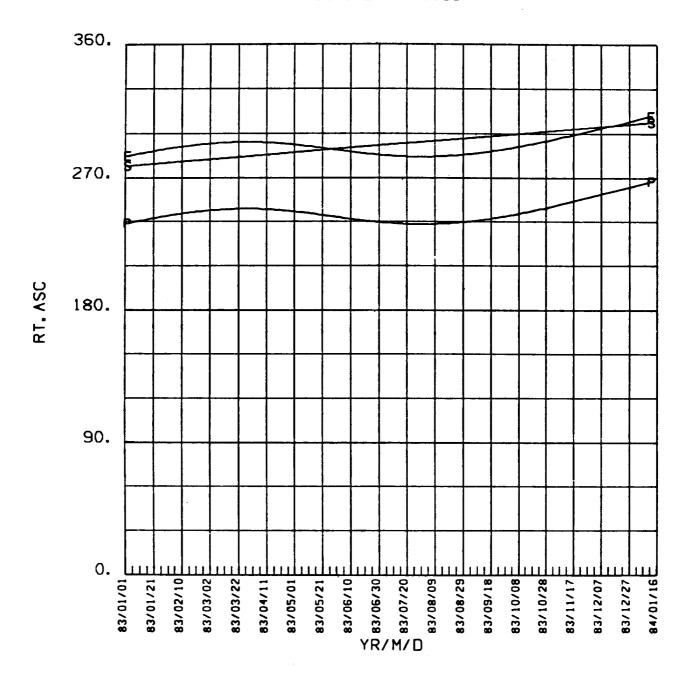


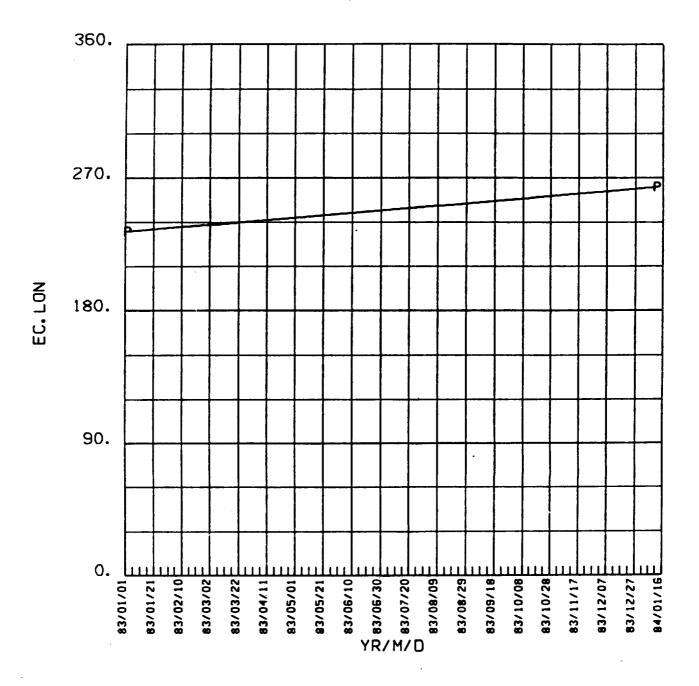
JUPITER 1982

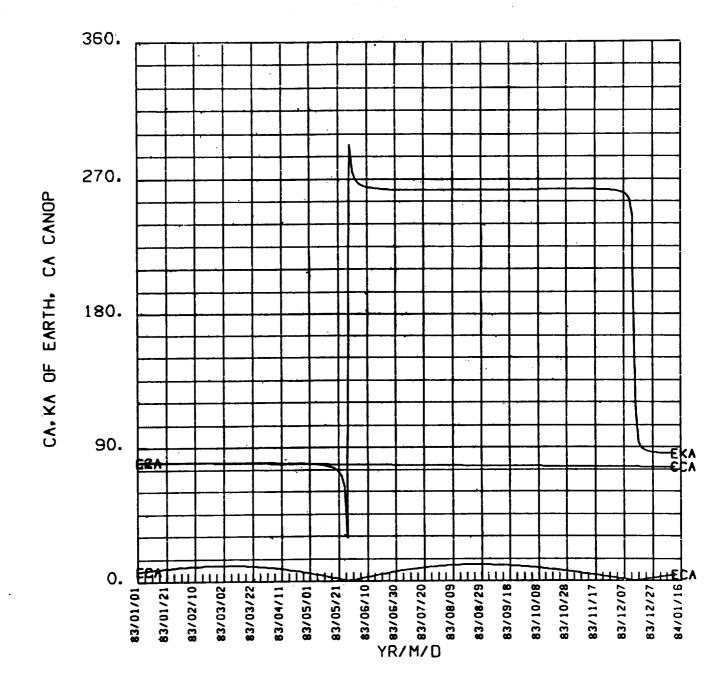




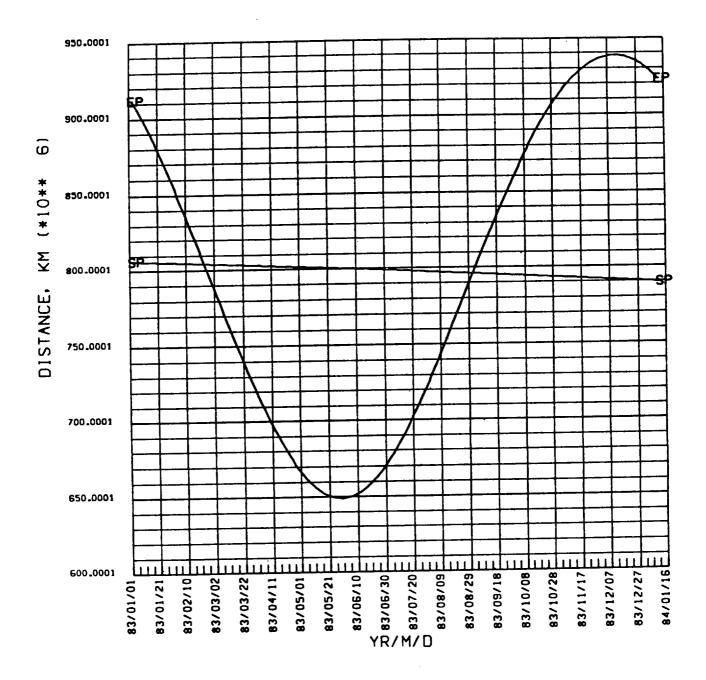




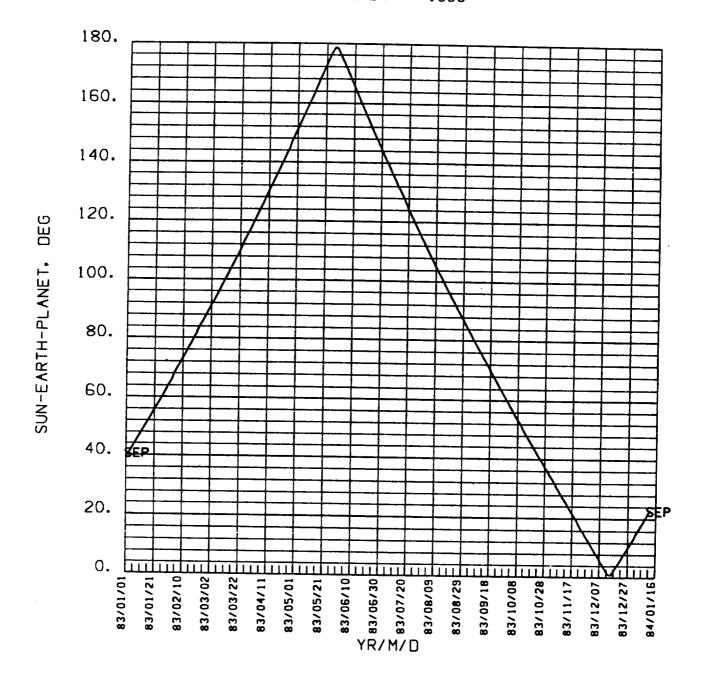


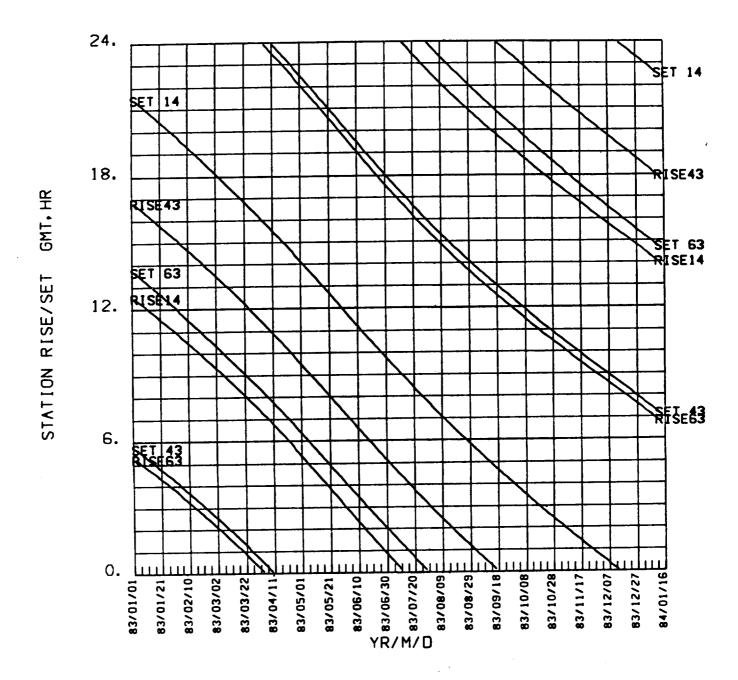


JUPITER 1983

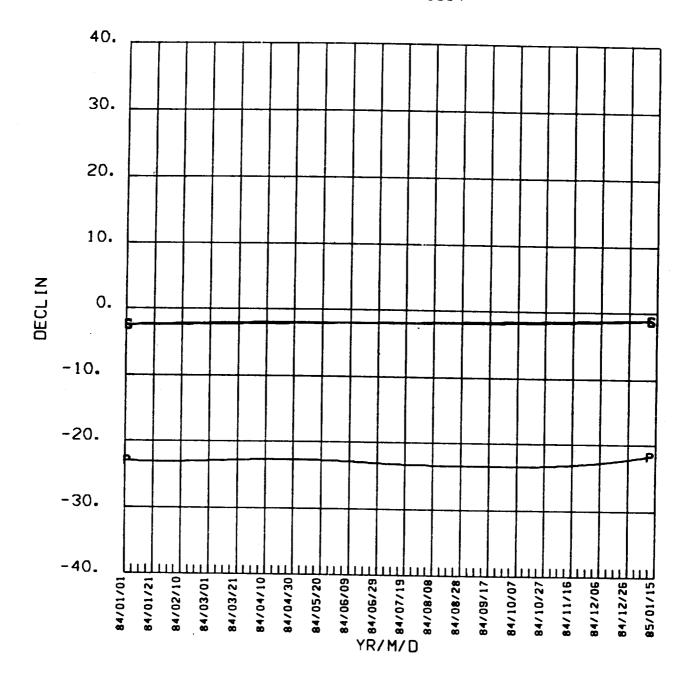


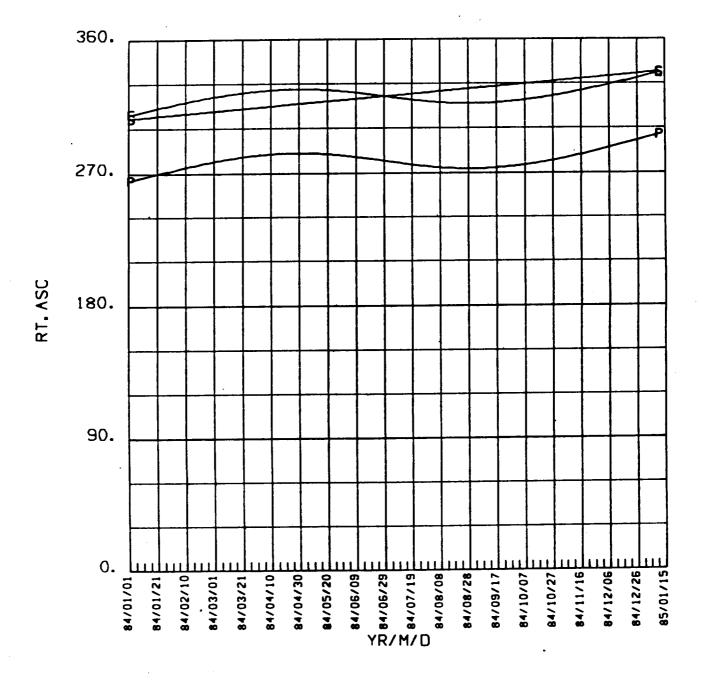
JUPITER 1983

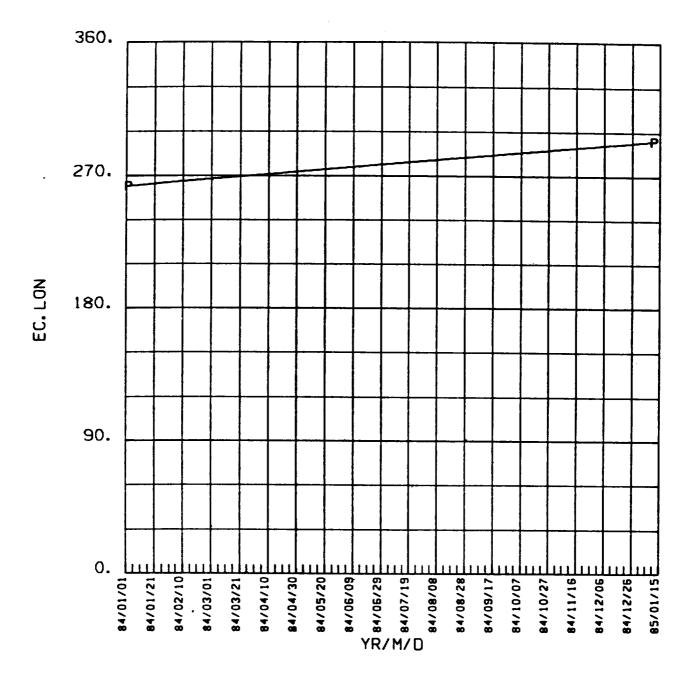


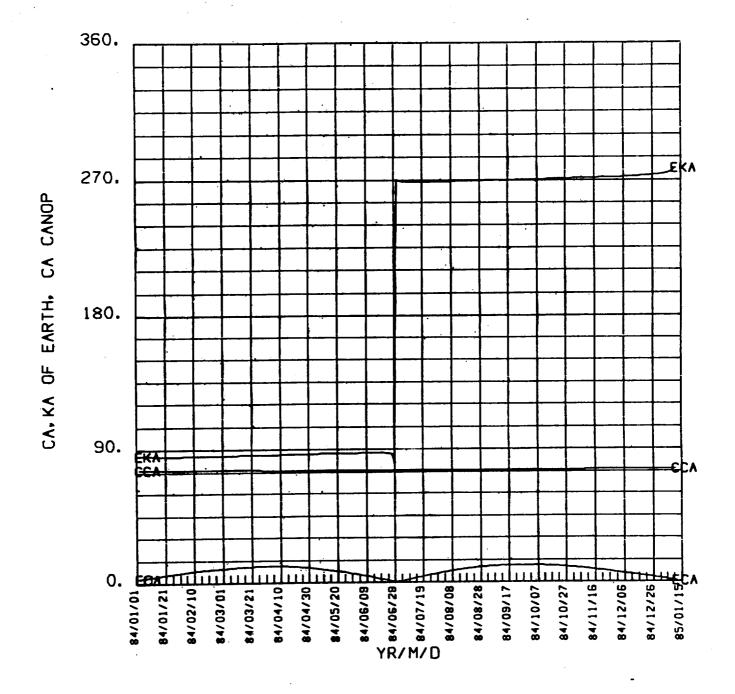


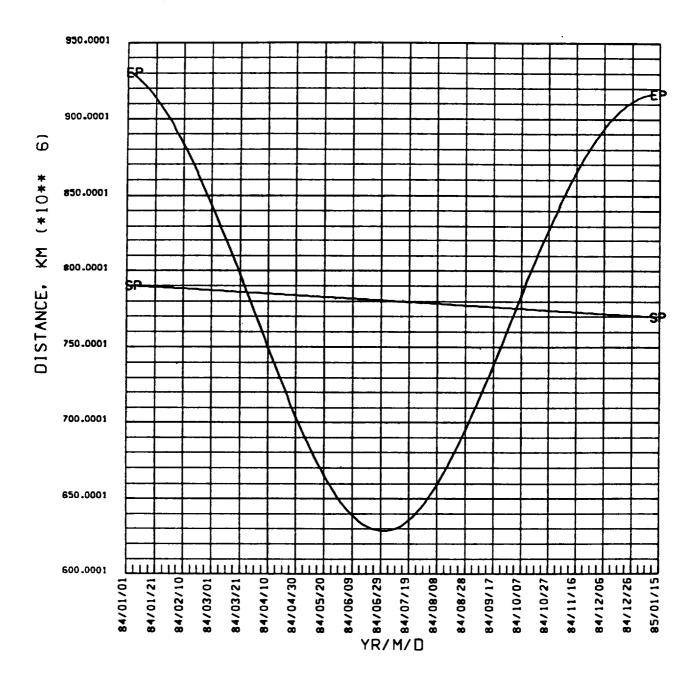


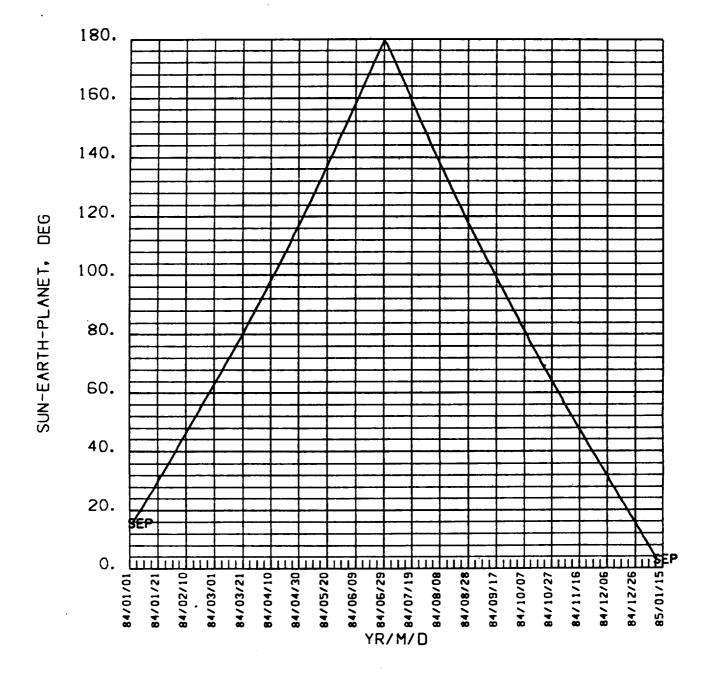




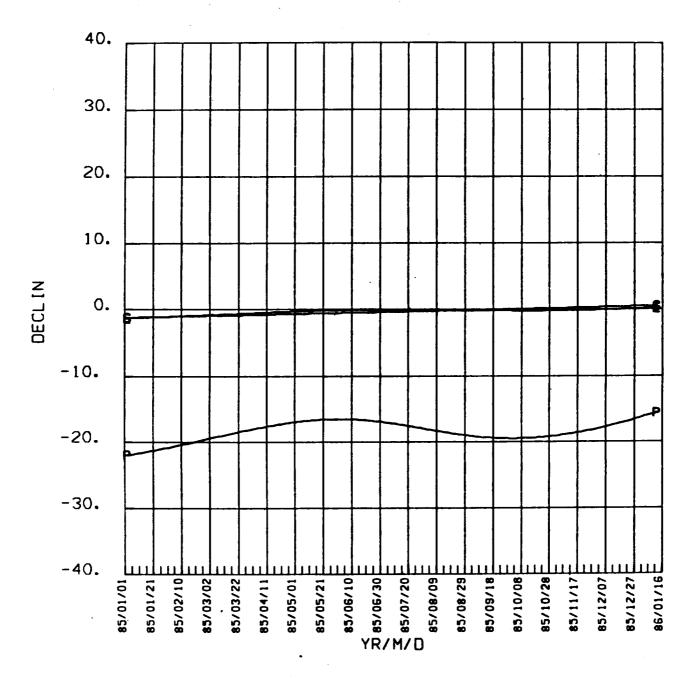


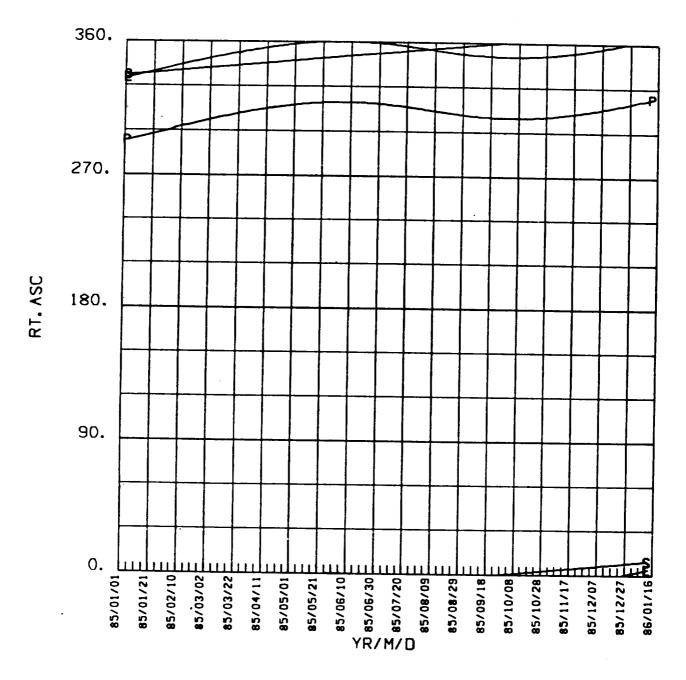


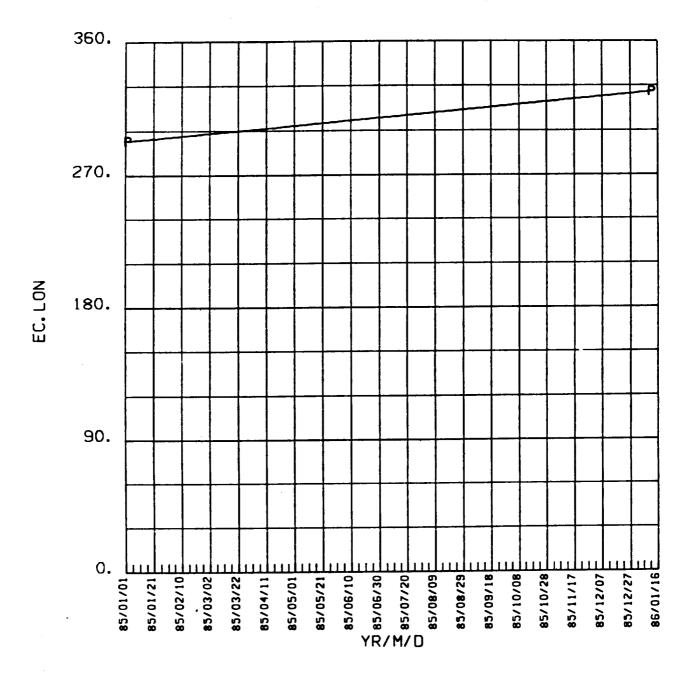


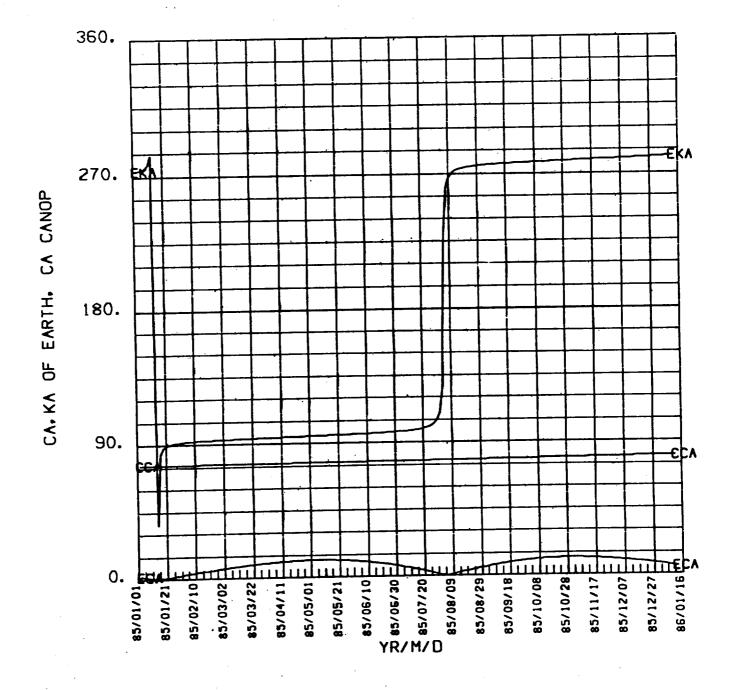




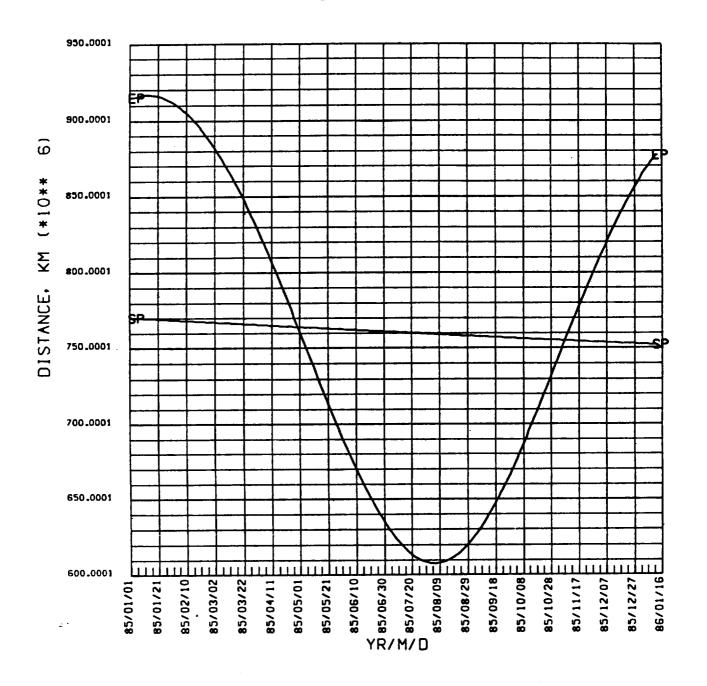


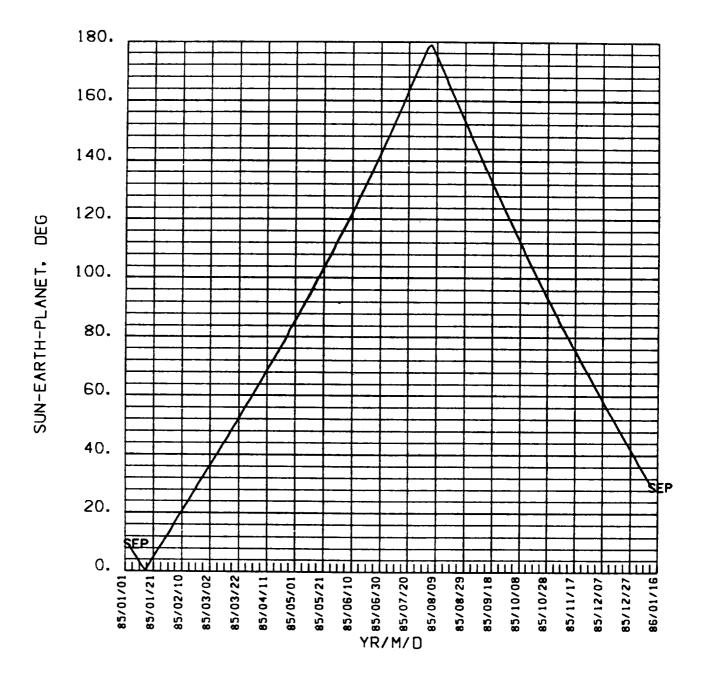




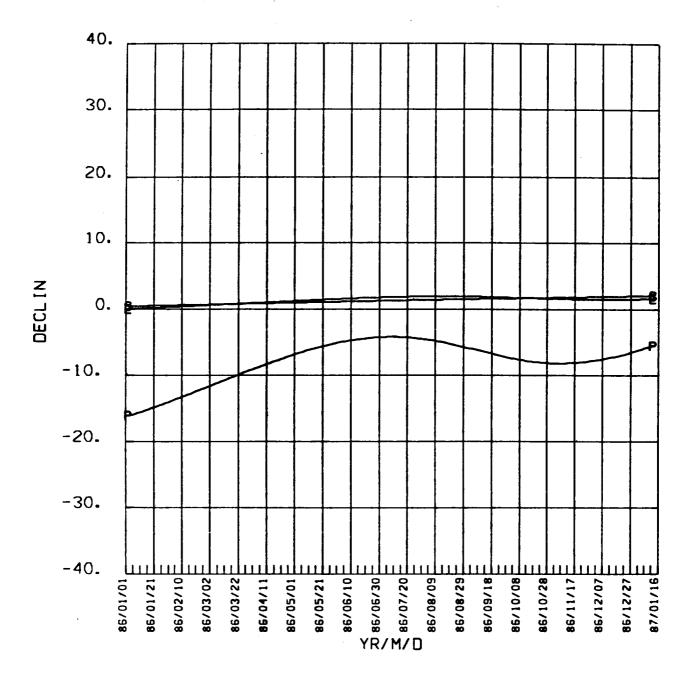


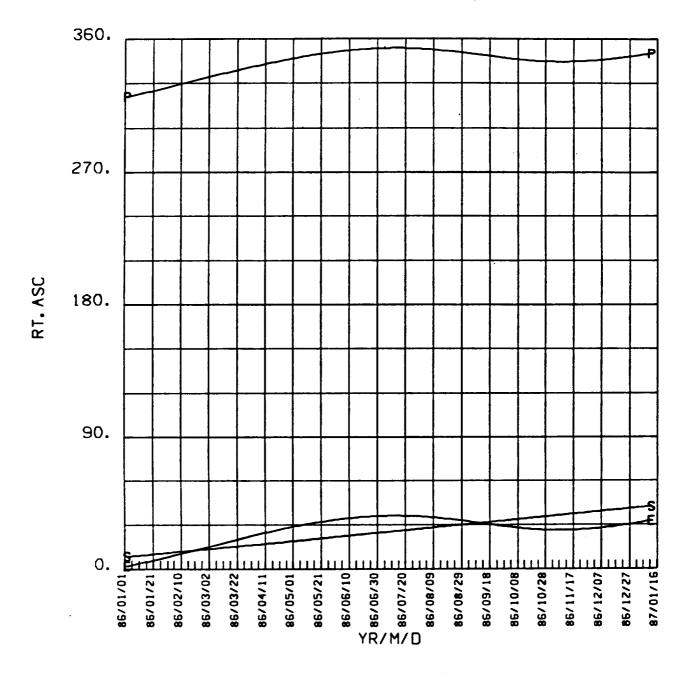
JUPITER 1985

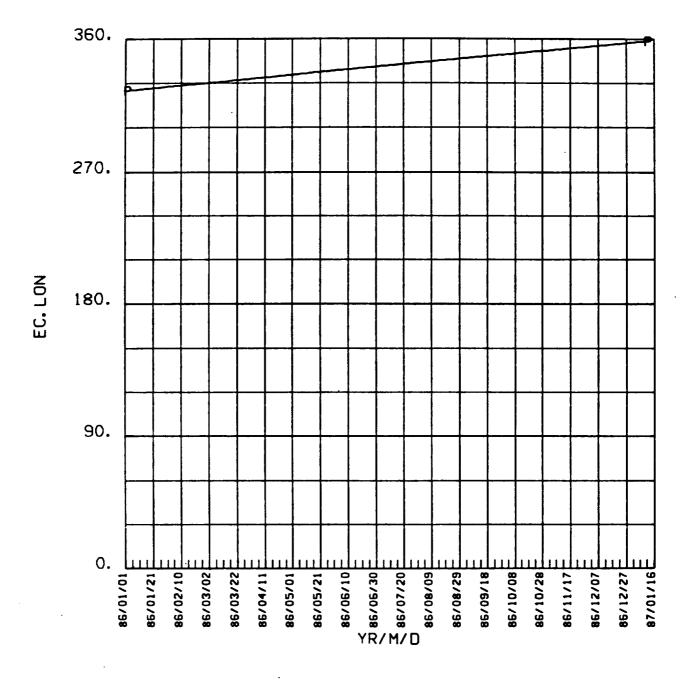


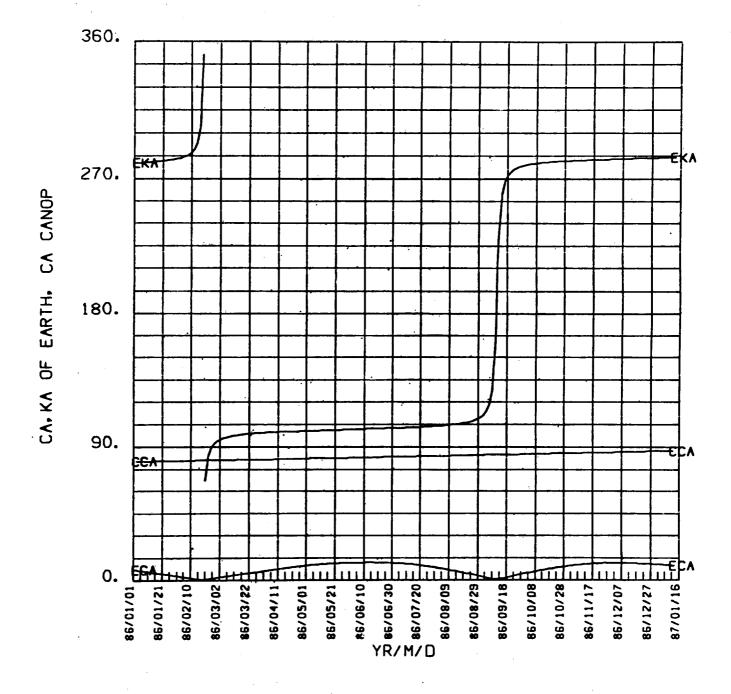


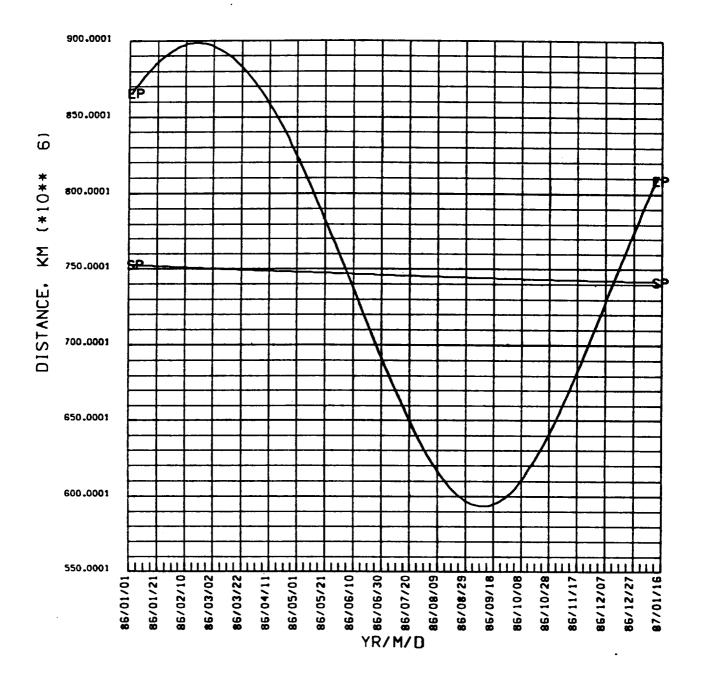




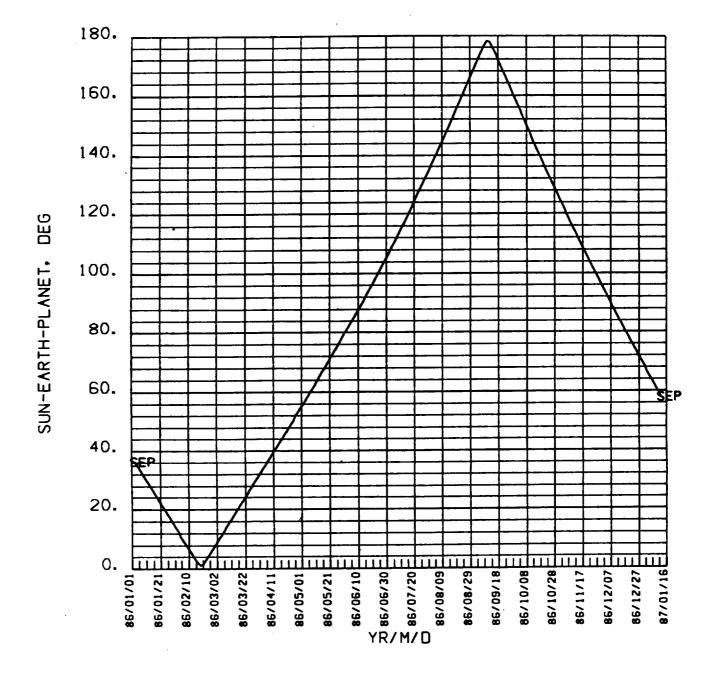


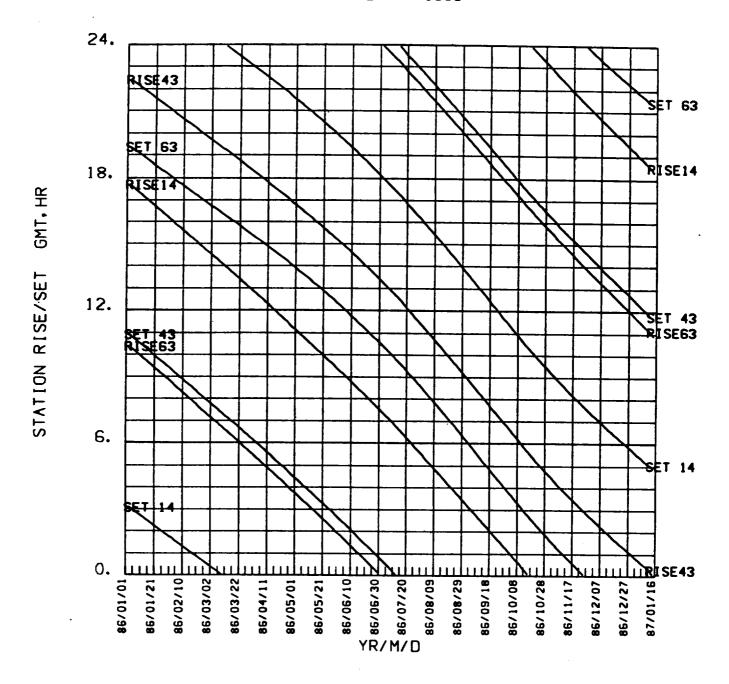




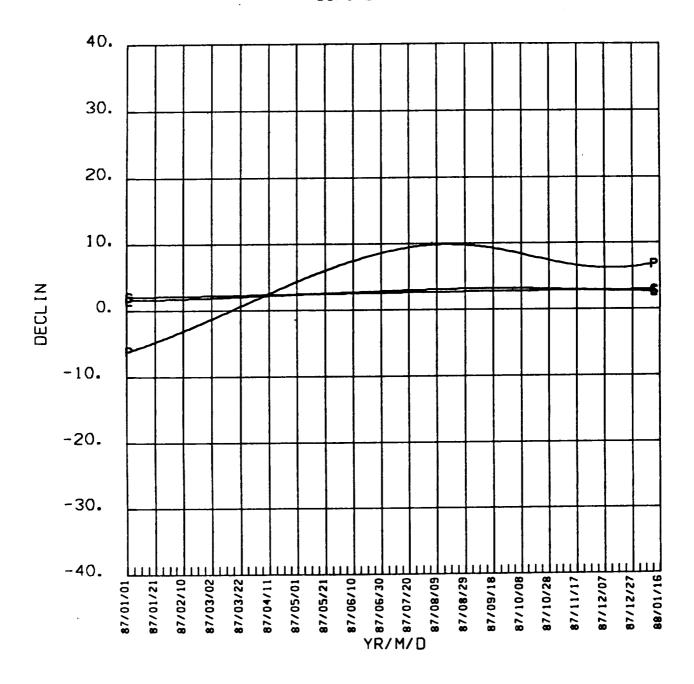


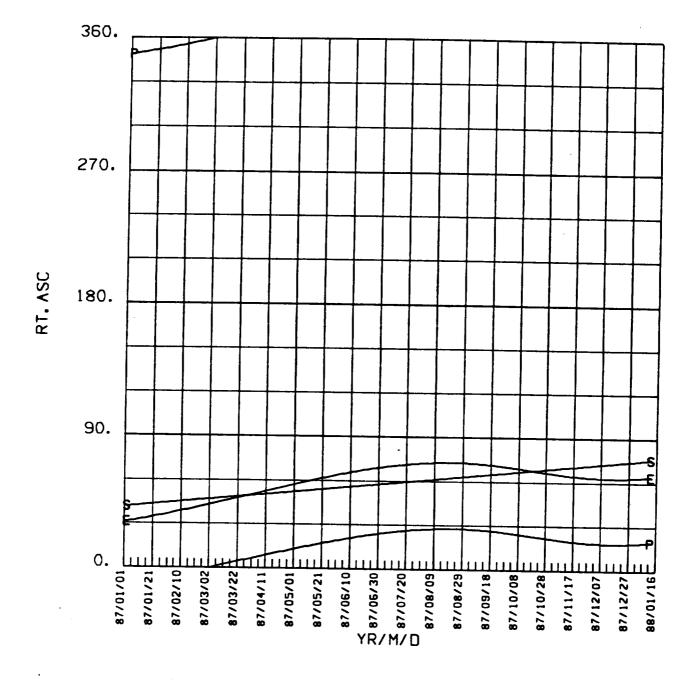
JUPITER 1986



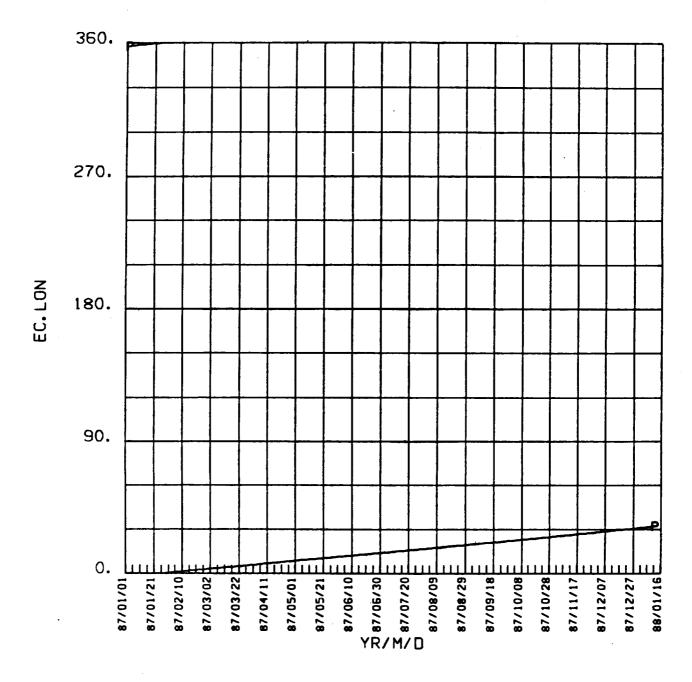


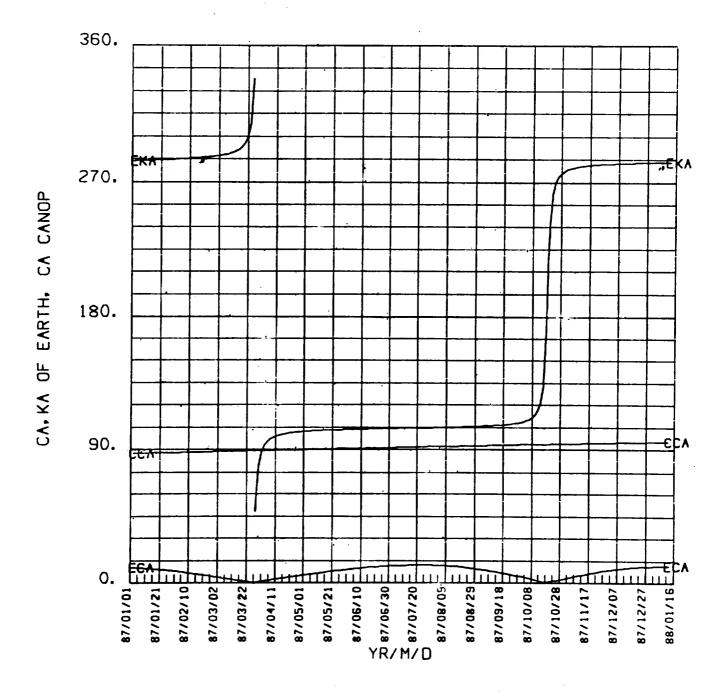




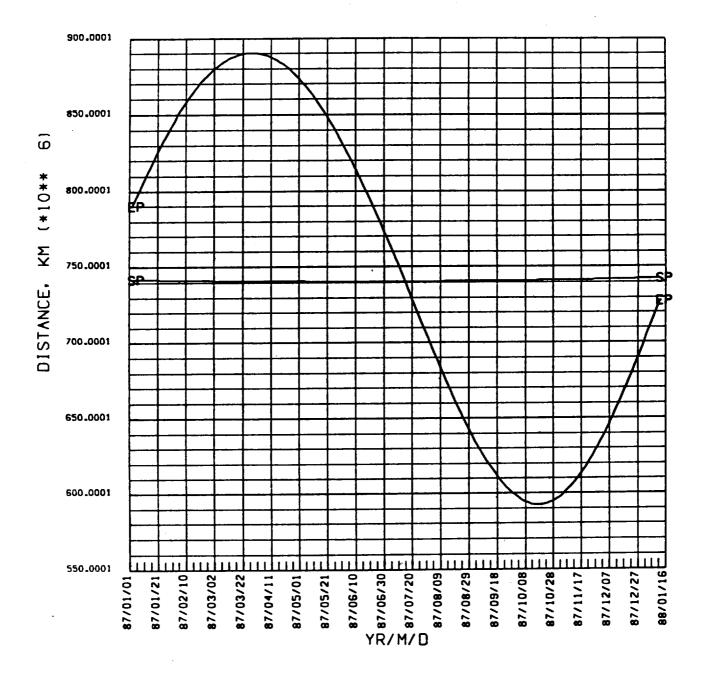


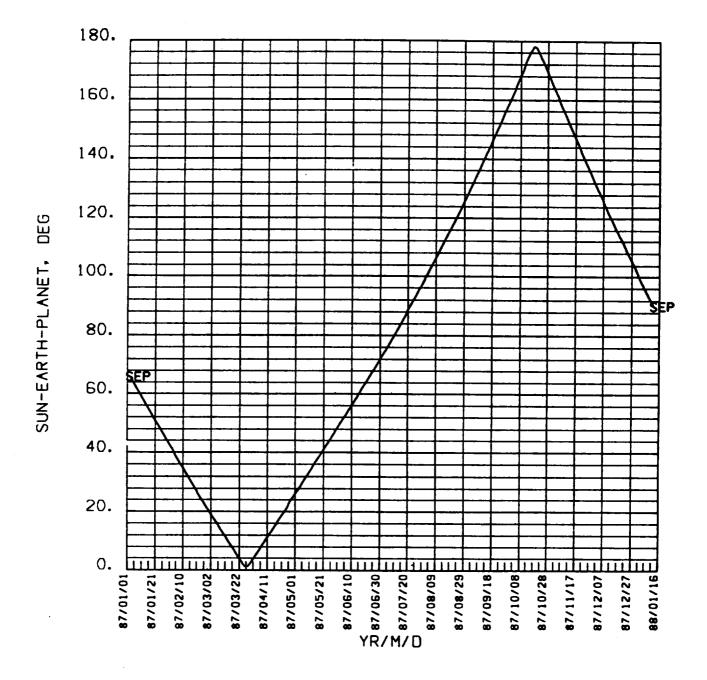


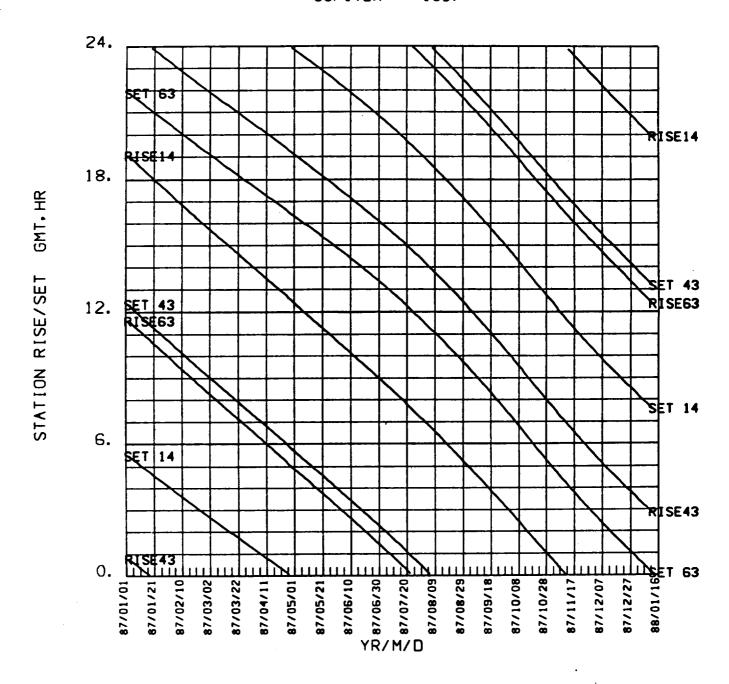


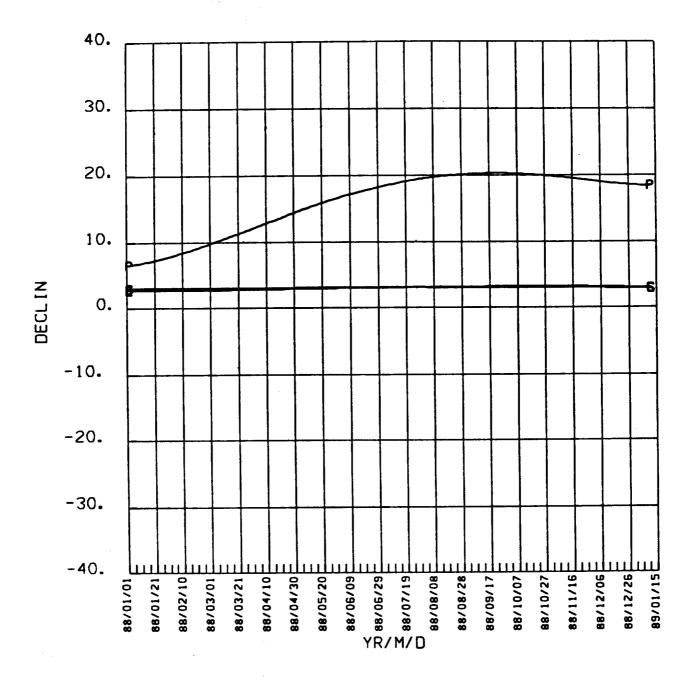


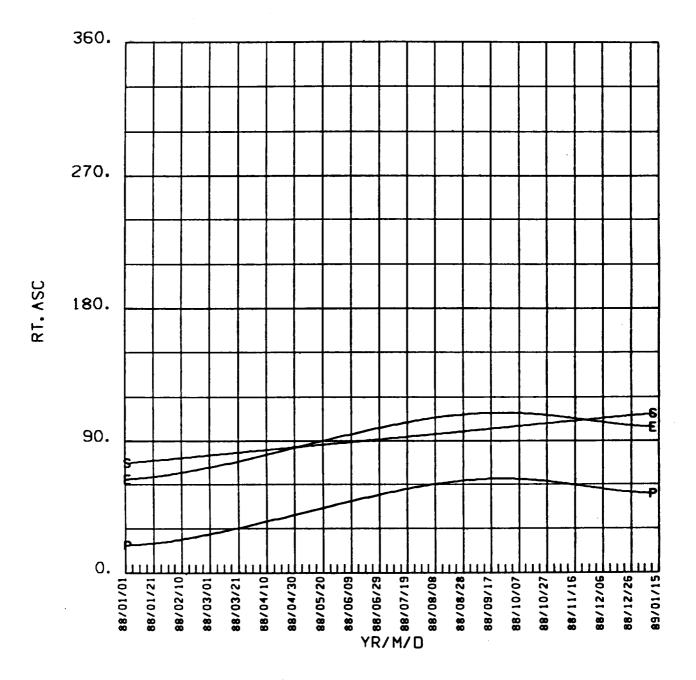
JUPITER 1987

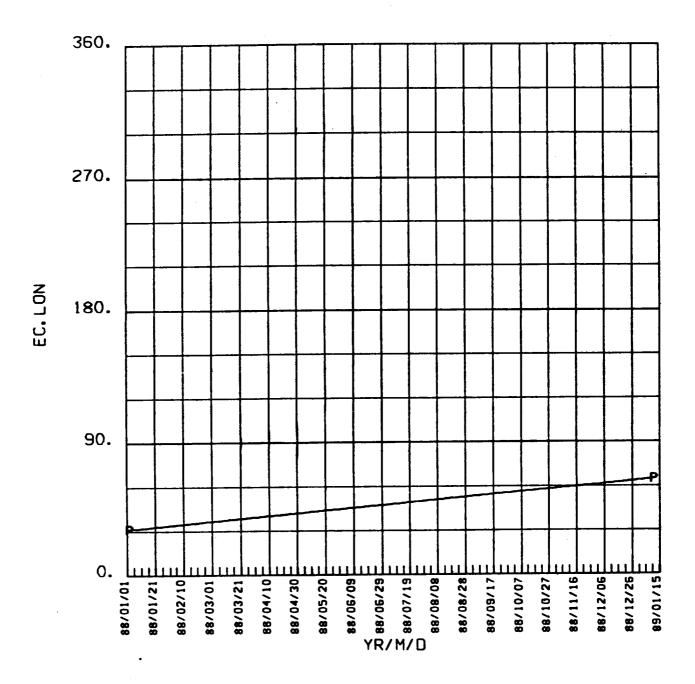


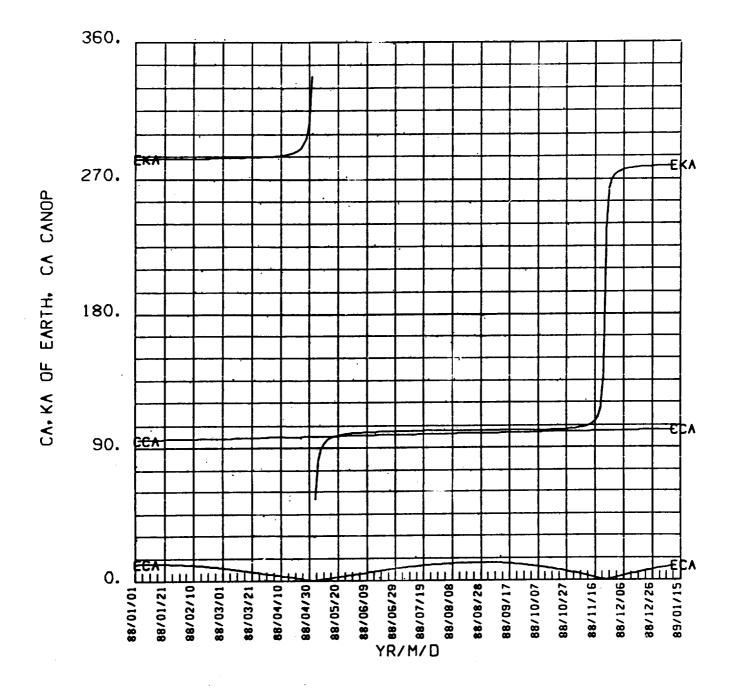


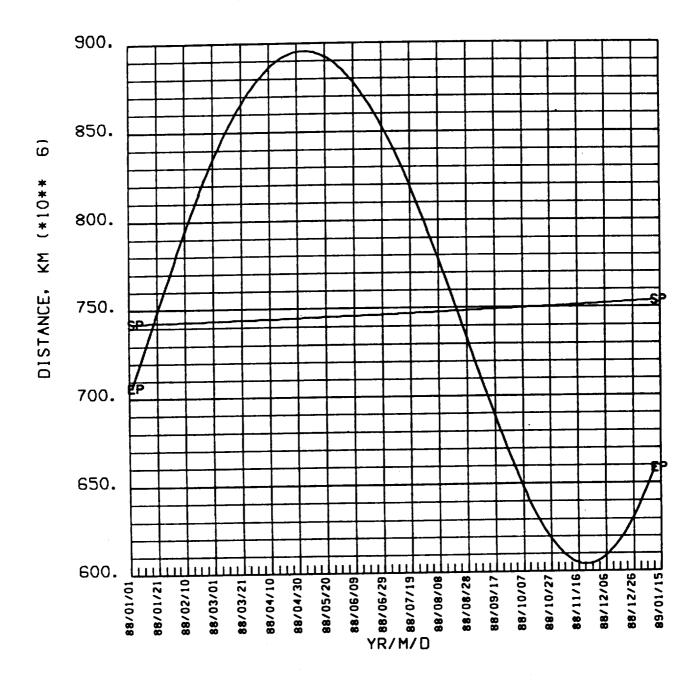




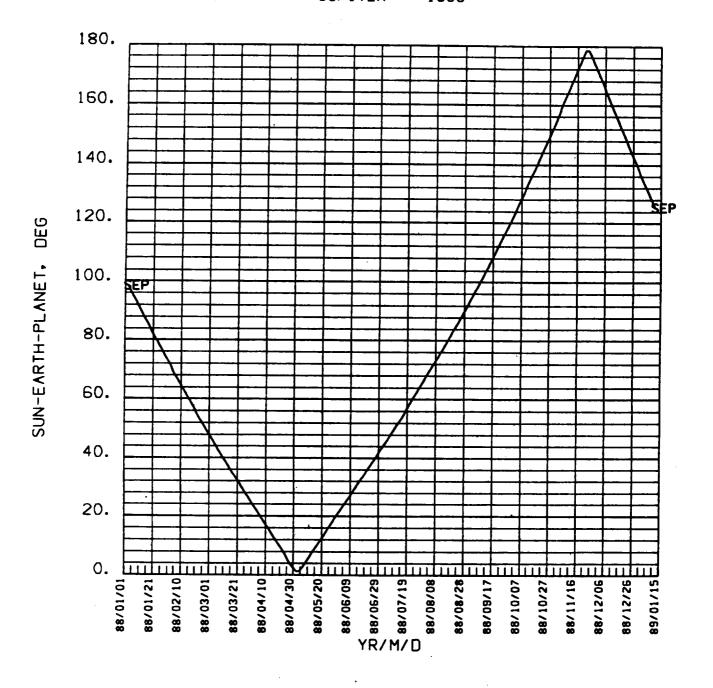


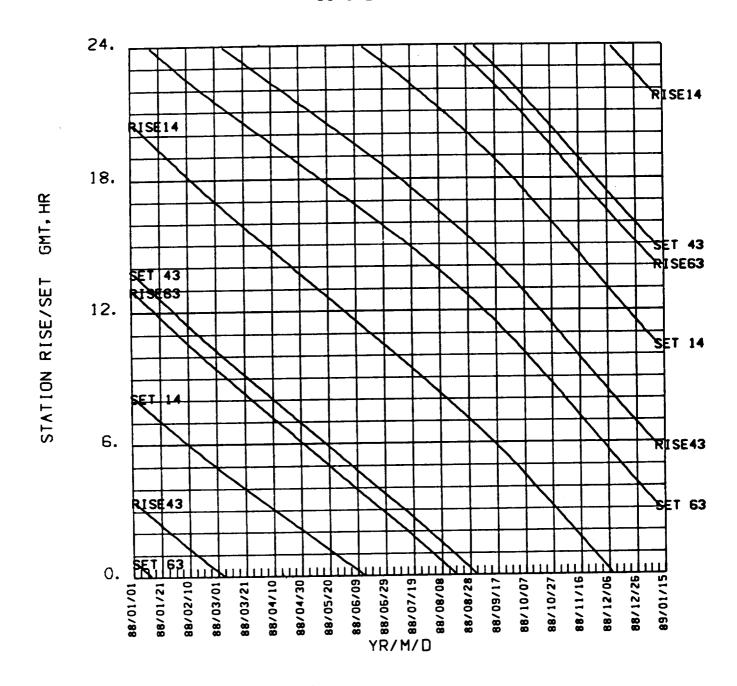




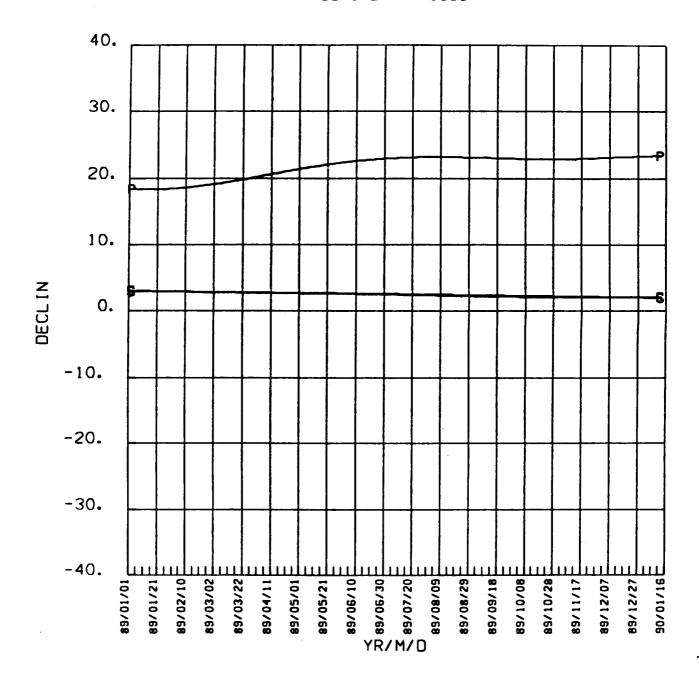


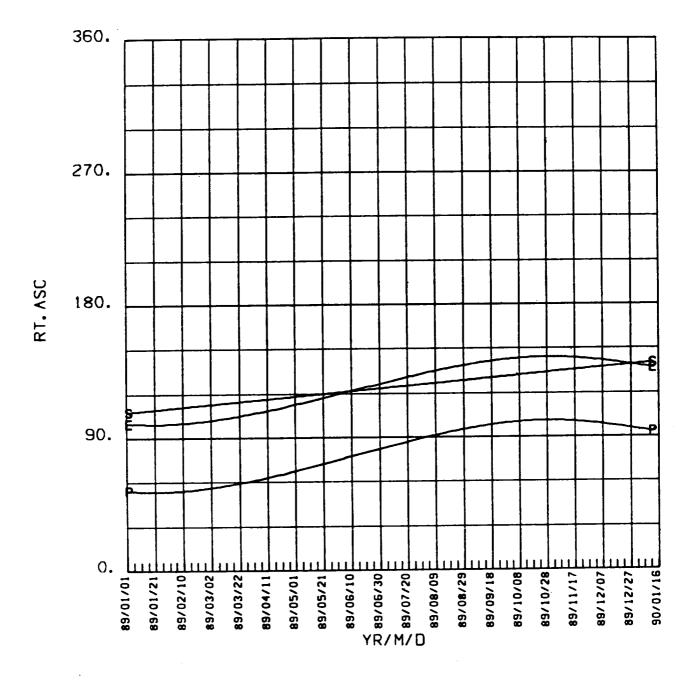
JUPITER 1988

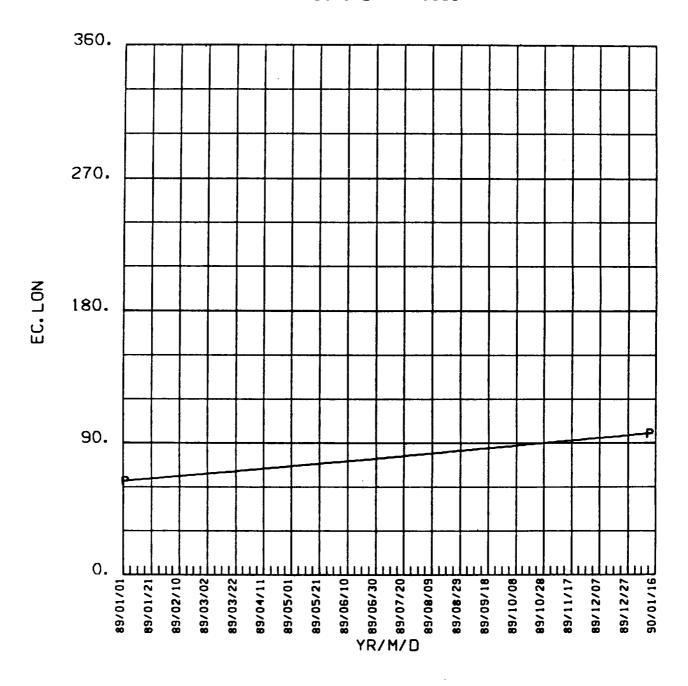


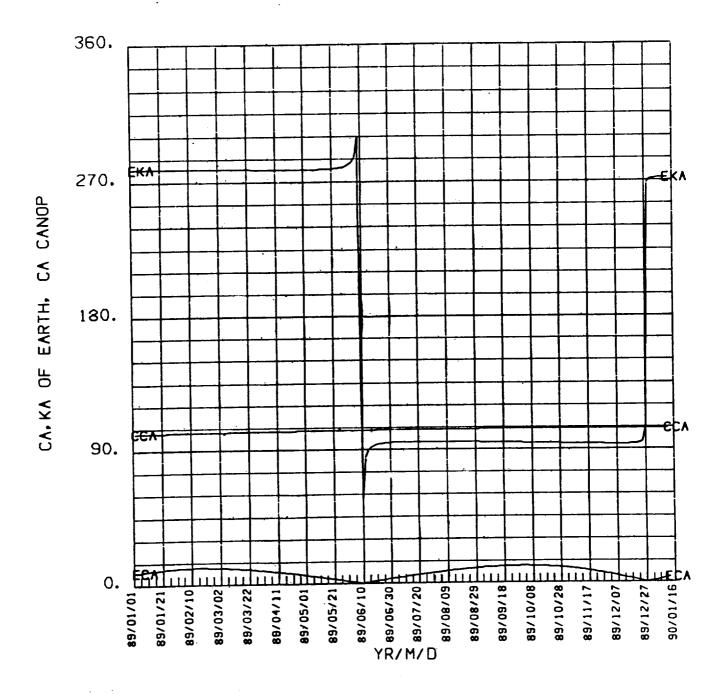


JUPITER . 1989

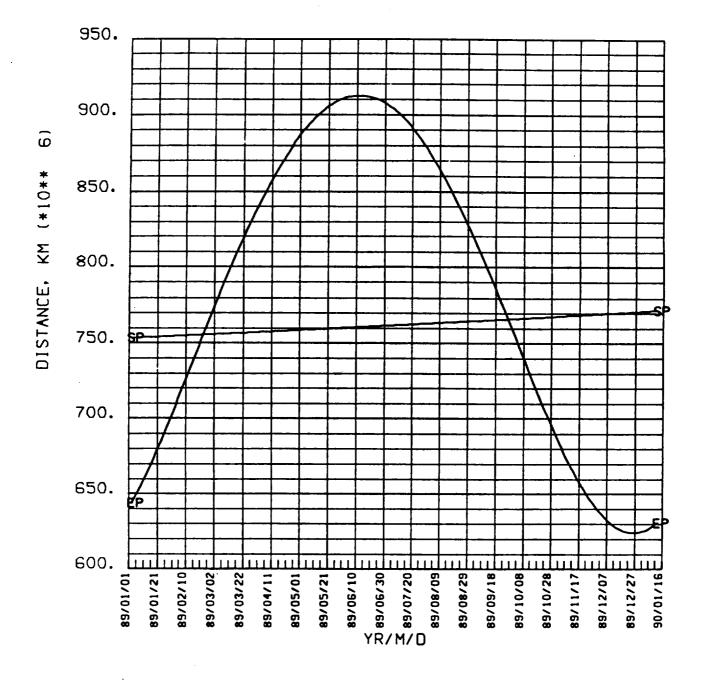


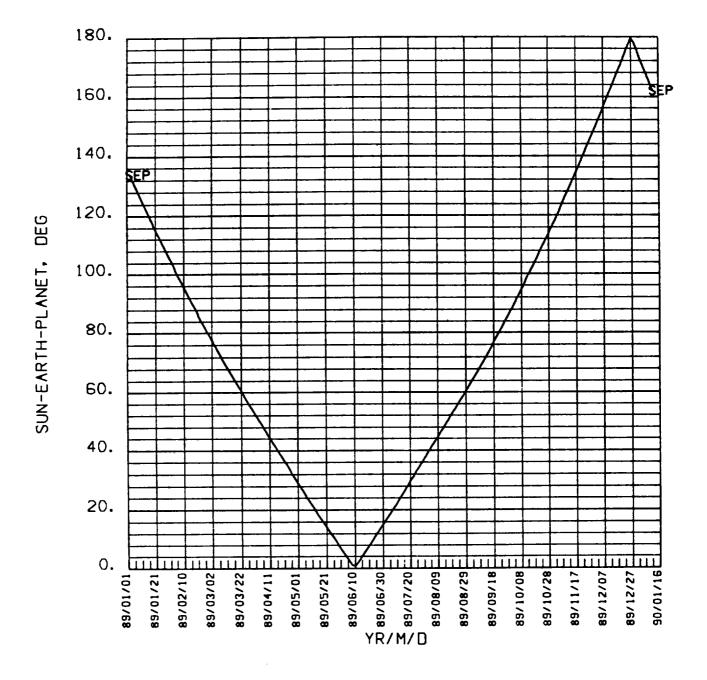




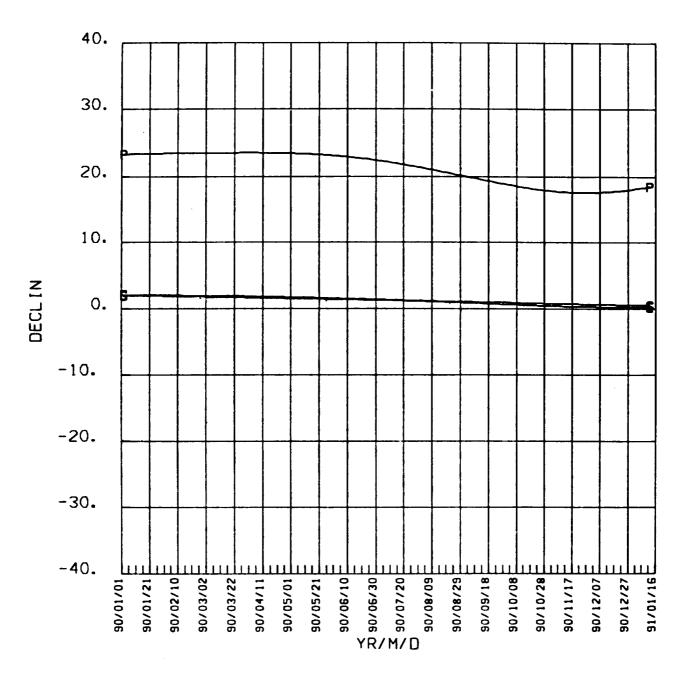


C-4

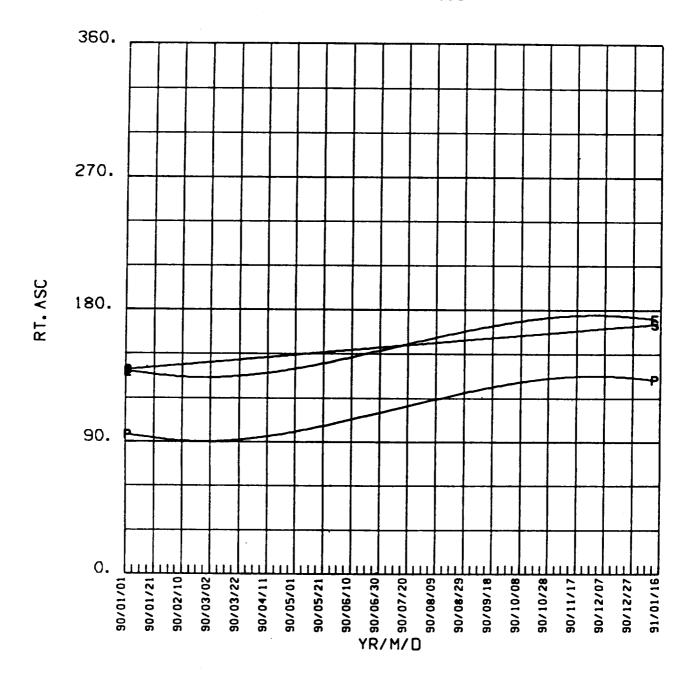


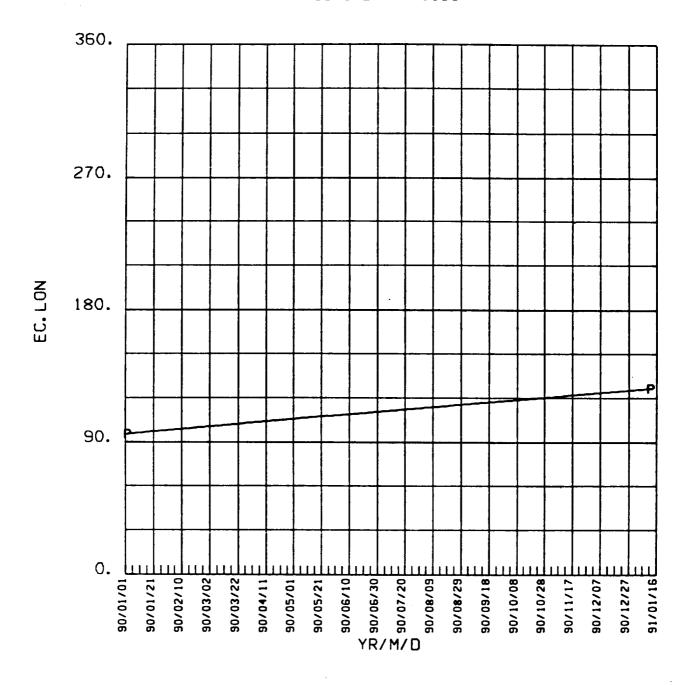


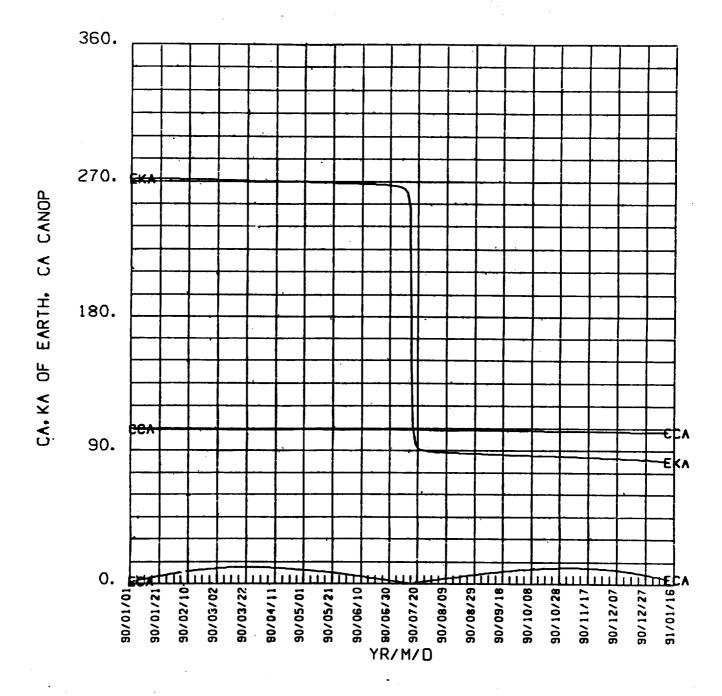


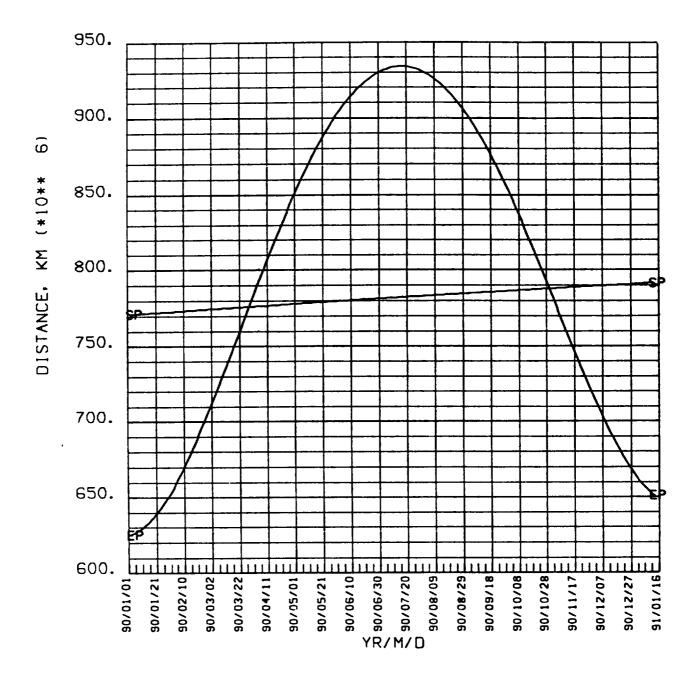


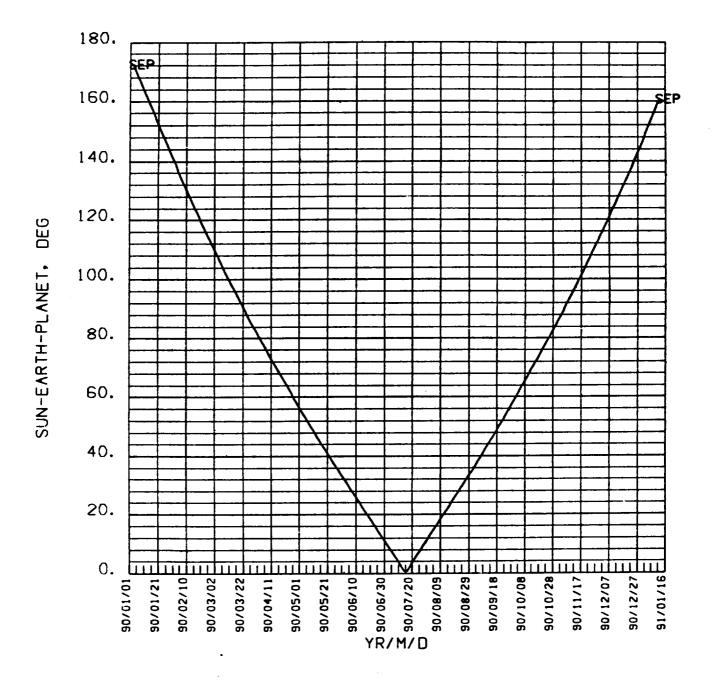


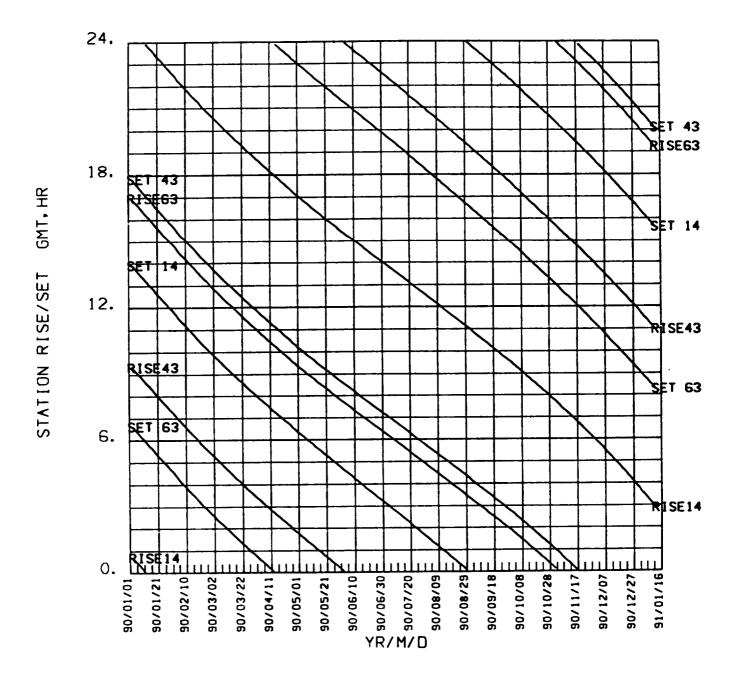




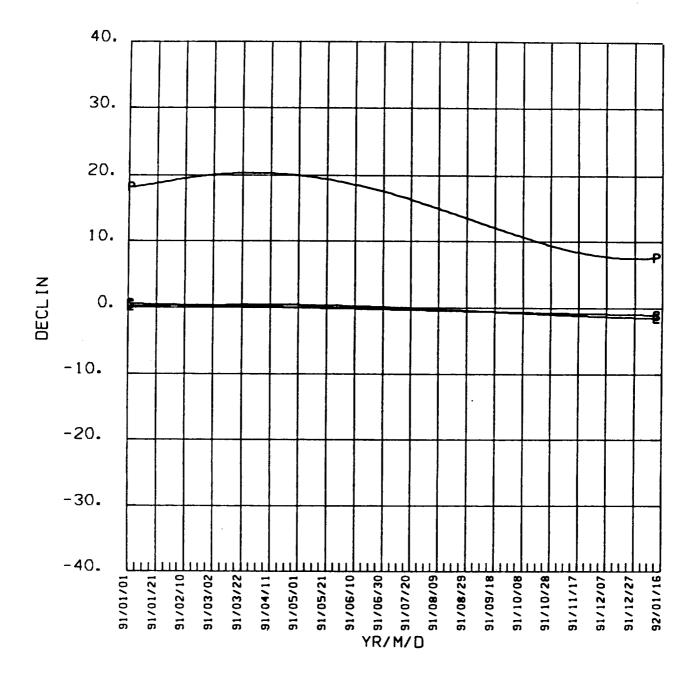


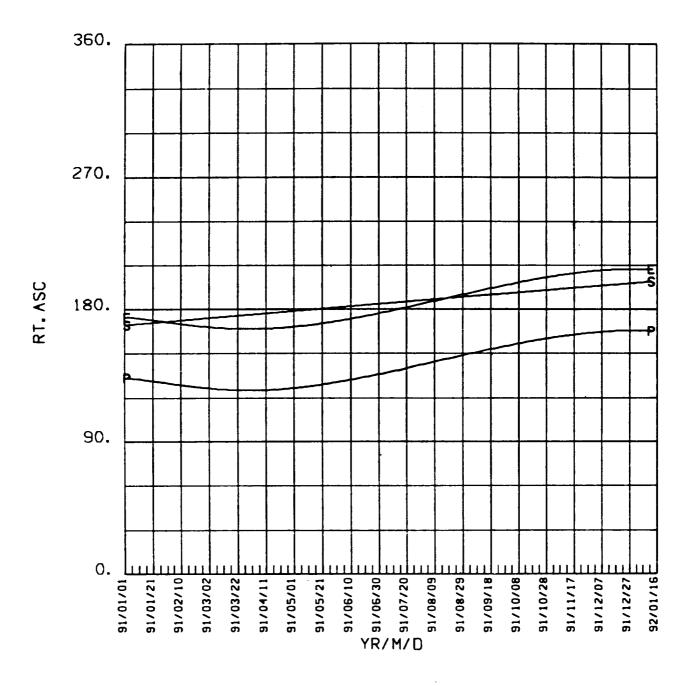


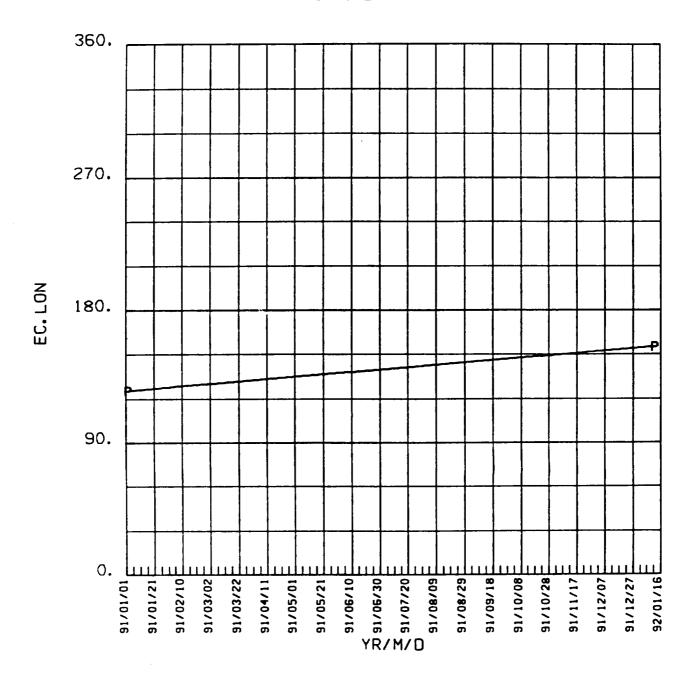


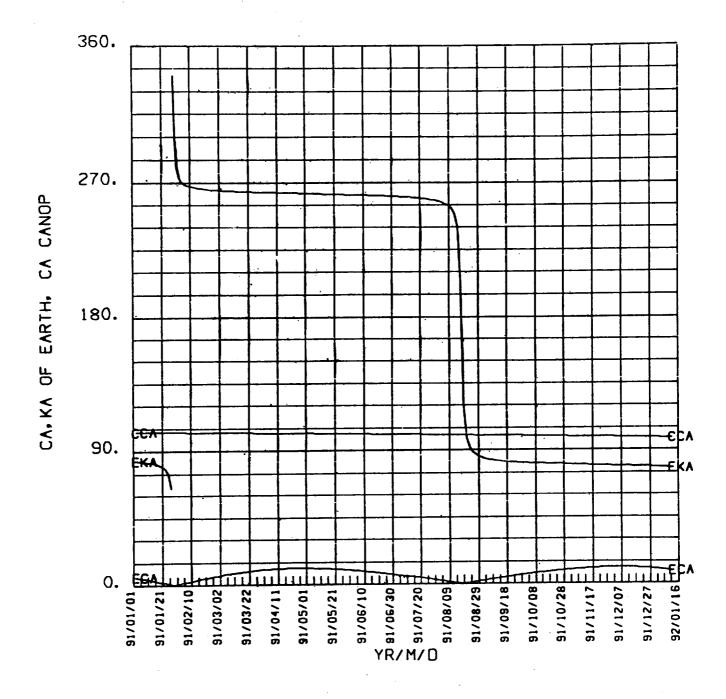




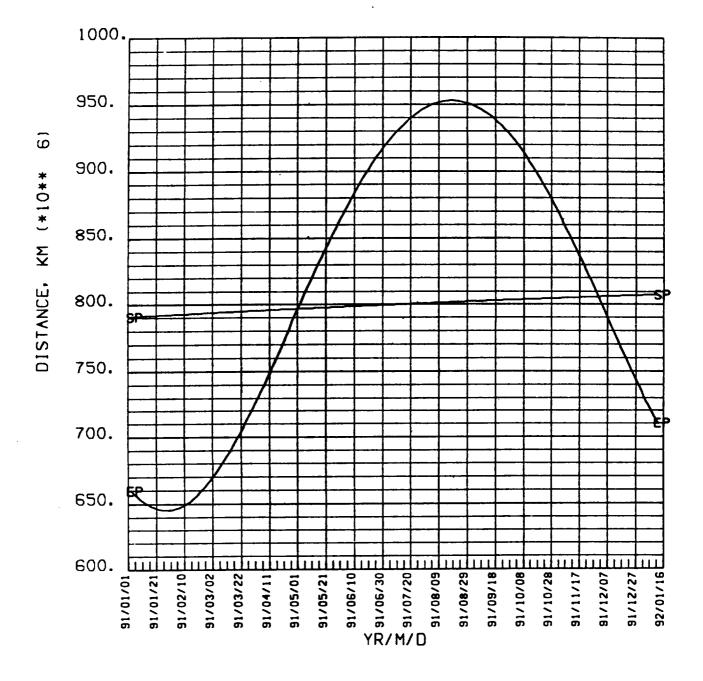


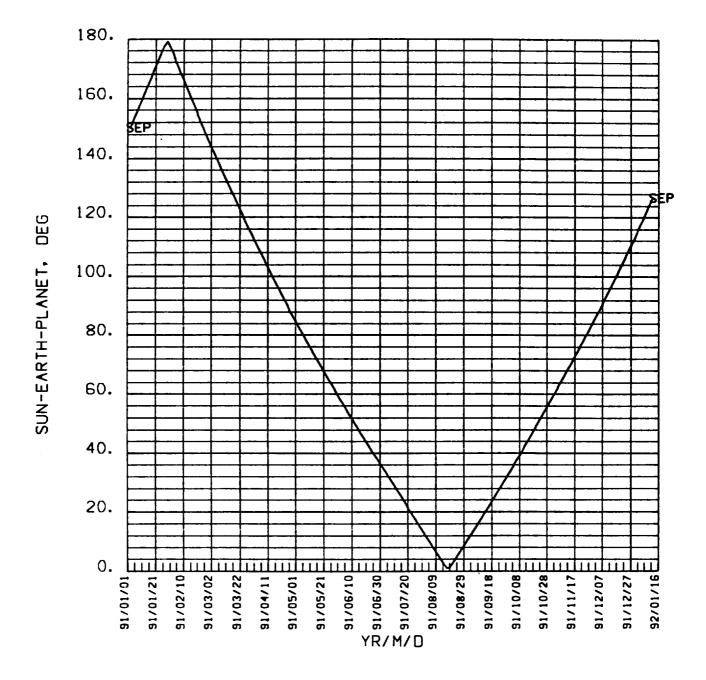


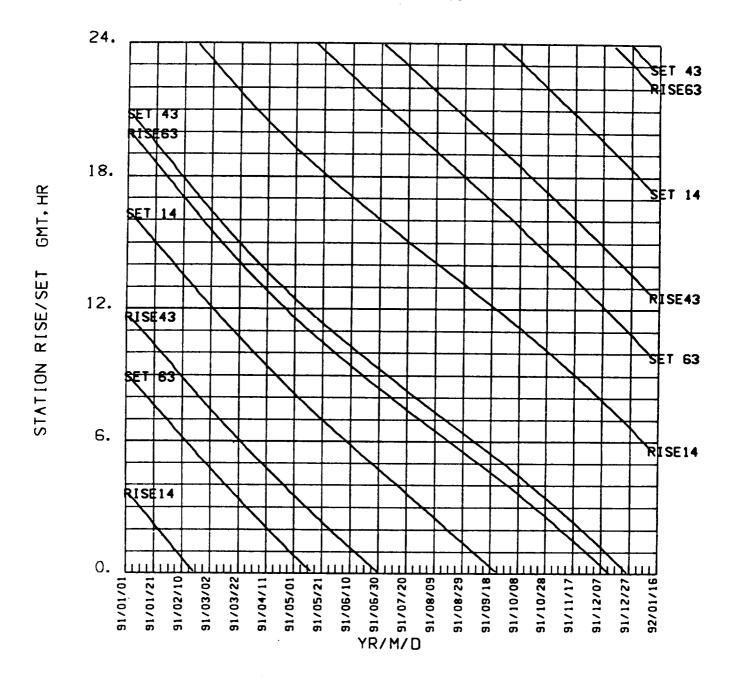




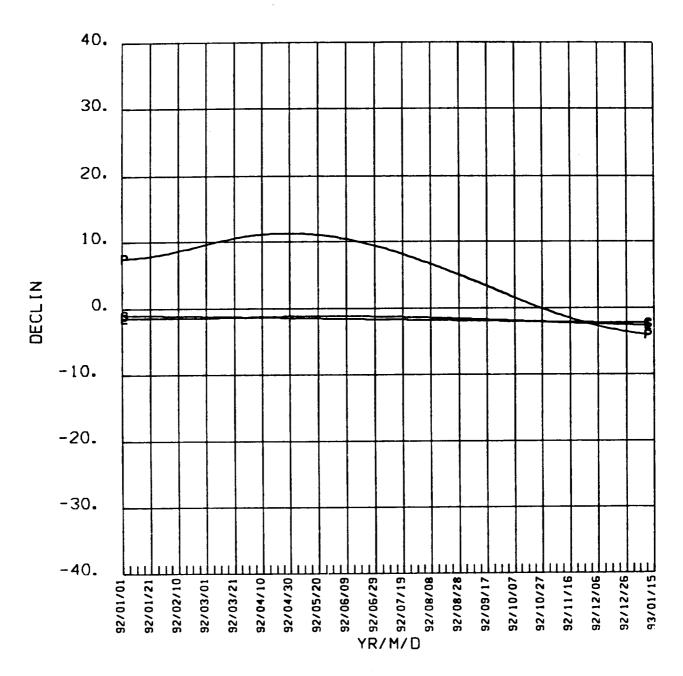




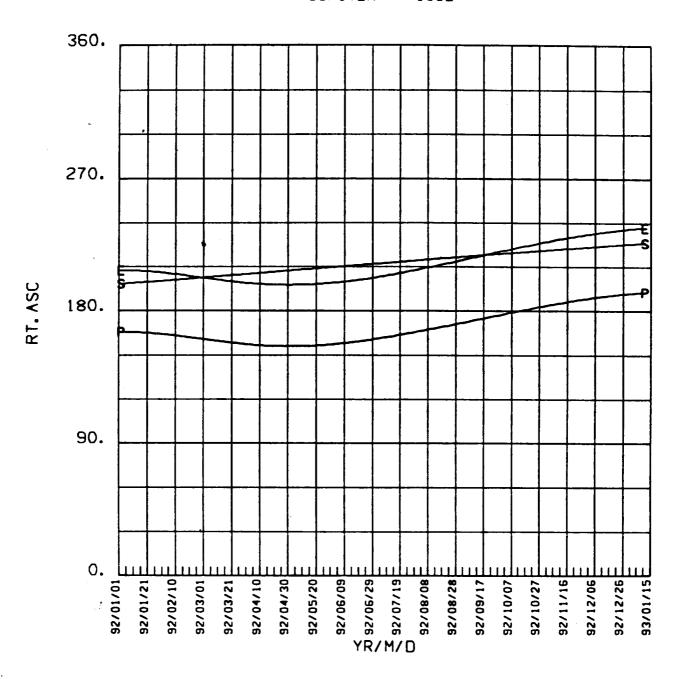




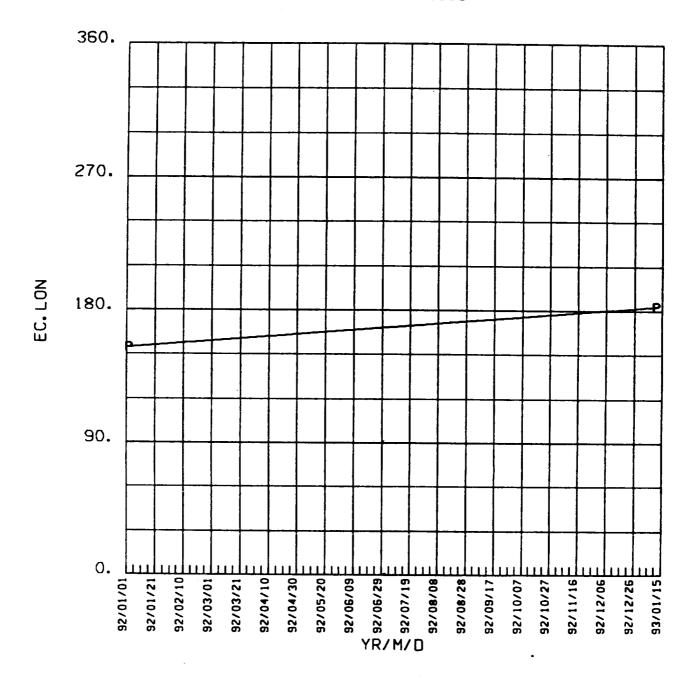


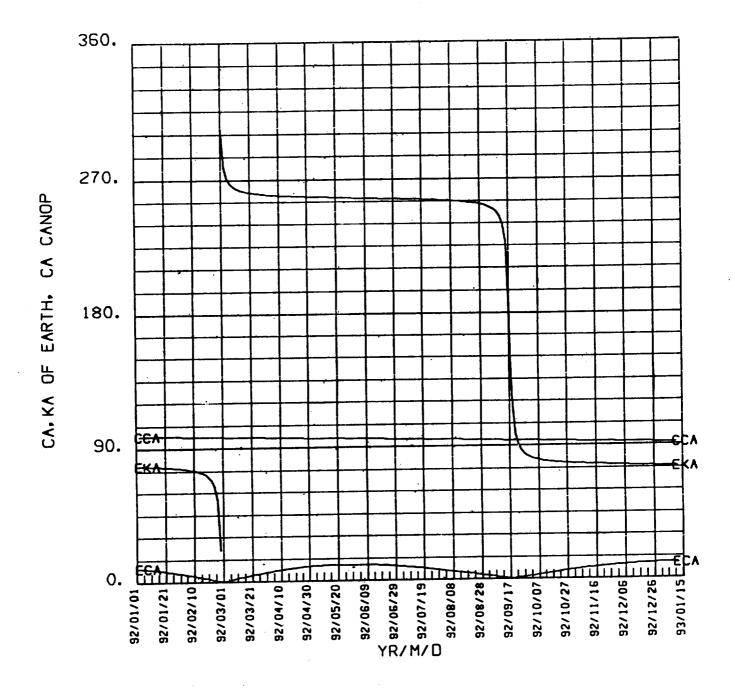


This page intentionally left blank.

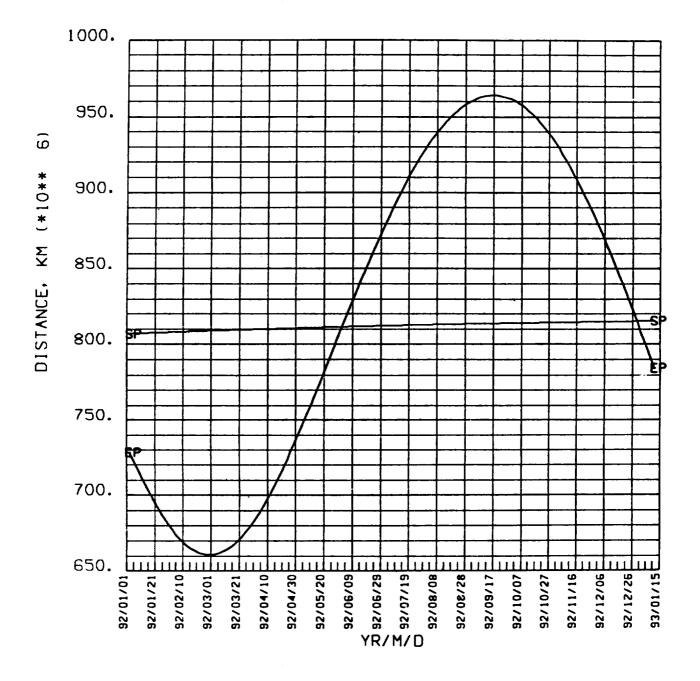


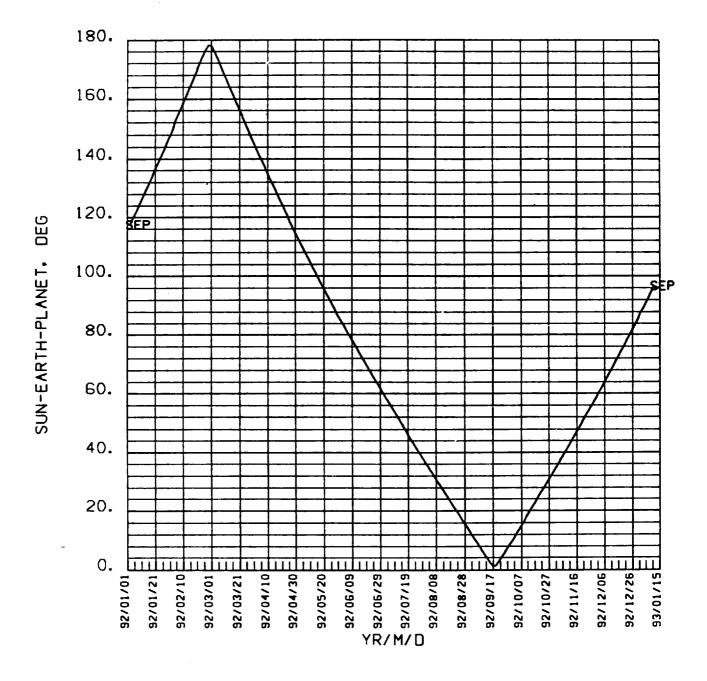
PRECEDING PAGE BLANK NOT FILMED

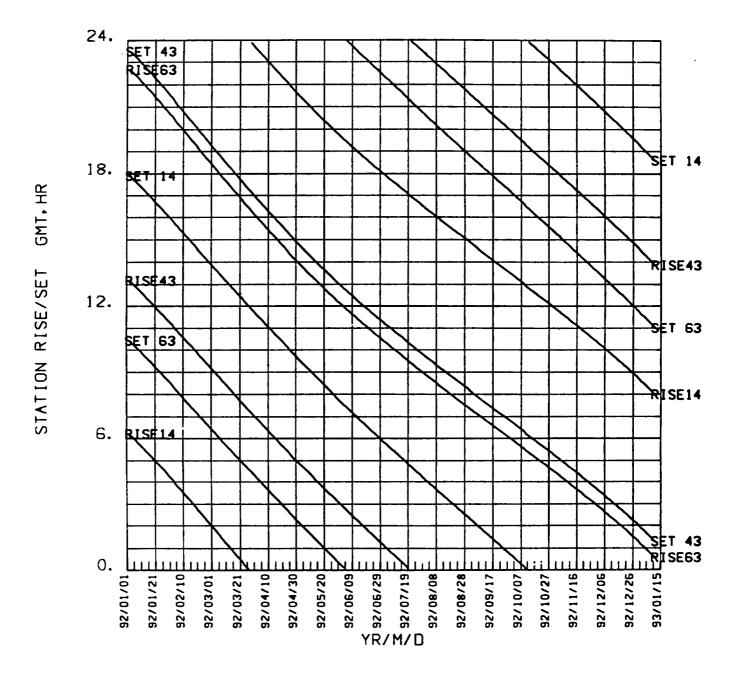




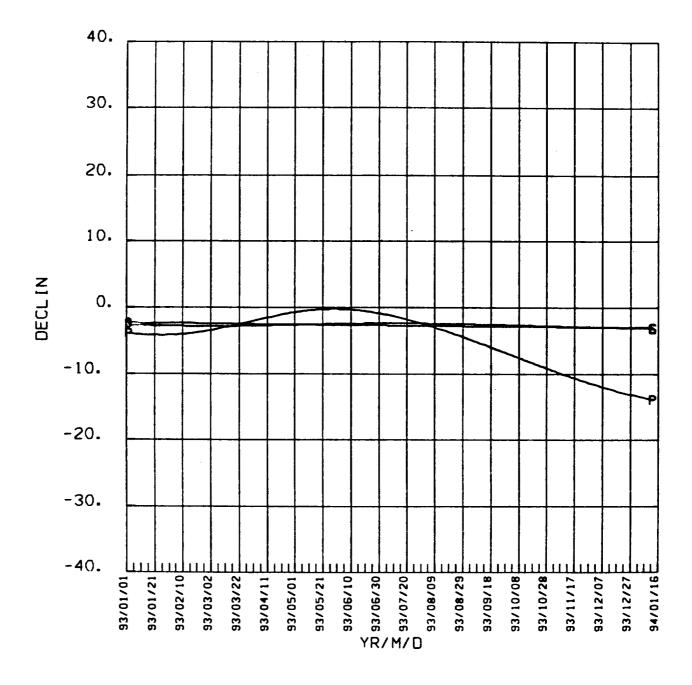


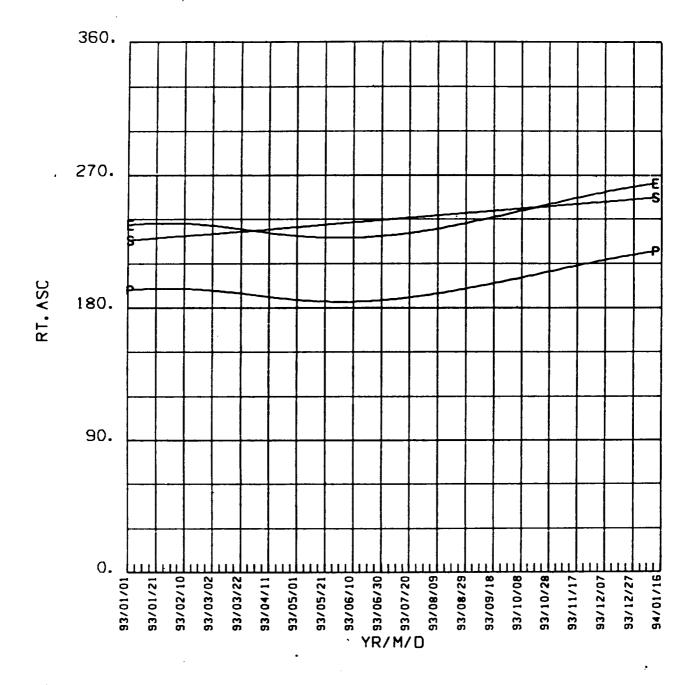


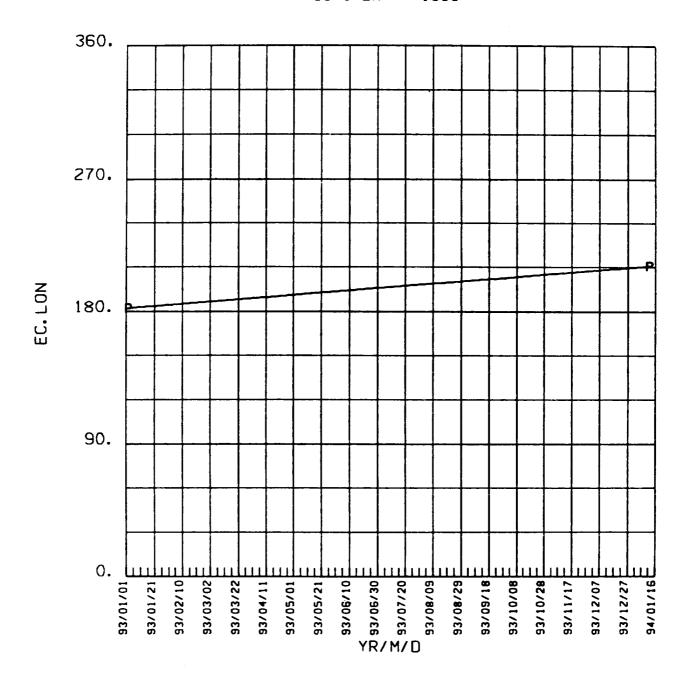




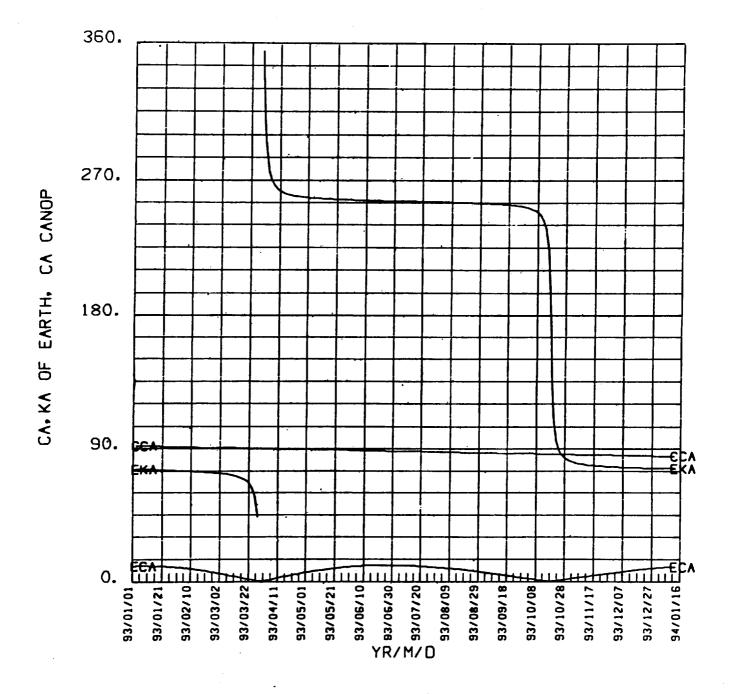


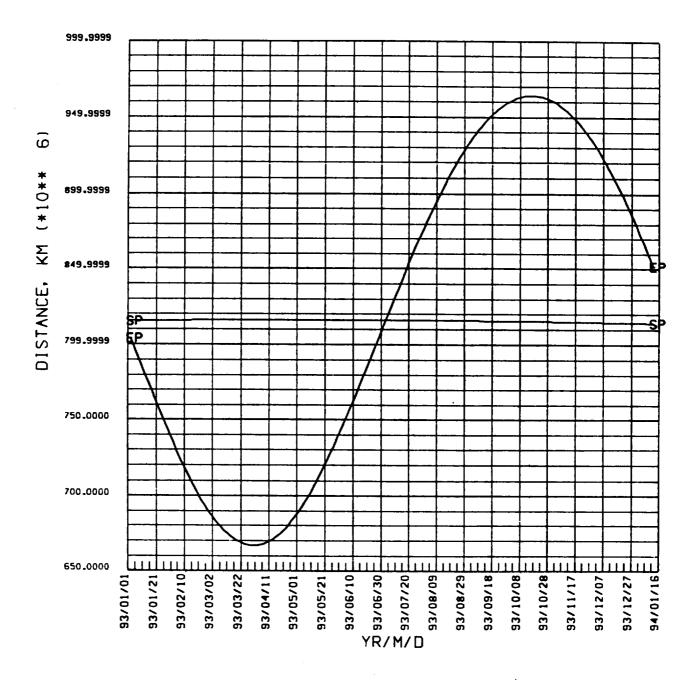


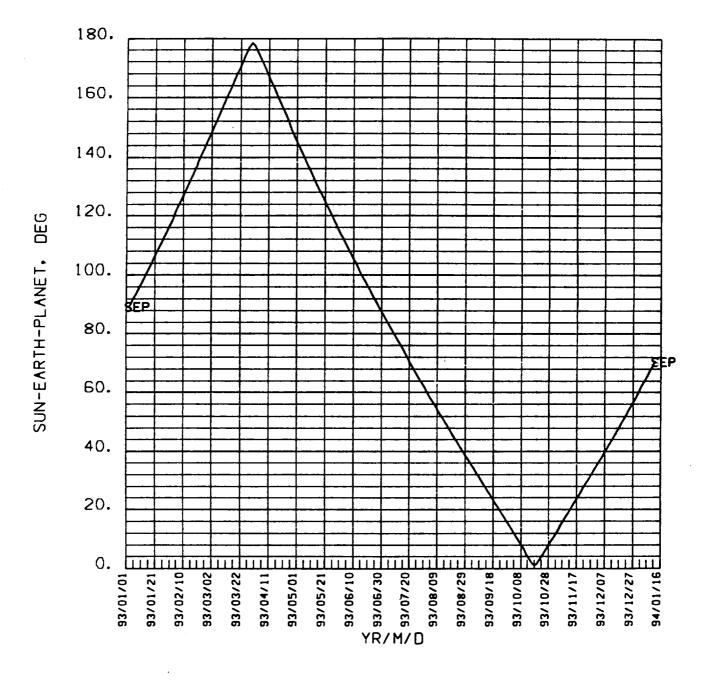




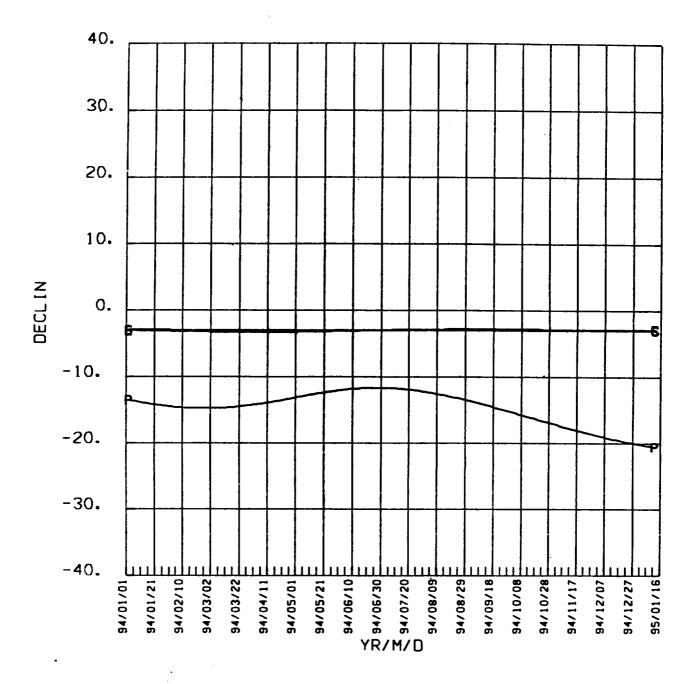


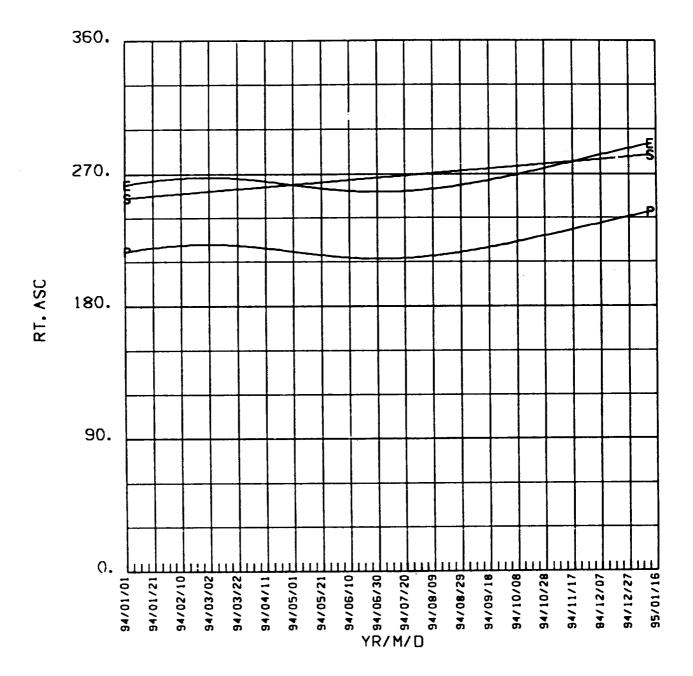


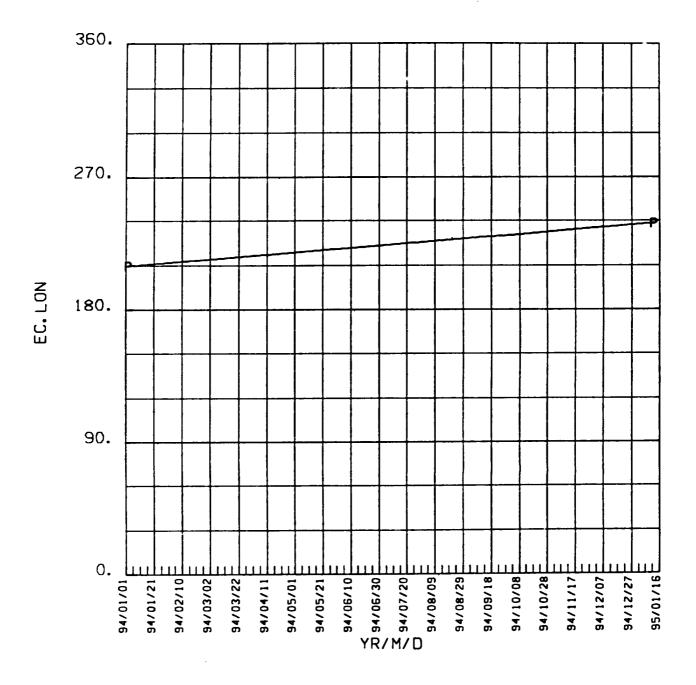


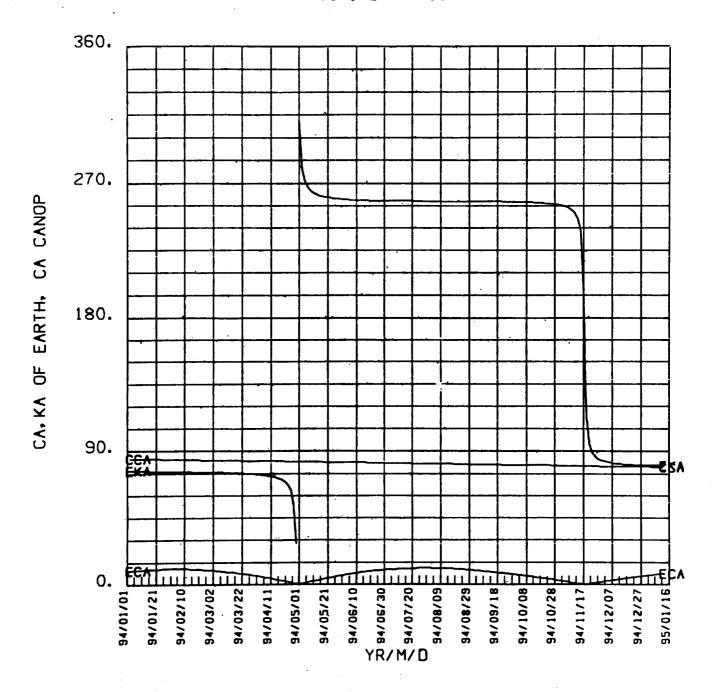












JUPITER 1994

